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**DEVELOPMENT OF  
ALUMINUM BASE ALLOYS**

**SECTION III**



**ALCOA RESEARCH LABORATORIES  
NEW KENSINGTON, PA.**

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ALUMINUM COMPANY OF AMERICA

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ALCOA

July 25, 1966

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Attn: Mr. Harold Markus  
SMUFA-1320

REF: CONTRACT NO. DA-36-034-ORD-3559RD  
DEVELOPMENT OF ALUMINUM BASE ALLOYS

Dear Mr. Markus:

Enclosed is one (1) copy of Section III of the Final Report of the subject contract covering the period September 29, 1961 to September 30, 1965. Section II of the captioned report has already been completed and distributed; Section I is being assembled into rough draft form for your approval.

Copies of Section III are being distributed according to the report's distribution list.

Very truly yours,

J. P. LYLE, JR., Assistant Chief  
Physical Metallurgy Division

JPL/mmk

Enclosure

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DEVELOPMENT OF ALUMINUM-BASE ALLOYS-Section III  
(Unclassified)  
FINAL REPORT

FOR THE PERIOD SEPTEMBER 29, 1961 to  
SEPTEMBER 30, 1965

CONTRACT NO. DA-36-034-ORD-3559RD  
MAY 31, 1966

by

A. P. HAARR

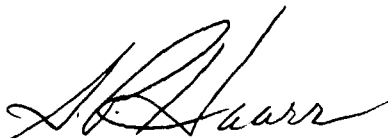
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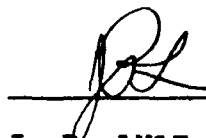
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DA PROJECT NO. 593-32-004

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### FOREWORD

New highs in the strengths of aluminum alloy products were the object of Contract No. DA-36-034-ORD-3559RD. This contract has been successfully completed. The final report covers all the work done on the contract since its beginning in September 1961, and is divided into three parts:

I. Outline of information on the most promising alloys. The preferred processes and compositions are described.

II. Fabricating Development.

The development of the processes which achieved the desired properties and products is described, and recommendations for further study are made.

III. Alloy Development.

The development of alloy compositions and thermal practices which achieved the desired properties is described, and recommendations for further study are made.



ABSTRACT - SECTION III

The objectives of this alloy development research program were: (1) aluminum alloys with 125,000 psi yield strength, (2) aluminum alloys with yield strengths at least 10% higher than commercial alloys with no sacrifice in resistance to stress corrosion cracking (SCC), and (3) determination of tensile properties at elevated and cryogenic temperatures, impact properties, tear properties, electrical conductivities, hardness, and fatigue strengths. All objectives have been met by APM (Aluminum Powder Metallurgy) extrusions made from prealloyed atomized powders using alloys which combine precipitation and dispersion hardening.

1. A yield strength of 124,000 psi was obtained in Alloy 50 containing Al, 9.8 Zn, 4.0 Mg, 0.8 Cu, 1.1 Mn, 1.0 Fe, 1.3 Ni, 0.01 Cr and 0.01 Ti after solution heat treating for 0.5 hours at 920°F, quenching at 2,000-25,000°F/sec., and aging at 225°F for 96 hours.
2. (a) - Alloy 87 containing Al, 7.6 Zn, 2.5 Mg, 1.1 Cu, 2.2 Fe, 2.3 Ni and 0.2 Cr, solution heat treated 2 hours at 860°F, quenched in cold water after aging for 6 hours at 250°F plus 8 hours at 330°F had a yield strength 11% higher than 7075-T7351 and did not fail in A.I. (alternate immersion) when stressed at 75% of Y.S.  
  
(b) - Alloy 71 containing Al, 9.2 Zn, 3.6 Mg, 0.6 Cu, 0.75 Co and solution heat treated for 2 hours at 860°F, quenched in cold water and after aging 24 hours at 250°F had a yield strength 14% higher than 7178-T651 and did not fail in A.I. when stressed at 25% of Y.S.  
  
(c) - Alloy 71 after aging 24 hours at 250°F plus 3 hours at 330°F had strengths equal to 7178-T651 and the threshold stress in A.I. was at least twice as high as 7178-T651.

The strengths of some APM alloys are higher than commercial alloys from -112°F to 350°F. Fatigue strengths of smooth specimens of those APM alloys are higher than commercial 7075-T6 and fatigue strengths of notched specimens are at least as high. Impact and tear properties are low but may be improved by further changes in composition, fabrication and heat treatment.

When heat treated to maximum strengths, dispersion hardeners raise strengths slightly but the elongation of such alloys is so low that the potential usefulness is very limited. When "overaged" to lower strengths, however, dispersion hardeners make a very important contribution to resistance to SCC.

The structures were studied by light and electron microscopy, X-ray diffraction and electron microprobe.

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## INTRODUCTION

The powder metallurgy approach to alloy development has several attractive features. Atomization of alloys from the melt permits the use of higher concentrations of alloy elements than is possible in ingot metallurgy where the cooling rate is slow relative to atomizing cooling rates. Thus compositions which result in such coarse constituents in ingots that properties or fabricability are impaired, and which result in ingot cracking, can be made by atomizing. The structure of atomized alloy powders is very similar to that of ingot except that powder structures are several orders of magnitude finer. This can be carried over into the products made from powder with the result that wrought products made from atomized alloys can have a very fine structure. Powder metallurgy makes possible the fabrication of dispersion hardened alloys and also alloys combining dispersion hardening with precipitation hardening.

Before this contract began, it was known that high strength aluminum alloy extrusions could be made from prealloyed atomized powder. The alloys fell into three classes: (1) dispersion hardened alloys which are characterized by a high liquid solubility and a low solid solubility at room and elevated temperatures; (2) precipitation hardened alloys which are characterized by a high liquid solubility, a high solid solubility at high temperatures and a low solid solubility at room and intermediate temperatures; and (3) alloys combining the characteristics of the dispersion hardened and precipitation hardened systems. Examples of the three classes from ARL work are:

Alloy	T.S.
(1) 4.5 Fe, 7.0 Ni	69,000 psi
(2) 6.0 Zn, 2.7 Mg, 1.6 Cu, .2 Cr	94,000 psi
(3) 5.7 Fe, 5.5 Ni, 5.3 Zn, 2.4 Mg, 1.7 Cu, .2 Cr	112,000 psi

Other examples can be cited from the work of S. G. Roberts (Ref. 1, 6, 7 and 8):

(1) 5.0 Mn, 2.0 Zr, 0.5 Ti, 0.5 V	59,000 psi
(2) 9.8 Zn, 3.2 Mg, 1.98 Cu, 0.2 Cr	114,000 psi
(3) 12.4 Zn, 2.5 Mg, 2.2 Cu, 3.3 Fe	122,000 psi

For the purposes of this investigation, Zn, Mg, and Cu are considered to be the elements which contribute the most to precipitation hardening. The dispersion forming elements include Mn, Fe, Ni, Cr, Ti, V, Zr, Co, Mo, and W.

It had been found in ARL investigations that high strength powder metallurgy alloys of the Al-Zn-Mg-Cu-Cr type sometimes had unusually good resistance to SCC (stress corrosion cracking) even in the short transverse direction when the alloys were heat treated and aged to their highest strengths, but the results were erratic. Roberts raised the question of SCC in alloys of this type because of their structure but reported no test results to support this speculation.

Roberts used extrusions which were too small to permit testing for transverse tensile properties. The extrusions used in ARL investigations prior to this contract were large enough to permit the determination of transverse properties, but low quality of the extrusions frequently vitiated the results. After the development of a method for making high quality extrusions, which is described

in detail in Section II of this report, it was possible to get reliable results in tensile tests and SCC tests in the transverse direction.

### OBJECT OF SECTION III

Develop an aluminum alloy with a yield strength of 125,000 psi.

Develop an aluminum alloy with a yield strength at least 10% higher than commercial alloys without a sacrifice in resistance to stress corrosion cracking.

Determine tensile properties at cryogenic and elevated temperatures, impact properties, tear properties, electrical conductivities, hardness, and fatigue properties.

### MATERIALS

1. Powders prepared as described in Section II of this report and having the compositions shown in Table I.
2. Two-inch diameter extrusions prepared as described in Section II - Fig. 6 and having the compositions shown in Table II.
3. 0.100 in. sheet prepared by rolling as described in Section II or machined from 2 inch dia. extrusions.

### RESULTS AND DISCUSSION

#### I. Aluminum Alloys With 125,000 psi Yield Strength

##### A. Selection of Alloys

Various approaches were used in the alloy selection. Alloys 1-3 were based on earlier work performed at ARL. Mr. S. G.

Roberts reported 0.25 in. dia. extrusions made from atomized powders of the same composition as Alloys 4-6 had very high strengths (Ref. 1). Compositions 7-18 contained two levels of Zn, 8% and 12%, and a fixed level of Mg and Cu to which various amounts of dispersion hardening elements were added. The base levels of Zn, Mg and Cu in these alloys were chosen from results of prior work at ARL.

Quaternary Alloys 19-31 constitute a broad survey of certain phase fields in the Al-Zn-Mg-Cu system (Table III). Some of these alloys are outside the limits of ingot metallurgy. Some of them also served as a base for evaluating the effects of dispersion hardening additions. Many of the earlier alloys lay in the  $\alpha + \beta$  and  $\alpha + M$  phase fields at 860°F (Ref. 2). The other phase fields investigated are listed in Table III.

Alloys 32-34 and 36-40 were suggested from observed trends. Alloy 35 was raised to a higher Mn level than Alloy 33, so that the Mn would exceed the solid solubility. Alloys 41 and 42 were selected to evaluate the effects of large amounts of Al-Mg-Cr constituents on tensile properties. Alloys 43-59 were selected on the basis of statistical analysis from data generated on Alloys 1-39. A discussion of this and other statistical analyses is found in Appendix A.

It was expected that the effect of omitting dispersion hardeners could be determined by comparing Alloys 60 and 61 with 39 and 50 (or 52) respectively. Alloy 60 is similar to Alloy 39 with a lower Mn level, 1.7% versus 0.77; Alloy 61 is similar to Alloys 50 and 52 without any dispersion hardeners present. However, the comparison was confused by unintentional variations in Zn and Mg.



Alloys 62-74 are modifications of Alloy 32. Alloys 75-84 are modifications of 7178. Alloys 85-87 and 90 are modifications of Alloy 34. Alloys 88 and 89 are the same compositions as 79 and 87 respectively, but the former are mixtures of powders instead of prealloyed powders.

## B. Properties of Extrusions

### 1. Tensile Properties

Table II summarizes the properties of the extrusions. Attention is called to the ultrasonic rating; over 75% of the sections examined exceeded SNT Class A with 95% meeting or exceeding Class A.

The densities of the extrusions, Table II, are probably essentially 100% of theoretical, based on the absence of porosity seen in metallographic examinations. The values in Table II were derived from the measured densities of the powders in Table I by multiplying the density of the powder by 1.014 after it was determined that the density of the extrusion was 1.1 to 1.7% higher than that of powder (Table IV). This difference between extrusions and powders is due to sealed pores in the atomized powder particles which are not filled by the liquid in the determination of density by the pycnometer method. This difference in density could also be due in part to techniques used to measure density of powder and extrusion and also to structural changes in fabrication and heat treatment. In Table IV it is seen that the density of the heat-treated and aged extrusions is about 0.3% less than that of the as-extruded material. This is typical of Al-Zn-Mg-Cu alloys produced in conventional ways (Ref. 3) and

gives further evidence of the soundness and low gas content of the extrusions made from powder by the process described in Section II.

The tensile properties of the alloys are compared in Table II in one or two heat treated conditions: (1) solution heat treated in 2 in. dia. sections for 2 hours at 860°F, quenched in cold water, aged for 24 hours at 250°F or (2) solution heat treated in 1 in. x 1 in. quadrants for 0.5 hours at 920°F, quenched in cold water, aged for 96 hours at 225°F. In general, the second heat treatment resulted in higher strengths. The highest yield strength for Heat Treatment #1 is 112 ksi for Alloy 38; the highest yield strength for Heat Treatment #2 is 117 ksi for Alloys 38 and 70. Incidentally, Alloys 73, 74, 76-78, 80, 81, 84, 88 and 89 were not tested after it became desirable to use available funds in other parts of the program.

The longitudinal tensile properties are generally uniform between the front and back of the extrusions; the greatest difference for Heat Treatment #1 is 5,200 psi. Transverse yield strengths were usually not determined since failure occurred before a 0.2% offset was reached. The strain followers were damaged in some cases due to premature failures, therefore only the tensile strengths were usually measured. The transverse tensile strength is much lower than the longitudinal value and greater variations exist between the front and back. The greatest differences are 63,200 psi for Alloy 66 and 39,100 psi for Alloy 9; these large differences are probably a result of internal defects. The higher property is probably more representative of the potential of the alloy.

The heat treatments reported in Table II do not necessarily develop the highest properties in these alloys. Examples of increases which can be obtained are shown in Table V where it is seen that

combinations of higher solution heat treat temperature, longer aging times, and lower aging temperatures tend to raise tensile strengths; the effectiveness of these treatments depends on alloy composition. Higher solution heat-treat temperatures, however, tend to increase cracking and splitting during quenching. The tensile strength of Alloy 39 was raised to 122 ksi by a solution heat treatment of 2 hours at 920°F followed by aging for 96 hours at 225°F. The time at solution heat treatment temperatures may also affect properties slightly as shown in Table VI, which suggests that highest strengths are obtained with shortest times.

Tensile and yield strengths increase with increasing quench rates as shown in Table VII, Figure 1, and Figure 2. The quench rates are only rough estimates, but the trend is at least qualitatively reliable. The high strength objective of this contract was substantially achieved by Alloy 50 having a tensile strength of 127,500 psi and a yield strength of 124,000 psi when solution heat treated for 0.5 hours at 920°F, quenched at 2,000-25,000°F/sec., and aged for 96 hours at 225°F. The highest Y.S.: density ( $1.17 \times 10^6$  in.) was found in Alloy 52 when solution heat treated for 0.5 hours at 920°F, quenched at approximately 25,000°F/sec., and aged for 96 hours at 225°F.

The compact preheat temperature can have an effect on quench sensitivity as shown in Figure 2 for Alloy 50. A 900°F preheat results in a higher quench sensitivity than a 1000°F preheat. The same trend was observed in Alloy 52.

## 2. Tensile Properties at Cryogenic and Elevated Temperatures

The ten alloys evaluated at cryogenic and/or elevated temperatures are listed in Table VIII. Only four alloys were tested below room temperature. Properties of 7075-T6 and X2020-T6 extrusions fabricated from ingot are included also for purposes of comparison. The effect of the temperature on tensile strengths is best illustrated in Figure 3. (Alloy 52 is the only APM alloy shown since the other alloys responded in a similar fashion.) The APM alloy has a definite strength advantage up to about 375°F, above which X2020-T6 takes over.

The effect of time at temperature on the tensile properties of Alloy 52 was also investigated; the results are given in Table IX and Figure 4 and 5. Again the APM alloy has the highest strength up to about 375°F at which time X2020-T6 surpasses it. The lack of data for Alloy 52 between 212°F and 400°F leaves some question as to whether this alloy has a curve as illustrated in Figure 3 or one more similar to that of 7075-T6.

## 3. Notch Toughness and Tear Resistance

The notch toughness and/or tear resistance of Alloys 38, 52, 62, 64 and 71 were determined. The former property was evaluated by use of the Izod impact test on heat treated 2 in. dia. extruded rod machined into test specimens as specified by ASTM Standard E-23-60T. The tear resistance was evaluated by means of a Kahn-Type tear test on 0.10 in. thick sheet. This sheet was produced by either rolling of 1 in. x 4-1/4 in. extruded slab or machining a "sheet type" specimen from a 2 in. dia. extrusion. This test measures the energy

necessary to initiate a crack in the specimen and the energy necessary to propagate this crack to complete failure and is described in Ref. 4.

The Izod impact tests revealed the brittleness of the APM alloys and the impact resistance is closer to that of casting alloys than to extruded material, Table X. The sheet rolled from extrusions was very notch sensitive, the energy required to initiate a crack ranging from 1.2 to 2.4 in.-lb. Once the crack was initiated, it propagated without additional energy being needed until complete failure occurs. This is much lower than the values usually obtained from sheet, and is even lower than values obtained for castings.

The tear resistance of certain step aged 0.10 in. thick sheet machined from 2 in. dia. extrusions compared favorably with commercial alloys, Table XI. The tear strength to yield strength ratios are generally lower than the commercial alloys because of the higher yield strengths obtained for the APM alloys. Alloy 71 looks promising when compared with 7178-T6.

#### 4. Electrical Conductivity

Heat treated slices of 2 in. dia. rod and rolled 1 in. x 4-1/4 in. extrusions were measured using the Magnatest type FM-100 Conductivity meter (Table XII). Conductivities follow patterns generally expected from compositions and thermal practices.

#### 5. Hardness

Hardness tests, both Brinell and Rockwell, were made on APM products to determine the values and to try to correlate a

possible relationship between hardness and tensile strength. The hardness values were higher than those usually obtained for aluminum but the test results were not reproducible. The data are summarized in Appendix B.

#### 6. Fatigue Tests

The fatigue endurance limits of certain alloys were investigated using a rotating beam specimen in both smooth and notched configurations. The results are discussed in Appendix C.

#### C. Effectiveness of Dispersion Hardening

One of the original objectives of this investigation was to determine the feasibility of obtaining high tensile strengths by combining dispersion hardening with precipitation hardening. That this approach is effective, at least to a limited extent, is seen from the fact that the highest yield strength obtained in the entire investigation was in an alloy containing significant amounts of Mn, Fe, and Ni in addition to Zn, Mg, and Cu. Furthermore, the highest Y.S. to density ratio was obtained in an alloy containing a large amount of Co in addition to Zn, Mg, and Cu.

It is well, however, to make a judgment as to the effectiveness of the combination of dispersion hardening and precipitation hardening by comparing that kind of alloy with precipitation hardened alloys using the results in Table II. The properties for Heat Treatment #1 in Table II have been arranged in groups according to Zn, Mg, and Cu content in Table XIII. It is seen that dispersion hardeners generally tend to raise tensile and yield strengths (at least up to a point) and to lower elongations.

It is also fairly clear that there is a maximum in the yield strength-amount of dispersoid relationship with that maximum depending on the Zn, Mg, and Cu content as well as on the specific dispersoids present. In all cases, however, the alloys containing dispersoids have such low elongations when heat treated to high strengths that the usefulness of the alloys will be very limited.

Another means of comparing the two types of alloys is on the basis of elongation at a given yield strength-to-density ratio. This is done in Table XIV in which it appears that the two types of alloys are about the same at Y.S.:density greater than  $0.95 \times 10^6$  and that the precipitation alloys tend to be more ductile at Y.S.:density values of 0.95 and lower.

An attempt was made to assess the effects of dispersion hardening elements on tensile properties by various computer analyses of results obtained in this investigation. While these were not very successful in predicting results, they are included in Appendix A for the record. The highest predicted yield strength for any powder metallurgy extrusion was 127,000 psi for Heat Treatment #1. It is interesting to compare this with a predicted maximum of 122,000 psi for 3/4 in. dia. extrusions made from ingot in another investigation. Allowing for the effect of difference in quench rate between 3/4 in. and 2 in. dia. rod, these analyses would indicate an advantage of 8-10 ksi for the powder metallurgy product. However, the very low elongations of all of the aluminum alloy materials in this strength range make these differences academic; in practice these high strengths are not obtained consistently, possibly due to the difficulties

associated with making tensile tests on extremely brittle materials.

The effectiveness of dispersion hardening in alloys which also contain precipitation hardening elements can be judged to some extent from the response of the alloy to factors affecting dispersion hardening. The strength of dispersion hardened alloys based solely on intermetallic compounds having low solid solubility at elevated temperatures, e.g.,  $\text{FeAl}_3$ ,  $\text{FeAl}_6$ ,  $\text{FeNiAl}_9$ ,  $\text{Co}_2\text{Al}_9$ ,  $\text{MnAl}_6$ , etc., is a function of the volume percent of the dispersed phase and an inverse function of the spacing between the particles of that phase. The interparticle spacing for a given volume percent can be minimized by keeping the times and temperatures at a minimum during preheating, fabrication and heat treatment, and by using finer atomized powders.

The effects of shorter preheat times and lower preheat temperatures were generally opposite to those expected as shown in Table XV. Die quenching, which minimizes times at elevated temperatures by eliminating a separate heat treating operation, also gave unexpected results as shown in Table XVI.

Finer powders gave higher longitudinal strengths than the normal powder in the case of Al-Zn-Mg-Cu alloys given Heat Treatment #2 in Table XVII, but the reverse was true for Al-Zn-Mg alloy and for both types of alloys for Heat Treatment #1. Finer powders (Table XVIII) may give higher transverse strengths than normal powders, but the wide scatter in data shown in Section II - Table VIII requires that this conclusion be treated with reservations.

#### D. Summary

Higher tensile strengths than ever reported before for Al



alloys have been achieved by combining dispersion hardening and precipitation hardening. These high strengths may not be obtainable with precipitation hardening alone. The extremely low ductility, low impact strength and low tear strength associated with these high tensile strengths and the specialized processes required to achieve them will seriously limit the usefulness of these alloys. Practical considerations probably would favor extrusions made from ingot over extrusions made from powder.

## II. Development of Aluminum Alloys With Yield Strength 10% Higher Than Commercial Alloys Without Sacrificing Resistance to Stress Corrosion Cracking

When it became apparent that alloys having yield strengths of 125,000 psi would have very low elongations, the emphasis of the investigation was changed. At the suggestion of Mr. Harold Markus of Frankford Arsenal, it was agreed to try to develop alloys and practices to make extrusions which had 10% higher yield strengths than currently available commercial extrusions, elongations of at least 5%, and resistance to SCC (stress corrosion cracking) at least equal to commercial alloys. This amounted to two targets depending on strength and degree of resistance to stress corrosion cracking, based on 7075-T7351 (the strongest commercial alloy with immunity to stress corrosion cracking) and on 7178-T651.

Alloy	Typical Tensile Properties (of Extrusions, 2-3" Diameter)					
	Longitudinal			Transverse		
	T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D
7075-T7351	75	66	11	62	56	4
7178-T651	92	84	8	83	74	5

<u>Targets</u>	<u>Longitudinal Tensile Properties</u>			<u>SCC*</u> <u>ksi</u>
	<u>T.S.</u> <u>ksi</u>	<u>Y.S.</u> <u>ksi</u>	<u>El. %</u> <u>in 4D</u>	
1	83	73	5	>44
2	101	92	5	7

\* Highest sustained tension stress in short transverse direction at which test specimen does not fail in the 3.5% NaCl alternate immersion test.

The attack on the targets was to be two pronged: (1) longer aging times and higher aging temperatures than were used to develop the highest tensile strengths, (2) alloys with lower amounts of dispersoids.

In the initial tests (Tables XIX and XX), it was found that these alloys responded to extended aging with much higher increases in elongation than were expected.

Systematic aging studies were made on a number of alloys using 2" diameter extrusions. An extensive effort was made to use Rockwell G hardness measurements to follow the aging, but it was found that the relation between hardness and strength was not very clear and that hardness values could not be interpolated accurately from existing hardness-aging time curves. (The hardness values are tabulated in Appendix B for the record.) Tensile property-aging time relationships for Alloys 2, 3, 4, 5, 6, 19, 20, 28, 33, 36, 38, 39, 49, 50, 52, 59, 60, 61, and 71 are tabulated in Tables XXI through XXXIX. Some of the data are plotted in Figures 6 through 11. The results of stress corrosion tests are given in Tables XXI through XLIII and Figures 12 through 34.

A summary of stress-corrosion cracking results is given in Table XLIV in which the alloys and thermal practices are listed in

order of decreasing stress levels.

The following alloys and aging treatments meet the strength and stress corrosion targets.

<u>Alloy</u>	<u>Aging</u>	<u>T.S.</u> <u>ksi</u>	<u>Longitudinal</u> <u>Y.S.</u> <u>ksi</u>	<u>El. %</u> <u>in 4D</u>	<u>SCC</u> <u>Stress</u> <u>ksi</u>	<u>Days to</u> <u>Failure</u>
<u>Target I: 7075-T7351 + 10% With No Failures at 75% Y.S.</u>						
		83	73	5	>48	OK 84
87	6 @ 250 + 8 @ 330	90	81	7	57	OK 84
87	16 @ 330	86	75	7	52	OK 84
79	6 @ 250 + 8 @ 330	86	79	8	54	OK 84
90	6 @ 250 + 8 @ 330	84	78	8	54	OK 84
<u>Target II: 7178-T651 + 10% Strength Improvement, Equal Stress, Corrosion, Resistance</u>						
		101	92	5	7-20	OK 84
71	24 @ 250	108	105	5	22	OK 84
<u>Target III: Better Stress Corrosion Resistance Than 7178-T651</u>						
		92	84	5	7-20	OK 84
71	24 @ 250	108	105	5	22	OK 84
90	24 @ 250	99	93	5	20	OK 28
71	24 @ 250 + 3 @ 330	94	88	7	40	OK 84

(The alloys were all SHT 2 hours at 860°F and immediately quenched in cold water prior to aging.) Target III was not included as one of the original two targets but was considered. A more complete summary is given in Table XLV which lists the composition of these alloys as well as alloys which may meet the targets with adjustments in aging practices.

The susceptibility of APM alloys to exfoliation was evaluated on 1 in. x 4-1/4 in. extruded sections of Alloy 52, S.No. 293690 using an improved accelerated test of acetic acid-sodium chloride intermittent spray at 95°F for a 2 week exposure (Ref. 5). The extrusions were sawed such that surface and mid-plane surfaces would be exposed in the as-extruded (-W) temper, given Heat Treatment #1 (SHT 2 hours at 860°F, C.W.Q., Aged 24 hours at 250°F) and given Heat Treatment #1 with an additional step aging of 4 hours at 330°F. All heat treatments were conducted prior to the sample preparation. Generally, the mid-plane and surface of the specimens were covered with small shallow pits and a mild form of corrosion. There was no indication of exfoliation on any of the panels. Extrusion lines were visible on the panel surface after exposure. Heat Treatment #1 panels resisted the corrosive environment best. The -W temper pieces exhibited slightly more pitting; the step-aged panels had more pits than the as-extruded panels.

### III. Structure

The structures of atomized alloy powders and extrusions made from them are fine relative to ingot and ingot extrusions, and the structures of extrusions are coarser than the powders from which they were made (Figures 35-37).

In the early stages of this investigation, large constituent particles which were present in some of the extrusions were analyzed by electron microprobe with the results shown in Table XLVI. It was concluded from the analyses that Fe, Ni, Co, and Mn were desirable elements for dispersion hardening because they

tended to be absent from (or present in relatively slight amounts in) the large constituent particles.

Guinier X-ray diffraction also revealed that the Zn, Mg, Cu, and Mn, were present in the same phases in powder metallurgy extrusions (Table XLVII) that occur in products made from ingot.

After it became apparent that Alloys 34, 87, 52, and 71 had good strengths combined with good resistance to SCC, a somewhat more extensive study was made of the structure.

Extrusions of these alloys all had essentially the same structures when viewed by the light microscope, and Figure 38 showing Alloy 71 in the -T6 condition serves as an example. The two-step aged condition, Figure 39, shows no obvious difference in structure, but may contain a little more precipitate.

The electron microprobe was used to analyze 14 to 15 of the larger particles in extrusions of Alloys 34, 87, 52 and 71, as illustrated by the black phases in Figure 38, Tables XLVIII and XLIX. It was determined that some of the particles were non-metallic, i.e., their composition could not be entirely accounted for in the microprobe analysis. It should be noted that the percentage figures reported for the non-metallic particles cannot be considered to be quantitatively correct because of uncompensated absorption effects and are given only to indicate order of magnitude. The percentage figures for the metallic particles are somewhat more meaningful because approximate absorption corrections were made; quantitative accuracy is still questionable. Sixty percent of the larger particles in Alloy 34 were approximately 40% Zr and 10% Si (probably  $ZrSiO_4$ ) while 29% of the

particles in Alloy 87 had a similar composition. No Zr was reported in the semiquantitative chemical analysis. These particles are probably part of the Zirconite paste wash which was applied to the crucible prior to melting a charge. The particles flaked off or were chipped off during stirring to become part of the atomized product. The higher  $\text{ZrSiO}_4$  concentration occurred in the larger charges. The presence of Cr, Fe, Si, and Ni in certain of the large constituent particles indicate more work should be done to minimize growth of these particles. Mn and Co do not occur as large constituent particles and are desirable. The effect of Zn and Mg varies; sufficient data are not available to come to any conclusions at this time.

Alloys 34, 87, 52, and 71 were also analyzed by the Guinier X-ray diffraction technique. Not only were attempts made to identify the phases present but also to determine the formation of additional phases when progressing from the powder to -T6 extrusion and finally after prolonged aging. Only 2 or 3 phases were indicated as being present in the powder, Table L,  $\text{Al}$ ,  $\text{Mg}_2\text{Si}$  and either  $\text{FeNiAl}_9$  or  $\text{Co}_2\text{Al}_9$ , depending on the alloy. After the powder is compacted, extruded and given a -T6 heat treatment, the structure becomes quite complex. Additional phases appear including one or more phases which cannot be identified. Prolonged aging results in an apparent increase in the amount of  $\text{FeNiAl}_9$  but no change in the amount of  $\text{CoAl}_9$ . Prolonged aging also results in the appearance of an  $\text{M}'$  precipitate. Other phases stay relatively unchanged.

Examination of the powders by electron microscope indicates the microstructures of Alloys 34 and 87 are somewhat coarser than

that of Alloys 52 and 71, Figures 40 to 43. Alloy 71, Figure 43, appeared to have a less uniform structure in terms of constituent dispersion than the other powders.

The microstructures of Alloys 34 and 37 extrusions are quite similar, Figures 44, 45, 46, and 47. Prolonged aging of both these alloys results in an increase in the large dark particles labeled  $MgZn_2$ . The gray particles are believed to be  $FeNiAl_9$  and the number does not change appreciably. The smaller, more spherical particles are probably the E-phase ( $Al_{12}MgCr$ ) dispersoid. The  $Mg_2Si$  precipitate would be present in small quantity and is not easily identified. A very fine, light background precipitate also appears to increase during step aging. This is similar to the  $\theta'$  phase which occurs in alloys containing Cu, but a positive identification was not made. This could also be the  $M'$  or unidentified phase noted in X-ray diffraction.

Alloys 52 and 71 are also similar in composition, but the amount of precipitate present in Alloy 52 is greater than for any of the other alloys, Figures 48, 49, 50 and 51. The differences between the  $MgZn_2$  and the  $Mg_3Zn_3Al_2$  phases are indistinguishable in the electron micrographs. The  $Co_2Al_9$  dispersoid is believed to be the light gray angular particles. Alloy 71 exhibits noticeable grain (or subgrain) boundary precipitation, especially after step aging. Note the increase in precipitate around the  $Co_2Al_9$  particles.

Small phases within the larger phases are noted in Figures 44 to 51. The compositions of the larger phases could be influenced by these smaller included phases when analyzed by the microprobe.

The microprobe has also been used to investigate tensile specimen fractures of two specimens of Alloy 59 in which low strengths and low elongations had been observed. The origin of failure appeared to be a region containing black inclusions. Test results indicate the black spot inclusions have a high Mg content, a moderate amount of Al and low-to-trace amounts of Zn, Mn, Ni, Fe and Cu. The level of each of these elements, with the exception of Mg, is far less in the dark region than in the surrounding region of the fracture. A thin layer of Mg-rich material, a spinel or possibly MgO, is present in the "black spot" regions. Thus low strengths and elongations can be the result of greater than normal oxidation.

#### CONCLUSIONS

1. The longitudinal tensile properties of certain APM alloys exceeded 100 ksi and were uniform along the length of the extrusion. The transverse tensile properties were lower than the longitudinal and failure frequently occurred before the yield strength at 0.2% offset was reached.

2. The ductility of the alloys generally decreased as the strength increased.

3. Alloy 50, Al, 9.8% Zn, 4.0% Mg, 0.8% Cu, 1.1% Mn, 1.0% Fe, 1.3% Ni, 0.01% Cr and 0.01% Ti, had the highest strength values with a yield strength of 124,400 psi and a tensile strength of 127,500 psi.

4. Alloy 52 containing 10.0% Zn, 4.0% Mn, 0.9% Cu, 0.4% Mn, 0.02% Ti, 0.01% Cr and 1.5% Co had the highest yield strength to density ratio,  $1.174 \times 10^6$  in.



5. The extrusion densities are higher than the powder densities due to the inability of the testing fluid to completely fill all voids and crevices.

6. The difference in the densities of the extrusions tested in the -F and -T6 conditions is similar to that of conventional wrought alloys.

7. Compact density and compact preheat temperature affect the tensile properties of a fabricated product from a given alloy.

8. The tensile strengths of an alloy increase with the quench rate. Quench rates are determined by the specimen size and the quenching medium.

9. Alloy composition affects the quench sensitivity of alloys.

10. Compact preheats of 1000°F result in extrusions with less quench sensitivity than 900°F preheats.

11. Elevated and cryogenic tensile tests on selected alloys indicate the properties are superior to conventional alloys up to 350°F. Above this temperature the tensile strength decreases rapidly.

12. Increasing the time at elevated temperatures from 1/2 to 100 hours results in a lowering of strengths and an improvement on the ductility. Comparison with other alloys for similar time at temperatures indicates that Alloy 52 again has superior properties up to about 350°F.

13. The tear strength values, while lower than some commercial alloys, can be improved by prolonged aging of the more ductile alloys.

14. The poor notch sensitivity of these extrusions is indicated by the low Izod impact test values.

15. The conductivity of these alloys varies depending on composition and aging practice.

16. The alloys are generally harder than commercial APM alloys.

17. Attempts to correlate hardness or conductivity data and tensile properties have not been successful.

18. The fatigue endurance limit on smooth specimens of Alloys 52, 62, 64, and 71 are greater than 7075-T6 and the data points generally fall above the upper scatter band limit for 7075-T6.

19. Dispersion hardeners generally tend to raise the tensile and yield strengths and lower the elongation.

20. The alloys respond to prolonged aging by increased ductility and resistance to stress corrosion cracking. The strengths decrease.

21. Alloy 71 responds very well to overaging. The strengths are almost as high as those obtained for Alloy 52, however, the material is much more ductile.

22. Alternate immersion stress corrosion tests have a varied response on the short transverse specimens from extrusions. Alloy content and heat treatments determine the effect.

23. Stress corrosion resistance of alloys given prolonged aging and containing high Fe and Ni (Alloys 34, 87 and 90) resulted in no failures at stress levels significantly greater than typical values of 7075-T73. Heat Treatment #1 resulted in very high strengths with better stress corrosion resistance than 7075-T6 or 7178-T6.

24. Alloys 52 and 71, Co containing series, had stress corrosion resistance better than that obtained for 7075-T73 + 10% after prolonged aging. Although not as good as the Fe-Ni series, the ductility of these alloys is greater.

25. No indication of exfoliation corrosion was present in Alloy 52. The general effect of exfoliation corrosion on APM material is not known.

26. The maximum dispersion strengthening effect was not achieved from the insoluble elements because the constituent particles in the extrusions were coarsened considerably over those in the atomized powder.

27. The electron microprobe indicated that the coarse constituents contained large amounts of Cr, Ti, V, Zr, Mo and W. These alloying elements are therefore considered undesirable in large amounts.

28. Fe, Ni, Co and Mn are desirable elements for dispersion hardeners since they are not present as large constituent particles.

#### FUTURE WORK

A total of 90 alloys have been partially evaluated. The results have indicated that at least two families of alloys should be given additional testing, namely, Al-Zn-Mg-Co-Cu (Alloys 52 and 71) and Al-Zn-Mg-Cu-Fe-Ni (Alloys 34, 87 and 90). Additional testing has shown that these alloys are indeed promising but many unanswered questions are still present, i.e., alloy limits, purity, fabrication techniques, more complete stress corrosion data in both alternate immersion and atmospheric tests, etc. Also, means of

improving some of the poor properties, i.e., ductility and tear strength, must be made. Both alloy systems should be included as part of an extensive program to obtain a high strength commercial alloy.

The alloys obtained by use of statistical analyses should be produced and fabricated to determine how accurate the model was and also to improve the model.

Other families of alloys could be included for further evaluation, however, the emphasis should be directed toward the alloys discussed in the earlier paragraphs.

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TABLE I

## COMPOSITION AND PROPERTIES OF POWDERS

Alloy No.	Series	Sub No.	Material Analyzed	Zn	Mg	Cu	Mn	Fe	Ni	Cr	Ti	V	Zr	Co	Mo	W	Si	Al <sub>2</sub> O <sub>3</sub> (a)	Powder Density lb./in. <sup>3</sup> (b)	Powder Size MMD, microns
1			Intended Powder	9.86	3.49	1.02	0.47												.1014	44
1	A	267948	Intended Powder	10.0	3.53	1.04	0.50	0.12		0.00	0.00							0.06	.1014	
1	A	277373	Extrusion	9.94	3.53	1.07	0.52	0.08		0.01	0.00							0.05	---	
2			Intended Powder	12.0	3.50	1.50	0.50												.1038	54
2	A	267949	Intended Powder	12.1	3.48	1.51	0.54	0.13		0.00	0.01							0.08	.1038	
3			Intended Powder	12.30	4.05	1.53	0.50												.1038	27
3	A	267970	Intended Powder	12.3	4.03	1.56	0.53	0.13		0.00	0.01							0.08	.1038	
4			Intended Powder	9.76	3.24	1.99				0.20									.1027	39
4	A	267971	Intended Powder	9.60	3.19	1.99	0.00	0.16		0.23	0.01							0.05	.1027	
5			Intended Powder	10.36	3.08	2.03	1.74			0.20									.1047	37
5	A	267972	Intended Powder	10.3	2.93	2.07	1.80	0.21		0.14	0.00							0.11	.1047	
6			Intended Powder	11.25	2.45	2.01				0.20									.1015	39
6	A	267973	Intended Powder	11.2	2.44	2.03	0.01	0.16		0.17	0.01							0.06	.1015	
7			Intended Powder	8.0	3.5	1.5			2.5										.1043	26
7	A	267960	Intended Powder	7.92	3.41	1.52	0.01	2.50	2.00									0.02	.1043	
8			Intended Powder	8.0	3.5	1.5	1.5	1.63	1.40										.1042	44
8	A	267961	Intended Powder	7.88	3.46	1.50	1.58	1.63	1.40									0.06	.1042	
9			Intended Powder	8.0	3.5	1.5				0.5	0.5	0.5	0.5						.1023	38
9	A	267962	Intended Powder	7.88	3.54	1.60	0.01	0.13		0.48	0.38	0.41	0.44					0.07	.1023	
10			Intended Powder	8.0	3.5	1.5				2.0	0.8	2.0	1.0						.1050	39
10	A	267963	Intended Powder	7.91	3.54	1.51	0.04	0.15		2.13	0.89	1.73	0.85					0.08	.1050	
11			Intended Powder	8.0	3.5	1.5								1.5	1.5	1.5			.1033	37
11	A	267964	Intended Powder	8.03	3.54	1.53	0.01	0.10						1.48	1.19	0.39		0.10	.1033	
11	A	277383	Extrusion	7.55	3.54	1.52	0.01	0.09						1.37	1.55	0.48		0.08	---	
12			Intended Powder	8.0	3.5	1.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5			.1046	41
12	A	267965	Intended Powder	7.81	3.53	1.56	0.61	0.51	0.55	0.53	0.44	0.39	0.56					0.09	.1046	
13			Intended Powder	12.0	3.5	1.5			2.5										.1081	37
13	A	267966	Intended Powder	11.7	3.43	1.52	0.02	2.43	2.23									0.05	.1081	
14			Intended Powder	12.0	3.5	1.5	1.5	1.64	1.45										.1083	35
14	A	267967	Intended Powder	11.8	3.47	1.52	1.65	1.64	1.45									0.07	.1083	
15			Intended Powder	12.0	3.5	1.5				0.5	0.5	0.5	0.5						.1054	33
15	A	267968	Intended Powder	11.4	3.40	1.50	0.02	0.14		0.54	0.51	0.46	0.55					0.07	.1054	

(Continued)

TABLE I (Cont'd.)

Alloy No.	Spec. No.	Int'l. Analysis	Cu	Fe	Mn	P	S	Cr	Ni	V	Zr	Co	Mo	V	Si	1-2-3-4	Powder Density lb./in. <sup>3</sup> (b)	Powder Blow V.D. lb./sq.in.
5	—	Intended Powder	12.0	3.5	1.5	—	—	2.0	0.8	2.0	1.0	—	—	—	—	—	.1080	—
6	267969	—	12.0	3.54	1.52	0.14	0.13	2.18	1.06	1.89	0.98	—	—	—	—	—	.1080	—
17	—	Intended Powder	12.0	3.5	1.5	—	—	—	—	—	—	1.5	1.5	1.5	—	—	.1087	41
17	267970	—	11.9	3.46	1.55	0.00	0.10	—	—	—	—	1.48	1.10	0.54	0.08	—	.1087	—
18	—	Intended Powder	12.0	3.5	1.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	0.5	—	—	.1090	39
18	267971	—	11.7	3.38	1.52	0.62	0.54	0.54	0.45	0.39	0.54	0.50	0.50	0.50	0.11	—	.1090	—
19	—	Intended Powder	7.6	3.0	1.5	—	—	—	—	—	—	—	—	—	—	—	.1004	42
19	267972	—	7.49	2.97	1.70	0.00	0.17	—	—	—	—	—	—	—	—	—	.1004	—
20	—	Intended Powder	7.9	3.6	2.3	—	—	—	—	—	—	—	—	—	—	—	.1007	43
20	267973	—	7.74	3.39	2.33	0.00	0.17	—	—	—	—	—	—	—	—	—	.1007	—
21	—	Intended Powder	8.0	5.6	7.5	—	—	—	—	—	—	—	—	—	—	—	.1079	42
21	267974	—	7.92	5.40	7.44	—	—	—	—	—	—	—	—	—	—	—	.1079	—
22	—	Intended Powder	8.0	8.5	15.0	—	—	—	—	—	—	—	—	—	—	—	.1085	43
22	267975	—	7.93	8.39	14.9	0.00	0.13	—	—	—	—	—	—	—	—	—	.1085	—
23	—	Intended Powder	8.0	10.3	20.0	—	—	—	—	—	—	—	—	—	—	—	.1130	34
23	267976	—	7.71	10.1	20.1	0.00	0.12	—	—	—	—	—	—	—	—	—	.1130	—
24	—	Intended Powder	8.0	7.5	20.0	—	—	—	—	—	—	—	—	—	—	—	.1151	27
24	267977	—	7.76	7.16	20.0	0.00	0.14	—	—	—	—	—	—	—	—	—	.1151	—
25	—	Intended Powder	—	10.8	2.3	—	—	—	—	—	—	—	—	—	—	—	.0927	39
25	267978	—	—	10.6	2.37	0.00	0.19	—	—	—	—	—	—	—	—	—	.0927	—
26	—	Intended Powder	3.0	7.5	2.3	—	—	—	—	—	—	—	—	—	—	—	.0944	30
26	267979	—	2.95	7.41	2.33	0.00	0.17	—	—	—	—	—	—	—	—	—	.0944	—
27	—	Intended Powder	7.0	4.0	2.1	—	—	—	—	—	—	—	—	—	—	—	.1000	38
27	267980	—	6.90	3.93	2.33	0.00	0.14	—	—	—	—	—	—	—	—	—	.1000	—
28	—	Intended Powder	10.0	4.0	2.3	—	—	—	—	—	—	—	—	—	—	—	.1028	34
28	267981	—	9.77	3.83	2.28	0.00	0.16	—	—	—	—	—	—	—	—	—	.1028	—
29	—	Intended Powder	14.0	4.4	2.3	—	—	—	—	—	—	—	—	—	—	—	.1054	42
29	267982	—	13.7	4.33	2.33	0.00	0.12	—	—	—	—	—	—	—	—	—	.1054	—
30	—	Intended Powder	9.0	15.0	—	—	—	—	—	—	—	—	—	—	—	—	.0928	27
30	267983	—	7.90	14.6	0.02	0.00	0.12	—	—	—	—	—	—	—	—	—	.0928	—

(continued)



TABLE 1 (Cont'd.)

Alloy No.	Section	St. No.	Material / Intended	Al	Si	Fe	Mn	P	S	Cr	Ni	Cu	Mo	Co	W	V	Zr	Bi	As	Pb	Sn	Ag	Cd	Other	Powder Density $10^3/\text{in}^3$ (b)	Powder Size MMD - microns
31	A	2677964	Intended Powder	3.0	15.0	2.3	0.00	0.13	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.0922	31
32	A	2677964	Intended Powder	2.94	14.6	2.38	0.00	0.13	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1011	44
33	A	2677964	Intended Powder	6.45	2.70	0.41	0.00	0.10	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1018	41
34	A	2677964	Intended Powder	6.59	2.71	0.44	0.00	0.10	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1018	41
35	A	2677964	Intended Powder	9.15	3.44	1.17	1.10	0.15	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1024	45
36	A	2677964	Intended Powder	8.95	3.47	1.18	1.13	0.15	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1024	45
37	A	2677964	Intended Powder	8.0	2.5	1.0	—	3.25	5.0	0.10	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1074	44
38	A	2677964	Intended Powder	7.79	2.50	1.00	0.02	3.20	4.92	0.19	0.02	—	—	—	—	—	—	—	—	—	—	—	—	—	.1074	44
39	A	2677964	Intended Powder	7.78	2.53	1.03	0.04	3.93	5.34	0.19	0.01	—	—	—	—	—	—	—	—	—	—	—	—	—	.1074	44
40	A	2677964	Intended Powder	7.77	2.48	1.03	0.02	3.49	4.94	0.09	0.01	—	—	—	—	—	—	—	—	—	—	—	—	—	.1074	44
41	A	2677964	Intended Powder	9.15	3.44	1.17	1.10	0.15	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1024	45
42	A	2677964	Intended Powder	9.01	3.41	1.16	1.13	0.19	0.00	0.01	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1024	45
43	A	2677964	Intended Powder	11.0	3.7	2.0	1.7	0.13	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1061	35
44	A	2677964	Intended Powder	10.7	3.82	2.04	1.66	0.20	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1061	35
45	A	2677964	Intended Powder	10.6	4.36	1.97	1.68	0.20	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1044	22
46	A	2677964	Intended Powder	6.5	3.5	—	2.0	4.0	6.0	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1112	43
47	A	2677964	Intended Powder	8.43	3.52	0.02	2.06	3.32	6.05	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1112	43
48	A	2677964	Intended Powder	6.5	3.5	—	1.0	3.0	3.0	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1056	39
49	A	2677964	Intended Powder	8.36	3.62	0.02	1.00	0.91	3.10	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1056	39
50	A	2677964	Intended Powder	8.39	3.56	0.02	1.02	0.76	2.80	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1056	39
51	A	2677964	Intended Powder	8.35	3.33	0.00	1.06	1.08	2.96	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1035	22
52	A	2677964	Intended Powder	8.48	3.33	0.01	1.08	1.21	2.95	0.00	0.01	—	—	—	—	—	—	—	—	—	—	—	—	—	.1045	24
53	A	2677964	Intended Powder	11.0	3.0	2.0	1.7	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1045	24
54	A	2677964	Intended Powder	10.9	4.09	2.04	1.74	0.33	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1053	35
55	A	2677964	Intended Powder	10.9	4.90	2.00	1.79	0.19	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1046	24
56	A	2677964	Intended Powder	8.5	5.0	—	1.0	1.0	0.72	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1044	42
57	A	2677964	Intended Powder	8.42	5.15	0.02	1.02	0.72	3.00	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1044	42
58	A	2677964	Intended Powder	8.0	5.0	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1044	42
59	A	2677964	Intended Powder	7.96	4.62	0.01	—	0.15	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1013	28
60	A	2677964	Intended Powder	8.0	4.3	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1013	28
61	A	2677964	Intended Powder	7.63	3.92	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	—	.1008	32

(Continued)

TABLE I (Cont'd.)

Alloy No.	Series	S <sub>1</sub> No.	Material Analyzed	Composition, %													Powder Density lb./in. <sup>3</sup> (b)	Powder Size ISO-Microns				
				Zn	Mg	Cu	Mn	Fe	Ni	Cr	Ti	V	Zr	Co	Mo	W			Si	Al <sub>2</sub> O <sub>3</sub> (a)		
43	A	278105	Intended Powder	12.0	4.5	1.0	1.0	—	—	0.20	—	—	—	0.35	—	—	—	—	0.09	0.28	.1041	18
43			—	12.0	4.66	0.90	0.86	0.14	0.01	0.20	0.01	—	—	0.23	—	—	—	—	—	—	—	—
44	A	278106	Intended Powder	12.0	4.5	1.0	2.4	—	—	0.25	—	—	1.00	—	—	—	—	0.24	0.33	.1077	20	
44			—	11.2	4.96	0.90	2.53	0.24	0.03	0.23	0.01	—	—	0.93	—	—	—	—	—	—	—	—
45	A	278107	Intended Powder	12.0	4.5	1.0	2.8	—	—	0.75	—	—	0.40	—	—	—	—	0.16	0.41	.1067	19	
45			—	11.7	4.58	0.90	3.02	0.21	0.01	0.75	0	—	—	0.46	—	—	—	—	—	—	—	—
46	A	278108	Intended Powder	12.0	4.5	1.0	1.1	—	—	1.0	—	—	0.80	—	—	—	—	0.07	0.42	.1054	23	
46			—	11.4	4.61	0.93	0.57	0.13	0.09	1.12	0.01	—	—	1.02	—	—	—	—	—	—	—	—
47	A	278109	Intended Powder	10.0	4.0	0.4	2.0	1.5	5.0	—	—	—	—	—	—	—	—	0.13	0.25	.1086	27	
47			—	9.82	4.20	0.42	2.05	1.57	4.56	0.09	0.01	—	—	—	—	—	—	—	—	—	—	—
48	A	278110	Intended Powder	15.5	4.7	—	1.75	—	—	0.00	—	—	—	—	—	—	—	0.10	0.30	.1064	23	
48			—	15.2	4.92	0.01	1.81	0.19	0.02	0.00	0.01	—	—	—	—	—	—	—	—	—	—	—
49	A	278111	Intended Powder	15.5	4.7	2.0	1.75	—	—	—	—	—	—	—	—	—	—	—	—	—	—	24
49			—	14.8	4.92	1.77	1.71	0.16	0.00	0.00	0.00	—	—	—	—	—	—	—	0.10	0.28	.1080	—
49	C	283272	—	15.3	4.64	2.02	1.87	0.21	0.00	0.00	0.01	—	—	—	—	—	0.08	0.63	.1085	—	—	
50	A	278112	Intended Powder	10.0	4.0	0.9	1.3	0.9	1.5	0.01	0.01	—	—	—	—	—	—	—	—	—	—	22
50			—	9.80	4.02	0.78	1.10	1.00	1.34	0.01	0.01	—	—	—	—	—	—	—	0.09	0.81	.1053	23
50	C	283273	—	9.95	4.11	0.72	1.39	1.11	1.44	0.01	0.02	—	—	—	—	—	0.06	0.69	.1053	—	—	
50	Special	284130	—	9.7	3.86	0.90	1.26	0.85	1.51	—	0.01	—	—	—	—	—	0.08	—	—	—	—	
51	A	278113	Intended Powder	10.0	4.0	0.9	1.3	0.9	1.5	0.01	0.01	—	—	1.5	—	—	—	—	—	—	—	22
51			—	9.55	4.12	0.75	1.23	1.02	1.37	0.01	0.01	—	—	—	1.34	—	—	—	0.08	0.75	.1059	—
52	A	278114	Intended Powder	10.0	4.0	0.9	—	—	—	0.01	0.01	—	—	—	—	—	—	—	—	—	—	22
52			—	9.65	4.14	0.83	0.04	0.13	0.03	0.01	0.01	—	—	—	1.5	—	—	—	0.04	0.73	.1034	24
52	C	283274	—	9.79	4.01	0.92	0.01	0.14	0.01	0.01	0.02	—	—	1.38	—	—	0.06	0.75	.1033	—	—	
52	Special	284131	—	10.1	4.0	0.93	—	0.1	0.02	—	—	—	0.01	1.49	—	—	0.3	—	—	—	—	
53	A	278115	Intended Powder	10.0	4.0	—	2.4	1.3	4.9	0.04	0.04	—	—	—	—	—	—	—	—	—	—	25
53			—	9.55	4.13	0.02	2.43	1.40	4.82	0.04	0.04	—	—	—	—	—	—	—	0.10	0.71	.1085	—
54	A	278116	Intended Powder	10.0	4.0	1.6	2.9	1.5	1.0	0.05	0.04	—	—	—	—	—	—	—	—	—	—	23
54			—	9.45	3.98	1.39	2.88	1.61	0.89	0.05	0.04	—	—	—	—	—	—	—	0.11	0.60	.1070	—
55	A	278117	Intended Powder	10.0	4.0	1.6	1.0	1.7	2.6	0.07	0.05	—	—	—	—	—	—	—	—	—	—	23
55			—	9.92	4.18	1.47	0.86	1.85	2.41	0.07	0.05	—	—	—	—	—	—	—	0.07	0.73	.1071	—

(Continued)

TABLE I (Cont'd.)

Alloy No.	Series	S. No.	Material Analyzed	Zn	Pb	Cu	Fe	Mn	Cr	Composition, %	Co	Mo	V	Si	Al <sub>2</sub> O <sub>3</sub> (a)	Powder Density lb/in. <sup>3</sup> (b)	Powder Size R <sub>90</sub> - microns
56			Intended	10.0	4.0	1.8	1.9	—	2.6	0.11	0.07	—	—	—	—	—	—
56	A	278118	Powder	9.76	4.28	1.67	1.99	0.21	2.50	0.11	0.07	—	—	0.09	0.66	.1065	22
57			Intended	10.0	4.0	2.1	1.9	1.3	2.9	0.00	0.11	—	—	—	—	—	—
57	A	278119	Powder	9.55	4.02	1.92	2.00	1.24	2.98	0.00	0.10	—	—	0.08	0.75	.1081	25
58			Intended	10.0	4.0	2.7	2.0	1.3	3.6	0.04	—	—	—	—	—	—	—
58	A	278120	Powder	10.0	4.12	2.51	2.18	1.38	3.24	0.04	0.01	—	—	0.09	0.61	.1096	25
59			Intended	10.0	4.0	1.5	1.5	1.0	4.0	—	—	—	—	—	—	—	—
59	A	277918C	Powder	10.2	3.85	1.57	1.62	1.03	4.10	—	—	—	—	0.10	0.28	.1068	26
60			Intended	11.0	5.0	2.0	0.7	—	—	—	—	—	—	—	—	—	—
60	A	283674	Powder	11.6	5.50	1.82	0.72	0.13	—	—	—	—	—	0.05	0.33	.1035	33
61			Intended	10.0	4.0	0.9	—	—	—	—	—	—	—	—	—	—	—
61	A	283675	Powder	10.4	4.48	0.76	0.00	0.09	—	—	—	—	—	0.03	0.45	.1018	26
62			Intended	10.0	4.0	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
62	A	283925	Powder	10.0	4.03	0.94	—	—	—	0.01	0.03	—	—	—	0.31	.1022	42
63			Intended	10.0	4.0	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
63	A	283926	Powder	10.1	3.95	0.94	0.00	0.1	0.07	0.01	0.03	0.01	—	0.07	—	.1029	50
64			Intended	9.0	4.0	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
64	A	283927	Powder	8.9	4.06	0.94	—	—	—	0.01	0.03	—	—	—	0.31	.1024	48
65			Intended	11.0	4.0	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
65	A	283928	Powder	11.1	4.13	1.02	—	—	—	0.01	0.03	—	—	—	0.29	.1038	43
66			Intended	10.0	3.5	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
66	A	283929	Powder	10.0	3.56	0.94	—	—	—	0.01	0.03	—	—	—	0.30	.1034	43
67			Intended	10.0	4.5	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
67	A	283930	Powder	10.3	4.67	0.94	—	—	—	0.01	0.03	—	—	—	0.29	.1021	47
68			Intended	10.0	4.0	1.5	—	—	—	0.01	0.02	—	—	—	—	—	—
68	A	283931	Powder	9.9	3.87	1.50	0.00	0.1	0.06	0.01	0.03	0.01	—	0.06	—	.1033	37
69			Intended	10.0	4.0	0.5	—	—	—	0.01	0.02	—	—	—	—	—	—
69	A	283932	Powder	9.8	4.14	0.65	—	—	—	0.01	0.03	—	—	—	0.35	.1029	41
70			Intended	11.0	4.5	1.5	—	—	—	0.01	0.02	—	—	—	—	—	—
70	A	283933	Powder	10.8	4.64	1.31	—	—	—	0.01	0.03	—	—	—	0.31	.1047	42
71			Intended	9.0	3.5	0.5	—	—	—	0.01	0.02	—	—	—	—	—	—
71	A	283934	Powder	9.2	3.62	0.63	—	—	—	0.01	0.03	—	—	—	0.26	.1016	40
71	C	293303	Powder	9.29	3.58	0.54	—	—	—	0.02	0.02	—	—	—	0.45	.1020	23
72			Intended	10.0	4.0	0.9	—	—	—	0.01	0.02	—	—	—	—	—	—
72	A	283935	Powder	10.1	4.09	0.93	—	—	—	0.01	0.03	—	—	0.5	0.35	.1027	35

(Continued)

TABLE I (Cont'd.)

Alloy No.	Series	S. No.	Material Analyzed	Composition, %										Powder Density lb/in. <sup>3</sup> (b)	Powder Size MMD-microns
				Si	Fe	Mn	Cu	Ni	Cr	Ti	V	Al	Al <sub>2</sub> O <sub>3</sub> (a)		
73	A	284202	Powder	10.2	4.10	0.94	--	--	0.01	0.03	--	--	0.29	.1032	41
74	A	284203	Powder	9.8	4.14	1.36	--	--	0.01	0.03	--	--	0.34	.1033	44
75	A	--	Intended	6.8	2.7	2.0	--	1.0	0.3	--	--	--	--	--	31
75	A	307586	Powder	7.06	2.79	2.04	0.01	0.93	4.72	0.01	--	--	0.08	.1056	31
76	--	--	Intended	6.8	2.7	2.0	--	3.0	0.3	--	--	--	--	--	31
76	A	307587	Powder	6.72	2.68	2.08	0.02	3.51	6.74	0.01	--	--	0.05	.1098	31
77	--	--	Intended	6.8	2.7	2.0	1.0	--	0.5	--	--	--	--	--	35
77	A	307588	Powder	7.03	2.71	2.08	1.06	0.16	0.68	0.01	0.01	--	0.54	.1022	35
78	--	--	Intended	6.8	2.7	2.0	3.0	--	1.0	--	--	--	--	--	37
78	A	207589	Powder	7.06	3.11	2.12	2.84	0.21	0.84	0.01	0.01	--	0.60	.1035	37
79	--	--	Intended	6.8	2.7	2.0	1.0	--	0.5	--	--	--	--	--	31
79	A	307590	Powder	6.85	2.76	2.06	1.02	0.15	0.52	0.01	0.01	--	0.61	.1023	31
80	--	--	Intended	6.8	2.7	2.0	3.0	--	1.0	--	--	--	--	--	33
80	A	307591	Powder	7.12	2.72	2.11	2.88	0.14	1.02	0.01	0.01	--	0.50	.1043	33
81	--	--	Intended	6.8	2.7	2.0	1.0	--	0.5	--	--	--	--	--	34
81	A	307592	Powder	6.96	2.74	2.12	1.02	0.16	0.58	0.01	0.01	--	0.59	.1060	34
82	--	--	Intended	6.8	2.7	2.0	3.0	--	1.0	--	--	--	--	--	34
82	A	307593	Powder	6.92	2.68	2.11	2.86	0.13	1.28	0.01	0.01	--	0.48	.1103	34
83	--	--	Intended	6.8	2.7	2.0	1.0	1.0	0.5	--	--	--	--	--	35
83	A	307594	Powder	6.85	2.65	2.06	1.02	1.25	0.43	0.01	0.01	--	0.57	.1065	35
84	--	--	Intended	6.8	2.7	2.0	3.0	3.0	1.0	--	--	--	--	--	29
84	A	307595	Powder	6.74	2.72	2.08	2.88	3.05	1.08	0.01	0.01	--	0.51	.1129	29
85	--	--	Intended	7.8	2.5	1.0	--	2.0	0.2	--	--	--	--	--	33
85	A	307596	Powder	7.58	2.55	1.14	0.02	2.16	0.00	0.01	0.01	--	0.59	.1060	33
86	--	--	Intended	7.8	2.5	1.0	--	4.0	0.2	--	--	--	--	--	36
86	A	307597	Powder	7.55	2.55	1.11	0.02	4.04	0.14	0.01	0.01	--	0.55	.1061	36
87	--	--	Intended	7.8	2.5	1.0	--	2.0	0.2	--	--	--	--	--	37
87	A	307598	Powder	7.58	2.52	1.06	0.02	2.16	0.16	0.01	0.01	--	0.63	.1043	37
88(c)	A	(c)	Intended	6.8	2.7	2.0	1.0	1.0	0.5	--	--	--	--	--	--
88	A	307599	Powder	--	--	--	--	--	--	--	--	--	--	--	--
89(c)	A	--(c)	Intended	7.8	2.5	1.0	--	2.0	0.2	--	--	--	--	--	--
89	A	307600	Powder	--	--	--	--	--	--	--	--	--	--	--	--
90	--	--	Intended	7.8	2.5	1.0	--	1.0	0.2	--	--	--	0.36	--	--
90	A	294013	Powder	7.52	2.44	1.00	0.01	1.08	0.20	0.00	0.00	--	0.02	.1030	44
90	A	294013(c)	Powder	7.74	2.44	1.01	--	0.96	0.20	0.01	--	--	0.06	.1027	41

(a) The HCl technique was modified depending on the alloying elements present.

(b) Density by pycnometer.

(c) Not prealloyed, but mixtures of elemental powders.

Table II

Properties of 2 in. Dia. Extrusions from Alloy Powders

Alloy No.	Ext. S-No.	Composition, % (b)										Powder Size μm - microns	Density lb/in. <sup>3</sup>	Loca- tion	Ultra- sonic Rating (G.d)	Heat Treatment #1 (a)				Heat Treatment #2 (h)			
		Zn	Pb	Cu	Fe	Ni	Cr	Ti	Other	T. S. ksi	Y. S. ksi					El. % in 4D	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. % in 4D	YS/Density x10 <sup>6</sup> in.		
1	277373	9.9	3.5	1.1	0.5	-	-	-	-	-	44	.1028	Front Back Avg.	A+ A+	106.8 107.2 107.0	104.2 (e)	2.1 2.1 2.1	1.014 -	88.6 83.4 76.0	(b) (b)	0.0 1.6 0.8		
2	277374	12.1	3.5	1.5	0.5	-	-	-	-	-	54	.1052	Front Back Avg.	A Failed	105.8 105.8 105.7	103.5 103.1 103.3	2.9 2.9 2.9	0.984 0.980 0.982	75.9 85.1 80.5	(b) (b)	0.0 0.0 0.0		
3	277375	12.3	4.0	1.6	0.5	-	-	-	-	-	27	.1052	Front Back Avg.	A+ A+	106.7 105.6 106.2	104.0 103.1 103.6	1.4 0.7 1.1	0.988 0.980 0.985	84.3 90.9 87.8	(b) (b)	(e) 0.0		
4	277376	9.6	3.2	2.0	-	-	0.2	-	-	-	39	.1041	Front Back Avg.	A+ A+	103.5 103.2 103.4	100.7 99.6 100.2	2.1 1.4 1.8	0.987 0.983 0.985	72.9 85.5 79.2	(b) (b)	0.0 0.0 0.0		
5	277377	10.3	2.9	2.1	1.8	-	0.1	-	-	-	37	.1062	Front Back Avg.	A+ A+	103.8 107.4 105.6	101.0 103.7 102.4	1.4 1.4 1.4	0.951 0.976 0.964	70.6 89.9 80.3	(b) (b)	0.0 0.0 0.0		
6	277378	11.2	2.4	2.0	-	-	0.2	-	-	-	39	.1060	Front Back Avg.	A+ A+	104.2 103.7 104.0	99.9 98.9 99.4	4.3 3.6 4.0	0.942 0.933 0.938	87.1 89.0 88.1	(b) (b)	0.0 1.9 1.0		
7	277379	7.9	3.4	1.5	-	2.5	2.0	-	-	-	26	.1058	Front Back Avg.	A+ A+	106.4 106.2 106.3	101.6 101.6 101.6	1.4 1.4 1.4	0.960 0.960 0.960	83.1 95.9 89.5	(b) (b)	0.0 1.9 1.0		
8	277380	7.9	3.5	1.5	1.6	1.6	1.4	-	-	-	44	.1057	Front Back Avg.	A Split	104.6 107.5 106.1	101.5 102.7 102.1	0.7 2.1 1.4	0.960 0.972 0.966	84.8 89.4 87.1	(b) (b)	0.0 0.0 0.0		
9	277381	7.9	3.5	1.6	-	-	0.5	0.4	V-0.4, Zr-0.4	38	.1037	Front Back Avg.	A+ A+	98.8 98.5 98.7	93.3 94.3 93.8	2.9 2.1 2.5	0.900 0.909 0.904	(1)41.4 80.5 80.5	(b) (b)	0.0 0.0 0.0			

continued .....

Table II (Continued)

Alloy No.	Ext. S.No.	Composition, % (b)										Powder Size μm	Density lb/in. <sup>3</sup>	Loca- tion	Ultra- sonic Rating (c,d)	Heat Treatment #1 (a)					Heat Treatment #2 (h)				
																Longitudinal (c)					Transverse (e)				
		Zn	Al	Cu	Fe	Mn	Cr	Ti	Other	T.S. ksi	Y.S. ksi					El. in 4D	YS/Density x10 <sup>4</sup> in.	T.S. ksi	Y.S. ksi	El. in 4D	Y.S. ksi	T.S. ksi	El. in 4D	Y.S./Density x10 <sup>4</sup> in.	
10	277382	7.9	3.5	1.5	-	-	2.1	0.9	V-1.7, Zr-0.6	39	.1066	Front Back Avg.	A	83.6 85.4 84.5	83.2 85.4 84.3	0.0 0.0 0.0	0.781 0.802 0.792	62.9 63.2 63.1	(b) (b) (b)	0.0 0.0 0.0					
11	277383	7.6	3.5	1.5	-	-	-	-	Co-1.4, Mo-1.6, W-0.5	37	.1047	Front Back Avg.	A+	97.8 97.5 97.7	94.2 95.2 94.7	0.7 0.7 0.7	0.900 0.909 0.904	71.3 80.3 75.8	(b) (b) (b)	0.0 0.0 0.0					
12	277384	7.8	3.5	1.6	0.6	0.5	0.6	0.5	0.4	V-0.4, Zr-0.6, Co-0.5, Mo-0.5, W-0.5	41	.1061	Front Back Avg.	A+	97.4 98.6 98.0	94.1 95.1 94.6	0.7 0.7 0.7	0.887 0.896 0.892	77.5 86.6 82.1	(b) (b) (b)	0.0 0.0 0.0				
13	277385	11.7	3.4	1.5	-	2.4	2.2	-	-	37	.1098	Front Back Avg.	A+	104.9 105.8 106.4	103.7 100.7 102.2	0.7 1.4 1.1	0.946 0.919 0.932	75.0 87.2 81.1	(b) (b) (b)	0.0 0.0 0.0					
14	277386	11.8	3.5	1.5	1.6	1.6	1.5	-	-	35	.1098	Front Back Avg.	A+	107.0 106.3 106.7	102.7 102.2 102.5	0.7 0.7 0.7	0.935 0.931 0.933	84.4 81.5 83.0	(b) (b) (b)	0.0 0.0 0.0					
15	277387	11.4	3.4	1.5	-	-	0.5	0.5	0.5	V-0.5, Zr-0.6	33	.1069	Front Back Avg.	A+	101.2 95.3 98.3	94.5 94.5 94.5	(e) 1.4 1.4	0.884 0.884 0.884	81.3 88.7 86.0	(b) (b) (b)	0.0 0.0 0.0				
16	277388	12.0	3.5	1.5	-	-	2.2	1.1	-	44	.1104	Front Back Avg.	A+	82.4 82.5 83.0	(f) (f) (f)	0.0 0.0 0.0	- - -	57.1 61.9 59.5	(b) (b) (b)	0.0 0.0 0.0					
17	277389	11.9	3.5	1.6	-	-	-	-	-	Co-1.5, Mo-1.1, W-0.5	41	.1082	Front Back Avg.	A+	100.2 101.7 101.0	96.9 97.9 97.4	0.7 0.7 0.7	0.896 0.905 0.900	79.4 81.8 80.6	(b) (b) (b)	0.0 0.0 0.0				
18	277390	11.7	3.4	1.5	0.6	0.5	0.6	0.5	0.5	V-0.4, Zr-0.6, Co-0.5, Mo-0.5, W-0.5	36	.1095	Front Back Avg.	A+	98.2 100.9 99.6	97.5 97.9 97.7	0.0 0.7 0.4	0.890 0.894 0.892	57.1 <sup>(1)</sup> 71.8 71.8	(b) (b) (b)	(e) (e) (e)				
19	277391	7.5	3.0	1.7	-	-	-	-	-	42	.1018	Front Back Avg.	A+	101.9 100.6 101.3	95.9 94.7 95.3	6.4 7.9 7.2	0.942 0.930 0.936	84.1 87.5 85.8	(b) (b) (b)	0.0 1.9 1.0					
																		105.8	101.2	7.1	0.984				

Continued . . .

Table II (Continued)

Alloy No.	Ext. S.No.	Composition, % (b)										Powder Size μm - microns	Density, lb/in <sup>3</sup>	Loca- tion	Ultra- sonic Rating (g.d)	Heat Treatment #1 (a)				Heat Treatment #2 (k)				
		Composition, % (b)														Longitudinal (c)		Transverse (c)		Longitudinal				
		Zn	Mn	Cu	Mo	Fe	Ni	Cr	Ti	Other	T. S. ksi					Y. S. ksi	El. % in 4D	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. in 4D	YS/Density x10 <sup>6</sup> in.		
20	277392	7.0	3.6	2.3	-	-	-	-	-	-	43	.1021	Front Back Avg.	A+ A+ Avg.		101.0 99.6 100.3	95.2 94.4 94.8	4.3 3.6 4.0	0.932 0.924 0.928	84.2 90.3 87.3	(b) (b) (b)	0.0 3.7 1.9	108.9 103.3 4.3	1.012
21	277393	7.9	5.6	7.4	-	-	-	-	-	-	42	.1053	Front Back Avg.	A+ A+ Avg.		98.0 94.5 96.3	93.7 90.6 92.2	1.4 0.7 1.1	0.890 0.860 0.875	85.4 83.2 84.3	(b) (b) (b)	0.0 0.0 0.0		
22	277394	7.9	8.4	14.9	-	-	-	-	-	-	48	.1100	Front Back Avg.	Failed A		93.2 93.7 93.5	{f} {f} {f}	0.7 0.0 0.4	- - -	68.0 70.3 69.2	(b) (b) (b)	0.0 0.0 0.0		
23	277395	7.7	10.1	20.1	-	-	-	-	-	-	34	.1146	Front Back Avg.	None None Avg.		72.2 88.2 70.2	{f} {f} {f}	0.0 0.0 0.0	- - -	17.2 13.6 15.4	(b) (b) (b)	0.0 0.0 0.0		
24	277396	7.8	7.2	20.0	-	-	-	-	-	-	27	.1167	Front Back Avg.	A+ A+ Avg.		79.1 83.6 81.5	{f} {f} {f}	0.0 0.0 0.0	- - -	74.5 82.4 78.6	(b) (b) (b)	0.0 (g) (g)		
25	277397	-	10.6	2.4	-	-	-	-	-	-	39	.0940	Front Back Avg.	A+ A+ Avg.		88.7 88.3 87.5	48.0 50.7 49.4	8.6 9.3 9.0	0.511 0.539 0.525	55.2 63.4 59.3	(b) (b) (b)	0.0 5.6 2.8		
26	277398	3.0	7.4	2.3	-	-	-	-	-	-	50	.0957	Front Back Avg.	A+ A+ Avg.		89.2 87.6 88.4	50.5 48.2 49.9	8.4 7.9 7.2	0.528 0.514 0.521	60.1 60.4 60.3	(b) (b) (b)	5.6 5.6 5.6		
27	277399	6.9	3.9	2.3	-	-	-	-	-	-	39	.1014	Front Back Avg.	A+ A+ Avg.		99.8 98.9 99.4	94.5 93.2 93.9	2.9 2.9 2.9	0.932 0.919 0.926	76.3 82.8 79.6	(b) (b) (b)	0.0 (g) (g)		
28	277400	9.8	3.8	2.3	-	-	-	-	-	-	34	.1042	Front Back Avg.	A+ A+ Avg.		102.7 101.2 102.0	99.9 97.7 98.6	2.9 2.1 2.5	0.959 0.938 0.948	89.0 89.2 89.1	(b) (b) (b)	0.0 1.9 1.0		
29	277401	13.7	4.3	2.3	-	-	-	-	-	-	42	.1069	Front Back Avg.	A+ A+ Avg.		108.2 106.3 107.3	104.8 101.8 103.3	1.4 1.4 1.4	0.980 0.952 0.966	84.9 90.8 87.9	(b) (b) (b)	0.0 1.9 1.0		

continued . . .

Table II (Continued)

Alloy No.	Ext. S.No.	Composition, % (b)										Powder Size microns	Density lb/in. <sup>3</sup>	Loca- tion	Ultra- sonic Rating (G.D.)	Heat Treatment #1 (a)										Heat Treatment #2 (h)																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																		
		Composition, % (b)														Longitudinal					Transverse (c)					Longitudinal					Transverse (c)																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																													
		Zn	Mn	Cu	Fe	Al	Cr	Ti	Other	T.S. ksi	Y.S. ksi					El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																													
30	277402	9.9	14.6	-	-	-	-	-	-	-	27	.0941	Front Back Avg.	Split Split	67.9 64.2 66.1	57.1 56.9 57.0	2.1 1.4 1.8	0.807 0.805 0.806	59.9 (b) 55.5 (b) 56.2	0.0 0.0 0.0																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																								

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Table II (Continued)

Alloy No.	Ext. S.No.	Composition, % (b)										Powder Size μm - atons	Density lb/in <sup>3</sup>	Loca- tion	Ultra- sonic Rating (C.O.)	Heat Treatment #1 (a)				Heat Treatment #2 (κ)										
		Zn		Pb		Cu		Fe		Al						Cr		Ti		Other		T.S. Y.S.		Longitudinal		Transverse (c)				
		El.	Gr.	El.	Gr.	El.	Gr.	El.	Gr.	El.	Gr.					El.	Gr.	El.	Gr.	El.	Gr.	ksi	ksi	ksi	ksi	ksi	ksi			
40	277428	8.4	5.2	-	1.0	0.7	3.0	-	-	-	-	42	.1059	Front Back Avg.	A Failed	108.0 102.5 105.3	104.3 99.2 101.7	0.7 0.7 0.7	0.985 0.936 0.980	81.4 84.6 83.0	{b} {h}	{g}	1.9							
41	273040	8.0	4.6	-	-	-	2.9	-	-	-	-	28	.1027	Front Back Avg.	A+ A+	90.5 93.5 92.0	85.9 89.4 87.7	2.1 2.1 2.1	0.836 0.870 0.854	70.1 86.1 78.1	{f}	{f}	0.0 0.0 0.0						0.918	
42	278041	7.6	3.9	-	-	-	2.0	-	-	-	-	32	.1022	Front Back Avg.	A+ A+	91.4 90.7 91.1	88.4 {f}	0.7 0.0 0.4	0.885 - -	68.0 81.0 74.5	{f}	{f}	0.0 0.0 0.0						0.931	
43	278241	12.0	4.7	0.9	0.9	-	0.20	-	-	-	-	18	.1058	Front Back Avg.	A+ A+	104.6 103.6 104.1	101.1 99.9 100.5	0.9 1.4 1.2	0.957 0.946 0.952	73.9 88.1 81.0	{f}	{f}	0.0 0.0 0.0						1.048	
44	278242	11.2	5.0	0.9	2.5	-	0.23	-	-	-	-	20	.1072	Front Back Avg.	A+ A+	102.2 96.7 99.5	{f}	0.0 0.0 0.0	- - -	71.3 80.8 76.1	{f}	{f}	0.0 0.0 0.0						---	
45	278243	11.7	4.6	0.9	3.0	-	0.75	-	-	-	-	19	.1062	Front Back Avg.	A+ A+	98.5 104.0 101.3	{f}	0.0 0.7 0.4	- 0.929 -	76.3 81.7 79.0	{f}	{f}	0.0 0.0 0.0						0.946	
46	278244	11.4	4.6	0.9	0.6	-	1.12	-	-	-	-	23	.1069	Front Back Avg.	A+ A+	98.8 100.2 99.5	94.5 96.7 95.6	0.9 0.7 0.8	0.884 0.804 0.894	83.7 92.0 87.9	{f}	{f}	0.0 0.0 0.0						0.980	
47	278245	9.8	4.2	0.4	2.1	1.6	4.6	-	-	-	-	27	.1101	Front Back Avg.	A+ A+	82.7 105.1 93.9	{f}	0.0 0.0 0.0	- - -	63.9 82.8 73.4	{f}	{f}	0.0 0.0 0.0						---	
48	278246	15.2	4.9	-	1.8	-	-	-	-	-	-	23	.1078	Front Back Avg.	A+ A+	105.5 105.8 105.7	101.8 102.3 102.1	0.8 0.6 0.7	0.944 0.949 0.947	76.1 91.1 83.7	{f}	{f}	0.0 0.0 0.0						1.002	
49	278247	14.8	4.9	1.8	1.7	-	-	-	-	-	-	21	.1095	Front Back Avg.	A+ A+	102.9 106.3 104.6	{f}	0.0 0.7 0.4	- 0.941 -	71.6 85.3 78.5	{f}	{f}	0.0 0.0 0.0						1.061	

continued . . .

continued . . .

Table II (Continued)

Alloy No.	Ext. A-E	Composition, % (b)										Powder Size microns	Density lb/in. <sup>3</sup>	Loca- tion	Ultra- sonic Rating (c, d)	Heat Treatment #1 (e)				Heat Treatment #2 (k)				
		Composition, % (b)														Heat Treatment #1 (e)				Heat Treatment #2 (k)				
		Zn	Al	Cu	Fe	Mn	Pb	Si	Cr	Other	T. S. ksi					Y. S. ksi	El. % in 4D	IS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. % in 4D	IS/Density x10 <sup>6</sup> in.		
50	278246	9.8	4.0	0.8	1.1	1.0	1.3	0.01	0.01	-	22	.1068	Front Back Avg.	A+ A+	106.8 108.6 107.7	103.3 104.1 103.7	1.1 1.1 1.1	0.937 0.976 0.971	79.5 83.3 86.4	{f} {f} {f}	118.4 115.0 116.7	0.9 0.9 0.9	1.077	
51	278249	9.6	4.1	0.8	1.2	1.0	1.4	0.01	0.01	Co-1.3	22	.1074	Front Back Avg.	A+ A+	107.7 107.9 107.8	104.0 103.6 103.8	0.9 0.9 0.9	0.968 0.965 0.966	82.5 86.1 86.8	{f} {f} {f}	110.4 107.4 108.9	0.4(g) 0.4(g) 0.4(g)	1.000	
52	278250	9.7	4.1	0.8	-	-	-	0.01	0.01	Co-1.4	22	.1048	Front Back Avg.	A+ A+	105.8 106.8 105.8	102.3 102.8 102.6	1.8 1.8 1.8	0.976 0.981 0.979	91.1 94.1 92.6	89.9 90.3 90.1	0.6 0.4 0.5	108.7 103.8 106.2	2.3 2.3 2.3	0.990
53	278251	9.6	4.1	-	2.5	1.4	4.8	0.04	0.04	-	25	.1100	Front Back Avg.	A+ A+	95.9 101.7 98.8	{f} {f} {f}	0.0 0.0 0.0	- - -	57.0 86.4 86.4	{f} {f} {f}	99.8 {f} {f}	0.0 0.0 0.0	---	
54	278252	9.5	4.0	1.4	2.9	1.6	0.9	0.05	0.04	-	23	.1085	Front Back Avg.	A+ A+	104.6 106.1 105.4	104.1 103.9 104.0	0.1 0.5 0.3	0.959 0.958 0.958	74.8 91.5 83.2	{f} {f} {f}	110.3 {f} {f}	0.0 0.0 0.0	---	
55	278253	9.9	4.2	1.5	0.9	1.9	2.4	0.07	0.05	-	23	.1086	Front Back Avg.	A+ A+	105.1 106.9 106.0	101.6 102.4 102.0	0.7 0.6 0.7	0.935 0.943 0.943	69.9 86.8 78.4	{f} {f} {f}	109.2 {f} {f}	0.0 0.0 0.0	---	
56	278254	9.8	4.3	1.7	2.0	-	2.5	0.11	0.07	-	22	.1080	Front Back Avg.	A+ A+	105.6 106.7 106.2	102.1 101.2 101.7	0.7 1.1 0.9	0.945 0.937 0.942	84.2 94.8 89.5	{f} {f} {f}	111.5 107.5 109.5	0.7 0.7 0.7	0.995	
57	278255	9.6	4.0	1.9	2.0	1.2	2.6	-	0.10	-	25	.1096	Front Back Avg.	A+ A+	104.9 107.2 106.1	104.3 103.3 103.8	0.4 (g) (g)	0.952 0.942 0.947	82.5 86.4 84.5	{f} {f} {f}	114.7 110.7 (e)	0.0 0.0 0.0	1.010	
58	278256	10.0	4.1	2.5	2.2	1.4	3.2	0.04	-	-	25	.1111	Front Back Avg.	A+ A+	102.3 104.6 103.5	{f} {f} {f}	0.0 0.0 0.0	- - -	59.4 85.8 85.8	{f} {f} {f}	106.8 {f} {f}	0.0 0.0 0.0	---	
59	278257	10.2	3.9	1.6	1.6	1.0	4.1	-	-	-	26	.1083	Front Back Avg.	A+ A+	96.4 102.1 99.5	{f} {f} {f}	0.0 0.0 0.0	- - -	61.3 74.6 68.0	{f} {f} {f}	83.9 {f} {f}	0.0 0.0 0.0	---	
60	278258	11.6	5.5	1.8	0.7	-	-	-	-	-	31	.1049	Front Back Avg.	A+	106.3 - -	104.2 - -	1.9 - -	0.993 - -	84.8 79.9 82.4	84.8 {f} -	0.4(f) 0.2 0.2	- - -	-	
61	283756	10.4	4.5	0.9	-	-	-	-	-	-	26	.1032	Front Back Avg.	A+	109.1 108.9 109.0	106.3 106.3 106.3	3.0 3.4 3.2	1.030 1.030 1.030	91.1 91.6 91.4	87.2 87.7 87.4	0.9 (g) -	- - -	-	

continued . . .

Table II (Continued)

Alloy No.	S-J-1	Composition, % (b)								Powder Size microns	Density lb/in. <sup>3</sup>	Loca- tion	Ultra- sonic Rating (C, D)	Heat Treatment #1 (a)				Heat Treatment #2 (k)					
		Zn	Pb	Cu	Mn	Fe	Ni	Cr	Ti					Other	Longitudinal (c)		Transverse (c)		Longitudinal				
		T. S. ksi	Y. S. ksi	El. % in 4D	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. % in 4D	YS/Density x10 <sup>6</sup> in.					T. S. ksi	Y. S. ksi	El. % in 4D	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. % in 4D	YS/Density x10 <sup>6</sup> in.		
62	307310	10.0	4.0	0.9	-	-	-	.01	.03	Co-0.8	42	.1036	Front	A+	106.6 (b)	2.9	-	-	95.6 (b)	116.2	113.7	2.1	1.097
													Front		107.2 103.2	2.9	0.996	-	96.7 (b)				
													Back		107.9 (b)	2.6	-	-	92.7 (b)				
													Avg.		107.5 103.5	2.1	0.999	-	91.1 (b)				
															107.4 103.4	2.6	0.998	-	91.5 (b)				
63	307311	10.1	4.0	0.9	-	-	-	.01	.03	Co-2.3	50	.1043	Front	A+	101.6 (b)	0.0	-	-	73.2 (b)	115.7	107.7	0.4	1.050
													Front		103.6 100.1	0.6	0.960	-	81.3 (b)				
													Back		104.8 (b)	0.8	-	-	77.9 (b)				
													Back		105.7 102.5	0.9	0.983	-	81.1 (b)				
													Avg.		103.9 101.2	0.6	0.971	-	78.4 (b)				
64	307312	8.9	4.1	0.9	-	-	-	.01	.03	Co-1.5	48	.1038	Front	A+	104.8 (b)	1.9	-	-	91.3 (b)	118.0	115.1	1.1	1.090
													Front		105.5 101.0	2.8	0.973	-	94.1 (b)				
													Back		104.7 (b)	1.8	-	-	86.9 (b)				
													Back		104.3 100.9	1.4	0.972	-	91.4 (b)				
													Avg.		104.8 101.0	2.0	0.973	-	91.2 (b)				
65	307313	11.1	4.1	1.0	-	-	-	.01	.03	Co-1.6	43	.1052	Front	A+	108.1 (b)	1.9	-	-	84.6 (b)	118.7	116.2	0.9	1.104
													Front		108.2 106.2	1.5	1.000	-	94.8 (b)				
													Back		107.6 (b)	1.2	-	-	87.3 (b)				
													Back		112.4 107.1	1.5	1.018	-	92.3 (b)				
													Avg.		109.3 106.2	1.5	1.010	-	89.8 (b)				
66	307314	10.0	3.6	0.9	-	-	-	.01	.03	Co-1.5	43	.1048	Front	A+	108.3 (b)	2.1	-	-	96.9 (b)	116.9	115.3	2.1	1.061
													Front		108.5 104.5	2.0	0.997	-	91.5 (b)				
													Back		107.1 (b)	1.9	-	-	81.1 (b)				
													Back		108.0 106.8	2.9	1.010	-	93.7 (b)				
													Avg.		108.2 105.2	2.2	1.004	-	89.8 (b)				
67	307315	10.3	4.7	1.0	-	-	-	.01	.03	Co-1.6	47	.1035	Front	A+	104.8 (b)	1.6	-	-	87.9 (b)	112.4	107.6	1.4	1.040
													Front		106.2 101.8	1.2	0.984	-	86.6 (b)				
													Back		106.8 (b)	2.9	-	-	81.7 (b)				
													Back		109.6 104.1	2.0	1.002	-	84.6 (b)				
													Avg.		107.3 103.0	1.9	0.996	-	85.2 (b)				

continued . . .

Table II (Continued)

Alloy No.	Ent. Size	Composition, % (b)										Powder Size No. 40 microns	IR/Inch Location	Ultra- sonic Rating (c, d)	Heat Treatment #1 (a)				Heat Treatment #2 (k)			
		Zn	Al	Cu	Mn	Pb	Fe	Si	Cr	Other	Longitudinal (f)				Transverse (g)		Longitudinal		Transverse			
		T. S. ksi	Y. S. ksi	El. % in 40	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. % in 40	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi				El. % in 40	YS/Density x10 <sup>6</sup> in.	T. S. ksi	Y. S. ksi	El. % in 40	YS/Density x10 <sup>6</sup> in.		
66	307319	9.9	3.9	1.5	-	-	-	.01	.03	Co-1.5	37	.1047	A+	104.3 (b) 106.5 (b) 107.1 (b) 106.6 (b) 105.9 (b) Avg.	1.4 1.4 2.1 1.5 1.6	- 0.957 - 0.978 0.968	87.8 (b) 86.8 (b) 91.5 (b) 87.3 (b) 86.1 (b) -	0.3(j) 0.4 0.7 0.4 0.4	114.8	110.0	1.6	1.051
69	307317	9.8	4.1	0.6	-	-	-	.01	.03	Co-1.5	41	.1043	A+	107.9 (b) 106.4 (b) 107.3 (b) 111.9 (b) 106.9 (b) Avg.	1.9 1.8 1.0 1.5 1.6	- 0.996 - 1.012 1.005	91.9 (b) 93.3 (b) 86.0 (b) 84.0 (b) 89.7 (b) -	{g} {g} {g} 0 -	120.8	116.5	1.2	1.117
70	307318	10.8	4.6	1.3	-	-	-	.01	.03	Co-2.4	42	.1062	A+	105.9 (b) 108.1 (b) 109.2 (b) 110.6 (b) 108.4 (b) Avg.	1.0 0.9 0.6 0.9 0.8	- 0.976 - 0.999 0.987	77.0 (b) 83.6 (b) 91.0 (b) 91.6 (b) 85.8 (b) -	0 0 {g} {g} -	121.3	117.3	0.4	1.104
71	307319	9.2	3.6	0.6	-	-	-	.01	.03	Co-0.8	40	.1030	A+	106.9 (b) 110.1 (b) 107.7 (b) 110.6 (b) 108.8 (b) Avg.	3.9 2.3 1.9 4.3 3.1	- 1.013 1.025 1.012	94.6 (b) 93.3 (b) 93.4 (b) 86.9 (b) 92.0 (b) -	1.3 0.7 0.9 0.2 0.8	115.3	112.0	3.6	1.087
72	307320	10.1	4.1	0.9	-	-	-	.01	.03	Co-1.5, 35 Si-0.4	35	.1044	A+	104.9 (b) 108.1 (b) 105.9 (b) 110.1 (b) 107.2 (b) Avg.	1.5 1.7 0.8 1.4 1.4	- 0.978 - 0.995 0.986	94.9 (b) 98.0 (b) 91.6 (b) 88.6 (b) 93.3 (b) -	0.7(j) 1.2 0.6(j) 0.3 0.7	116.9	113.4	0.8	1.086
75	293184	7.1	2.8	2.0	-	1.0	4.7	0.3	-	-	31	.1071	A+	101.4 (b) 100.2 (b) 100.8 (b) Avg.	0.6 0.6 0.6	0.916 0.911 0.913	89.6 (b) 87.3 (b) 88.4 (b) -	{g} {g} -	-	-	-	-
79	293188	6.8	2.8	2.1	1.0	-	-	0.5	-	0.1 2x	31	.1037	A+	97.2 (b) 96.3 (b) 96.8 (b) Avg.	3.9 2.7 3.3	0.848 0.856 0.852	90.2 (b) - 90.2 (b) -	3.4 - 3.4	-	-	-	-

Table II (Continued)

Alloy No.	S.No.	Composition, % (b)										Powder Size MPD - microns	Density, lb/in. <sup>3</sup>	Loca- tion	Ultra- sonic Rating (c,d)	Heat Treatment #1 (a)					Heat Treatment #2 (c)				
		Zn	Mg	Cu	Mn	Fe	Ni	Cr	Ti	Other	Longitudinal (c)					Transverse (c)		Longitudinal (c)		Transverse (c)					
											Y.S. ksi					El. % in 4D	YS/ Density x10 <sup>3</sup> in.	T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D		
82	293191	6.9	2.7	2.1	2.9	-	7.1	1.3	-	0.5	2r	Front Back Avg.	A+	92.6 98.3 95.4	(f) (f) -	0 0 0	- - -	82.4 81.7 82.0	(f) (f) -	(g) (g) -	(g) (g) -				
83	293192	6.8	2.6	2.1	1.0	1.2	4.5	0.4	-	0.1	2r	Front Back Avg.	A+	101.2 99.7 100.4	96.6 95.6 96.1	0.6 0.6 0.6	0.894 0.885 0.890	96.9 86.5 91.7	91.7 -	1.0 (g) -	1.0 (g) -				
85	293194	7.6	2.6	1.1	-	2.2	4.8	0.0	-	-	-	Front Back Avg.	A+	88.5 88.2 88.4	83.7 83.4 83.6	1.7 1.3 1.5	0.779 0.776 0.777	81.5 81.1 81.3	76.0 75.8 75.9	1.3 1.2 1.3	1.3 (g) 1.3				
86	293195	7.6	2.6	1.1	-	4.0	2.1	0.1	-	-	-	Front Back Avg.	A+	107.4 106.8 107.1	101.3 100.9 101.1	1.3 0.7 1.0	0.941 0.938 0.940	88.5 94.9 91.7	(f) 91.0 -	0.3 0.6 0.4	0.3 (f) -				
87	293196	7.6	2.5	1.1	-	2.2	2.3	0.2	-	-	-	Front Back Avg.	A+	104.1 103.7 103.9	99.1 99.3 99.2	1.7 1.0 1.4	0.937 0.928 0.938	97.6 92.2 91.9	91.3 92.2 91.8	1.5 0.7 1.1	1.5 (g) 1.1				
90	294014	7.5	2.4	1.0	-	1.1	1.0	0.2	-	-	-	Front Back Avg.	A+	98.8 99.1 99.0	92.4 93.6 93.0	5.4 5.4 5.4	0.885 0.897 0.891	87.6 86.5 87.0	80.5 80.7 80.6	2.4 2.3 2.4	2.4 (g) 2.4				

(a) Heat Treatment #1 - SHT 2 hrs at 860°F, CWQ, aged 24 hrs at 250°F-2" dia.

(b) Based on analysis of powder or extrusion.

(c) One test per value of heat treated rod to Alloy 59. The tests from 60-90 are for an extruded material.

(d) A+ = met standards higher than Class A. A = met Class A Standards.  
Failed = Class A standards not met. Split = rod cracked during CWQ.  
None = not measured due to rough surface.

(e) Not measured.

(f) Failed before reaching 0.2% offset.

(g) Not obtained - specimen shattered.

(h) Strain follower not used to avoid breaking it.

(i) Data not included in average.

(j) Failed outside gauge length.

(k) Heat Treatment #2 - SHT 1/2 hr at 920°F, CWQ, aged 96 hrs at 225°F - 1" x 1" quadrants.

(l) S.No. 277645.

(m) S.No. 277646.

TABLE III

COMPOSITIONS IN VARIOUS PHASE FIELDS OF THE  
Al-Zn-Mg-Cu SYSTEM AT 860°F (460°C)

Alloy No.	% Zn	% Mg	% Cu	Phase Field*
19	7.6	3.0	1.5	$\alpha + S + T + M$
20	7.9	3.6	2.3	$\alpha + S + T + M$
21	8.0	5.6	7.5	$\alpha + S + T + M$
22	8.0	8.5	15.0	$\alpha + S + T + M$
23	8.0	10.3	20.0	$\alpha + S + T + M$
24	8.0	7.5	20.0	$\alpha + S + Z + M$
25	---	10.8	2.3	$\alpha + S + T$
26	3.0	7.5	2.3	$\alpha + S + T$
27	7.0	4.0	2.3	$\alpha + S + T$
28	10.0	4.0	2.3	$\alpha + M + T$
29	14.0	4.4	2.3	$\alpha + M + T$
30	8.0	15.0	---	$\alpha + T$
31	3.0	15.0	2.3	$\alpha + T$

\* Phase identification

$\alpha$  = Al

S =  $\alpha$ (Al-Cu-Mg)

T =  $\left\{ \begin{array}{l} \alpha(\text{Al-Mg-Cu}) \\ \text{Mg-Zn-Al} \end{array} \right\}$  isomorphous

M =  $\left\{ \begin{array}{l} \beta(\text{Zn-Mg}) \text{ or } \text{MgZn}_2 \\ \gamma(\text{Al-Cu-Mg}) \end{array} \right\}$  isomorphous

=  $\left\{ \begin{array}{l} \alpha(\text{Zn-Mg}) \\ \beta(\text{Al-Cu-Mg}) \end{array} \right\}$  isomorphous

TABLE IV

## COMPARISON OF DENSITY OF POWDERS AND EXTRUSIONS

Alloy No.	S. No.	Location	Powder Density, gm/cc <sup>(b)</sup>	Extrusion Density, gm/cc <sup>(c)</sup>	Difference in Densities, gm/cc (ED-PD)	Variation in Density, % ((ED-PD)/ED)x100
39 (a)	283462	F	2.894	2.9304	.036	1.2
"	"	F	2.894	2.9315	.038	1.3
"	"	B	2.894	2.9298	.036	1.2
"	"	Avg.	2.894	2.9306	.037	1.2
50 (a)	283481	F	2.914	2.9465	.032	1.1
"	"	F	2.914	2.9459	.032	1.1
"	"	B (d)	2.914	2.9481	.034	1.2
"	"	Avg.	2.914	2.9468	.033	1.1
52 (a)	283492	F	2.858	2.8974	.039	1.3
"	"	F	2.858	2.8979	.040	1.4
"	"	B	2.858	2.8975	.038	1.3
"	"	Avg.	2.858	2.8970	.039	1.3
52 (e)	307454	F	2.855	2.9007	.046	1.6
"	"	FM	2.855	2.9005	.046	1.6
"	"	M	2.855	2.9010	.046	1.6
"	"	BM	2.855	2.8998	.045	1.6
"	"	B	2.855	2.8992	.044	1.6
"	"	Avg.	2.855	2.9001	.045	1.6
52 (e)	307455	F	2.855	2.8978	.043	1.5
"	"	B	2.855	2.8977	.043	1.5
"	"	Avg.	2.855	2.8978	.043	1.5
62 (e)	307321	F	2.830	2.8785	.048	1.7
"	"	B	2.830	2.8772	.047	1.6
"	"	Avg.	2.830	2.8778	.048	1.7
64 (e)	307323	F	2.833	2.8770	.044	1.5
"	"	B	2.833	2.8764	.043	1.5
"	"	Avg.	2.833	2.8767	.044	1.5
71 (e)	307330	F	2.813	2.8566	.044	1.5
"	"	FM	2.813	2.8567	.044	1.5
"	"	M	2.813	2.8568	.044	1.5
"	"	BM	2.813	2.8565	.044	1.5
"	"	B	2.813	2.8568	.044	1.5
"	"	Avg.	2.813	2.8567	.044	1.5

(a) Two in. dia. tensile specimens SHT 1/2 hr. at 920°F, CWQ, aged 96 hrs. at 225°F.

(b) By pycnometer method on powders. (PD)

(c) By water displacement method on a, d, and e. (ED)

(d) Two in. dia. broken tensile specimens SHT 2 hrs. at 860°F, CWQ, aged 96 hrs. at 225°F.

(e) One in. x 4-1/4 in. extruded slab in -F temper. Location of specimens in sketch below.

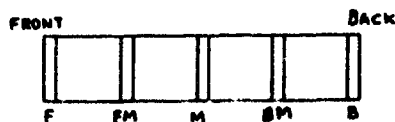


TABLE V  
EFFECT OF HEAT TREATMENT CONDITIONS ON LONGITUDINAL TENSILE PROPERTIES (a) OF 2 IN. DIA. EXTRUDED ROD

S. No.	Composition, % (b)					SHT(c)		ARE	T.S.		El. % in SD	TS/Density x10 <sup>5</sup> in.
	Zn	Mn	Cu	Pb	Fe	Si	Cr		ksi	ksi		
277557	10.9	4.9	2.0	1.7	—	—	—	96 144 24	225 225 250	122.4 118.9 110.7	(d) 0.7 0.9	1.105 1.107 1.002
277426	8.4	3.6	—	1.0	.8	2.8	—	96 { 6 +48 24	225 200 225 250	118.5 115.8 118.8 112.3	1.0 0.9 (d)	1.081 1.076 1.024
277424	10.7	3.8	2.0	1.7	—	—	—	144 96 24	225 225 250	117.9 118.2 109.4	1.9 1.7 1.4	1.066 1.061 0.993
277401	13.7	4.3	2.3	—	—	—	—	96 144 24	225 200 250	118.1 117.4 108.2	0.8 1.4 1.4	1.062 1.058 0.980
277573	9.9	3.5	1.1	.5	—	—	—	96 144 24	225 200 225 250	112.6 109.5 112.9 109.4 106.8	2.1 2.7 2.7 2.1	1.065 1.064 1.064 1.014
277406	7.8	2.5	1.0	—	3.9	5.3	.2	96 96 24	225 250 250	112.3 (e) 109.7 104.4	0.3 0.4 0.7	— 0.982 0.941

(a) based on one tensile test on front of extrusion.  
 (b) based on analysis of powder or extrusion.  
 (c) quenched in cold water after solution heat treatment and before aging.  
 (d) not obtained - specimen shattered.  
 (e) specimen failed before reaching 0.2% offset.



TABLE VI  
EFFECT OF HOLDING TIME AT THE SOLUTION HEAT TREATING TEMPERATURE ON TENSILE PROPERTIES

Alloy No.	Extr. S. No.	Composition, %			Density, lb./in. <sup>3</sup>	Compact Preheat Hr.	SHT (c)		T.S.		El. % in. 4D	TS/Density $\times 10^6$ in.
		Al	Cu	Mn			Time Hr.	Temp. °F.	ksi	ksi		
36	277645	10.7	3.8	1.7	0.1076	.75	.5	920	118.5	115.9	1.1	1.077
		10.7	3.8	2.0	"	.75	1.	920	118.6	115.0	(a)	1.069
		10.7	3.8	2.0	"	.75	2.	920	114.2	111.2	0.3(b)	1.033
36	277646	8.4	3.6	1.0	0.1071	1.5	.5	920	119.6	117.0	1.2	1.092
		8.4	3.6	1.0	"	1.5	1.	920	119.4	116.9	1.3	1.091
		8.4	3.6	1.0	"	1.5	2.	920	119.4	116.9	(a)	1.091

Note: Each result represents one tensile test.

- (a) Not obtained. Specimen shattered.
- (b) Specimen failed outside gage length.
- (c) Quarter sections of 2 in. dia. rod from front of extrusions were solution heat treated, C40 and aged immediately 96 hrs at 225°F.

Table VII

## EFFECT OF COOLING RATE ON TENSILE PROPERTIES

Alloy No.	S. No.	Composition, %					Compact Preheat Temp., °F	Heat Treatment		Specimen Size, in Dia.	Approx. (h) Cooling Rate, °F/Sec.	Longitudinal			Transverse				
		Zn	Mg	Cu	Mn	P		SWT T/T (s)	Quench (s)			T.S., ksi	El., % in LD	Y.S./Density x 10 <sup>6</sup> in	T.S., ksi	El., % in LD	Y.S., ksi	El., % in LD	
Al	277378	11.2	2.4	2.0	---	0.2	---	2/860	CMQ	2L/250	1x1(c) 2	350	108.4	104.9	4.3	0.990	---	---	---
									CMQ			150	104.2	99.9	4.3	0.942	---	---	
Al	277379	7.5	3.2	1.7	---	---	900	2/860	CMQ	2L/250	1x1 2	350	104.3	98.4	7.1	0.966	---	---	---
									CMQ			150	101.9	95.9	6.4	0.942	---	---	
Al	277372	7.8	3.6	2.3	---	---	900	2/860	CMQ	2L/250	1x1 2	350	104.3	99.9	4.3	0.978	---	---	---
									CMQ			150	101.0	95.2	4.3	0.932	---	---	
Al	277424	6.4	2.7	0.6	---	0.1	900	2/860	CMQ	2L/250	1x1 2	350	97.4	94.2	4.3	0.919	---	---	---
									CMQ			150	96.8	92.4	5.7	0.901	---	---	
Al	277938	10.6	4.4	2.0	1.7	---	550	2/860	CMQ	2L/250	1x1/5/16(d) 2	350/1500(d)	107.0	104.9	0.8	0.991	64.0	(e)	0
									CMQ			150	102.4	99.5	0.9	0.940	70.5	(e)	0
Al	277939(f)								CMQ	1x1/5/16(d) 2		350/1500(d)	105.1	102.9	0.8	0.973	88.6	(e)	0
									CMQ			150	102.1	98.0	1.3	0.911	80.4	(e)	0.2
Al	277938	8.2	3.3	---	1.1	---	850	2/860	CMQ	2L/250	1x1/5/16(d) 2	350/1500(d)	113.3	112.7	0.1	1.074	92.3	(e)	0
									CMQ			150	104.7	104.5	0.5	0.946	86.2	(e)	0
Al	277937(f)								CMQ	1x1/5/16(d) 2		350/1500(d)	113.6	111.2	0.6	1.060	96.4	(e)	0
									CMQ			150	102.7	100.3	1.3	0.956	93.1	(e)	0
Al	277936								CMQ	2L/250	1x1/5/16(d) 2	350/1500(d)	83.5	77.7	2.2	0.741	80.9	77.4	1.4
									CMQ			150	85.6	76.1	5.0	0.725	72.8	70.6	0.8
Al	277937(f)								CMQ	1x1/5/16(d) 2		350/1500(d)	85.0	77.0	3.6	0.734	82.2	76.6	1.8
									CMQ			150	82.9	74.4	3.4	0.709	73.9	70.3	1.2
Al	277938								CMQ	2L/250	1x1/5/16(d) 2	350/1500(d)	79.0	67.3	5.1	0.612	74.0	66.8	2.0
									CMQ			150	79.8	68.2	6.8	0.650	73.4	64.6	2.6
Al	277937(f)								CMQ	1x1/5/16(d) 2		350/1500(d)	78.1	67.2	4.6	0.611	75.8	68.2	2.3
									CMQ			150	77.8	66.6	4.0	0.635	72.4	63.8	2.8
Al	277938								CMQ	2L/250	1x1/5/16(d) 2	350/1500(d)	75.1	62.1	5.8	0.592	70.0	62.6	2.5
									CMQ			150	74.8	61.1	5.8	0.582	65.7	58.2	2.0
Al	277937(f)								CMQ	1x1/5/16(d) 2		350/1500(d)	74.1	61.2	6.4	0.585	72.0	64.4	2.9
									CMQ			150	73.4	60.1	5.8	0.573	66.5	57.4	2.6
Al	24462	10.9	4.9	2.0	1.0	---	950	0.5/920	CMQ	96/225	0.250 2	2000	122.5	120.2	0.8	1.133	---	---	---
									CMQ			150	112.8	120.2	0.0	1.053	---	---	---
Al									SMQ			23	109.7	111.7	1.2	0.995	---	---	---
									A			1.7	99.6	105.6	1.4	0.870	---	---	---
Al									A			0.21	62.5	48.5	3.3	0.457	---	---	---
									A										

Continued

Table VII (continued)

## EFFECT OF COOLING RATE ON TENSILE PROPERTIES

Alloy No.	S. No.	Composition, %					Compact Preheat Temp, °F		Heat Treatment		Specimen Size, in Dia.	Approx. Cooling Rate, °F/Sec.	Longitudinal		Transverse	
		Cu	Mg	Al	Mn	Fe	°F	°F	SRF t/T (a)	Quench t/T (b)			T.S. ksi	EL, %	T.S. ksi	EL, %
50	278217	11.0	4.9	1.8	1.7	---	---	---	0.5/920	CAQ 96/225	0.250	25,000	118.1	(e)	---	---
							900			CAQ	0.250	2,000	121.3	(e)	---	---
										CAQ	1x1	350	116.5	116.2	0.5	1.061
										CAQ	2	150	112.9	(e)	0.0	---
50	283471	15.1	4.6	2.0	1.9	---	---	---	---	CAQ	0.250	2,000	106.7	101.2	0.8	0.124
							900			CAQ	0.250	2,000	120.2	(e)	---	---
										CAQ	0.250	25,000	127.5	124.4	(e)	---
							900			CAQ	0.250	2,000	127.5	124.4	(e)	---
50	278218	9.8	4.0	0.8	1.1	0.1	1.0	1.3	0.01	0.01	0.250	2,000	118.9	115.6	1.0	1.083
										CAQ	0.750	500	118.4	115.0	0.9	1.077
										CAQ	1-1	350	102.0	94.8	2.0	0.888
										CAQ	0.250	23	91.9	81.5	4.0	0.763
50	283481	10.0	4.1	0.9	1.4	0.01	1.1	1.4	0.02	0.02	0.250	2,000	115.3	109.0	(e)	1.020
							900			CAQ	2	150	121.9	119.8	1.4	1.122
										CAQ	0.250	2,000	114.4	111.9	2.3	1.048
							1000			CAQ	2	150	108.2	103.9	1.4	0.973
52	278250	9.7	4.1	0.8	---	0.01	---	---	0.01	0.01	0.250	2,000	99.3	93.5	1.4	0.875
										CAQ	2	0.21	63.6	51.6	2.4	0.485
										CAQ	0.250	25,000	124.8	123.0	2.4	1.174
							900			CAQ	0.250	2,000	123.3	121.5	1.8	1.160
52	283491	10.0	4.0	0.9	---	0.01	---	---	0.01	0.01	0.250	2,000	117.9	114.9	1.4	1.008
										CAQ	0.750	500	108.7	103.8	2.2	0.991
										CAQ	1x1	350	99.6	92.3	4.6	0.882
										CAQ	0.250	23	89.4	80.9	6.0	0.773
52	283491	10.0	4.0	0.9	---	0.01	---	---	0.01	0.01	0.250	2,000	124.0	121.6	2.8	1.163
										CAQ	2	150	115.4	111.2	(e)	1.062
										CAQ	0.250	2,000	122.0	119.4	1.8	1.039
							1000			CAQ	2	150	115.1	111.1	1.9	1.060
59	277931	10.2	3.9	1.6	1.6	---	---	---	---	---	0.250	23	104.1	102.0	2.2	0.973
										CAQ	2	1.7	96.7	88.6	3.8	0.845
										CAQ	0.250	0.21	61.9	46.0	8.6	0.439
										CAQ	1x1/5/16(d)	350/1500(d)	87.8	(e)	0.0	---
60	283475	11.6	5.5	1.8	0.7	---	---	---	---	---	0.250	2,000	111.1	110.7	0.8	---
							950			CAQ	0.250	150	107.9	---	0.3	---
										CAQ	0.250	23	105.5	103.9	0.9	0.990
										CAQ	0.250	1.7	100.6	97.2	1.0	0.927
62	281759	10.4	4.5	0.8	---	---	---	---	---	---	0.250	2,000	119.5	115.9	2.6	1.123
							1000			CAQ	0.250	150	114.9	111.0	2.9	1.076
										CAQ	0.250	23	109.4	103.7	3.3	1.005
										CAQ	0.250	1.7	98.4	91.1	4.6	0.883

NOTE: (a) Time in hours, T = temperature in °F  
 (b) CAQ - Cold water spray quenched  
 CAQ - Cold water quench  
 BAQ - Boiling water quench  
 A - Coiled in air  
 (c) 1/2 quarter sections from 2 in dia. extrusions  
 (d) Longitudinal processed as 1x1 quadrants, approx. cooling rate 350°F/sec.  
 Transverse processed as 5/16" thick slices, approx. cooling rate 1500°F/sec.  
 (e) Failed before reaching 0.2% offset  
 (f) Classified powder - Fine fraction  
 (g) Specimen shattered - Not determined  
 (h) Approximate cooling rate from 750° to 550°F

Table VIII

## TENSILE PROPERTIES OF APM EXTRUSIONS AT CRYOGENIC AND ELEVATED TEMPERATURES

Alloy No.	S-No.	Heat Treat Mo.(a)	-112°F		-180°F		75° (Room Temp.)			1/2 hr. at 212°F			1/2 hr. at 400°F		
			F.S. ksi	Y.S. ksi	F.S. ksi	Y.S. ksi	F.S. ksi	Y.S. ksi	F.S. ksi	Y.S. ksi	F.S. ksi	Y.S. ksi	F.S. ksi	Y.S. ksi	
34	203442	2	-	-	-	-	109.8	109.1	0.0	96.8	(b)	48.4	40.4	17.7	
		-	-	-	-	103.5	(b)	0.0	93.5	(b)	47.7	36.7	14.9		
		AVG	-	-	-	-	106.7	-	0.0	95.4	-	48.1	38.6	16.3	
38	203452	2	-	-	-	-	115.6	111.3	(c)	100.4	97.0	46.4	39.8	21.1	
		-	-	-	-	112.9	112.2	0.0	104.5	100.2	45.6	41.8	22.9		
		AVG	-	-	-	-	114.6	111.8	-	102.5	98.6	46.0	40.8	22.0	
39	203462	2	-	-	-	-	112.2	112.2	0.0	105.7	102.8	56.4	51.9	9.7	
		-	-	-	-	113.4	111.2	0.0	108.6	104.0	55.7	51.4	19.4		
		AVG	-	-	-	-	112.8	111.7	0.0	107.2	103.4	56.1	51.7	14.6	
49	203472	3	-	-	-	-	107.0	104.7	0.0	99.9	95.0	50.8	48.0	22.3	
		2	-	-	-	-	108.2	(b)	0.0	103.3	102.2	55.9	52.1	10.7	
		AVG	-	-	-	-	103.5	(b)	0.0	107.3	106.1	55.6	52.6	13.1	
50	203483	3	-	-	-	-	105.9	-	0.0	105.3	104.2	55.8	52.4	11.9	
		2	-	-	-	-	111.0	107.7	0.6	99.5	97.9	51.6	47.7	10.9	
		AVG	-	-	-	-	114.7	112.7	0.0	106.4	104.0	54.2	52.1	12.0	
52	203493	2	-	-	-	-	114.0	111.1	0.6	104.7	101.5	54.7	51.8	15.4	
		-	-	-	-	114.4	111.9	0.3	105.6	102.8	54.5	52.0	13.7		
		AVG	-	-	-	-	116.2	112.5	2.3	105.1	102.1	48.5	47.0	23.4	
52	203494	2	124.6	122.2	2.0	120.2	119.2	2.0	109.8	103.7	4.0	56.3	52.8	16.0	
		126.2	126.2	1.0	119.8	117.2	2.0	(e) 106.6	101.2	5.0	53.2	50.9	17.5		
		AVG	125.4	124.2	1.5	120.0	118.2	2.0	116.7 (d) 108.2	102.4	4.5	54.8	51.8	19.0	
61	203759	2	-	-	-	-	114.1	110.1	2.2	105.5	102.9	52.4	50.6	18.3	
		-	-	-	-	115.7	111.5	3.5	104.7	102.3	51.3	50.8	21.7		
		AVG	-	-	-	-	114.9	111.0	2.9	105.1	102.6	51.9	50.7	20.0	
62	307310	2	125.2	124.2	2.0	121.2	119.2	2.0	106.1	101.8	7.0	53.1	49.9	22.0	
		124.6	124.2	2.0	122.2	120.2	2.0	117.5 (d)	103.2	5.0	53.1	51.5	23.0		
		AVG	124.9	124.2	2.0	121.7	119.7	2.0	116.8	102.5	6.0	53.1	50.7	22.5	
64	307312	2	123.2	122.2	(c)	119.2	117.1	2.0	106.0	101.4	5.0	51.0	47.5	26.0	
		122.2	122.2	2.0	118.1	116.1	3.0	118.0	113.1	1.1	106.9	101.9	50.6	46.0	
		AVG	122.7	122.2	2.0	118.6	116.6	2.5	117.1	106.4	6.5	50.8	46.8	27.0	
71	307319	2	123.6	123.2	3.0	117.1	116.1	4.0	103.6	99.2	10.0	48.0	46.1	26.0	
		125.2	125.2	2.0	118.1	116.1	3.0	115.3	112.0	3.6	103.7	100.4	47.2	24.0	
		AVG	124.4	124.2	2.5	117.6	116.1	3.5	114.4	99.8	10.5	47.6	45.2	25.0	
7075-76 (Extrusions)			97.0	91.0	8.0	93.0	86.0	9.0	83.0	78.0	42.0	40.0	18.0		
X2020-76			88.0	81.0	5.0	86.0	79.0	6.0	77.0	72.0	57.0	54.0	8.0		

(a) 81 S.N.T. 1/2 hr. 320°F, CWQ, 96 hrs. at 225°F.  
 83 S.N.T. 2 hrs. 860°F, CWQ, 96 hrs. at 225°F.

(b) Failed before reaching 0.2% offset.

(c) Not obtained - specimen shattered.

(d) Not obtained.

(e) Not included in average.

Table IX

## THE EFFECT OF TIME AT ELEVATED TEMPERATURE ON LONGITUDINAL TENSILE PROPERTIES OF ALLOY 52

S. No.	Heat Treatment No. (a)	Time at Temp.	212°F			400°F		
			T.S. ksi	Y.S. ksi	El. % in 4D	T.S. ksi	Y.S. ksi	El. % in 4D
283493 (b)	2	1/2	105.1 103.5	102.1 100.6	3.0 3.0	48.5 48.2	47.7 45.6	23.4 19.4
283494 (c)	2	1/2	109.8 106.6 106.2	103.7 101.2 101.9	4.0 5.0 3.8	56.3 53.2 51.6	52.8 50.9 49.2	16.0 19.0 19.4
AVG								
283497 (d)	2	1/2	106.4	103.9	1.14	47.9	43.8	22.1
283497 (d)	2	100	106.1	102.9	5.0 (f)	25.3	22.9	49.3
283497 (d)	1	1/2	97.2	93.9	7.9 (e)	46.2	41.5	22.9
283497 (d)	1	100	97.4	92.9	1.4	23.7	22.1	57.1
7075-T6		1/2 100	83.0 84.0	78.0 79.0	15.0 14.0	42.0 22.0	40.0 20.0	18.0 35.0
X2020-T6		1/2 100	77.0 78.0	72.0 73.0	9.0 9.0	57.0 38.0	54.0 34.0	8.0 15.0

(a) Heat Treatment #1 - SHT 2 hrs. at 860°F, CWQ, Aged 24 hrs. at 250°F.  
Heat Treatment #2 - SHT 1/2 hr. at 920°F, CWQ, Aged 96 hrs. at 225°F.

(b) Compact preheat temperature 1000°F, 81% dense green compact.

(c) Compact preheat temperature 1050°F, 81% dense green compact.

(d) Compact preheat temperature 950°F, 74% dense green compact.

(e) Specimen fractured in three places.

(f) Reduced section splintered longitudinally.

Table X

IMPACT AND TEAR PROPERTIES OF APM MATERIAL

<u>Alloy Number</u>	<u>Sample Number</u>	<u>H.T. No. (a)</u>	<u>Izod Impact Energy At Failure ft-lb (b)</u>	<u>Tear Test Unit Propagation Energy in.lb/in.<sup>2</sup> (c)</u>	
				<u>Long.</u>	<u>Trans.</u>
52	283492	2	1.0	-	-
	307454	1	-	0	0
	307455	1	-	0	0
	307454	2	-	0	0
	307455	2	-	0	0
62	307321	1	1.1	0	0
	307321	2	-	0	0
64	307323	1	1.1	0	0
	307323	2	-	0	0
71	307330	1	1.3	0	0
	307330	2	-	0	0
7178-T6	Typical	-	-	140 (d)	130 (d)
7075-T6 (e)	Typical	-	4.5	290 (d)	220 (d)
195-T6	Typical	-	2	-	-
356-T6	Typical	-	1	75 (e)	-

(a) H.T. #1, SHT 2 hrs. at 860°F, CWQ, Aged 24 hrs. at 250°F.  
H.T. #2, SHT 1/2 hr. at 920°F, CWQ, Aged 96 hrs. at 225°F.

(b) Obtained from 2 in. dia. extrusions.

(c) .10 in. Thick rolled stock from 1" x 4-1/4" extrusions of alloys covered in this contract.

(d) Results of tear tests from 0.063 in. Al sheet.

(e) Unpublished data.

Table XI

## COMPARISON OF TEAR TEST PROPERTIES AFTER STEP AGING

Alloy No.	S. No.	Heat Treatment	Test Specimen (b) Thickness in.	Y.S. ksi	Tear Strength ksi	Tear/Yield Ratio	Unit Energy in-lb/in <sup>2</sup>	Product Tested
38	283451	#1	0.1000	108.2	20.8	0.19	65	Extrusion
	283451	#1A	0.1000	69.9	28.4	0.41	95	Extrusion
52	283490	#1	0.1002	105.6	29.1	0.28	100	Extrusion
	283490	#1B	0.0987	69.6	32.4	0.47	115	Extrusion
71	283387	#1C	0.1006	88.1	63.2	0.72	230	Extrusion
	293387	#1D	0.1010	71.1	44.4	0.62	120	Extrusion
7075 7075 7172	(c)	-T6	0.1000	74.7	65.0	0.87	300	Extrusion
	(c)	-T6	0.063	75.1	78.1	1.04	290	Sheet
	(c)	-T6	0.063	81.3	61.8	0.76	140	Sheet

Note: (a) Kahn Type Tear Test, Ref. 4

(b) #1 SHT 2 hours at 860°F, C<sub>40</sub>, Aged 24 Hours at 250°F.

1A Age #2 18 Hrs. at 330°F

1B Age #2 24 Hrs. at 330°F

1C Age #2 3 Hrs. at 330°F

1D Age #2 20 Hrs. at 330°F

-T6 similar to #1 except SHT---870°F

(c) Specific test values from available data. Not to be considered as typical.

Table III

## ELECTRICAL CONDUCTIVITY DATA

Alloy	S. No.	SHT (t)		Age (a)				TS ksi	YS ksi	EL, %	Electrical Conductivity, % IACS (c)
		t	T (°F)	t	T	t	T				
38	283451	2	860	24	250	--	---	112	108	1.2	< 25
38	283453	2	860	24	250	--	---	108	105	0.8	< 25
38	283453	2	860	24	250	2	330	97	93	2.1	25.9
38	283453	2	860	24	250	4	330	91	84	3.0	26.7
38	283453	2	860	24	250	8	330	87	80	3.0	27.6
38	283453	2	860	24	250	16	330	82	72	4.0	28.0
38	283451	2	860	24	250	18	330	82	70	4.0	29.3
38	283453	2	860	24	250	32	330	76	64	5.0	29.1
38	283453	2	860	24	250	48	330	73	61	5.5	29.6
38	283453	2	860	24	250	1	350	93	87	2.5	26.3
38	283453	2	860	24	250	2	350	88	82	2.5	27.3
38	283453	2	860	24	250	4	350	84	75	4.0	27.7
38	283453	2	860	24	250	8	350	79	68	4.5	28.5
38	283453	2	860	24	250	12	350	76	64	5.5	28.9
39	283460	2	860	24	250	--	---	107	104	1.3	< 25
39	283460	2	860	24	250	6	330	81	71	1.8	27.8
39	283460	2	860	24	250	32	330	73	62	2.3	28.7
39	283460	2	860	24	250	48	330	74	61	4.0	29.2
39	283460	2	860	24	250	4	350	82	75	1.8	27.3
39	283460	2	860	24	250	8	350	79	68	3.0	28.2
39	283460	2	860	24	250	12	350	77	65	3.5	28.7
49	283471	2	860	24	250	--	---	107	104	1.0	< 25
49	283471	2	860	24	250	16	330	81	77	2.1	28.2
49	283471	2	860	24	250	32	330	75	67	3.0	29.3
49	283471	2	860	24	250	48	330	72	63	3.0	29.8
49	283471	2	860	24	250	4	350	84	78	1.5	27.4
49	283471	2	860	24	250	8	350	71	66	1.4	28.7
49	283471	2	860	24	250	12	350	71	66	2.4	29.3
50	283480	2	860	24	250	--	---	100	96	1.5	< 25
50	283480	2	860	24	250	16	330	85	77	1.6	27.9
50	283480	2	860	24	250	32	330	78	70	2.4	29.0
50	283480	2	860	24	250	48	330	74	66	2.0	29.2
50	283480	2	860	24	250	4	350	85	79	1.9	27.4
50	283480	2	860	24	250	8	350	80	73	2.4	28.2
50	283480	2	860	24	250	12	350	78	69	2.4	28.8
52	307454(b)	2	860	24	250	--	---	100	95	2	28.6
52	307455(b)	2	860	24	250	--	---	103	98	2	28.2
52	283490	2	860	24	250	--	---	109	106	2.9	28.9
52	283496	2	860	24	250	--	---	107	105	2.6	28.2
52	283496	2	860	24	250	2	330	95	93	1.9	31.5
52	283496	2	860	24	250	4	330	94	90	4.2	32.5
52	283496	2	860	24	250	8	330	91	84	6.5	33.8
52	283496	2	860	24	250	16	330	84	76	6.5	34.7
52	283490	2	860	24	250	24	330	79	70	6.0	36.2
52	283496	2	860	24	250	32	330	77	68	6.0	36.0
52	283496	2	860	24	250	48	330	73	64	7.0	36.7
52	283496	2	860	24	250	1	350	95	90	5.0	32.1
52	283496	2	860	24	250	2	350	91	85	5.0	33.1
52	283496	2	860	24	250	4	350	85	77	6.5	34.1
52	283496	2	860	24	250	8	350	80	70	8.0	35.1
52	283496	2	860	24	250	12	350	77	66	8.0	36.1
52	307454(b)	0.5	920	96	225	--	---	96	---	---	27.6
52	307455(b)	0.5	920	96	225	--	---	102	97	2	28.6
62	307321(b)	2	860	24	250	--	---	104	97	3	28.4
62	307321(b)	0.5	920	96	225	--	---	79	---	---	27.3
64	307323(b)	2	860	24	250	--	---	96	90	2	27.4
64	307323(b)	0.5	920	96	225	--	---	83	---	---	26.3
71	307330(b)	2	860	24	250	--	---	100	94	4	29.2
71	293387	2	860	24	250	--	---	108	103	4.5	31.0
71	293387	2	860	24	250	2	300	103	100	5.2	32.1
71	293387	2	860	24	250	4	300	101	98	5.8	32.5
71	293387	2	860	24	250	8	300	99	95	6.5	33.5
71	293387	2	860	24	250	24	300	91	86	7.8	35.4
71	293387	2	860	24	250	48	300	86	79	9.2	36.8
71	293387	2	860	24	250	2	330	97	93	6.0	33.6
71	293387	2	860	24	250	4	330	93	88	8.0	34.6
71	293387	2	860	24	250	8	330	87	81	9.2	36.3
71	293387	2	860	24	250	16	330	82	74	9.5	37.4
71	293387	2	860	24	250	32	330	76	66	11.2	38.7
71	293387	2	860	24	250	48	330	73	62	11.5	39.2
71	307330(b)	0.5	920	96	225	--	---	98	93	3	29.0
7075-T651 (d)(e)		--	---	--	---	--	---	88	80	11	32.0
7178-T651 (d)(e)		--	---	--	---	--	---	92	84	8	32.5

## NOTE:

- (a) t = time in hours, T = temperature in °F.  
 (b) Data from 1" x 4-1/4" extrusions rolled to sheet. All other tests - 2" diameter.  
 (c) Reading taken midway between edge and center.  
 (d) Typicals from R. R. Sons 3/23/65, 2 to 3 inch diameter extrusions.  
 (e) Typicals from C. F. Babilon 4/5/66, plate and rolled rod.  
 (f) Quenched in cold water immediately after SHT



Table XIII

## EFFECT OF DISPERSOIDS ON TENSILE PROPERTIES

REF. TABLE II HEAT TREATMENT NO. 1

Alloy	Composition			Longitudinal					
	Dispersoids	Wt. %	Total	Zn	Mg	Cu	T.S., ksi	Y.S., ksi	EL. % in 4D
90	1.1 Fe, 1.0 Ni,	(0.9 Ni/Fe)	2.1	7.5	2.4	1.0	99	93	5.4
87	2.2 Fe, 2.3 Ni,	(1.0 Ni/Fe)	4.5	7.6	2.5	1.1	104	99	1.4
86	4.0 Fe, 2.1 Ni,	(0.5 Ni/Fe)	6.1	7.6	2.6	1.1	107	101	1.0
85	2.2 Fe, 4.8 Ni,	(2.2 Ni/Fe)	7.0	7.6	2.6	1.1	88	84	1.5
34	3.9 Fe, 5.3 Ni,	(1.4 Ni/Fe)	9.2	7.8	2.5	1.1	106	105	0.7
19	-----		0	7.5	3.0	1.7	101	95	7.2
9	0.5 Cr, 0.4 V, 0.4 Zr, 0.4 Ti		1.7	7.9	3.5	1.6	99	94	2.5
11	1.4 Co, 1.6 Mo, 0.5 W,		3.5	7.6	3.5	1.5	98	95	0.7
8	1.6 Mn, 1.6 Fe, 1.4 Ni,		4.3	7.9	3.5	1.5	106	102	1.4
7	2.5 Fe, 2.0 Ni,		4.5	7.9	3.4	1.5	106	102	1.4
12	0.6 Mn, 0.5 Fe, 0.6 Ni, 0.5 Cr, 0.4 Ti,								
	0.4 V, 0.6 Zr, 0.5 Co, 0.5 Mo, 0.5 W		4.8	7.8	3.5	1.6	98	95	0.7
10	2.1 Cr, 0.9 Ti, 1.7 V, 0.9 Zr,		5.6	7.9	3.5	1.5	84	84	0.0
62	0.8 Co,		0.8	10.0	4.0	0.9	107	103	2.6
52	1.4 Co,		1.4	9.7	4.1	0.8	106	103	1.8
63	2.3 Co,		2.3	10.1	4.0	0.9	104	101	0.6
68	1.5 Co,		1.5	9.9	3.9	1.5	106	101	1.6
57	2.0 Mn, 1.2 Fe, 2.6 Ni, 0.1 Ti,		5.5	9.6	4.0	1.9	106	164	0.4
28	-----		0						
58	2.2 Mn, 1.4 Fe, 3.2 Ni,		6.5	9.8	3.8	2.3	102	99	2.5
				10.0	4.1	2.5	104	--	0.0
46	0.6 Mn, 1.1 Cr, 1.0 Zr,		2.4	11.4	4.6	0.9	100	96	0.8
45	3.0 Mn, 0.8 Cr, 0.5 Zr,		4.0	11.7	4.6	0.9	101	100	0.4

Table XIV

COMPARISON OF ALLOY TYPES WITH RESPECT TO ELONGATION

REF: TABLE II, HEAT TREATMENT NO. 1

Y.S./p	Elongation, % (Alloy No.)	
	Precipitation Alloys	Dispersion & Precipitation Alloys
1.03 x 10 <sup>6</sup>	3.2 (61)	---
1.02	---	3.1 (71)
1.02	---	0 (38)
1.01	2.1 (1)	1.8 (33)
1.01	---	1.6 (69)
1.01	---	1.5 (65)
1.00	---	2.6 (62)
1.00	---	2.2 (66)
1.00	---	1.9 (67)
0.99	1.1 (3)	1.9 (60)
0.99	---	1.4 (36)
0.99	---	1.4 (99)
0.99	---	0.8 (39)
0.99	---	0.8 (70)
0.99	---	0 (35)
0.98	2.9 (2)	1.8 (52)
0.97	1.8 (4)	2.0 (64)
0.97	1.4 (29)	1.6 (68)
0.97	---	1.4 (8)
0.97	---	1.1 (50)
0.97	---	0.9 (51)
0.97	---	0.6 (63)
0.96	---	1.4 (5)
0.96	---	1.4 (7)
0.96	---	0.7 (34)
0.96	---	0.7 (40)
0.96	---	0.3 (54)
0.95	2.5 (28)	1.2 (43)
0.95	---	0.7 (48)
0.95	---	0.7 (55)
0.94	7.2 (19)	1.4 (87)
0.94	4.0 (6)	1.0 (86)
0.94	---	0.9 (56)
0.94	---	0.0 (57)
0.93	4.0 (20)	1.1 (13)
0.93	2.9 (27)	0.7 (14)

TABLE XV  
EFFECT OF COMPACT PREHEAT TIME AND TEMPERATURE  
LONGITUDINAL TENSILE PROPERTIES (2)

ALLOY (1)	S. No.	COMPACT PREHEAT		HEAT TREATMENT #1			HEAT TREATMENT #2		
		Time Hr.	TEMP. °F.	T.S. ksi	Y.S. ksi	EL. %	T.S. ksi	Y.S. ksi	EL. (%)
36	278038	1.25	750	99.9	97.2	1.2	56.4(d)	(c)	(b)
	278042	21.5	750	103.4	100.4	1.4	113.6	113.3	(b)
	277644	.25	850 (a)	102.1	97.4	1.9	110.5	107.9	(b)
	277645	.75	850 (a)	105.9	102.1	1.4	118.5	115.9	1.1
	277424	12	900	109.4	106.9	1.4	-	-	-
38	278039	2.25	750	106.5	105.0	0.7	117.7	117.2	(b)
	278043	22.25	750	107.9	104.4	1.6	119.5	118.4	(b)
	277646	1.5	850 (a)	111.1	108.4	1.0	119.6	117.0	1.2
	277647	4	850 (a)	109.9	106.9	1.8	117.4	115.7	(b)
	277426	12	900	112.3	109.7	(b)	-	-	-

Notes: (1)

ALLOY	Zn	Mg	Cu	Mn	Fe	Ni
36	10.7	3.8	2.0	1.7	--	--
38	8.4	3.6	-	1.0	0.8	2.8

(2) Each result represents one test

(3) Thermal Treatments:

#1 Pieces 2" dia. x 6.5" long from front of extrusion solution heat treated 2 hrs at 860°F, CWQ, aged immediately 24 hrs. at 250°F (H.T. #1)

#2 Quadrants of 2" dia. rod x 4.75" long were solution heat treated .5 hr at 920°F, CWQ, and aged immediately 96 hrs at 225°F (H.T. #2)

- (4)
- (a) Compact extruded without hot pressing against a blind die
  - (b) Not obtained. Specimen shattered
  - (c) Not obtained. Specimen failed before reaching 0.2% offset
  - (d) Melting occurred during solution heat treatment

Table XVI

EFFECT OF DIE QUENCHING ON TENSILE PROPERTIES

S. No.	Alloy	Composition %				Compact Preheat Hr	Quench	Thermal Treat- ment (a)	Longitudinal		
		Zn	Mg	Cu	Mn	Fe	Ni		T.S. ksi	Y.S. ksi	El. in LD %
278180	36	10.7	3.8	2.0	1.7	---	---	.75 920	Die	A	100.3 95.2 1.0
277424	36	10.7	3.8	2.0	1.7	---	---	12. 900	CWQ	B	109.4 106.9 1.4
278181(b)	38	8.4	3.6	---	1.0	.8	2.8	1.75 920	Die	A	102.8 99.5 0.7
277426	38	8.4	3.6	---	1.0	0.8	2.8	12. 900	CWQ	B	112.3 109.7 (c)

NOTE: Each result represents one tensile test

(a) Thermal treatment to -T6 temper as follows:

A. Pieces from middle of die quenched 2" dia. extrusion was aged 24 hrs. at 250°F

B. Pieces from front of 2" dia. extrusion was solution heat treated 2 hrs. at 860°F, CWQ, and aged 24 hrs. at 250°F

(b) Extruded with 7075 alloy leader

(c) Not obtained. Specimen shattered

Table XVII

## EFFECT OF POWDER SIZE ON LONGITUDINAL TENSILE PROPERTIES

Alloy	S. No.	Compact Preheat		Powder Size ( $\phi$ )	Heat Treatment #1			Heat Treatment #2		
		Time Hr.	Temp. $^{\circ}$ F.		T.S. ksi	Y.S. ksi	HL %	T.S. ksi	Y.S. ksi	HL %
34	277938	5.5	850 (a)	14	95.4	95.2	0.5	108.1	106.3	(c)
	277939	2.5	850 (a)	5	94.5	92.2	0.6	110.4	109.4	(c)
38	277936	2.75	850 (a)	13	101.5	100.5	0.6	113.1	112.6	(c)
	277937	1.75	850 (a)	6	100.5	97.7	1.1	109.1	105.2	(c)
59	277934	2.00	850 (a)	15	90.9	(b)	0.4	91.9	(b)	(c)
	277935	1.00	850 (a)	6	90.4	(b)	0.4	107.4	(b)	(c)

Notes: (1) Composition:

Alloy	Zn	Mg	Cu	Mn	Fe	Ni
36	10.6	4.4	2.0	1.7	-	-
38	8.2	3.3	-	1.1	1.1	3.0
59	10.2	3.9	1.6	1.6	1.0	4.1

(2) Each result represents one tensile test.

- (3) a. Compact extruded without hot pressing against blind die.  
 b. Not obtained. Specimen failed before reaching 0.2% offset.  
 c. Not obtained. Specimen shattered.  
 d. Thermal treatment as follows:

#1 Pieces 2" dia. x 6.5" long from front of extrusions were solution heat treated 2 hrs at 860 $^{\circ}$ F, CWQ, and aged immediately 24 hrs at 250 $^{\circ}$ F.

#2 Quarter sections of 2" dia. rod x 4.75" long were solution heat treated .5 hr at 920 $^{\circ}$ F, CWQ, and aged immediately 96 hrs at 225 $^{\circ}$ F.

e. Microns, by Fisher SSS.

Table XVIII

EFFECT OF POWDER SIZE ON TRANSVERSE TENSILE PROPERTIES

Alloy	S. No.	Time Hr.	Compact Preheat Temp. ° F.	Powder Size (d)	2" Dia. x 6.5" Long (c)			2" Dia. x .31" Long (c)		
					T.S.	Y.S.	HL	T.S.	Y.S.	HL
					kai	kai	g	kai	kai	g
36	277938	3.5	850 (a)	14	70.5	(b)	0.0	64.0	(b)	0
	277939	2.5	850 (a)	5	80.4	(b)	0.2	88.6	(b)	0
38	277936	2.75	850 (a)	13	86.2	(b)	0.0	92.3	(b)	0
	277937	1.75	850 (a)	6	93.1	(b)	0.0	96.4	(b)	0
59	277934	2.00	850 (a)	15	64.6	(b)	0.0	61.2	(b)	0
	277935	1.00	850 (a)	6	85.0	(b)	0.0	91.5	(b)	0

Notes: (1) See Section II, Table VIII for details  
(2) Composition:

Alloy	Zn	Mg	Cu	Mn	Fe	Ni
36	10.6	4.4	2.0	1.7	-	-
38	8.2	3.3	-	1.1	1.1	3.0
59	10.2	3.9	1.6	1.6	1.0	4.1

- (3) (a) Compact extruded without pressing against blind die  
(b) Failed before reaching 0.2% offset  
(c) Specimens of indicated size solution heat-treated 2 hrs @ 860°F, C&Q, aged immediately for 24 hrs at 250°F.  
(d) Microns, by Fisher 5SS.

Table XIX

EFFECT OF HEAT TREATMENT AND AGING ON TENSILE PROPERTIES OF EXTRUSIONS (b)

Longitudinal Properties	Alloy 52 S-No. 282494		Alloy 62 S-No. 307310		Alloy 64 S-No. 307312		Alloy 71 S-No. 307319	
	Front	Back	Front	Back	Front	Back	Front	Back
	Heat Treatment #2 S.H.T. 1/2 hr at 920°F, CWQ, Aged 96 hrs at 225°F		Heat Treatment #3 S.H.T. 2 hrs at 860°F, CWQ, Aged 96 hrs at 225°F		Heat Treatment #4 S.H.T. 2 hrs at 830°F, CWQ, Aged 96 hrs at 225°F (f)		Heat Treatment #5 S.H.T. 2 hrs at 830°F, CWQ, Aged 48 hrs at 250°F	
T.S., ksi	116.6	81.7(c)	116.6	116.8(g)	116.8	117.1(g)	117.1	114.4(g)
Y.S., ksi	112.2	(a)	112.2	113.7	113.7	113.1	113.1	112.0
El., % In LD	1.4	0.0(c)	1.4	2.1	2.1	1.5(g)	1.5	3.6
Y.S./Density, x 10 <sup>6</sup> in.	1.072	-	1.072	1.097	1.097	1.090	1.090	1.087
	Heat Treatment #6 S.H.T. 2 hrs at 860°F, CWQ, Aged 96 hrs at 225°F		Heat Treatment #7 S.H.T. 2 hrs at 830°F, CWQ, Aged 96 hrs at 225°F (f)		Heat Treatment #8 S.H.T. 2 hrs at 830°F, CWQ, Aged 48 hrs at 250°F		Heat Treatment #9 S.H.T. 2 hrs at 830°F, CWQ, Aged 24 hrs at 250°F	
T.S., ksi	-	-	112.2(g)	-	112.2	109.6(g)	109.6	114.7
Y.S., ksi	-	-	110.3	-	110.3	107.4	107.4	111.4
El., % In LD	-	-	1.8(g)	-	1.8	1.1(g)	1.1	2.4
Y.S./Density, x 10 <sup>6</sup> in.	-	-	1.065	-	1.065	1.035	1.035	1.082
	Heat Treatment #10 S.H.T. 2 hrs at 830°F, CWQ, Aged 96 hrs at 225°F (f)		Heat Treatment #11 S.H.T. 2 hrs at 830°F, CWQ, Aged 48 hrs at 250°F		Heat Treatment #12 S.H.T. 2 hrs at 830°F, CWQ, Aged 24 hrs at 250°F		Heat Treatment #13 S.H.T. 2 hrs at 830°F, CWQ, Aged 6 hrs at 330°F	
T.S., ksi	103.2	107.1	91.6(c)	107.1	107.1	106.2	105.9	106.0
Y.S., ksi	103.2	103.4	89.6(c)	104.9	104.9	103.5	103.2	103.4
El., % In LD	0.9	2.9	1.4(bc)	2.9	2.9	2.1	4.0	3.0
Y.S./Density, x 10 <sup>6</sup> in.	0.986	0.990	0.865(c)	1.012	1.012	0.997	0.994	0.996
	Heat Treatment #14 S.H.T. 2 hrs at 830°F, CWQ, Aged 48 hrs at 250°F		Heat Treatment #15 S.H.T. 2 hrs at 830°F, CWQ, Aged 24 hrs at 250°F		Heat Treatment #16 S.H.T. 2 hrs at 830°F, CWQ, Aged 6 hrs at 330°F		Heat Treatment #17 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F	
T.S., ksi	105.0	104.2	88.9(c)	105.2	105.2	105.2	88.1(c)	105.2
Y.S., ksi	102.5	101.8	(a,c)	103.2	102.5	102.5	(a,c)	102.5
El., % In LD	3.2	3.0	(b,c)	4.1	4.1	3.0	(b,c)	3.7
Y.S./Density, x 10 <sup>6</sup> in.	0.979	0.965	-	0.996	0.996	0.987	-	0.987
	Heat Treatment #18 S.H.T. 2 hrs at 830°F, CWQ, Aged 24 hrs at 250°F		Heat Treatment #19 S.H.T. 2 hrs at 830°F, CWQ, Aged 6 hrs at 330°F		Heat Treatment #20 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F		Heat Treatment #21 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F	
T.S., ksi	103.2	102.7	103.2	105.4	104.3	102.4	102.7	102.6
Y.S., ksi	101.0	100.5	100.5	102.9	101.7	99.8	99.5	99.6
El., % In LD	3.2	2.7	3.0	3.3	3.2	3.8	2.9	3.4
Y.S./Density, x 10 <sup>6</sup> in.	0.965	0.955	0.960	0.993	0.982	0.961	0.958	0.960
	Heat Treatment #22 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 330°F		Heat Treatment #23 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F		Heat Treatment #24 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F		Heat Treatment #25 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F	
T.S., ksi	91.8	90.5	91.2	91.3	91.6	91.7	91.8	91.8
Y.S., ksi	88.1	87.1	87.6	86.6	86.8	86.5	86.9	86.7
El., % In LD	5.4	5.8	5.6	7.7	6.2	6.6	7.0	6.8
Y.S./Density, x 10 <sup>6</sup> in.	0.841	0.832	0.837	0.836	0.841	0.838	0.837	0.835
	Heat Treatment #26 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F		Heat Treatment #27 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F		Heat Treatment #28 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F		Heat Treatment #29 S.H.T. 2 hrs at 860°F, CWQ, Aged 6 hrs at 350°F	
T.S., ksi	84.2	84.2	83.7	83.3	84.7	84.1	85.1	84.6
Y.S., ksi	78.2	78.2	77.4	76.5	78.0	77.2	77.2	76.7
El., % In LD	7.1	7.1	6.8	8.5	9.3	8.0	7.3	7.6
Y.S./Density, x 10 <sup>6</sup> in.	0.747	0.747	0.739	0.738	0.753	0.745	0.744	0.739

(a) Failed before reaching 0.5% offset.  
 (b) Failed through gage mark.  
 (c) Premature failures - not included in average.  
 (d) Average of 3 values.  
 (e) Specimen shattered.  
 (f) Letters assigned non standard heat treatments for ease in reference.  
 (g) Average 2 values.  
 (h) Heat treated as quadrants.

TABLE IX  
TENSILE PROPERTIES OF ALLOYS CONTAINING DECREASING AMOUNTS OF  
IRON AND NICKEL AND GIVEN VARIOUS AGING TREATMENTS

Alloy No. (a)	S.No.	Heat Treatment (M)			Longitudinal (b)			Transverse (b)		
		T <sub>1</sub>	T <sub>2</sub>	Quench	YS ksi	TS ksi	EL, % in 10	YS ksi	TS ksi	EL, % in 10
34	(c)	2	860	CMQ	24	250	0.997	103.1 (d)	90.7 (d)	9.4 (g)
AVE					106.5 (d)	104.5 (d)	0.7 (d)	84.1 (d)	70.5 (c)	1.5 (d)
87	293196	2	860	CMQ	24	250	0.938 (d)	90.5 (c)	82.2 (d)	1.5 (d)
AVE					101.7 (c)	103.6 (c)	0.7 (d)	87.9 (d)	89.2 (c)	2.1 (c)
90	294014	2	860	CMQ	24	250	0.931 (c)	92.5 (c)	87.6 (c)	2.1 (c)
AVE					104.6 (c)	99.1 (c)	2.6 (c)	86.5 (c)	80.7 (c)	2.3 (c)
79	293188	2	860	CMQ	24	250	0.891 (c)	87.0 (d)	80.6 (d)	4.1 (d)
AVE					99.0 (c)	93.6 (c)	5.4 (c)	90.2 (d)	81.6 (d)	3.4 (e)
34	(e)	2	860	CMQ	6/250+ 8/330		0.764 (i)	91.6 (d)	88.5 (d)	1.3 (d)
AVE					88.7 (i)	83.2 (i)	0.7 (i)	79.3 (c)	77.6 (c)	0.2 (e)
87	293196	2	860	CMQ	6/250+ 8/330		0.755 (i)	84.1 (d)	75.8 (c)	0.2 (e)
AVE					87.9 (i)	82.2 (i)	0.7 (i)	81.9 (c)	73.6 (c)	2.7 (c)
90	294014	2	860	CMQ	6/250+ 8/330		0.746 (i)	84.0 (d)	76.0 (c)	2.9 (c)
AVE					84.4 (d)	80.7 (d)	7.0 (d)	79.5 (c)	71.3 (c)	5.8 (c)
79	293188	2	860	CMQ	6/250+8/330		0.748 (i)	75.2 (c)	70.9 (c)	2.9 (c)
AVE					84.4 (d)	78.1 (d)	7.9 (d)	77.4 (c)	71.1 (c)	4.4 (c)
34	283440	2	860	CMQ	16	330	0.700 (i)	86.2 (d)	80.3 (d)	2.4 (d)
AVE					82.6 (i)	76.2 (i)	0.7 (i)	76.5 (c)	71.8 (c)	0.6 (c)
87	292196	2	860	CMQ	16	330	0.698 (i)	71.5 (c)	70.3 (c)	2.7 (c)
AVE					85.7 (i)	74.6 (i)	7.0 (i)	79.9 (c)	70.3 (c)	3.9 (c)
90	294014	2	860	CMQ	16	330	0.705 (i)	78.7 (c)	70.2 (c)	3.3 (c)
AVE					85.7 (i)	74.6 (i)	7.0 (i)	78.7 (c)	70.2 (c)	3.3 (c)
79	293188	2	860	CMQ	16	330	0.661 (i)	75.8 (c)	64.5 (c)	4.5 (c)
AVE					80.2 (i)	68.5 (i)	9.0 (i)	75.8 (c)	64.5 (c)	5.4 (c)
34	(c)	2	860	CMQ	46	315	0.671 (i)	83.4 (d)	76.8 (d)	2.2 (d)
AVE					79.1 (i)	73.1 (i)	0.7 (i)	73.0 (d)	69.1 (d)	0.5 (e)
87	293196	2	860	CMQ	48	315	0.670 (i)	77.9 (c)	72.5 (c)	1.2 (c)
AVE					82.1 (i)	73.0 (i)	0.7 (i)	77.9 (c)	72.5 (c)	1.2 (c)
90	294014	2	860	CMQ	48	315	0.657 (i)	77.6 (c)	65.9 (c)	3.8 (c)
AVE					82.9 (i)	69.5 (i)	10.0 (i)	77.6 (c)	65.9 (c)	3.8 (c)
79	293188	2	860	CMQ	48	315	0.653 (i)	71.7 (c)	61.5 (c)	3.9 (c)
AVE					82.9 (i)	69.5 (i)	10.0 (i)	71.7 (c)	61.5 (c)	3.9 (c)

(a) Alloy 34, Al 7.8 Zn, 2.5 Mg, 1.0 Cu, 3.5 Fe, 4.9 Ni, 0.09 Cr  
87, Al 7.6 Zn, 2.5 Mg, 1.0 Cu, 2.2 Fe, 4.3 Ni, 0.16 Cr  
90, Al 7.5 Zn, 2.5 Mg, 1.0 Cu, 1.0 Fe, 4.3 Ni, 0.20 Cr  
79, Al 6.8 Zn, 2.8 Mg, 2.1 Cu, 1.0 Ni, 0.5 Cr, 0.01 Zn

(b) One specimen per test unless noted.

(c) Avg. of 3 or more tests.

(d) Highest value of 3 or more test results.

(e) Lowest value of 3 or more test results.

(f) Failed before reaching 0.2% offset.

(g) Not obtained-specimen shattered.

(h) t = time in hours, T = Temp. in °F

(i) Data from S. No. 283445



TABLE XXI

THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 2

S. NO. 277374 (a)

Additional Aging Time at 330°F-Hrs. (b)	Data Source	Longitudinal				Transverse			Unstressed			Stressed to 25% TS			Stressed to 50% TS			Stressed to 75% TS		
		TS, ksi	YS, ksi	El, % In 4D	YS/Density x10 <sup>6</sup> In.	TS, ksi	YS, ksi	El, % In 4D	TS, ksi	% Change	TS, ksi	% Change	TS, ksi	% Change	TS, ksi	Days to Failure	% Change	TS, ksi	Days to Failure	% Change
0	(c)	105.6	103.5	2.9	0.984	75.9	(d)	0.0	--	--	--	--	--	--	--	--	--	--	--	--
	(e)	105.6	103.1	2.9	0.980	85.1	(d)	0.0	--	--	--	--	--	--	--	--	--	--	--	--
	(e)	106.0	103.2	3.2	0.981	83.1	(f)	0	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	105.7	102.7	3.2	0.976	73.0	(f)	0 (h)	--	--	--	--	--	--	--	--	--	--	--	--
4	(e)	91.5	89.7	4.7	0.853	77.0	(f)	0 (h)	--	--	--	--	--	--	--	--	--	--	--	--
	(e)	91.5	89.7	4.3	0.853	73.7	(f)	0	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	91.6	89.7	4.5	0.853	75.4	--	0	--	--	--	--	--	--	--	--	--	--	--	--
	Predicted	86	85	5	--	74	--	1	--	--	--	--	--	--	--	--	--	--	--	--
6	(g)	--	--	--	--	76.3	(f)	0.4	64.0	-17	41.2	-47	68 Da	-100	13 Da	-100	13 Da	-100	13 Da	-100
	(g)	--	--	--	--	78.1	(f)	0.6	57.4	-26	59.2	-23	40 Da	-100	11 Da	-100	11 Da	-100	11 Da	-100
	Avg.	--	--	--	--	77.2	--	0.5	60.7	-22	50.2	-35	--	--	--	--	--	--	--	--
	Predicted	85	81	6	--	73	71	1	54.6	-27	58.5	-22	39.7	-47	13 Da	-100	13 Da	-100	13 Da	-100
16	(e)	79.4	73.2	7.5	0.696	69.8	66.2	1.2	54.6	-17	58.5	-22	39.7	-47	13 Da	-100	13 Da	-100	13 Da	-100
	(e)	79.4	72.5	8.0	0.689	68.1	65.8	0.4	61.8	-17	60.7	-19	38.9	-48	15 Da	-100	15 Da	-100	15 Da	-100
	Avg.	79.4	72.8	7.6	0.692	69.0	66.0	1.2	58.2	-22	59.6	-20	39.3	-47	--	--	--	--	--	--
	Predicted	78	71	8	--	68	65	1	--	--	--	--	--	--	--	--	--	--	--	--
18	(g)	--	--	--	--	69.8	67.4	1.1	58.0	-17	59.7	-14	55.9	-20	57 Da	-100	57 Da	-100	57 Da	-100
	(g)	--	--	--	--	69.8	67.6	0.9	53.5	-23	58.1	-17	56.6	-19	41 Da	-100	41 Da	-100	41 Da	-100
	Avg.	--	--	--	--	69.8	67.5	1.0	55.8	-20	58.9	-16	56.2	-19	--	--	--	--	--	--
	Predicted	74	66	9	--	65	62	1	--	--	--	--	--	--	--	--	--	--	--	--
26	(e)	--	--	--	--	65.6	63.2	0.9	49.6	-25	54.1	-18	53.4	-19	61 Da	-100	61 Da	-100	61 Da	-100
	(e)	--	--	--	--	67.0	63.2	1.7	57.4	-13	56.6	-15	53.7	-19	46.1	-30	46.1	-30	46.1	-30
	Avg.	--	--	--	--	66.3	63.2	1.3	54.5	-19	55.4	-17	53.6	-19	--	--	--	--	--	--
	Predicted	74	66	9	--	65	62	1	--	--	--	--	--	--	--	--	--	--	--	--
40	(e)	70.8	61.4	6.5	0.584	61.5	57.1	1.3	--	--	--	--	--	--	--	--	--	--	--	--
	(e)	71.2	61.7	9.5	0.587	61.6	57.0	1.4	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	71.0	61.6	9.0	0.586	61.6	57.0	1.4	--	--	--	--	--	--	--	--	--	--	--	--
	Predicted	71.0	61.6	9.0	0.586	61.6	57.0	1.4	--	--	--	--	--	--	--	--	--	--	--	--

Notes: (a) 12-1 Zn, 3.5 Mg, 1.5 Cu, 0.5 Mn  
 (b) 500 hrs. at 330°F, CWG, Aged #1 24 hrs. at 250°F as 2 in. dia. piece.  
 (c) Table is original tests  
 (d) Strain follower not used to avoid breaking it.  
 (e) Step Aged Data  
 (f) Failed before reaching 0.2% Offset.  
 (g) From stress corrosion data.  
 (h) Failed at or outside gage length.

TABLE XXII

THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 3

Additional Aging Time at 310°P-Hrs. (b)	Data Source	S.No. 271375 (a)									
		Longitudinal					Transverse				
		YS		El. %		VS/Density x10 <sup>6</sup> In.	YS		El. %		TS
		ksi	ksi	In 4D	In 4D		ksi	ksi	In 4D	In 4D	ksi
0	(c)	106.7	104.0	1.4	1.4	0.988	84.2	(d)	(e)		84.2
	(c)	105.6	103.1	0.7	0.7	0.980	90.2	(d)	0.0		90.2
	(f)	107.6	104.1	2.2	0.9	0.990	87.7	86.4	0.9		87.7
	(f)	108.2	104.8	2.3	0	0.996	82.8	(h)	0		82.8
	Avg.	107.0	104.0	1.9	0.3	0.988	86.4	86.4	0.3		86.4
3	Predicted	102	99	3	3	--	82	81	1		82
	(g)	--	--	--	0.4	--	77.8	(h)	0.4		77.8
	(g)	--	--	--	0.4	--	81.6	81.3	0.4		81.6
	(g)	--	--	--	0.4	--	79.7	--	0.4		79.7
	Avg.	--	--	--	--	--	79.7	--	--		79.7
4	(f)	92.8	89.8	3.6	0.854	0.854	51.2	79.8	0.7		51.2
	(f)	92.6	89.7	3.0	0.853	0.853	80.7	78.7	1.0		80.7
	(f)	92.7	89.8	3.3	0.854	0.854	81.0	79.2	0.8		81.0
	Predicted	89	85	4	--	--	77	74	1		77
	(g)	--	--	--	--	--	80.8	77.9	0.7		80.8
7	(g)	--	--	--	--	--	80.2	77.9	0.5		80.2
	(g)	--	--	--	--	--	80.5	77.9	0.6		80.5
	(g)	--	--	--	--	--	69.3	65.2	1.0		69.3
	(g)	--	--	--	--	--	69.7	65.7	0.8 (1)		69.7
	Avg.	--	--	--	--	--	69.6	65.4	0.9		69.6
16	(f)	79.8	73.0	5.8	0.694	0.694	69.3	65.2	1.0		69.3
	(f)	79.8	72.9	5.4	0.693	0.693	69.6	65.4	0.9		69.6
	(f)	79.8	73.0	5.6	0.694	0.694	69.6	65.4	0.9		69.6
	Predicted	79	72	6	--	--	69	65	1		69
	(g)	--	--	--	--	--	72.0	68.1	0.7 (1)		72.0
17	(g)	--	--	--	--	--	71.5	68.2	0.7		71.5
	(g)	--	--	--	--	--	71.8	68.2	0.7		71.8
	(g)	--	--	--	--	--	66	62	1		66
	(g)	--	--	--	--	--	71.3	65.3	2.0		71.3
	Avg.	--	--	--	--	--	69.3	65.7	1.1		69.3
20	(f)	75	67	6	--	--	70.3	65.5	1.6		70.3
	(f)	75	67	6	--	--	66	62	1		66
	(f)	75	67	6	--	--	71.3	65.3	2.0		71.3
	(f)	75	67	6	--	--	69.3	65.7	1.1		69.3
	Avg.	--	--	--	--	--	70.3	65.5	1.6		70.3
40	(f)	71.2	61.4	6.5	0.584	0.584	61.5	57.5	1.2		61.5
	(f)	71.9	61.9	6.5	0.588	0.588	63.6	57.0	1.7		63.6
	(f)	71.6	61.6	6.5	0.586	0.586	62.6	57.2	1.4		62.6
	Predicted	75	67	6	--	--	66	62	1		66
	(g)	--	--	--	--	--	71.3	65.3	2.0		71.3

Notes: (a) 12.3 Zn, 4.0 Mg, 1.6 Cu, 0.5 Mn

(b) SH: 2 hrs. at 860°P, CWG, Age #1 24 hrs. at 250°P as 2 in. dia. pieces.

(c) Original evaluation, Table II

(d) Strain follower not used to avoid breaking it.

(e) Not obtained - specimen shattered.

(f) Step aging tests.

(g) Stress corrosion data from (f)

(h) Failed before reaching of 2% offset.

(i) Specimen failed outside gage length.

TABLE XXIII

THE EFFECT OF STEP AGING OF TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 4

Additional Aging Time at 1300° F.	Data Source	S.No. 277376 (a)									
		Longitudinal					Transverse				
		Yield		El. %		Yield Density x10 <sup>3</sup> in.	Yield		El. %		Yield Density x10 <sup>3</sup> in.
		ksi	ksi	ksi	ksi		ksi	ksi	ksi	ksi	
0		103.5	105.7	2.1	2.1	0.967	72.9	(d)	0.0	0.0	
		101.1	99.6	1.9	1.9	0.963	55.5	(d)	0.0	0.0	
		102.7	100.8	2.0	2.0	0.966	72.7	(h)	0.1(g)	0.1(g)	
		105.7	103.9	2.1	2.1	0.968	82.5	(h)	0.1(g)	0.1(g)	
Avg.		103.1	103.4	2.0	2.0	0.964	80.2	--	0.0	0.0	
4		94.7	89.1	5.5	5.5	0.856	77.7	77.7	0.4	0.4	
		93.7	88.7	6.1	6.1	0.852	72.4	(h)	0.0	0.0	
Avg.		94.2	88.9	5.8	5.8	0.854	75.0	77.7	0.2	0.2	
6	Predicted	86	83	--	--	--	77	74	1	1	
		--	--	--	--	--	71.6	(h)	0.1(g)	0.1(g)	
		--	--	--	--	--	73.0	(h)	0.2	0.2	
Avg.		--	--	--	--	--	72.3	--	0.2	0.2	
8	Predicted	86	81	6	6	--	72	71	1	1	
		--	--	--	--	--	76.6	75.2	0.5	0.5	
		--	--	--	--	--	76.1	75.2	0.7	0.7	
Avg.		--	--	--	--	--	76.4	75.2	0.6	0.6	
16		86	71.4	7.1	7.1	0.686	70.7	67.1	1.2	1.2	
		82.2	72.3	7.5	7.5	0.695	68.1	66.1	1.1	1.1	
Avg.		84.0	71.8	7.3	7.3	0.690	69.4	66.6	1.2	1.2	
24	Predicted	76	71	--	--	--	68	65	1	1	
		--	--	--	--	--	69.7	67.2	0.9	0.9	
		--	--	--	--	--	69.2	66.7	0.9	0.9	
Avg.		--	--	--	--	--	69.4	67.0	0.8	0.8	
40	Predicted	76	65	7	7	--	65	62	1	1	
		--	--	--	--	--	63.4	61.6	0.9(g)	0.9(g)	
		--	--	--	--	--	70.7	64.1	0.3	0.3	
Avg.		--	--	--	--	--	67.0	63.0	0.6	0.6	
48		73.3	61.6	7.5	7.5	0.681	63.3	57.3	2.0	2.0	
		73.4	61.1	4.5	4.5	0.654	58.0	57.0	0.2 (g)	0.2 (g)	
Avg.		73.4	61.4	6.0	6.0	0.667	60.6	57.0	1.1	1.1	

Notes: (a) 2.6 Zn, 3.2 Mg, 0.3 Cu, 0.2 Sn.  
 (b) 2 hrs. at 1300° F., 24 hrs. at 2500° F. as 2 in. dia. piece.  
 (c) Original Evaluation - Table 17  
 (d) Grain follower not used to avoid breaking it.  
 (e) Step Aging Data.  
 (f) Stress corrosion data.  
 (g) Specimen failed at or outside gage mark.  
 (h) Specimen failed before reaching 0.2% offset.

Table XXIV

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING: RESISTANCE OF ALLOY 5

S. No. 277377 (4)

Additional Aging Time at 330°F, Hrs(b)	Data Source	Longitudinal				Transverse				Unstressed		Stressed		Stressed 50% T.S.		Stressed 75% T.S.	
		T.S. ksi	Y.S. ksi	El. %	in LD	T.S. ksi	Y.S. ksi	El. %	in LD	T.S. ksi	Change %	T.S. ksi	Change %	T.S. ksi	Days to Failure	T.S. ksi	Days to Failure
0	(c)	103.8	101.0	1.4	0.951	70.6	(d)	0.0	0.0	---	---	---	---	---	---	---	---
	(c)	107.4	103.7	1.4	0.976	69.9	(d)	0.0	0.0	---	---	---	---	---	---	---	---
	(e)	107.3	103.7	1.4	0.976	74.8	(d)	0	0	---	---	---	---	---	---	---	---
	(e)	106.8	100.6	3.0	0.947	5.6	(h)	0	0	---	---	---	---	---	---	---	---
	Avg.	106.3	102.2	1.6	0.962	77.7	---	---	---	---	---	---	---	---	---	---	---
4	(e)	93.5	86.0	3.1(g)	0.810	70.5	(d)	0	0	---	---	---	---	---	---	---	---
	(e)	96.1	88.8	4.0	0.836	74.9	(d)	0	0	---	---	---	---	---	---	---	---
	Avg.	94.8	87.4	3.6	0.823	72.7	---	---	---	---	---	---	---	---	---	---	---
16	(e)	85.2	72.8	6.0	0.685	56.1	66.1	0.4	0.4	---	---	---	---	---	---	---	---
	(e)	84.5	72.0	3.5(g)	0.584	67.5	57.8	0.8	0.8	---	---	---	---	---	---	---	---
	Avg.	84.8	72.7	4.8	0.685	66.8	66.0	0.6	0.6	---	---	---	---	---	---	---	---
17	Predicted	84	72	5	---	66	65	<1	<1	---	---	---	---	---	---	---	---
	(f)	---	---	---	---	67.0	(h)	0.7	0.7	54.6	-19	40.6	-40	22.9	3 days	---	-100
	Avg.	---	---	---	---	67.6	67.5	0.4(g)	0.6	49.8	-26	38.9	-42	47.4	9 days	---	-100
25	Predicted	81	67	5	---	64	62	1	1	---	---	---	---	---	---	---	---
	(f)	---	---	---	---	64.4	63.4	0.7(g)	0.8	51.6	-21	39.6	-39	51.9	44 days	---	-100
	Avg.	---	---	---	---	65.5	63.6	0.8	0.8	55.3	-15	52.5	-19	41.2	11 days	---	-100
40	(e)	76.8	61.8	7.0	0.582	61.8	57.1	1.5	1.5	---	---	---	---	---	---	---	---
	(e)	77.8	62.3	7.0	0.587	59.5	56.6	1.3	1.3	---	---	---	---	---	---	---	---
	Avg.	77.3	62.0	7.0	0.584	60.6	56.8	1.4	1.4	---	---	---	---	---	---	---	---

NOTE: (a) Al, 10.3 Zn, 2.9 Mg, 2.1 Cu, 1.8 Mn, 0.1 Cr  
 (b) SHT 2 hrs at 840°F, OAO, Age #1 24 hrs at 250°F as 2 in. dia. pieces  
 (c) Original evaluation - Table II  
 (d) Strain follower not used to avoid breaking it  
 (e) Step aging data  
 (f) Stress corrosion data  
 (g) Specimen failed outside gage length  
 (h) Specimen failed before reaching 0.2% offset

Table XXV

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 6

S. No. 277378 (a)

Additional Aging Time at 250°F, Hrs(b)	Data Source	Longitudinal		Transverse		Unstressed		Stressed 25% T.S.		Stressed 50% T.S.		Stressed 75% T.S.	
		T.S. ksi	El., % in 10 <sup>6</sup> in	T.S. ksi	El., % in 10 <sup>6</sup> in	T.S. ksi	% Change	T.S. ksi	% Change	T.S. ksi, or Days to Failure	% Change	T.S. ksi, or Days to Failure	% Change
0	(c)	104.2	99.9	4.3	0.942	87.1	(d)	0.0	---	---	---	---	---
	(c)	103.7	98.9	3.6	0.933	89.0	85.7	1.2	---	---	---	---	---
	(e)	106.3	100.6	7.0	0.949	83.9	82.8	0.2(g)	---	---	---	---	---
	(e)	---	---	---	---	84.2	83.2	0.4(g)	---	---	---	---	---
	Avg.	104.7	99.8	5.0	0.942	86.0	83.9	0.6	---	---	---	---	---
4	(e)	93.1	89.0	7.0	0.840	78.0	76.6	0.2(g)	---	---	---	---	---
	(e)	---	---	---	---	79.6	77.9	0.4(g)	---	---	---	---	---
	Avg.	93.1	89.0	7.0	0.840	78.8	77.2	0.3	---	---	---	---	---
6	Predicted	90	85	8	---	76	74	1	---	---	---	---	---
	(f)	---	---	---	---	78.8	76.3	1.4	---	---	---	---	---
	(f)	---	---	---	---	79.1	76.4	1.8	---	---	---	---	---
	Avg.	---	---	---	---	79.0	76.4	1.6	---	---	---	---	---
16	(e)	80.8	73.0	10.5	0.689	70.2	65.8	2.0	---	---	---	---	---
	(e)	---	---	---	---	69.8	65.8	2.8	---	---	---	---	---
	Avg.	80.8	73.0	10.5	0.689	70.0	65.8	2.4	---	---	---	---	---
	Predicted	80	72	10	---	69	65	2	---	---	---	---	---
17	(f)	---	---	---	---	68.8	67.1	1.7	---	---	---	---	---
	(f)	---	---	---	---	71.5	67.1	2.6(g)	---	---	---	---	---
	(f)	---	---	---	---	70.2	67.1	2.2	---	---	---	---	---
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---
24	Predicted	77	69	12	---	68	62	3	---	---	---	---	---
	(f)	---	---	---	---	69.9	64.6	2.1(g)	---	---	---	---	---
	(f)	---	---	---	---	68.7	64.2	1.9(g)	---	---	---	---	---
	Avg.	---	---	---	---	69.3	64.4	2.0	---	---	---	---	---
40	(e)	75.6	65.9	10.5	0.622	66.8	60.0	3.7	---	---	---	---	---
	(e)	---	---	---	---	67.4	59.5	3.9	---	---	---	---	---
	(e)	---	---	---	---	67.1	59.8	3.8	---	---	---	---	---
	Avg.	75.6	65.9	10.5	0.622	---	---	---	---	---	---	---	---

NOTE: (a) Al, 11.2 Zn, 2.4 Mg, 2.0 Cu, 0.2 Cr  
 (b) 5HT 2 hrs at 860°F, OAG, Age #1 24 hrs at 250°F as 2 in dia. piece  
 (c) Original evaluation - Table II  
 (d) Strain follower not used to avoid breaking it  
 (e) Step aging data  
 (f) Stress corrosion data  
 (g) Specimen failed outside gage length

Table XVI

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 19

S. No. 277391 (a)

Additional Aging Time at 350°F, hrs (b)	Data Source	Longitudinal			Transverse			Unstressed		Stressed		Stressed 50% T.S.		Stressed 75% T.S.	
		Y.S. ksi	El., in	Y.S./Density $\times 10^6$ in	Y.S. ksi	El., in	Y.S. ksi	Y.S. ksi	% Change	Y.S. ksi	% Change	Y.S., ksi, or Days to Failure	% Change	Y.S., ksi, or Days to Failure	% Change
0	(c)	101.9	95.9	6.4	84.1	(d)	0.0	---	---	---	---	---	---	---	---
	(c)	100.6	94.7	7.9	87.5	(d)	1.9	---	---	---	---	---	---	---	---
	(e)	102.4	97.0	7.0	86.6	78.0	2.2	---	---	---	---	---	---	---	---
	(e)	102.2	96.9	7.0	85.9	78.5	0.2	---	---	---	---	---	---	---	---
	Avg.	101.8	96.1	7.1	86.0	78.2	1.1	---	---	---	---	---	---	---	---
4	(e)	93.3	88.4	9.0	84.0	77.5	3.1(g)	---	---	---	---	---	---	---	---
	(e)	93.4	88.4	9.5	83.3	77.5	3.3	---	---	---	---	---	---	---	---
	Avg.	93.4	88.4	9.2	83.6	77.5	3.2	---	---	---	---	---	---	---	---
	Predicted	88	83	10	81	74	4	---	---	---	---	---	---	---	---
	(f)	---	---	---	79.4	74.0	2.8	72.5	-10	67.7	-16	60 days	-100	24 days	-100
12	(f)	---	---	---	81.3	74.4	4.2	71.6	-11	67.9	-16	44 days	-100	32 days	-100
	(f)	---	---	---	80.4	74.2	3.5	72.0	-10	67.8	-16	---	-100	---	-100
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	85	79	10	78	71	5	---	---	---	---	---	---	---	---
	(f)	---	---	---	76.3	72.0	2.9	70.1	-9	68.6	-11	54 days	-100	34 days	-100
16	(e)	82.0	74.5	10.5	76.4	67.4	6.6	---	---	---	---	---	---	---	---
	(e)	83.1	75.9	10.5	74.8	66.5	3.5(g)	---	---	---	---	---	---	---	---
	Avg.	82.6	75.2	10.5	75.6	67.0	5.0	---	---	---	---	---	---	---	---
	Predicted	81	73	11	74	65	6	---	---	---	---	---	---	---	---
	(f)	---	---	---	76.3	67.9	5.2	66.3	-13	60.7	-20	34.9	-54	44 days	-100
19	(f)	---	---	---	76.0	67.9	4.2	67.1	-12	64.8	-15	49.8	-35	50 days	-100
	(f)	---	---	---	76.2	67.9	4.7	66.7	-12	62.8	-18	42.4	-44	---	-100
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	78	70	11	72	62	7	---	---	---	---	---	---	---	---
	(f)	---	---	---	74.6	65.2	5.7	64.8	-12	60.6	-18	42.7	-42	57 days	-100
25	(f)	---	---	---	73.4	65.6	3.9	65.1	-12	59.5	-20	46.0	-38	47 days	-100
	(f)	---	---	---	74.0	65.4	4.8	65.0	-12	60.0	-19	44.4	-40	---	-100
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	75.2	65.2	12.0	70.7	59.1	8.4	---	---	---	---	---	---	---	---
	(e)	74.1	63.8	12.5	70.7	59.1	9.2	---	---	---	---	---	---	---	---
40	(e)	74.6	64.5	12.2	70.7	59.1	8.8	---	---	---	---	---	---	---	---
	(e)	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	(f)	---	---	---	---	---	---	---	---	---	---	---	---	---	---

NOTE: (a) Al, 7.5 Zn, 3.0 Mg, 1.7 Cu

(b) SHT 2 hrs at 660°F, CAG, Age #1 24 hrs at 250°F as 2 in dia. pieces

(c) Original evaluation - Table II

(d) Strain follower not used to avoid breaking it

(e) Step aging data

(f) Stress corrosion data

(g) Specimen failed outside gage length

Table XXV

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 20

S. No. 277392 (a)

Additional Aging Time at 300°F, Hrs(b)	Data Source	Longitudinal			Transverse			Unstressed		Stressed 25% T.S.		Stressed 50% T.S.		Stressed 75% T.S.	
		T.S. ksi	El., % in 10 in	Y.S./Density x 10 <sup>6</sup> in	T.S. ksi	Y.S. ksi	El., % in 10 in	T.S. ksi	% Change	T.S. ksi	% Change	T.S., ksi, or Days to Failure	% Change	T.S., ksi, or Days to Failure	% Change
0	(c)	101.0	95.2	4.3	84.2	(d)	0.0	---	---	---	---	---	---	---	---
	(c)	99.6	94.4	3.6	90.3	(d)	3.7	---	---	---	---	---	---	---	---
	(e)	102.4	97.2	6.5	86.6	79.5	2.2	---	---	---	---	---	---	---	---
	(e)	102.4	97.1	6.5	85.3	79.5	1.2	---	---	---	---	---	---	---	---
	Avg.	101.4	96.0	5.2	86.6	79.5	1.8	---	---	---	---	---	---	---	---
4	(e)	95.0	90.1	7.0	81.6	78.3	1.1	---	---	---	---	---	---	---	---
	(e)	95.4	90.1	7.0	83.6	77.9	1.6	---	---	---	---	---	---	---	---
	(e)	95.2	90.1	7.0	82.6	78.1	1.3	---	---	---	---	---	---	---	---
	Avg.	95.2	90.1	7.0	82.6	78.1	1.3	---	---	---	---	---	---	---	---
8	Predicted	91	85	8	80	74	2	---	---	---	---	---	---	---	---
	(f)	---	---	---	79.5	76.4	1.2	---	---	---	---	---	---	---	---
	(f)	---	---	---	79.4	76.4	1.0	---	---	---	---	---	---	---	---
	Avg.	---	---	---	79.4	76.4	1.1	---	---	---	---	---	---	---	---
13	Predicted	87	81	8	78	71	3	---	---	---	---	---	---	---	---
	(f)	---	---	---	80	73.2	3.9	---	---	---	---	---	---	---	---
	(f)	---	---	---	77.1	73.2	1.6	---	---	---	---	---	---	---	---
	Avg.	---	---	---	78.9	73.2	2.8	---	---	---	---	---	---	---	---
16	(e)	85.5	78.6	8.0	76.4	69.3	2.9	---	---	---	---	---	---	---	---
	(e)	85.3	78.3	9.0	77.2	69.5	3.0	---	---	---	---	---	---	---	---
	(e)	85.4	78.4	8.5	76.8	69.4	3.0	---	---	---	---	---	---	---	---
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---
28	Predicted	81	72	9	73	65	4	---	---	---	---	---	---	---	---
	(f)	---	---	---	74.4	66.7	3.1	---	---	---	---	---	---	---	---
	(f)	---	---	---	75.0	66.7	3.1	---	---	---	---	---	---	---	---
	Avg.	---	---	---	74.7	66.7	3.1	---	---	---	---	---	---	---	---
40	(e)	77.8	68.5	10.0	71.4	62.1	4.6	---	---	---	---	---	---	---	---
	(e)	76.6	66.9	10.5	71.2	62.1	3.7	---	---	---	---	---	---	---	---
	(f)	---	---	---	72.8	63.9	4.4	---	---	---	---	---	---	---	---
	(f)	---	---	---	72.8	63.4	4.4	---	---	---	---	---	---	---	---
	Avg.	77.2	67.7	10.2	72.0	62.9	4.3	---	---	---	---	---	---	---	---

NOTE: (a) Al, 7.0 Zn, 3.6 Mg, 2.3 Cu  
 (b) SHT 2 hrs at 860°F, CQ, Age #1 24 hrs at 250°F as 2 in dia. piece  
 (c) Original data - Table II

(d) Strain follower not used to avoid breaking it  
 (e) Step aging data  
 (f) Stress corrosion data

Table XXVIII

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 28

S. No. 277400 (a)

Additional Aging Time at 330°C-Hrs. (b)	Data Source	Longitudinal			Transverse			Unstressed			Stressed 25% TS			Stressed 50% TS			Stressed 75% TS		
		TS ksi	YS ksi	El. % In. 4D	YS ksi	TS ksi	YS ksi	TS ksi	% Change	TS, ksi or Days to Failure	% Change	TS, ksi or Days to Failure	% Change	TS, ksi or Days to Failure	% Change	TS, ksi or Days to Failure	% Change	TS, ksi or Days to Failure	% Change
0	(c)	102.7	99.9	2.9	89.0	89.2	(d)	0.0	--	--	--	--	--	--	--	--	--	--	--
	(e)	101.2	97.7	2.1	89.2	89.2	(d)	1.9	--	--	--	--	--	--	--	--	--	--	--
	(e)	103.8	100.2	2.8	87.4	87.4	83.1	1.3	--	--	--	--	--	--	--	--	--	--	--
	(e)	103.5	99.9	3.1	85.6	85.6	81.5	1.1	--	--	--	--	--	--	--	--	--	--	--
	Avg.	102.8	99.4	2.7	87.6	87.6	82.3	1.1	--	--	--	--	--	--	--	--	--	--	--
1	Predicted	101	97	3	85	85	81	1.1	--	--	--	--	--	--	--	--	--	--	--
	(f)	--	--	--	84.8	84.8	81.3	0.9(g)	-20	68.8	-20	50.6	-41	1 Da	-100	1 Da	-100	1 Da	-100
	(f)	--	--	--	86.7	86.7	81.4	1.0	-17	71.2	-17	65.1	-24	1 Da	-100	1 Da	-100	1 Da	-100
	Avg.	--	--	--	85.8	85.8	81.4	1.0	-18	70.0	-18	57.8	-32	--	-100	--	-100	--	-100
4	(e)	93.1	89.7	5.0	81.7	81.7	78.8	1.3	--	--	--	--	--	--	--	--	--	--	--
	(e)	93.2	90.1	3.8	80.6	80.6	78.3	0.9	--	--	--	--	--	--	--	--	--	--	--
	Avg.	93.2	89.9	4.4	81.2	81.2	78.6	1.1	--	--	--	--	--	--	--	--	--	--	--
8	Predicted	87	83	5	77	77	74	1	--	--	--	--	--	--	--	--	--	--	--
	(f)	--	--	--	79.8	79.8	76.2	1.4	-21	63.0	-21	62.0	-23	--	-100	11 Da	-100	11 Da	-100
	(f)	--	--	--	80.4	80.4	76.9	1.4	-26	59.5	-26	64.9	-19	37 Da	-100	11 Da	-100	11 Da	-100
	Avg.	--	--	--	80.1	80.1	76.6	1.4	-24	61.2	-24	63.4	-21	--	-100	--	-100	--	-100
16	(e)	81.5	75.0	6.5	72.3	72.3	67.9	0.9	--	--	--	--	--	--	--	--	--	--	--
	(e)	82.4	75.6	5.5	72.9	72.9	68.3	1.9	--	--	--	--	--	--	--	--	--	--	--
	Avg.	82.0	75.3	6.0	72.6	72.6	68.1	1.4	--	--	--	--	--	--	--	--	--	--	--
22	Predicted	79	71	6	70	70	65	1	--	--	--	--	--	--	--	--	--	--	--
	(f)	--	--	--	73.0	73.0	68.2	2.5	-20	57.6	-20	58.8	-19	57.1	-21	32 Da	-100	32 Da	-100
	(f)	--	--	--	71.5	71.5	68.2	1.5	-21	57.1	-21	56.3	-22	42.7	-41	29 Da	-100	29 Da	-100
	Avg.	--	--	--	72.2	72.2	68.2	2.0	-21	57.4	-21	57.6	-20	49.9	-31	--	-100	--	-100
32	Predicted	76	67	7	67	67	62	1	--	--	--	--	--	--	--	--	--	--	--
	(f)	--	--	--	65.1	65.1	63.2	0.9	-21	53.6	-21	56.2	-17	54.1	-20	26 Da	-100	26 Da	-100
	(f)	--	--	--	71.1	71.1	64.5	2.8	-21	53.9	-21	55.7	-18	53.4	-21	49 Da	-100	49 Da	-100
	Avg.	--	--	--	68.0	68.0	63.8	1.8	-21	53.8	-21	56.0	-18	53.8	-21	--	-100	--	-100
40	(e)	74.7	65.7	8.5	66.0	66.0	60.9	1.6(g)	--	--	--	--	--	--	--	--	--	--	--
	(e)	73.5	64.4	8.0	65.2	65.2	60.0	1.7	--	--	--	--	--	--	--	--	--	--	--
	Avg.	74.1	65.0	8.2	65.6	65.6	60.4	1.6	--	--	--	--	--	--	--	--	--	--	--

Notes: (a) Al-2.8 Zn, 3.6 Mg, 2.3 Cu.

(b) SHT 2 hrs. at 860°C, CMQ, Age #1, 24 hrs. at 350°C as 2 in. dia. piece.

(c) Original Data - Table II, Section III

(d) Strain follower not used to avoid breaking it.

(e) Step Aging Data.

(f) Stress corrosion data.

(g) Specimen failed through gage mark.



Table XXIX

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE ON ALLOY 33

S. No. 277405 (a)

Additional Aging Time at 300°F, Hrs(b)	Data Source	Longitudinal			Transverse			Unstressed		Stressed		Stressed 50% T.S.		Stressed 75% T.S.	
		T.S., ksi	Y.S., ksi	El., % in 10 <sup>2</sup> in	T.S., ksi	Y.S., ksi	El., % in 10 <sup>2</sup> in	T.S., ksi	Change %	T.S., ksi	Change %	T.S., ksi, or Days to Failure	Change %	T.S., ksi, or Days to Failure	Change %
0	(c)	107.7	124.0	2.1	81.8	(d)	(e)	---	---	---	---	---	---	---	---
	(c)	108.2	104.4	1.4	82.9	(d)	0.0	---	---	---	---	---	---	---	---
	(f)	109.6	105.1	2.7	74.4	(h)	0	---	---	---	---	---	---	---	---
	(f)	108.4	104.5	2.4	76.9	(h)	0	---	---	---	---	---	---	---	---
	Avg.	108.5	104.5	2.2	79.0	---	0	---	---	---	---	---	---	---	---
4	(f)	96.5	90.6	5.3	75.0	(h)	0	---	---	---	---	---	---	---	---
	(f)	56.3	89.8	4.9	67.1	(h)	0	---	---	---	---	---	---	---	---
	Avg.	56.4	90.2	5.1	71.0	---	0	---	---	---	---	---	---	---	---
12	Predicted	87	79	7	71	---	<1	---	---	---	---	---	---	---	---
	(g)	---	---	---	70.1	(h)	1.2	60.4	-13	57.0	-18	75 days	-100	11 days	-100
	(g)	---	---	---	68.9	(h)	0.3(d)	63.9	-8	60.2	-13	58.1	-16	11 days	-100
	Avg.	---	---	---	69.5	---	0.8	62.2	-11	58.6	-16	---	-58	---	-100
16	(f)	85.8	76.4	7.0	70.0	69.2	1.2	---	---	---	---	---	---	---	---
	(f)	85.5	75.9	7.5	69.8	68.3	0.7	---	---	---	---	---	---	---	---
	Avg.	85.6	76.2	7.2	69.9	68.7	1.0	---	---	---	---	---	---	---	---
	Predicted	81	71	7	67	65	1	---	---	---	---	---	---	---	---
26	(g)	---	---	---	69.6	66.0	2.1	62.0	-10	60.2	-13	59.9	-13	24 days	-100
	(g)	---	---	---	68.4	66.4	1.3	62.7	-9	58.9	-15	54.9	-22	21 days	-100
	Avg.	---	---	---	69.0	66.2	1.7	62.4	-10	59.6	-14	57.0	-17	---	-100
	Predicted	79	67	7	64	62	1	---	---	---	---	---	---	---	---
35	(g)	---	---	---	64.1	62.3	0.5	57.5	-11	56.7	-12	59.4	-8	41 days	-100
	(g)	---	---	---	64.6	62.7	1.0	58.9	-9	57.5	-11	59.6	-7	35 days	-100
	Avg.	---	---	---	64.4	62.5	0.8	58.2	-10	57.1	-11	59.5	-8	---	-100
	Predicted	77	65	7	62.6	60.9	1.2	---	---	---	---	---	---	---	---
40	(f)	77.6	65.2	7.0	61.8	59.8	1.2	---	---	---	---	---	---	---	---
	(f)	77.3	65.2	8.0	62.2	60.4	1.2	---	---	---	---	---	---	---	---
	Avg.	77.4	65.2	7.5	62.2	60.4	1.2	---	---	---	---	---	---	---	---

NOTE: (a) Al, 9.0 Zn, 3.5 Mg, 1.2 Cu, 1.1 Mn  
 (b) SHF 2 hrs at 860°F, CaQ, Age #1 24 hrs at 250°F as 2 in dia. pieces  
 (c) Original evaluation - Table II  
 (d) Strain follower not used to avoid breaking it  
 (e) Not determined - Specimen shattered  
 (f) Step aging data  
 (g) Stress corrosion data  
 (h) Failed before reaching 0.2% offset  
 (i) Failed outside of at gage length

Table III  
THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY X6  
S. No. 277938 (a)

Additional Aging Time at 350°F, Hrs.(b)	Data Source	Longitudinal			Transverse			Unstressed		Stressed 25% T.S.		Stressed 50% T.S.		Stressed 75% T.S.		
		T.S. ksi	Y.S. El., % ksi in 10	T.S./Density x 10 <sup>6</sup> in	T.S. ksi	Y.S. El., % ksi in 10	T.S. ksi	Change ksi	T.S. ksi	Change ksi	T.S. ksi	Days to Failure	Change %	T.S. ksi	Days to Failure	Change %
0	(c)	95.4	95.2 0.5	0.900	---	---	---	---	---	---	---	---	---	---	---	---
	(d)	105.2	101.0 1.3	0.955	74.8 (e)	0.0	---	---	---	---	---	---	---	---	---	---
	(d)	104.1	101.4 0.7	0.958	55.6(f)	0(f)	---	---	---	---	---	---	---	---	---	---
	(g)	104.7	100.4 1.1(1)	0.949	85.2 (e)	0	---	---	---	---	---	---	---	---	---	---
	(g)	102.4	99.5 0.9	0.940	66.3 (e)	0	---	---	---	---	---	---	---	---	---	---
4	Avg.	102.4	99.5 0.9	0.940	70.5	0	---	---	---	---	---	---	---	---	---	---
	(g)	92.0	87.7 1.4	0.829	79.1 (e)	0(1)	---	---	---	---	---	---	---	---	---	---
	(g)	---	---	---	68.7 (e)	0	---	---	---	---	---	---	---	---	---	---
	Avg.	92.0	87.7 1.4	0.829	73.9	0	---	---	---	---	---	---	---	---	---	---
	(g)	84.1	74.8 3.0	0.707	70.0	0.1(1)	---	---	---	---	---	---	---	---	---	---
16	(g)	---	---	---	70.1	0.6	---	---	---	---	---	---	---	---	---	---
	(g)	84.1	74.8 3.0	0.707	70.1	0.6	---	---	---	---	---	---	---	---	---	---
	Avg.	---	---	---	70.0	0.4	---	---	---	---	---	---	---	---	---	---
	Predicted	80	70 3	---	68	65	1	---	---	---	---	---	---	---	---	---
	(h)	---	---	---	68.6	66.4	0.8	55.7	-20	50.9	48.7	-30	7 days	-100	7 days	-100
28	(h)	---	---	---	69.9	66.3	1.2	49.8	-28	46.5	48.5	-30	7 days	-100	7 days	-100
	(h)	---	---	---	69.2	66.4	1.0	52.8	-24	48.7	48.6	-30	7 days	-100	7 days	-100
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	79	67 3	---	67	62	1	---	---	---	---	---	---	---	---	---
	(h)	---	---	---	65.4	63.4	0.9	52.3	-21	53.4	51.7	-22	21 days	-100	21 days	-100
36	(h)	---	---	---	67.8	64.2	1.4	51.9	-22	54.2	52.2	-19	69 days	-100	69 days	-100
	(h)	---	---	---	66.6	63.8	1.2	52.1	-22	53.8	52.0	-22	69 days	-100	69 days	-100
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	78.0	66.2 3.2(1)	0.626	65.2	61.5	0.9	---	---	---	---	---	---	---	---	---
	(g)	---	---	---	67.4	62.0	1.6	---	---	---	---	---	---	---	---	---
40	(g)	78.0	66.2 3.2	0.626	66.3	61.8	1.2	---	---	---	---	---	---	---	---	---
	(g)	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Avg.	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	Predicted	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---
	(h)	---	---	---	---	---	---	---	---	---	---	---	---	---	---	---

NOTE: (a) Al, 10.6 Zn, 4.4 Mg, 2.0 Cu, 1.7 Mn-14 microns  
(b) SHT 2 hrs at 860°F, OAG, Age #1 24 hrs at 250°F 2 in. dia. pieces  
(c) Table VII, Section II  
(d) Table II, Section III  
(e) Failed before reaching 0.25 offset  
(f) Specimen contained discontinuity  
(g) Step aging data  
(h) Stress corrosion data  
(i) Specimen failed at or outside of gage marks

Table XXXI  
THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 36

Additional Aging Time at 330°F-Hrs.	Data Source	S.No. 277939 (a)									
		Longitudinal					Transverse				
		YS		El. %		YS/Density x10 <sup>6</sup> In.	YS		El. %		TS
		ksi	In. 4D	ksi	In. 4D		ksi	In. 4D	ksi	In. 4D	
0	(c)	94.5	0.6	92.2	0.871		87.2				
	(d)	103.7	1.4	99.4	0.940		74.6(r)				
	(f)	104.2	1.1	99.9	0.944		69.5				
	(g)	106.0	2.1	100.3	0.948		80.2				
	(h)	--	--	--	--		88.5				
	Avg.	102.1	1.3	98.0	0.926		80.4				
4	(g)	93.7	2.4(1)	86.8	0.620		82.1				
	(h)	--	--	--	--		81.0				
	Avg.	93.7	2.4	86.8	0.620		82.7				
10	Predicted	88	3	79	--		76				
	(h)	--	--	--	--		74.5				
	(i)	--	--	--	--		74.4				
	Avg.	--	--	--	--		74.4				
16	(g)	85.0	3.5	74.8	0.707		72.9				
	(h)	--	--	--	--		71.5				
	Avg.	85.0	3.5	74.8	0.707		72.2				
28	Predicted	78	4	65	--		65				
	(h)	--	--	--	--		65.5				
	(i)	--	--	--	--		63.5				
	Avg.	--	--	--	--		65.4				
40	(g)	77.6	4.1	65.0	0.614		65.0				
	(h)	--	--	--	--		66.1				
	Avg.	77.6	4.1	65.0	0.614		65.6				

Notes: (a) Al, 10.6 Zn, 4.4 Mg, 2.0 Cu, 1.7 Mn, 5 microns.  
(b) SHT, 2 hrs. at 860°F, CWQ, Aged 24 hrs. at 250°F as 2 in. dia. piece.  
(c) Table VII, Section II  
(d) Table II, Section III  
(e) Failed before reaching 0.2% offset.  
(f) Specimen contained discontinuity.  
(g) Step Aging Data  
(h) Stress Corrosion Data  
(i) Failed at or outside gage marks.

S.No. 277939 (a)

Longitudinal

YS

El. %

In. 4D

YS/Density

x10<sup>6</sup> In.

YS

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

TS

ksi

El. %

In. 4D

Table XXII  
THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 36 (a)

Age 62 (b)	S. No.	Longitudinal				Transverse				Unstressed			Stressed 25% YS			Stressed 50% YS			Stressed 75% YS		
		TS ksi	YS ksi	El. % in. in.	YS/Density x10 <sup>6</sup> in.	TS ksi	YS ksi	El. % in. in.	TS ksi	TS ksi	% Change	Days to Failure	% Change	Days to Failure	% Change	Days to Failure	% Change	Days to Failure	Days to Failure	% Change	Days to Failure
--	283453	108.4	104.9	1.6	0.990	95.5	95.1	0.6	95.5	--	--	--	--	--	--	--	--	--	--	--	--
	108.4	105.4	105.4	0.1	0.994	91.9	91.9	0	91.9	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	108.4	105.2	0.8	0.992	93.7	95.1	0.3	93.7	--	--	--	--	--	--	--	--	--	--	--	--
2	283451	111.4	106.2	1.2	1.021	95.6	95.6	(d)	95.6	60.6	-37	17 Da	-100	2 Da	-100	1 Da	-100	1 Da	-100	-100	-100
	111.4	106.3	106.3	1.2	1.022	96.8	96.8	(d)	96.8	65.8	-32	6 Da	-100	1 Da	-100	1 Da	-100	1 Da	-100	-100	-100
	Avg.	111.6	108.2	1.2	1.021	96.2	96.2	(d)	96.2	63.2	-35	--	-100	--	-100	--	-100	--	-100	-100	-100
4	283453	97.9	97.4	2.1	0.876	86.0	84.0	0.7	86.0	--	--	--	--	--	--	--	--	--	--	--	--
	96.9	92.7	92.7	2.2	0.875	86.4	86.4	0.6	86.4	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	97.4	92.4	2.1	0.875	86.2	86.2	0.6	86.2	--	--	--	--	--	--	--	--	--	--	--	--
8	283453	84.2	83.3	1.4	0.766	85.0	83.1	1.2	85.0	--	--	--	--	--	--	--	--	--	--	--	--
	81.5	84.4	84.4	1.3	0.765	84.1	79.7	0.8	84.1	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	82.8	83.8	1.3	0.765	84.6	81.4	1.0	84.6	--	--	--	--	--	--	--	--	--	--	--	--
16	283453	87.9	86.3	2.5	0.756	82.1	76.4	2.0	82.1	--	--	--	--	--	--	--	--	--	--	--	--
	87.9	86.3	86.3	2.3	0.756	79.7	75.2	1.3	79.7	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	87.4	86.3	2.4	0.757	80.9	75.8	1.6	80.9	--	--	--	--	--	--	--	--	--	--	--	--
24	283453	81.5	72.3	5.0	0.682	74.4	69.5	1.6	74.4	--	--	--	--	--	--	--	--	--	--	--	--
	81.5	72.3	72.3	4.0	0.682	76.8	70.1	1.9	76.8	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	81.5	72.3	4.5	0.682	75.5	69.8	1.8	75.5	--	--	--	--	--	--	--	--	--	--	--	--
32	283453	81.5	69.9	4.0	0.659	75.9	65.2	3.1	75.9	66.0	-13	66.0	-13	66.5	-12	26 Da	-100	26 Da	-100	-100	-100
	81.5	69.9	69.9	4.0	0.659	75.2	65.6	2.8	75.2	65.2	-14	67.3	-13	65.7	-13	84 Da	-100	84 Da	-100	-100	-100
	Avg.	81.5	69.9	4.0	0.659	75.6	65.4	3.0	75.6	65.6	-13	66.6	-12	66.1	-13	--	--	--	--	--	--
48	283453	73.2	61.7	5.0	0.579	69.3	59.4	3.1	69.3	--	--	--	--	--	--	--	--	--	--	--	--
	73.2	61.7	61.7	5.0	0.579	68.9	59.4	2.8	68.9	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	73.2	61.2	5.0	0.577	69.1	59.4	3.0	69.1	--	--	--	--	--	--	--	--	--	--	--	--
1	283453	93.3	87.3	2.4	0.824	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	93.3	87.3	87.3	2.6	0.819	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	93.3	87.3	2.5	0.821	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
2	283453	88.4	82.3	1.5	0.776	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	88.4	82.3	82.3	1.5	0.776	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	88.4	82.3	1.5	0.776	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
4	283453	83.6	75.2	4.0	0.709	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	83.6	75.2	75.2	4.0	0.709	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	83.6	75.2	4.0	0.709	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
12	283453	76.1	64.2	5.0	0.666	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	76.1	64.2	64.2	5.0	0.666	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	Avg.	76.1	64.2	5.0	0.666	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--

(a) Al, 8.5 Zn, 3.5 Mg, 1.1 Mn, 1.2 Fe, 3.0 Ni.  
(b) All specimens SHT 2 hrs. at 660°F, CWQ, Age #1 24 hrs. at 250°F.  
(c) Not obtained, specimen shattered.  
(d) Failed before reaching 0.2% offset.

Table XXXIII

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES OF ALLOYS 39, 49 AND 50

Age #2 (d) t	Alloy 39				Alloy 49				Alloy 50			
	T.S. ksi	Y.S. ksi	El, % in 1/2	Y.S./Density x 10 <sup>6</sup> in	T.S. ksi	Y.S. ksi	El, % in 1/2	Y.S./Density x 10 <sup>6</sup> in	T.S. ksi	Y.S. ksi	El, % in 1/2	Y.S./Density x 10 <sup>6</sup> in
--	---	106.9	103.5	1.3	0.975	106.4	102.5	1.1	98.4	97.0	1.4	0.908
---	---	97.0(e)	(f)	0(e)	---	107.9	104.9	1.0	101.7	96.0	1.6	0.899
Avg.	Avg.	106.9	103.5	1.3	0.975	107.2	103.7	1.0	100.0	96.5	1.5	0.904
16	330	81.8	72.6	2.3	0.684	81.2	76.0	1.6	85.0	76.7	1.7	0.718
---	---	79.5	68.8	1.4	0.648	81.6	78.0	2.6	84.3	77.6	1.5	0.727
Avg.	Avg.	80.6	70.7	1.8	0.666	81.4	77.0	2.1	84.6	77.2	1.6	0.723
32	330	68.2	58.2	2.9	0.549	74.1	67.3	3.1	79.0	70.5	2.7	0.660
---	---	77.1	65.3	1.7	0.615	75.2	66.4	3.0	77.2	69.7	2.1	0.653
Avg.	Avg.	72.6	61.8	2.3	0.582	74.6	66.8	3.0	78.1	70.1	2.4	0.656
48	330	73.9	60.8	4.4	0.573	72.3	62.7	2.7	72.1	66.2	1.3	0.620
---	---	73.4	60.7	3.5	0.572	71.8	63.4	3.3	76.5	66.0	2.7	0.618
Avg.	Avg.	73.6	60.8	4.0	0.573	72.0	63.0	3.0	74.3	66.1	2.0	0.619
4	350	84.3	75.7	2.4	0.713	83.9	77.7	1.7	84.7	78.3	1.8	0.733
---	---	79.0	74.9	1.1	0.706	83.4	78.7	1.3	84.7	78.9	2.0	0.739
Avg.	Avg.	81.6	75.3	1.8	0.710	83.6	78.2	1.5	84.7	78.6	1.9	0.736
8	350	76.3	67.2	2.8	0.633	77.1	70.6	2.5	80.5	73.0	2.5	0.683
---	---	80.2	69.3	3.1	0.653	65.0	60.8	0.3	80.0	72.3	2.2	0.677
Avg.	Avg.	79.2	68.2	3.0	0.643	71.0	65.7	1.4	80.2	72.6	2.4	0.680
12	350	77.2	64.7	3.4	0.615	73.5	65.8	3.2	77.9	69.2	2.8	0.648
---	---	77.4	65.3	3.6	0.610	69.2	66.0	1.5	78.2	68.8	2.1	0.644
Avg.	Avg.	77.3	65.0	3.5	0.612	71.4	65.9	2.4	78.0	69.0	2.4	0.646

NOTE: (a) Al, 10.9 Zn, 4.9 Mg, 2.0 Cu, 1.8 Mn  
 (b) Al, 15.3 Zn, 4.6 Mg, 2.0 Cu, 1.9 Mn  
 (c) Al, 10.0 Zn, 4.1 Mg, 0.9 Cu, 1.4 Mn, 1.1 Fe, 1.4 Ni, 0.01 Cr, 0.02 Ti  
 (d) All specimens SHT 2 hrs @ 860°F, CwQ, Age #1 24 hrs @ 250°F  
 (e) Not included in average  
 (f) Failed before reaching 0.2% offset



Table XXXV

THE EFFECT OF STEP AGING ON TENSILE  
PROPERTIES OF ALLOY 59.S. No. 277934 (a)

Additional Aging Time At 330 F-Hrs. (b)	Data Source	Longitudinal			Transverse		
		T.S. ksi	Y.S. ksi	EL, % in 4D	Y.S./Density x 10 <sup>6</sup> in.	T.S. ksi	Y.S. ksi
0	(c)	90.9	(d)	0.4	-	-	-
	(e)	96.2	(f)	0.0	-	73.8 (g)	(d)
	(e)	89.7	(f)	0.0	-	50.1	(d)
	(h)	84.4	(d)	0.0	-	70.8	(d)
	(h)	90.5	(d)	0.0	-	63.5	(d)
	Avg.	90.3	-	0.0	-	64.6	-
4	(h)	90.2	89.8	0.5 (i)	0.829	52.2	(d)
	(h)	91.8	90.1	0.5 (i)	0.832	62.3	(d)
	Avg.	91.0	90.0	0.5	0.831	57.2	-
	(h)	83.4	78.4	1.2 (i)	0.724	67.9	(d)
16	(h)	82.7	78.9	1.0	0.729	64.2	(d)
	Avg.	83.0	78.6	1.1	0.726	66.0	-
	(h)	76.5	71.3	1.3 (i)	0.658	63.4	(d)
	(h)	74.1	69.0	0.6 (i)	0.637	60.4	(d)
40	Avg.	75.3	70.2	1.0	0.648	61.9	-

- (a) Al, 10.2 Zn, 3.9 Mg, 1.6 Cu, 1.6 Mn, 1.0 Fe, 4.1 Ni, 15 Microns  
(b) SHT 2 Hrs, at 860°F. CWQ, Ag: #1 24 Hrs at 250°F. as 2 in. Dia. Piece  
(c) Table VII, Section I  
(d) Failed before reaching 0.2% offset  
(e) Table VIII, Section II  
(f) Strain follower not used to avoid breaking it  
(g) Specimen contains discontinuity  
(h) Step aging data  
(i) Specimen failed at or outside gage mark





Table XXVII

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 60

S. No. 283758 (a)

Additional Aging Time at 330°F-Hrs. (b)	Data Source	Longitudinal			Transverse			Unstressed		Stressed 25% TS		Stressed 50% TS		Stressed 75% TS	
		TS ksi	YS ksi	El. % In 4D	TS ksi	YS ksi	El. % In 4D	TS ksi	Change %	TS ksi	Change %	TS ksi	Change %	TS ksi	Days to Failure
0	(c)	106.3	104.2	1.9	84.8	84.6	0.4(r)	--	--	--	--	--	--	--	--
	(c)	--	--	--	79.9	(e)	0	--	--	--	--	--	--	--	--
	Avg.	106.3	104.2	1.9	82.4	84.8	0.2	--	--	--	--	--	--	--	--
4	(c)	95.9	93.7	2.3	77.4	(e)	0	--	--	--	--	--	--	--	--
	(c)	--	--	--	80.2	(e)	0	--	--	--	--	--	--	--	--
	Avg.	95.9	93.7	2.3	78.8	--	0	--	--	--	--	--	--	--	--
16	(c)	82.8	80.3	2.2	62.6	(e)	0	--	--	--	--	--	--	--	--
	(c)	--	--	--	69.9	(e)	0	--	--	--	--	--	--	--	--
	Avg.	82.8	80.3	2.2	66.2	--	0	--	--	--	--	--	--	--	--
40	(c)	74.8	70.7	5.2	68.1	64.6	0.8	--	--	--	--	--	--	--	--
	(c)	--	--	--	66.4	64.2	0.3(r)	--	--	--	--	--	--	--	--
	(c)	--	--	--	68.4	65.3	0.6	46.5	-32	40.8	-41	50.1	-27	6 Da	-100
	(d)	--	--	--	68.9	65.6	0.9	43.4	-37	47.3	-31	40.4	-41	5 Da	-100
	Avg.	74.8	70.7	5.2	68.0	64.9	0.6	45.0	-35	44.0	-36	45.2	-34	--	--

Notes: (a) Al, 11.6 Zn, 5.5 Mg, 1.8 Cu, 0.7 Mn.

(b) SHT 2 hrs. at 860°F, CWQ, Age #1 24 hrs. at 250°F as 2 in. dia. piece.

(c) Step Aging Data.

(d) Stress Corrosion Data.

(e) Failed before reaching 0.2% offset.

(f) Specimen failed at or outside gage mark.

Table XXVIII

## THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 61

S. No. 283/59 (a)

Additional Aging Time at 250°F, Hrs (b)	Data Source	Longitudinal			Transverse			Unstressed		Stressed		Stressed 50% T.S.		Stressed 75% T.S.	
		T.S. ksi	Y.S. ksi	El., % in 10 in	T.S. ksi	Y.S. ksi	El., % in 10 in	T.S. ksi	Change	T.S. ksi	Change	T.S., ksi, or Days to Failure	Change	Days to Failure	Change
0	(c)	109.1	106.3	3.0	91.1	87.2	0.9	---	---	---	---	---	---	---	---
	(c)	108.9	106.3	3.4	91.6	87.7	(e)	---	---	---	---	---	---	---	---
	Avg.	109.0	106.3	3.2	91.4	87.4	0.9	---	---	---	---	---	---	---	---
4	(c)	93.9	91.3	5.5	80.5	80.1	C.1(f)	---	---	---	---	---	---	---	---
	(c)	94.4	91.6	5.5	83.1	80.4	1.1	---	---	---	---	---	---	---	---
	Avg.	94.2	91.4	5.5	81.8	80.2	C.6	---	---	---	---	---	---	---	---
8	Predicted	88	83	7	78	74	1	---	---	---	---	---	---	---	---
	(c)	---	---	---	81.5	75.2	1.1(f)	59.4	-27	50.9	-38	35 days	-100	7 days	-100
	Avg.	---	---	---	81.7	77.2	0.8(f)	60.0	-22	48.5	-41	21 days	-100	6 days	-100
16	(c)	82.6	76.6	9.0	74.0	68.4	2.0	---	---	---	---	---	---	---	---
	(c)	83.0	77.0	7.5	73.8	68.4	1.4(f)	---	---	---	---	---	---	---	---
	Avg.	82.8	76.8	8.2	73.9	68.4	1.7	---	---	---	---	---	---	---	---
22	Predicted	80	72	9	71	55	2	---	---	---	---	---	---	---	---
	(d)	---	---	---	70.7	68.3	0.7	62.9	-14	56.0	-23	58.7	-19	35 days	-100
	Avg.	---	---	---	74.9	68.3	2.7	62.4	-14	60.6	-17	56.4	-23	24 days	-100
30	(c)	77	69	10	69	62	2	---	---	---	---	---	---	---	---
	(d)	---	---	---	71.1	64.6	2.4	60.5	-15	54.4	-23	49.6	-30	35 days	-100
	Avg.	---	---	---	71.0	64.9	2.3	60.8	-14	57.3	-19	49.3	-31	24 days	-100
40	(c)	74.1	65.5	10.0	69.0	59.5	4.3	---	---	---	---	---	---	---	---
	(c)	74.3	65.5	11.0	63.0	59.5	1.4	---	---	---	---	---	---	---	---
	Avg.	74.2	65.5	10.5	66.0	59.5	2.8	---	---	---	---	---	---	---	---

NOTE: (a) Al, 10.4 Zn, 4.5 Mg, 0.8 Cu  
 (b) SH 2 hrs at 860°F, CQ, Age #1 24 hrs at 250°F  
 (c) Step aging data

(c) Stress corrosion data  
 (e) Not determined, specimen shattered  
 (f) Specimen failed at or outside gage length

Table XXXIX  
THE EFFECT OF STEP AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING RESISTANCE OF ALLOY 71  
S. No. 293387

Age, (b)	Longitudinal				Transverse				Unstressed				Stressed 25%YS				Stressed 50%YS				Stressed 75%YS			
	TS ksi	YS ksi	El. % in./in.	YS/Density x10 <sup>6</sup> in.	TS ksi	YS ksi	El. % in./in.		TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change
--	109.0	103.9	4.0	1.005	93.2	89.0	2.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	107.1	102.9	5.0	0.995	94.0	89.0	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	106.0	103.4	4.5	1.000	93.6	89.0	1.5		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	107.1	105.0	4.8	1.015	92.0	87.8	1.0		58.1	-37	53.5	-42	18 Da	-100	13 Da	-100	22 Da	-100	7 Da	-100	13 Da	-100	7 Da	-100
AVG.	107.7	105.0	4.6	1.015	92.0	87.8	1.0		61.9	-33	56.1	-35	22 Da	-100	7 Da	-100	22 Da	-100	7 Da	-100	22 Da	-100	7 Da	-100
2	103.2	100.3	4.5	0.970	91.9	87.1	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	102.2	99.8	6.0	0.965	87.8	85.5	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	102.8	100.0	5.2	0.967	89.8	86.3	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
4	101.2	98.3	5.5	0.951	88.6	85.6	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	101.4	97.8	6.0	0.946	90.3	85.4	2.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	101.3	98.0	5.8	0.948	89.4	85.5	1.5		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
8	99.3	95.2	6.5	0.921	87.7	84.0	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	98.4	94.0	6.5	0.909	85.8	83.2	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	98.8	94.6	6.5	0.915	86.8	83.6	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
24	91.4	86.2	6.0	0.834	80.2	77.2	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	91.0	85.9	7.8	0.831	85.3	77.4	3.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	91.2	86.0	7.8	0.832	82.8	77.4	3.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
48	85.5	79.2	9.0	0.766	79.9	72.1	5.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	82.5	79.2	9.5	0.766	79.8	72.3	4.5		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	85.5	79.2	9.2	0.766	79.8	72.3	4.5		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
2	97.1	93.4	7.0	0.903	86.4	82.1	2.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	94.1	92.2	6.0	0.892	81.7	78.1	0.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	96.6	92.8	6.0	0.896	84.0	82.1	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
3	93.5	88.1	7.0	0.852	82.5	79.9	2.0		76.0	-10	75.2	-11	60.3	-28	31 Da	-100	60.3	-28	31 Da	-100	60.3	-28	31 Da	-100
AVG.	93.5	88.1	7.0	0.852	84.0	79.4	3.0		76.3	-9	71.8	-15	51.0	-39	31 Da	-100	51.0	-39	31 Da	-100	51.0	-39	31 Da	-100
4	92.9	88.6	6.0	0.857	83.1	79.9	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	92.7	88.2	8.0	0.853	83.5	79.7	1.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
8	87.3	81.3	9.0	0.786	80.2	73.4	4.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	86.9	80.6	9.5	0.781	79.1	73.8	4.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	87.1	81.0	9.2	0.783	79.6	73.6	4.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
16	87.2	80.1	8.9	0.775	76.6	72.0	2.0		71.3	-8	71.3	-8	70.7	-8	45 Da	-100	70.7	-8	45 Da	-100	70.7	-8	45 Da	-100
AVG.	87.2	80.1	8.9	0.775	77.0	71.6	3.0		71.4	-7	71.6	-7	69.7	-10	54 Da	-100	69.7	-10	54 Da	-100	69.7	-10	54 Da	-100
20	82.2	74.6	9.0	0.721	76.7	67.5	5.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	81.7	73.8	10.0	0.714	73.4	67.6	4.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	82.0	74.2	9.5	0.718	75.0	67.6	4.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
32	80.9	71.1	7.6	0.688	75.5	63.5	8.0		65.7	-12	66.4	-11	65.2	-13	56.2	-25	65.2	-13	56.2	-25	65.2	-13	56.2	-25
AVG.	80.9	71.1	7.6	0.688	74.6	63.6	6.5		66.2	-11	66.4	-11	65.1	-13	55.8	-25	65.1	-13	55.8	-25	65.1	-13	55.8	-25
48	76.4	66.7	11.0	0.645	68.9	61.0	4.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	75.6	65.7	11.5	0.635	72.1	61.5	6.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	76.1	66.2	11.2	0.640	70.5	61.2	5.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
48	72.8	62.1	12.0	0.601	70.9	57.8	6.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
AVG.	72.7	62.1	11.0	0.601	70.5	57.8	6.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
	72.8	62.1	11.5	0.601	70.7	57.8	6.0		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--

(a) Al-9.3 Zn, 3.6 Mg, 0.5 Cu, 0.02 Cr, 0.02 Ti, 0.7 Co.  
(b) All specimens SHF 2 hrs. at 860°F, CVD, Age #1 24 hrs. at 250°F.  
(c) Failed before reaching 0.2% offset.  
(d) Not obtained, specimen shattered.  
(e) Premature failure - Disregard

Table XL  
THE EFFECT OF PROLONGED AGING AND QUENCHING RATES ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING OF ALLOY 24 (1)

Specimen	Age (h)	S. No.	Longitudinal			Transverse			Stress Value Used	Unstressed		Stressed 25%		Stressed 50%		Stressed 75%	
			TS ksi	YS ksi	El. In./in.	TS ksi	YS ksi	El. In./in.		TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change
24-250	203441	203441	97.2(f)	101.9	0.7(f)	101.9	101.9	0.7(f)	--	50.4(h)	-50(h)	54.8(h)	-46(h)	1 Da	-100	1 Da	--
24-250	106.1(f)	106.1(f)	106.1(f)	100.6	0.7(f)	100.6	100.6	0.7(f)	--	55.4(h)	-45(h)	52.6(h)	-48(h)	1 Da	-100	1 Da	--
24-250	101.6(f)	101.6(f)	101.6(f)	101.2	0.7(f)	101.2	101.2	0.7(f)	101.2	52.9(h)	-48(h)	50.0(h)	-51(h)	--	-100	--	--
24-250	104.4(g)	104.4(g)	104.4(g)	99.4	0.7(g)	99.4	99.4	0.7(g)	--	57.5	-43	69.5	-31	2 Da	-100	2 Da	-100
24-250	106.6(g)	106.6(g)	106.6(g)	101.3	0.7(g)	101.3	101.3	0.7(g)	--	67.3	-33	66.2	-34	2 Da	-100	2 Da	-100
24-250	105.5(g)	105.5(g)	105.5(g)	100.4	0.7(g)	100.4	100.4	0.7(g)	100.4	62.4	-38	67.8	-32	--	-100	--	-100
24-250	--	--	--	--	--	--	--	--	--	--	--	--	--	2 Da	-100	2 Da	-100
24-250	--	--	--	--	--	--	--	--	--	51.1	-49	53.2	-46	2 Da	-100	2 Da	-100
24-250	--	--	--	--	--	--	--	--	99.5	37.8	-62	55.0	-44	1 Da	-100	1 Da	-100
24-250	--	--	--	--	--	--	--	--	--	75.2	-22	62.2	-36	1 Da	-100	1 Da	-100
24-250	--	--	--	--	--	--	--	--	--	57.0	-41	63.3	-35	3 Da	-100	3 Da	-100
24-250	--	--	--	--	--	--	--	--	97.0	66.1	-32	62.8	-35	--	-100	--	-100
24-250	--	--	--	--	--	--	--	--	--	50.0	-41	57.7	-31	2 Da	-100	2 Da	-100
24-250	--	--	--	--	--	--	--	--	--	51.2	-39	65.5	-28	2 Da	-100	2 Da	-100
24-250	--	--	--	--	--	--	--	--	84.2	50.6	-40	59.1	-30	--	-100	--	-100
24-250	103.6(c)	103.6(c)	103.6(c)	96.4	0.5(c)	96.4	96.4	0.5(c)	--	54.2	-43	61.7	-35	--	-100	--	-100
24-250	--	--	--	--	--	--	--	--	--	65.6	-27	77.8	-13	74.8	-16	3 Da	-100
24-250	--	--	--	--	--	--	--	--	89.4	72.9	-18	74.4	-15	71.5	-20	3 Da	-100
24-250	--	--	--	--	--	--	--	--	--	--	--	--	--	73.2	-18	--	-100
24-250	88.7(f)	88.7(f)	88.7(f)	83.2(f)	0.7(f)	83.2(f)	83.2(f)	0.7(f)	--	77.2	-14	74.6	-17	74.6	-17	61 Da	-100
24-250	87.9(f)	87.9(f)	87.9(f)	82.2(f)	0.7(f)	82.2(f)	82.2(f)	0.7(f)	--	68.0	-27	68.5	-26	68.5	-26	31 Da	-100
24-250	88.3(f)	88.3(f)	88.3(f)	82.7(f)	0.7(f)	82.7(f)	82.7(f)	0.7(f)	90.0	71.6	-20	77.4	-14	70.6	-22	--	-100
24-250	--	--	--	--	--	--	--	--	--	67.6	-22	69.5	-19	63.5	-26	59.5	-31
24-250	--	--	--	--	--	--	--	--	--	63.6	-26	64.6	-25	67.5	-22	55.9	-35
24-250	--	--	--	--	--	--	--	--	82.0	65.6	-24	67.0	-22	65.5	-24	57.7	-33
24-250	--	--	--	--	--	--	--	--	--	70.1	-18	73.1	-14	72.8	-15	64.3	-25
24-250	--	--	--	--	--	--	--	--	81.0	45.9	-46	71.1	-17	66.7	-18	9 Da	-100
24-250	--	--	--	--	--	--	--	--	--	58.0	-32	72.1	-15	69.8	-18	--	-62
24-250	--	--	--	--	--	--	--	--	--	73.5	-12	74.5	-14	67.8	-19	70.6	-15
24-250	--	--	--	--	--	--	--	--	--	71.9	-14	73.1	-12	71.0	-15	71.8	-14
24-250	--	--	--	--	--	--	--	--	80.0	72.7	-13	72.3	-13	69.4	-17	--	-14
24-250	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
24-250	--	--	--	--	--	--	--	--	--	64.8	-22	65.6	-21	67.4	-19	70.5	-15
24-250	--	--	--	--	--	--	--	--	--	70.8	-14	70.6	-15	72.3	-13	68.2	-18
24-250	--	--	--	--	--	--	--	--	79.2	67.8	-18	68.1	-18	69.8	-16	69.2	-16
24-250	--	--	--	--	--	--	--	--	--	67.1	-22	71.4	-16	69.0	-19	--	-45
24-250	82.7	82.7	82.7	82.0	0.7	82.0	82.0	0.7	--	--	--	--	--	--	--	--	--
24-250	0.753	0.753	0.753	0.753	0.753	0.753	0.753	0.753	--	--	--	--	--	--	--	--	--
24-250	0.753	0.753	0.753	0.753	0.753	0.753	0.753	0.753	--	--	--	--	--	--	--	--	--

(Continued)

Table XL (continued)

THE EFFECT OF PROLONGED AGING AND QUENCHING RATES ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING OF ALLOY 34 (1)

Quench Age (a) t Y	S. No.	Longitudinal			Transverse			Stress Value Used	Unstressed			Stressed 25%			Stressed 50%			Stressed 75%		
		TS ksi	YS ksi	El., % In 100	YS ksi	El., % In 100	TS ksi		TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS ksi	% Change
CVQ 16 330	283440	82.6(f)	76.2(f)	0.7(f)	86.2	0.3	80.3	1.1	69.7	-18	72.0	-16	68.4	-20	68.4	-20	62 Da	-100	62 Da	-100
CVQ 16 330		83.0(f)	75.7(f)	0.7(f)	85.2	0.5	80.5	0.6	71.4	-16	72.4	-15	70.5	-15	70.5	-15	9 Da	-100	9 Da	-100
CVQ 16 330		82.8(f)	76.0(f)	0.7(f)	85.2	0.4	80.4	0.8	70.6	-17	72.4	-15	69.4	-19	69.4	-19	--	--	--	--
CVQ 6.25-8/350	283443	--	--	--	83.3	0.3	75.6	1.7	70.1	-14	69.1	-15	68.9	-14	68.9	-14	72.5	-11	72.5	-11
CVQ 6.25-8/350		--	--	--	80.1	0.1	75.6	1.4	71.3	-13	71.0	-13	71.1	-13	71.1	-13	70.1	-14	70.1	-14
CVQ 6.25-8/350		--	--	--	81.7	0.1	75.6	1.4	71.3	-13	71.0	-13	70.0	-14	70.0	-14	71.3	-13	71.3	-13
CVQ 48 315	283444	79.1(f)	73.1(f)	0.7(f)	83.4	0.8	76.8	2.0	67.2	-18	75.0	-9	68.5	-16	68.5	-16	68.6	-16	68.6	-16
CVQ 48 315		82.1(f)	73.0(f)	0.7(f)	86.5	0.1	76.0	1.2	72.0	-12	72.6	-11	69.1	-16	69.1	-16	67.8	-17	67.8	-17
CVQ 48 315		--	--	--	--	--	--	--	72.4	-12	--	--	--	--	--	--	--	--	--	--
CVQ 48 315		--	--	--	82.0	0.1	76.4	1.6	71.0	-13	73.8	-10	68.8	-17	68.8	-17	68.2	-17	68.2	-17
CVQ 48 315	283440	--	--	--	79.7	0.8	74.8	0.9	66.2	-15	64.5	-20	64.7	-19	64.7	-19	66.0	-18	66.0	-18
CVQ 48 315		--	--	--	80.5	0.9	75.2	0.9	67.1	-16	65.2	-19	65.1	-19	65.1	-19	63.3	-21	63.3	-21
CVQ 48 315		--	--	--	80.1	0.9	75.2	0.9	67.6	-16	65.5	-19	64.7	-19	64.7	-19	64.3	-19	64.3	-19
CVQ 48 315	283444	--	--	--	74.6	0.7	71.0	0.7	66.5	-17	64.9	-19	66.6	-17	66.6	-17	65.8	-18	65.8	-18
CVQ 48 315		--	--	--	80.6	0.1	71.1	2.2	64.6	-19	64.8	-20	65.4	-19	65.4	-19	65.5	-19	65.5	-19
CVQ 48 315		--	--	--	77.6	0.1	71.0	1.4	65.1	-19	64.8	-20	66.0	-18	66.0	-18	65.6	-19	65.6	-19
CVQ 48 315	283446	--	--	--	72.5	0.6	72.7	0.6	62.6	-18	63.2	-18	64.6	-16	64.6	-16	63.4	-17	63.4	-17
CVQ 48 315		--	--	--	76.2	0.1	72.7	1.2	63.8	-17	62.5	-19	61.8	-20	61.8	-20	64.6	-16	64.6	-16
CVQ 48 315		--	--	--	76.8	0.1	72.7	1.0	63.3	-18	62.5	-18	63.2	-18	63.2	-18	64.0	-17	64.0	-17
CVQ 48 315	283447	--	--	--	74.6	0.1	70.1	0.5	63.2	-17	66.3	-13	64.8	-15	64.8	-15	59.1	-23	59.1	-23
CVQ 48 315		--	--	--	78.7	0.1	71.1	0.6	64.1	-16	64.5	-12	67.1	-12	67.1	-12	67.5	-25	67.5	-25
CVQ 48 315		--	--	--	76.6	0.6	70.6	0.6	63.7	-17	65.1	-12	68.1	-14	68.1	-14	68.3	-24	68.3	-24
CVQ 48 315		--	--	--	78.6	0.1	73.2	1.1	67.1	-16	66.4	-15	65.8	-17	65.8	-17	64.2	-19	64.2	-19
CVQ 48 315	Overall Avg. 50.6	73.0	0.7	0.665	78.6	0.1	73.2	1.1	67.1	-16	66.4	-15	65.8	-17	65.8	-17	64.2	-19	64.2	-19
CVQ 32 330	283440	--	--	--	75.6	0.2	72.5	1.0	77.7	-1	65.0	-17	67.3	-14	67.3	-14	61.8	-21	61.8	-21
CVQ 32 330		--	--	--	81.2	0.2	73.6	1.6	77.7	-1	70.3	-10	68.8	-13	68.8	-13	63.1	-23	63.1	-23
CVQ 32 330		--	--	--	78.4	0.2	73.2	1.3	77.7	-1	67.6	-14	68.0	-13	68.0	-13	61.0	-22	61.0	-22
CVQ 8 350	283440	--	--	--	78.2	0.9	71.7	0.9	68.4	-13	63.9	-18	70.1	-11	70.1	-11	65 Da	-100	65 Da	-100
CVQ 8 350		--	--	--	78.2	0.9	71.7	1.1	69.3	-12	69.0	-12	69.3	-11	69.3	-11	63.6	-60	63.6	-60
CVQ 8 350		--	--	--	78.2	0.9	71.7	1.0	68.8	-12	66.4	-15	69.7	-11	69.7	-11	--	--	--	--
BMQ 24 250	283443	--	--	--	78.6	0.4	72.5	0.4	60.7	-21	60.7	-21	69.0	-10	69.0	-10	3 Da	-100	3 Da	-100
BMQ 24 250		--	--	--	75.0	0.4	73.4	1.1	56.1	-24	68.6	-11	65.9	-14	65.9	-14	3 Da	-100	3 Da	-100
BMQ 24 250		--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--	--
BMQ 24 250		--	--	--	76.8	0.1	73.0	1.0	59.4	-23	64.6	-16	66.5	-13	66.5	-13	--	--	--	--
BMQ 24 250	283440	--	--	--	75.7	0.9	69.5	1.0	58.3	-24	58.8	-23	59.0	-23	59.0	-23	24 Da	-100	24 Da	-100
BMQ 24 250		--	--	--	77.1	0.9	69.3	0.9	63.5	-17	63.1	-17	65.0	-15	65.0	-15	28 Da	-100	28 Da	-100
BMQ 24 250		--	--	--	76.4	0.9	69.6	1.0	60.9	-20	61.0	-20	62.0	-19	62.0	-19	--	--	--	--
BMQ 24 250	283444	--	--	--	72.6	0.9	72.4	0.9	58.0	-21	62.8	-15	62.4	-15	62.4	-15	53.7	-27	53.7	-27
BMQ 24 250		--	--	--	71.8	0.9	72.2	0.9	64.5	-11	63.1	-16	57.9	-21	57.9	-21	27.4	-35	27.4	-35
BMQ 24 250		--	--	--	73.7	0.9	73.7	0.9	61.7	-16	63.0	-15	60.2	-15	60.2	-15	55.6	-75	55.6	-75
BMQ 24 250	Overall Avg.	--	--	--	75.6	0.9	71.6	0.9	60.5	-20	62.8	-17	63.4	-16	63.4	-16	--	--	--	--
BMQ 8 330	283443	--	--	--	68.2	1.2	61.2	1.2	58.5	-14	59.1	-13	55.7	-18	55.7	-18	57.1	-22	57.1	-22
BMQ 8 330		--	--	--	68.3	0.9	62.4	0.9	59.7	-12	60.3	-12	56.4	-18	56.4	-18	55.9	-27	55.9	-27
BMQ 8 330		--	--	--	68.2	1.2	61.8	1.2	59.1	-13	59.7	-13	56.0	-18	56.0	-18	55.8	-27	55.8	-27

(a) All samples CHT 2 hrs at 860°F

(b) Failed before reaching 0.2% offset

(c) Avg. of 3 or more values

(d) Not obtained

(e) Does not include data from 28 day AI test

(f) S. No. 283445 data

(g) S. No. 277406 data

(h) Data for 28 day AI test

(i) AI 7.6 Zn, 2.5 Mg, 1.0 Cu, 3.5 Fe, 4.9 Ni, 0.09 Cr.

Table XLI  
THE EFFECT OF PROLONGED AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING OF ALLOY 79  
S. No. 283188 (b)

Age (a) t	Longitudinal					Transverse		Stressed Value, ksi	Unstressed		Stressed 25% YS		Stressed 50% YS		Stressed 75% YS	
	TS ksi	YS ksi	El, % In 4D	YS/Density x100 in.	TS ksi	YS ksi	El, % In 4D		TS ksi	% Change	TS ksi	% Change	TS ksi	% Change	TS, ksi or Days to Failure	% Change
24	97.2(d)	88.6(d)	5.5(d)	0.856(d)	90.2(d)	81.6(d)	4.1(d)	--	74.9	-12	61.1	-28	84 Da	-100	52 Da	-100
	96.3(e)	87.3(e)	2.7(e)	0.842(e)	84.1(e)	75.4(e)	3.4(e)	--	75.1	-12	59.9	-22	48.6	-43	52 Da	-100
Avg.	96.9(c)	86.0(c)	3.7(c)	0.849(c)	86.7(c)	77.5(c)	3.8(c)	75.5	75.0	-12	63.5	-25	--	-72	--	-100
6	86.4	77.0	7.5	0.743	79.3	71.1	3.5	--	71.3	-10	72.7	-8	70.9	-10	70.6	-11
8	--	--	--	--	78.7	70.1	3.4	--	71.3	-10	72.9	-8	70.6	-11	72.5	-8
Avg.	86.4	77.0	7.5	0.743	79.0	70.6	3.4	70.6	71.3	-10	72.8	-8	70.5	-10	71.6	-9
16	80.2	66.5	9.0	0.661	74.4	64.5	4.5	--	68.0	-9	69.3	-8	68.0	-9	66.5	-11
	--	--	--	--	75.8	64.5	6.4	--	68.1	-9	69.4	-8	64.5	-14	63.9	-15
Avg.	80.2	68.5	9.0	0.661	75.1	64.5	5.4	64.5	68.0	-9	69.4	-8	66.2	-12	65.2	-13
48	78.4	66.2	9.0	0.638	71.7	61.5	3.9	--	62.8	-14	63.6	-13	62.0	-15	65.9	-10
	--	--	--	--	74.6	61.7	7.2	--	63.2	-14	63.7	-13	63.7	-13	60.4	-17
Avg.	78.4	66.2	9.0	0.638	73.2	61.6	5.6	61.6	63.0	-14	63.6	-13	62.8	-14	63.2	-14

Notes: (a) SHT 2 hrs. at 860°F, CWQ, Aged as noted.  
(b) AL 6.6 Zn, 2.8 Mg, 2.1 Cu, 1.0 Mn, 0.5 Cr, 0.1 Zr.  
(c) Avg. of 3 or more test values.  
(d) Highest test value.  
(e) Lowest test value.

Table XLII

## THE EFFECT OF PROLONGED AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING OF ALLOY 87

S. No. 283196 (b)

Age (a)	Longitudinal			Transverse		El., %		Stress Value Used, ksi	Unstressed		Stressed 25% Y.S.		Stressed 50% Y.S.		Stressed 75% Y.S.	
	TS, ksi	YS, ksi	El., % In 4D	YS, ksi	TS, ksi	In 4D			TS, ksi	% Change	TS, ksi	% Change	TS, ksi	Days to Failure	TS, ksi	Days to Failure
24	106.6(d)	99.3(d)	5.1(d)	92.2(d)	97.6(d)	1.5(d)	--	--	60.6	-33	58.5	-35	82 Da	1 Da	-100	-100
250	103.7(e)	96.8(e)	1.0(e)	86.9(e)	87.9(e)	0.7(e)	--	--	58.5	-35	55.2	-39	58	1 Da	-100	-100
AVG.	104.8(c)	99.1(c)	2.6(c)	89.5(c)	92.5(c)	1.1(c)	87.2	87.2	59.6	-34	56.8	-37	--	--	-68	-100
6	90.1	80.7	7.0	75.6	83.9	2.7	--	--	67.5	-20	70.0	-17	68.7	59.8	-18	-29
8	90.1	80.7	7.0	75.8	84.0	2.8	75.8	75.8	67.5	-20	70.8	-16	67.1	50.6	-20	-40
AVG.	85.7	74.6	7.0	70.3	77.5	2.7	--	--	67.5	-20	70.4	-16	67.9	55.2	-19	-34
16	85.7	74.6	7.0	70.2	78.7	3.3	70.2	70.2	62.7	-20	66.6	-15	63.5	66.2	-19	-16
AVG.	85.7	74.6	7.0	70.2	78.7	3.3	70.2	70.2	65.2	-17	65.8	-16	67.6	64.4	-14	-18
48	82.9	69.5	10.0	65.9	77.6	3.8	--	--	64.0	-19	66.2	-16	65.6	65.3	-17	-17
AVG.	82.9	69.5	10.0	66.2	77.9	3.5	66.2	66.2	60.5	-22	60.2	-23	62.4	61.0	-20	-22
	82.9	69.5	10.0	66.2	77.8	3.6	66.2	66.2	60.9	-22	61.1	-21	61.0	61.8	-22	-21
									60.7	-22	60.6	-22	61.7	61.4	-21	-21

Notes: (a) SHT 2 hrs. at 860, CWQ, Aged as noted.

(b) Al, 7.6 Zn, 2.5 Mg. 1.1 Cu, 2.2 Fe, 2.3 Ni, 0.16 Cr.

(c) Avg. of 3 or more test values.

(d) Highest test value.

(e) Lowest test value.

Table XLVII:  
THE EFFECT OF PROLONGED AGING ON TENSILE PROPERTIES AND STRESS CORROSION CRACKING OF ALLOY 9C  
S. No. 294014

Longitudinal		Transverse				Distressed				Streamed 55% TS (b)				Streamed 50-55% TS (b)				Streamed 75% TS (b)				Streamed 75-80% TS (b)			
		TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS	TS ES	ES TS		
Ave.	90.0	92.4	5.4	80.5	76.2	-1.3	72.0	-1.7	20	72.6	-1.7	73.7	-1.5	1.1	67.5	-2.2	5 Da	-100	3 Da	-100	3 Da	-100	3 Da	-100	3 Da
	91.1	93.6	5.4	80.7	76.1	-1.3	74.6	-1.1	20	66.7	-2.3	68.9	-2.1	1.1	50.2	-1.9	16 Da	-100	3 Da	-100	3 Da	-100	3 Da	-100	3 Da
	94.0	93.0	5.4	80.6	76.2	-1.3	73.3	-1.6	20	66.7	-2.3	72.2	-1.7	1.1	62.9	-2.6	16 Da	-100	3 Da	-100	3 Da	-100	3 Da	-100	3 Da
	88.4	77.9	8.6	78.5	71.1	-6	82.4	-2.2	23	66.7	-2.3	70.4	-6	4.6	68.6	-1.1	69	73 Da	-100	3 Da	-100	3 Da	-100	3 Da	
Ave.	88.4	78.1	7.9	78.8	70.3	-9	66.3	-1.7	23	69.2	-1.3	66.9	-1.1	4.6	64.4	-1.1	69	73 Da	-100	3 Da	-100	3 Da	-100	3 Da	
	88.4	78.0	8.2	77.4	70.7	-7	64.4	-1.6	23	68.3	-1.1	69.7	-1.0	4.6	63.5	-1.5	69	73 Da	-100	3 Da	-100	3 Da	-100	3 Da	
	88.4	78.0	8.2	77.4	70.7	-7	64.4	-1.6	23	68.3	-1.1	69.7	-1.0	4.6	63.5	-1.5	69	73 Da	-100	3 Da	-100	3 Da	-100	3 Da	
	88.4	78.0	8.2	77.4	70.7	-7	64.4	-1.6	23	68.3	-1.1	69.7	-1.0	4.6	63.5	-1.5	69	73 Da	-100	3 Da	-100	3 Da	-100	3 Da	
Ave.	77.7	78.7	9.3	65.0	64.0	-1.0	64.1	-9	63.7	-12	65.5	-11	27	64.3	-12	64.0	-20	55	58.1	-20	63.8	-12	62	63.1	
	78.2	78.0	10.0	65.3	67.0	-8	64.5	-11	64.5	-11	65.5	-10	27	64.7	-12	64.3	-16	55	58.1	-16	63.0	-12	62	63.1	
	78.7	64.4	9.6	66.5	66.6	-6	64.2	-12	64.2	-12	65.3	-10	27	64.0	-12	64.0	-16	55	58.1	-16	63.0	-12	62	63.1	
	76.9	67.7	10.0	64.2	65.4	-5	67.2	-5	67.2	-5	65.0	-9	30	63.2	-11	63.2	-11	55	60.7	-17	62.8	-14	62.8	61.1	
Ave.	76.9	66.2	7.9	64.2	65.1	-5	64.1	-10	64.1	-10	65.0	-9	30	63.2	-11	63.2	-11	55	60.7	-17	62.8	-14	62.8	61.1	
	78.6	68.0	9.0	66.5	65.2	-5	65.6	-6	65.6	-6	63.6	-8	30	62.6	-12	62.6	-12	55	60.7	-17	62.8	-14	62.8	61.1	
	78.6	68.0	9.0	66.5	65.2	-5	65.6	-6	65.6	-6	63.6	-8	30	62.6	-12	62.6	-12	55	60.7	-17	62.8	-14	62.8	61.1	
	78.6	68.0	9.0	66.5	65.2	-5	65.6	-6	65.6	-6	63.6	-8	30	62.6	-12	62.6	-12	55	60.7	-17	62.8	-14	62.8	61.1	

(a) Exposed to 3/25 NaCr Al test for 20 days  
(b) Exposed to 3/25 NaCr Al test for 24 days  
(c) All samples SRT 2 hrs. at 600°F, CWQ. Aged as noted  
(d) Al, 7.5 Zn, 2.4 Mg, 1.0 Cu, 1.1 Fe, 1.0 Ni, 0.2 Cr



Table XLIV

## STRESS CORROSION CRACKING RESISTANCE OF APM ALLOYS EXPOSED TO

3% NaCl: ALTERNATE IMMERSION TEST BATH

Alloy No.	S. No.	Age #2 (a, b) Time At 300° F. Hrs.	Longitudinal		Transverse		75% (A)		50% (A)		25% (A)		
			T.S. ksi	EL, % in LD	T.S. ksi	Y.S. ksi	EL, % in LD	Stress ksi	Time Days	Stress ksi	Time Days	Stress ksi	Time Days
34	283412	-D-	110	0	102	(g)	<1	---	---	51	1,1	26	OK28(2)
	283413	-D-	116	<1(c)	102	98	<1	---	---	51	1,2	26	9, OK28
	283414	0	106(c)	<1(c)	101	(g)	(h)	---	---	50	1,1,1	25	OK28(3)
	283415	0	106(c)	<1(c)	100	(g)	(h)	75	1,1	50	1,2	25	OK84(2)
34	283416	0	106(c)	<1(c)	100	(g)	(h)	75	1,1	50	2,2	25	OK84(2)
52	283417	0	106(c)	2(c)	98	(g)	(h)	---	---	49	1,1,1	24	OK84(3)
	283418	-D-	116	<1	98	(g)	0	---	---	49	1,1	24	5,5
	283419	0	112	1	97	97	(h)	74	1,1	48	1,2	24	6,17
	283420	0	112	0	97	(g)	(h)	---	---	48	2,2	24	1,5
50	283421	-D-	115	0	97	(g)	(h)	74	1,2,2	48	1,3	24	OK84(2)
34	283422	0	106(c)	<1(c)	97	(g)	(h)	---	---	---	---	---	---
87	293423	0	105	3	92	90	1	68	1,1	45	82, OK84	22	OK84(2)
	293424	0	109	3	90	90	(h)	68	5,6	45	12,15	22	OK84(2)
	283425	-A-	88(c)	<1(c)	90	88	1	68	31,61	45	OK84(2)	22	OK84(2)
	283426	-F-	(d)	(d)	89	(g)	(h)	67	3,3	44	OK84(2)	22	OK84(2)
71	293427	0	108	5	92	88	1	66	7,13	44	18,22	22	OK84(2)
28	277400	1	101(e)	3(e)	86	81	1	64	<1, <1	43	<1, <1	22	OK84(2)
	283428	-B-	83(c)	<1(c)	85	80	1	64	9,62	42	OK84(2)	21	OK84(2)
	283429	0	106(c)	<1(c)	84	(g)	(b)	63	1,2	42	2,2	21	OK84(2)
	283430	0	88(c)	<1(c)	86	82	1	62	OK84(2)	41	OK84(2)	20	OK84(2)
61	283431	-A-	88(e)	7(e)	82	76	1	62	6,7	41	21,35	20	OK84(2)
34	283432	-C-	(d)	(d)	82	76	1	62	OK84(2)	41	OK84(2)	20	OK84(2)
	283433	-C-	81(c)	<1(c)	82	76	2	62	OK84(2)	41	OK84(2)	20	OK84(2)
	294011	0	99	5	87	81	2	61	2,3,3	40	5,5,16	20	OK28(3)
	283434	-A-	(d)	(d)	85	81	1	61	9, OK84	40	OK84(2)	20	OK84(2)
52	283435	4	94	1	84	80	1	60	25,33	40	OK84(2)	20	OK84(2)
34	283436	-A-	88(c)	<1(c)	83	80	1	60	OK84(2)	40	OK84(2)	20	OK84(2)
	277375	3	102(e)	3(e)	80	(g)	<1	60	2,3	40	24,27	20	OK84(2)
	277376	7	89(e)	4(e)	80	78	<1	60	9,9	40	29,53	20	OK84(2)
	277377	8	87(e)	5(e)	80	77	1	60	11,11	40	37	20	OK84(2)
19	277400	8	88(e)	10(e)	80	74	1	60	24,32	40	44,60	20	OK84(2)
90	294011	-C-	77	9	69	64	3	60(k)	OK84(3)	40(k)	OK84(3)	20(k)	OK84(3)
	293437	3	94	7	84	79	3	59	31,31	40	OK84(2)	20	OK84(2)
	283438	-A-	88(c)	<1(c)	79	79	1	59	OK84(2)	40	OK84(2)	20	OK84(2)
	277392	8	91(e)	8(e)	79	76	1	59	7,9	40	24,35	20	OK84(2)
6	277378	5	90(e)	8(e)	79	76	2	59	5,6	40	18,30	20	OK84(2)

(Continued)

Table CV Summary

STRESS COR. STOP CRACKING PREVENTION OF APM ABOVE EXP. 10

DATE: 10/10/2011

Allot No.	S. No.	Age of (a, b) Year	Longitudinal			T.R. Averages			75% (a)			50% (a)			25% (a)		
			7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg	7.5 kg
1	27735	13	67(e)	51(e)	5(e)	71	76	71	59	21,26	45,47	10	20	20	20	20	20
2	27736	13	51	51	5	64	76	76	58	52,52	45,47	39	39	39	39	39	39
3	27737	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
4	27738	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
5	27739	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
6	27740	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
7	27741	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
8	27742	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
9	27743	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
10	27744	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
11	27745	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
12	27746	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
13	27747	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
14	27748	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
15	27749	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
16	27750	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
17	27751	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
18	27752	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
19	27753	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
20	27754	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
21	27755	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
22	27756	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
23	27757	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
24	27758	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
25	27759	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
26	27760	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
27	27761	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
28	27762	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
29	27763	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
30	27764	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
31	27765	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
32	27766	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
33	27767	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
34	27768	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
35	27769	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
36	27770	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
37	27771	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
38	27772	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
39	27773	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
40	27774	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
41	27775	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
42	27776	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
43	27777	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
44	27778	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
45	27779	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
46	27780	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
47	27781	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
48	27782	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
49	27783	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
50	27784	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
51	27785	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
52	27786	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
53	27787	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
54	27788	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
55	27789	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
56	27790	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
57	27791	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
58	27792	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
59	27793	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
60	27794	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
61	27795	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
62	27796	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
63	27797	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
64	27798	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
65	27799	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
66	27800	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
67	27801	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
68	27802	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
69	27803	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
70	27804	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
71	27805	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
72	27806	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
73	27807	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
74	27808	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
75	27809	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
76	27810	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
77	27811	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
78	27812	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
79	27813	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
80	27814	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
81	27815	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
82	27816	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
83	27817	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
84	27818	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
85	27819	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
86	27820	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
87	27821	13	51	51	5	71	76	76	58	52,52	45,47	39	39	39	39	39	39
88	27822	13	51	51	5	71	76	76	58	52,52	45,47	39	3				

(Continued)

Table XLIV (Continued)

## STRESS CORROSION CRACKING RESISTANCE OF APM ALLOYS EXPOSED TO

3% NaCl: ALTERNATE IMMERSION TEST BATH

Alloy No.	S. No.	Age #2 (a, b) hrs	Longitudinal		Transverse		75% (d)		SCC (Stress-Failure Time) (e)	
			T.S. ksi	EL, S in/in	T.S. ksi	EL, S in/in	Stress ksi	Time hrs	Stress ksi	Time hrs
36	217936	36	79(e)	3(e)	67	64	50	21,69	34	ORBU(2)
4	277376	27	76(e)	7(e)	67	63	50	9,11	34	ORBU(2)
67	291196	-C-	83	20	78	66	50	ORBU	33	ORBU(2)
2	277376	28	77(e)	8(e)	66	65	50	61, ORBU	33	ORBU(2)
90	291011	-A-	81	6	77	71	49(m)	73, 75, ORBU	33(m)	ORBU(3)
36	291051	16	82	4	76	65	49	26, 84	32	ORBU(2)
90	291051	-B-	78	10	71	65	49	ORBU(3)	32	ORBU(2)
5	277377	28	83(e)	5(e)	67	64	49	11, 14	32	ORBU(2)
36	277379	36	78(e)	4(e)	67	65	49	11, 34	32	ORBU(2)
90	291051	0	83	5	87	81	48(n)	9, 16, 21	32(n)	ORBU(3)
71	291187	20	81	6	75	64	48	ORBU(2)	32	ORBU(2)
79	291188	-B-	80	9	75	64	48	(2)	32	ORBU(2)
90	291051	-C-	77	9	69	64	48	3(2)	32	ORBU(2)
31	277206	35	79(e)	7(e)	64	62	48	22, 41	32	ORBU(2)
52	281051	24	79	6	71	63	47	ORBU(2)	32	ORBU(2)
79	291188	-C-	76	9	73	62	46	ORBU(2)	31	ORBU(2)
49	281051	-	120	4.1	51	(g)	--	-----	26	4, 6
7178-7652 (1)			92	8	63	74	56	1 to 12	37	1 to 13
7075-7652 (1)			88	11	75	65	49	1 to 12	32	1 to 13
7075-7652 (1)			75	11	62	54	42	ORBU	28	ORBU

(a) All specimens SNT 2 Hrs at 800°F, OR, Age #1, 24 Hrs at 250°F, Age #2, as noted (exceptions are listed (b)).

(b) Heat Treatment Used

- A- Age #1, 1 Hrs at 250°F, Age #2 6 Hrs at 150°F.
- B- Age #1, 1 Hrs at 150°F.
- C- Age #1, 24 Hrs at 150°F.
- D- SNT 2 Hrs at 800°F, OR, Age #1 24 Hrs at 250°F.
- E- Age #1, 6 Hrs at 150°F.
- F- Age #1, 6 Hrs at 150°F.
- G- Age #1, 6 Hrs at 250°F, Age #2, 6 Hrs at 150°F.
- H- Age #1, 12 Hrs at 150°F.
- I- OR, Age #1, 24 Hrs at 250°F.
- J- OR, Age #1, 6 Hrs at 150°F.
- Q- No Age #2

(c) Tensile properties were determined on another extrusion.

(d) Not determined.

(e) Predicted values from available data.

(f) Each value represents one specimen except for run cuts where number of specimen are given in parentheses.

(g) Called before reaching 0.2% offset.

(h) Specimen fractured.

(i) Typical from 0.2% to 0.25% to 0.3% A. extrusion.

(j) Stress levels except when noted.

(k) Stress levels used, 175, 615, and 335, respectively.

(l) Stress levels used, 425, 535, and 275, respectively.

(m) Stress levels used, 495, 645, and 245, respectively.

(n) Stress levels used, 615, 615, and 245, respectively.

Table XLV

A. P. M. ALLOYS MEETING STRESS CORROSION AND STRENGTH TARGETS

Alloy	Composition, %			Other	Aging (a)	Longitudinal			SCC	
	Zn	Mg	Cu			T.S. ksi	Y.S. El., ksi in 4D	(b) Stress ksi	Days to Failure	
<u>Target I: 7075-T7351 + 10% with no failures at 75% Y.S.</u>						83	73	5	>48	O K 84
<u>A. Meets Target</u>										
87	7.6	2.5	1.1	2.2 Fe, 2.3 Ni, .2 Cr.	6 @ 250 + 8 @ 330	90	81	7	57	O K 84
87	7.6	2.5	1.1	2.2 Fe, 2.3 Ni, .2 Cr.	16 @ 330	86	75	7	52	O K 84
77	6.8	2.8	2.1	1.0 Mn, .5 Cu, .1 Zr.	6 @ 250 + 8 @ 330	86	79	8	54	O K 84
90	7.5	2.4	1.0	1.1 Fe, 1.0 Ni, .2 Cr.	6 @ 250 + 8 @ 330	84	78	8	54	O K 84
<u>B. May Meet Target After Changes in Aging Practices</u>										
71	9.3	3.6	.5	.7 Co, .02 Cr, .02 Ti.	24 @ 250 + 8 @ 330	87	80	9	60	45,54
52	10.0	4.0	.9	1.5 Co, .01 Cr, .02 Ti	24 @ 250 + 4 @ 330	94	89	4	54	33,33
34	7.8	2.5	1.0	3.5 Fe, 4.9 Ni, .09 Cr	6 @ 250 + 8 @ 330	88	83	<1	61	O K 84
<u>Target II: 7178-T651 + 10% Strength Improvement, Equal Stress Corrosion Resistance</u>						101	92	5	7-20	O K 84
<u>A. Meets Target</u>										
71	9.3	3.6	.5	.7 Co, .02 Cr, .02 Ti	24 @ 250	108	105	5	22	O K 84
<u>B. May Meet Target After Changes in Aging Practices</u>										
52	10.0	4.0	.9	1.5 Co, .01 Cr, .02 Ti	24 @ 250	106	103	2	45	12,15
87	7.6	2.5	1.1	2.2 Fe, 2.3 Ni, .2 Cr.	24 @ 250	105	99	3	45	82, OK84
52	10.0	4.0	.9	1.5 Co, .01 Cr, .02 Ti	24 @ 250	109	106	3	49	1,1
<u>Target III: Better Stress Corrosion Resistance than 7178-T651</u>						92	84	5	7-20	O K 84
<u>A. Meets Target</u>										
71	9.3	3.6	.5	.7 Co, .02 Cr, .02 Ti	24 @ 250	108	105	5	22	O K 84
90	7.5	2.4	1.0	1.1 Fe, 1.0 Ni, .2 Cr	24 @ 250	99	93	5	20	O K 28
71	9.3	3.6	.5	.7 Co, .02 Cr, .02 Ti	24 @ 250 + 3 @ 330	94	88	7	40	O K 84
<u>B. May Meet Target After Changes in Aging Practices</u>										
52	10.0	4.0	.9	1.5 Co, .01 Cr, .02 Ti	24 @ 250	106	103	2	24	O K 84
87	7.6	2.5	1.1	.02 Mn, 2.2 Fe, 2.3 Ni, .2 Cr.	24 @ 250	105	99	3	22	O K 84
52	10.0	4.0	.9	1.5 Co, .01 Cr, .02 Ti	24 @ 250 + 4 @ 330	94	89	4	40	O K 84
3	12.3	4.0	1.6	.5 Mn	24 @ 250 + 3 @ 330	102	99	3	20	O K 84
79	6.5	2.7	2.1	1.0 Mn, .5 Cr.	24 @ 250	97	88	4	20	O K 84

(a) All Specimens SH1 & Hru at 860°F, Cold water quenched, aged as noted.

(b) Exposed in 3% NaCl AI bath at room temperature. Stressed in short transverse direction.

TABLE XLVI

## ELECTRON MICROPROBE SEMI-QUANTITATIVE ANALYSIS OF LARGE CONSTITUENT PARTICLES

Alloy No.	S. No.	(e)	Alloying Elements	Constituent Composition, %									
				Zn	Mg	Cu	Mn	Fe	Ni	Cr	Ti	V	Al
14	277386		A	10	5	2	2	6	10	3(d)	—	—	57
15	277387		B	2	3	0	—	—	—	1	18	7	55
16	277388		C	1	0	0	—	—	—	7	12	17	59
17(b)	277389		D	16	9	4	—	—	—	—	—	—	47
17(c)	277389		D	3	3	0	—	—	—	—	—	—	51
18	277390		E	1	0	0	0	0	0	1	15	6	56

Note: Accuracy  $\pm 10\%$  of indicated value for amounts above 10%,  $\pm 20\%$  of indicated value for amounts below 10%.

(a) Base composition, Al-12% Zn-3.5% Mg-1.5% Cu to which were added

A - 1.5% each Mn, Fe, Ni  
 B - 0.5% each Cr, Ti, V, Zr  
 C - 2.0% Cr-0.8% Ti-2.0% V-1.0% Zr  
 D - 1.5% each Co, Mo, W  
 E - 0.5% each Mn, Fe, Ni, Cr, Ti, V, Zr, Co, Mo, W

(b) Light colored particle.

(c) Dark colored particles.

(d) Not an alloying element. Presence unexplained.

(e) Extrusions SHT 2 hrs at 860°F, CMQ, Aged 24 hrs at 250°F.



ELECTRON MICROPROBE SEMI-QUANTITATIVE ANALYSIS

# OF LARGE CONSTITUENT PARTICLES IN ALLOYS 34 AND 87

(a) Alloy 34, Al 7.77 Zn, 2.48 Mg, 1.03 Cu, 0.02 Mn, 3.49 Fe, 4.94 Ni, 0.09 Cr, 0.01 Ti, 0.07 Si, 0.59 Al<sub>2</sub>O<sub>3</sub>  
Alloy 87, Al 7.58 Zn, 2.52 Mg, 1.06 Cu, 0.02 Mn, 2.16 Fe, 2.26 Ni, 0.16 Cr, 0.01 Ti, 0.01 V, 0.01 Co, 0.05 Si, 0.63 Al<sub>2</sub>O<sub>3</sub>

Table XLIX

ELECTRON MICROPROBE SEMI-QUANTITATIVE ANALYSIS  
OF LARGE CONSTITUENT PARTICLES IN ALLOYS 52 AND 71

Alloy (a)	Particles Particular Type	Constituent Composition %											Remarks		
		Al	Mg	Zn	Fe	Mn	Cr	Cu	Pb	Ti	Si	Co		$\frac{Ca}{(b)}$	$\frac{K}{(b)}$
52	2,5,7,9-15	Nonmetallic	-	-	-	-	-	-	-	40	10	-	-	-	Probably Zr SiO <sub>4</sub>
	1	Nonmetallic	-	10	-	7	-	-	-	-	-	-	-	-	
	3	Nonmetallic	20	-	-	1	-	-	-	-	-	-	-	-	
	4	Nonmetallic	10	-	-	3	-	-	-	-	10	-	-	1	
	8	Nonmetallic	25	-	-	2	-	-	-	-	10	-	L	L	
	16B {c}	Nonmetallic	-	-	-	72	-	-	-	-	-	-	M	-	
	16A {c}	Metallic	56	-	2	42	-	-	-	2	-	-	-	-	Fe <sub>3</sub> O <sub>4</sub> FeAl <sub>3</sub> Fe
	16C {c}	Metallic	-	-	-	>95	-	-	-	2	-	-	-	-	
71	3,C,B,H	Nonmetallic	-	-	-	-	-	-	-	40	10	-	-	-	Zr SiO <sub>4</sub>
	2	Nonmetallic	10	5	-	-	-	-	-	-	-	-	-	-	
	5	Nonmetallic	7	5	-	<4	-	-	-	-	15	-	-	2	
	A	Nonmetallic	12	20	2	1	-	-	-	-	3	-	-	-	
	B	Nonmetallic	-	1	1	-	-	-	-	-	30	-	-	-	
	D	Nonmetallic	15	-	1	<1	-	-	-	-	10	-	-	M	
	G	Nonmetallic	15	15	2	<1	-	<1	-	-	3	-	-	-	
	I	Nonmetallic	15	15	2	<1	-	-	-	-	4	-	-	-	
	J	Nonmetallic	30	2	1	<1	-	-	-	-	4	-	-	-	
	1	Metallic	55	-	10	20	2	13	-	<1	4	-	-	-	
	4	Metallic	59	4	5	30	-	2	-	-	-	-	-	-	
	F	Metallic	55	10	7	25	-	3	-	-	-	-	-	-	<1

(a) Alloy 52, Al, 9.99 Zn, 4.01 Mg, 0.92 Cu, 0.01 Mn, 0.14 Fe, 0.01 Ni, 0.01 Cr, 0.02 Ti, 1.49 Co, 0.06 Si, 0.75 Al<sub>2</sub>O<sub>3</sub>

Alloy 71 Al, 9.2 Zn, 3.62 Mg, 0.63 Cu, 0.02 Cr, 0.03 Ti, 0.75 Co, 0.26 Al<sub>2</sub>O<sub>3</sub>

(b) Since per cent standards are not readily available for these elements, L for Low and M for Moderate were used.

(c) Particle 16 is very large and is composed of three phases--a dark nonmetallic phase (16B--Fe<sub>3</sub>O<sub>4</sub>) one side, a light metallic phase (16A--FeAl<sub>3</sub>) on the other side and a small amount of a second metallic phase (16C--Metallic Fe) in the center.



Table L  
IDENTIFICATION OF PHASES IN POWDERS AND EXTRUDED SECTIONS

Alloy	S. No.. (a)	Product	Age (b) t/T Hrs/°F	Phases									
				Al	Mg <sub>2</sub> Si	FeNiAl <sub>9</sub>	MgZn <sub>2</sub>	Al <sub>12</sub> Mg <sub>2</sub> Cr	Co <sub>2</sub> Al <sub>9</sub>	Mg <sub>3</sub> Zn <sub>3</sub> Al <sub>2</sub>	(e)	M'ppt	
34	283269	Powder	----	✓	✓	(c) ✓	---	---	---	---	---	---	
	283441	2" Ø Extrusion	24/250	✓	✓	(c) ✓	✓	✓	---	---	✓	---	
	283441	2" Ø Extrusion	48/315	✓	✓	(c)(d) ✓	✓	✓	---	---	✓	✓	
87	307598	Powder	----	✓	✓	(c) ✓	---	---	---	---	---	---	
	293196	2" Ø Extrusion	24/250	✓	✓	(c) ✓	✓	✓	---	---	✓	---	
	293196	2" Ø Extrusion	48/315	✓	✓	(c)(d) ✓	✓	✓	---	---	✓	✓	
52	283274	Powder	----	✓	✓	---	---	---	✓	---	---	---	
	283490	2" Ø Extrusion	24/250	✓	✓	---	✓	---	✓	✓	---	---	
	283490	2" Ø Extrusion	24/250 + 20/330	✓	✓	---	✓	---	✓	✓	---	✓	
71	283303	Powder	----	✓	✓	---	---	---	✓	---	---	---	
	293387	2" Ø Extrusion	24/250	✓	✓	---	✓	---	✓	✓	---	---	
	293387	2" Ø Extrusion	24/250 + 20/330	✓	✓	---	✓	---	✓	✓	---	✓	
(a) Alloy				Zn	Mg	Si	Cr	Ti	Co	V	Sn		
34				7.77	2.46	0.02	0.09	0.01	---	---	0.01	0.07	
87				7.58	2.52	0.02	0.16	0.01	0.01	0.01	0.05		
52				9.99	4.01	0.01	0.01	0.02	1.49	---	0.06		
71				9.29	3.58	---	0.02	0.02	0.68	---	---		

(b) Powders were as atomized, extrusions were SHT 2 Hrs at 860°F, CWQ, aged as noted.

(c) FeNiAl<sub>9</sub> finer in powder, amount increased in extrusion.

(d) The amount of FeNiAl<sub>9</sub> increases in extrusion.

(e) Four lines unidentified.

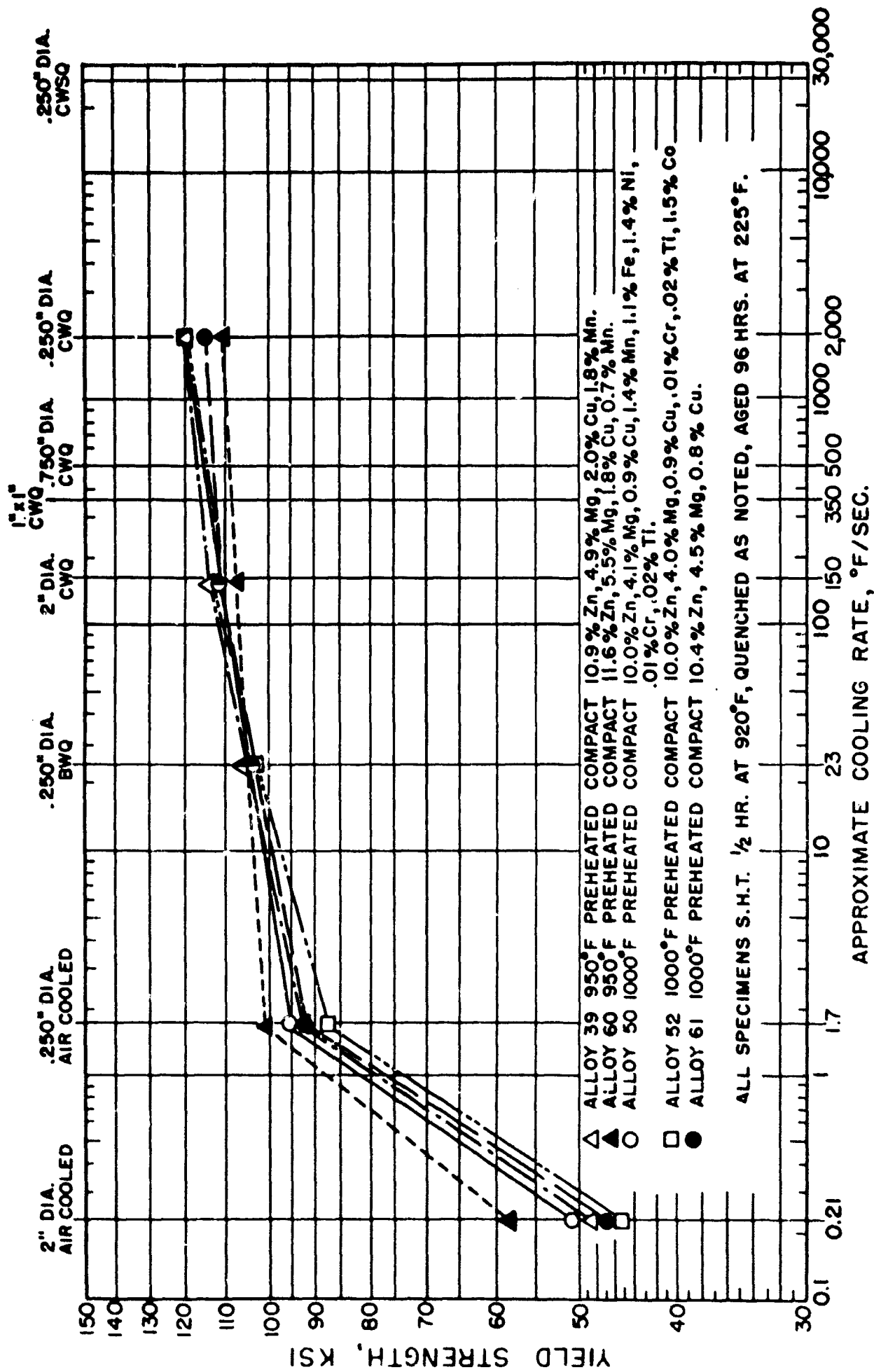


FIGURE 1

THE EFFECT OF COOLING RATE ON YIELD STRENGTHS OF EXTRUSIONS

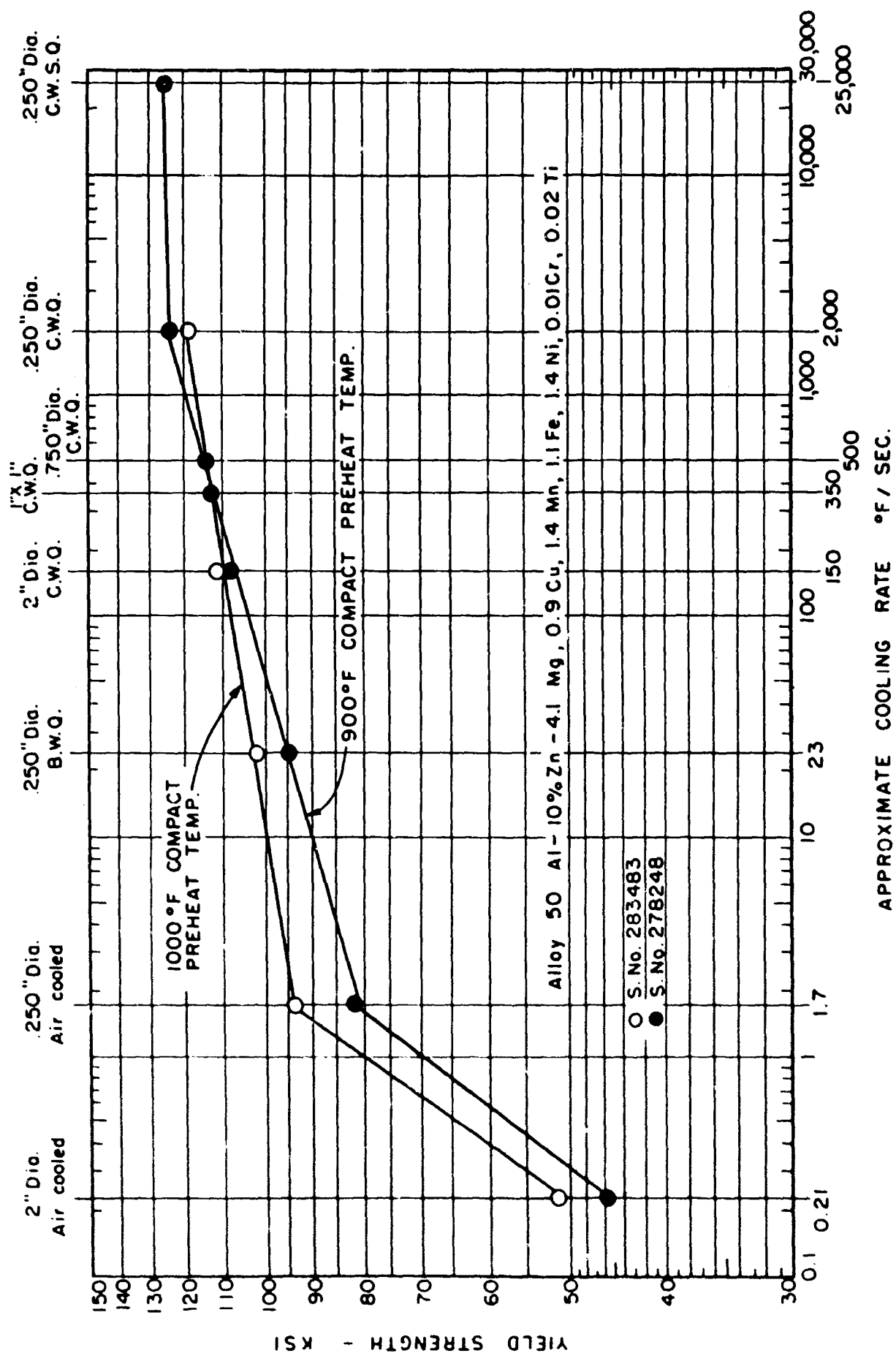


FIGURE 2  
THE EFFECT OF COMPACT PREHEAT TEMPERATURE AND COOLING RATE ON ROOM TEMPERATURE LONGITUDINAL YIELD STRENGTH

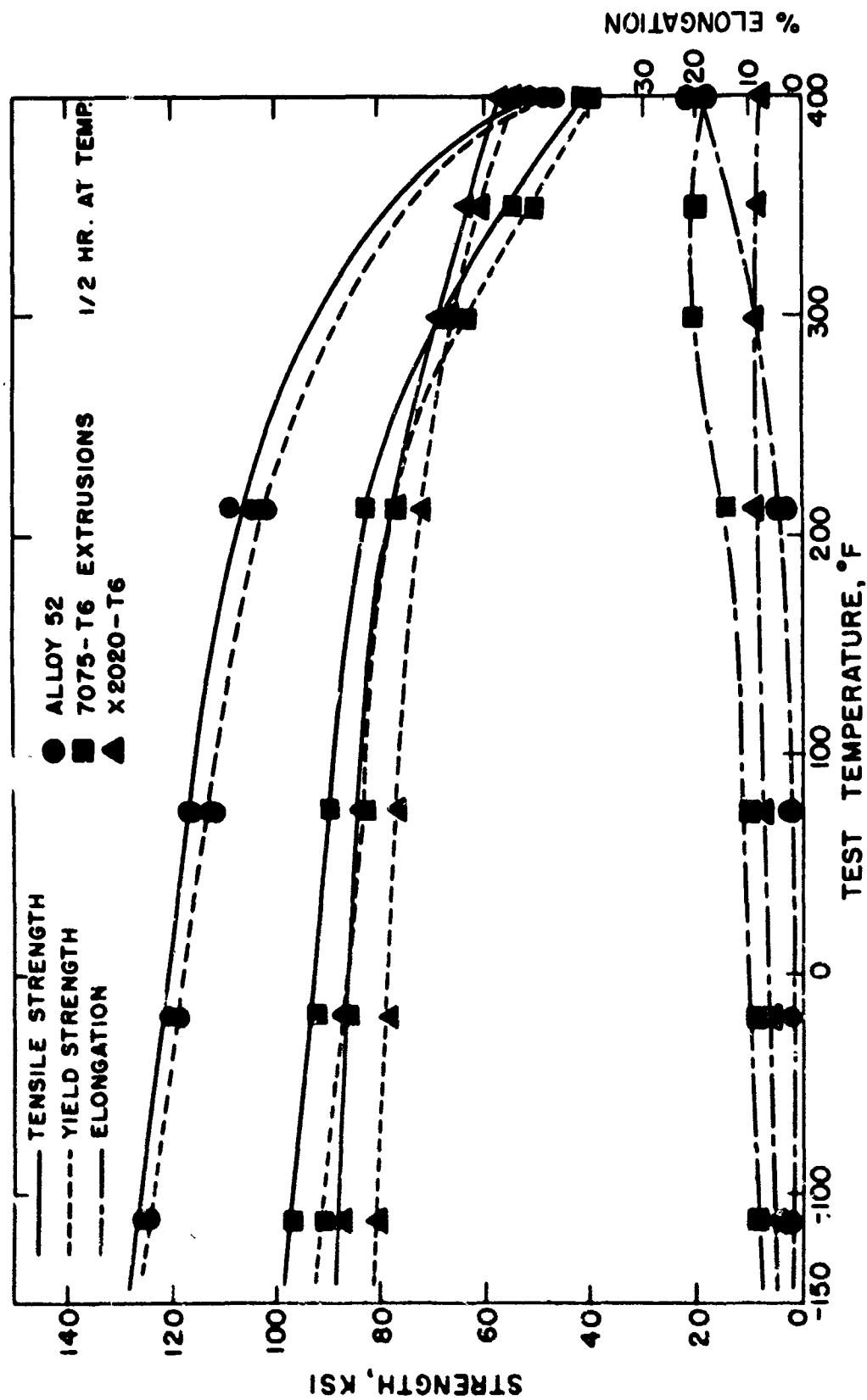
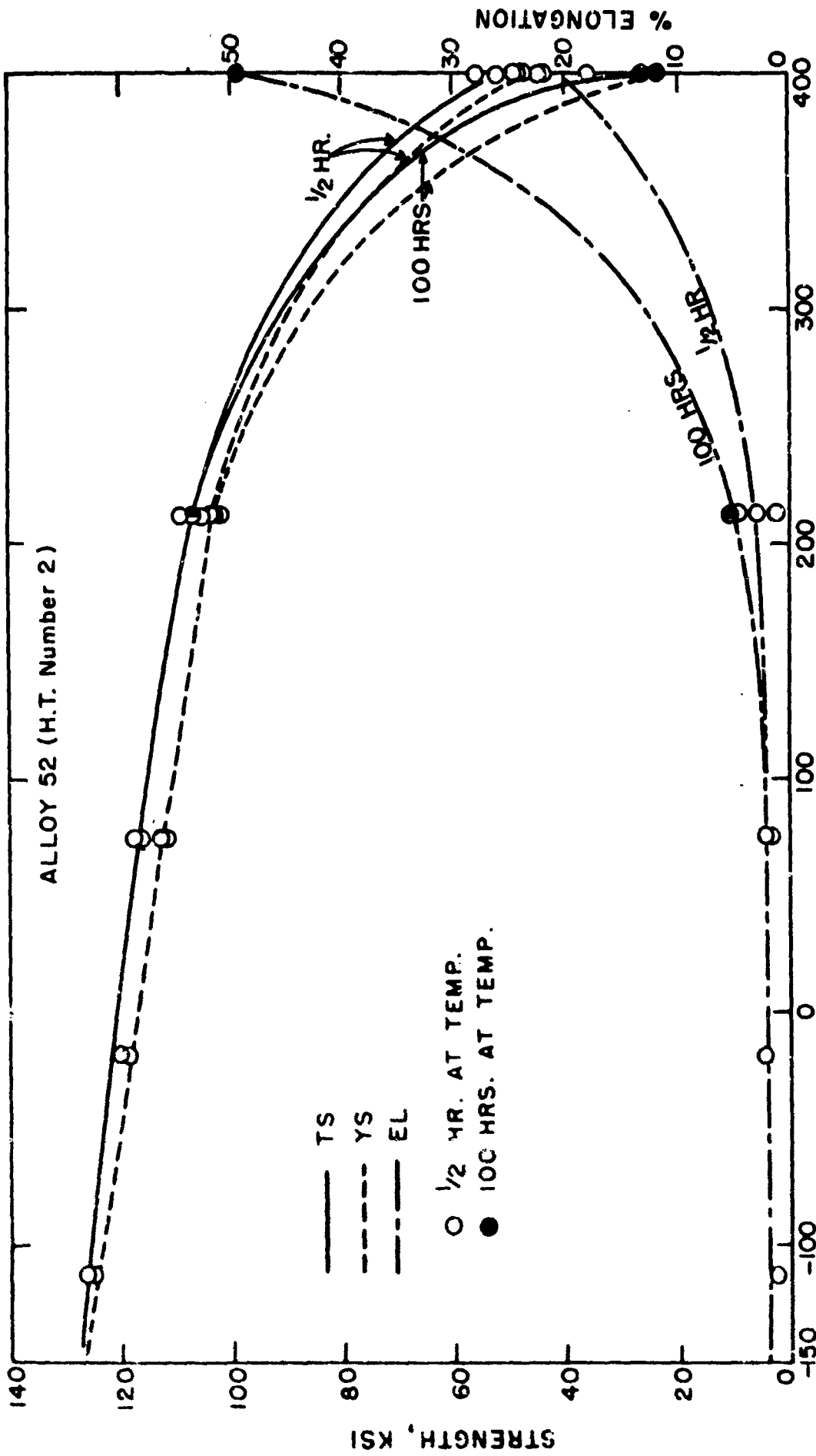


FIGURE 3  
LONGITUDINAL TENSILE PROPERTIES OF ALLOYS AT  
CRYOGENIC AND ELEVATED TEMPERATURES



TEMPERATURE, °F

FIGURE 4

THE EFFECT OF TIME AT TEMPERATURE ON LONGITUDINAL TENSILE PROPERTIES

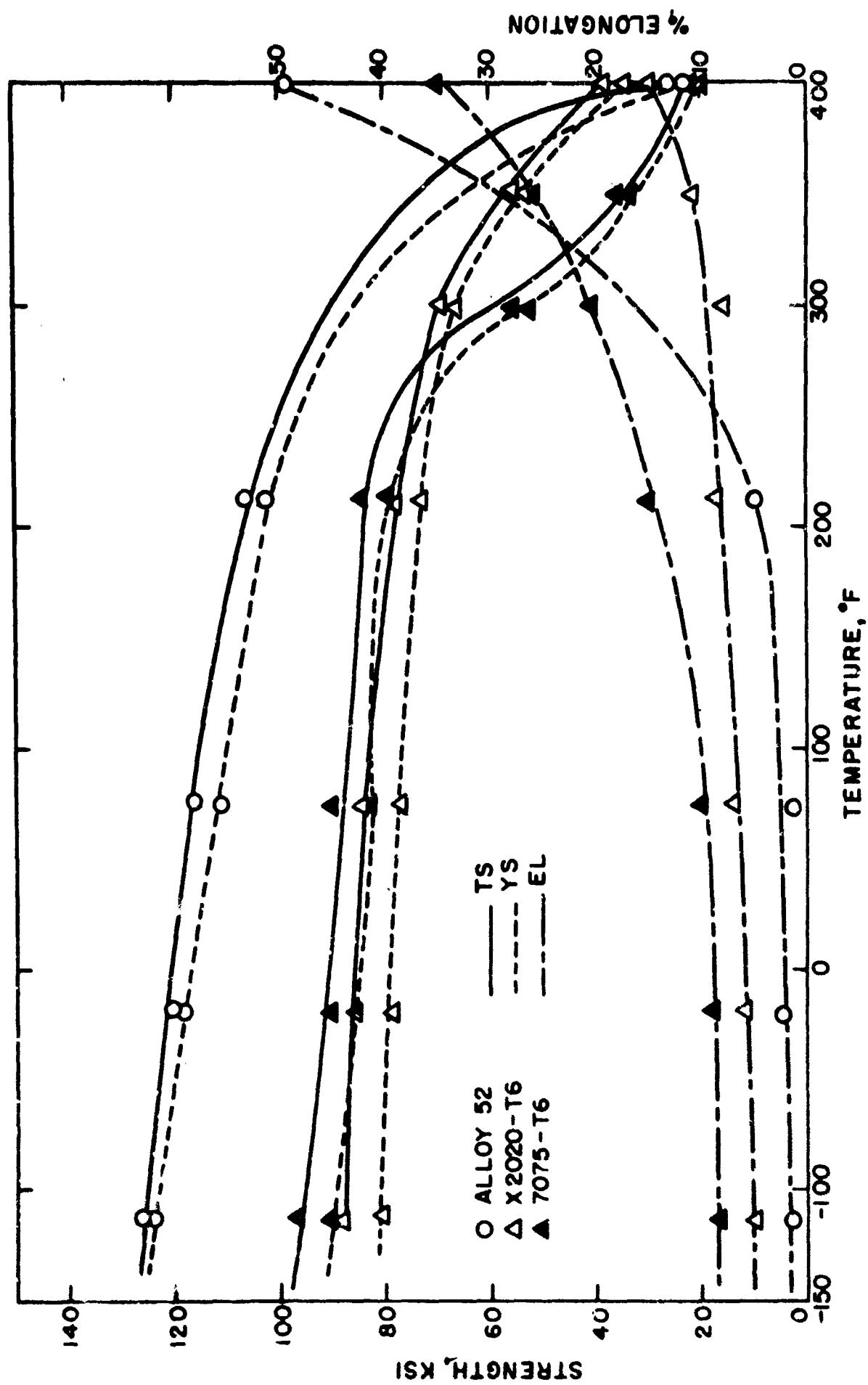


FIGURE 5

LONGITUDINAL TENSILE PROPERTIES AFTER 100 HOURS AT TEMPERATURE

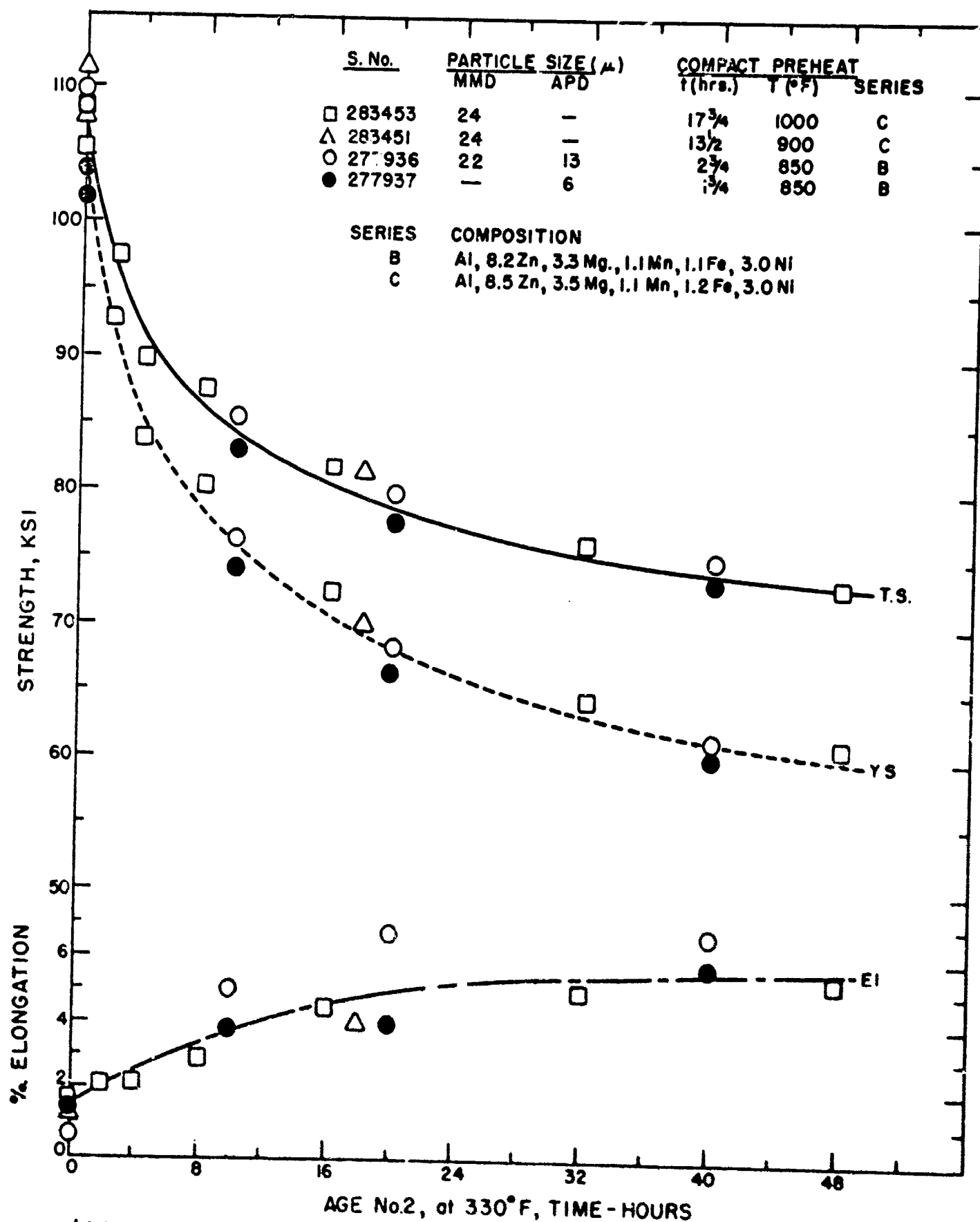
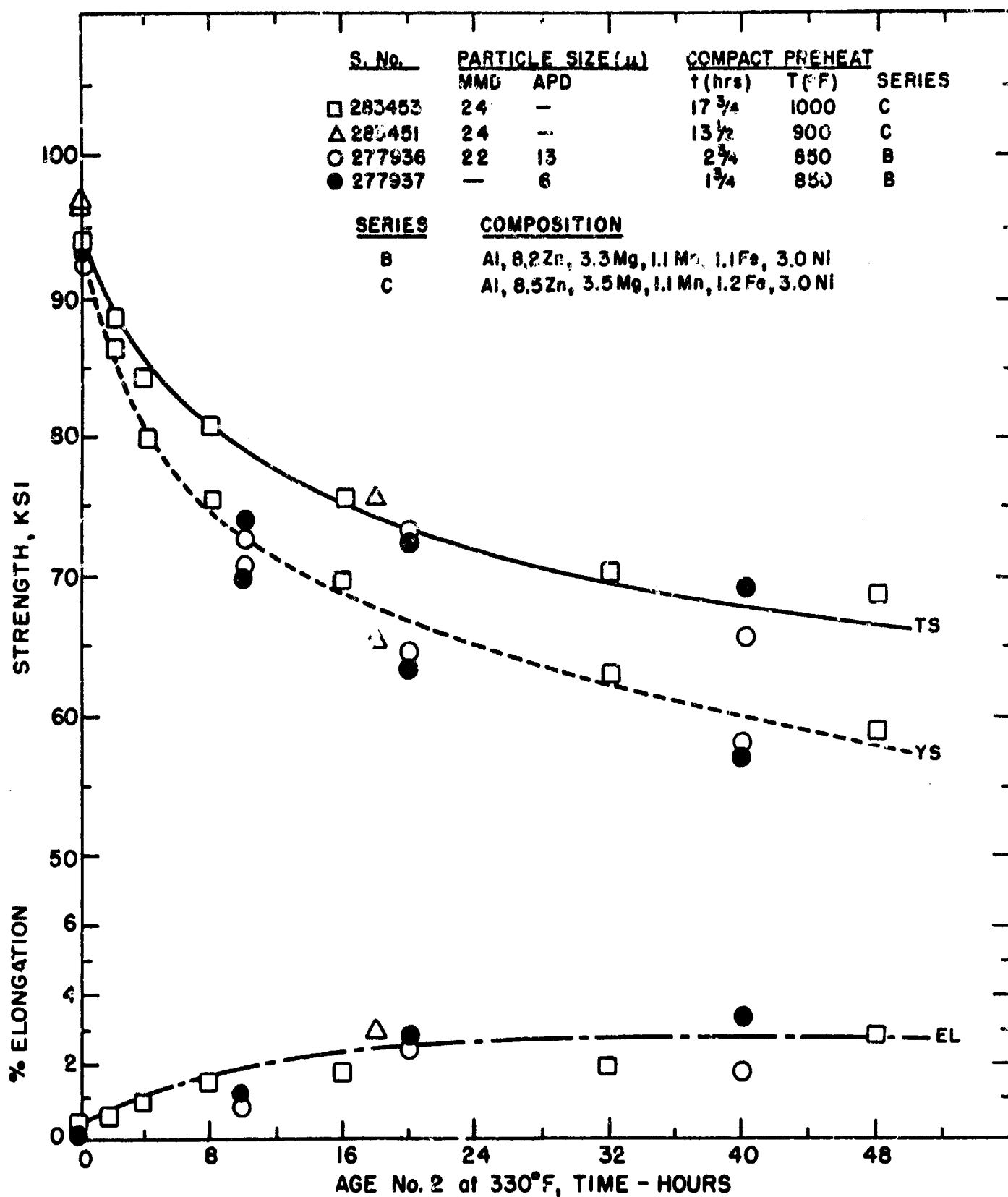


FIGURE 6  
THE EFFECT OF STEP AGING ON ALLOY 38 - LONGITUDINAL

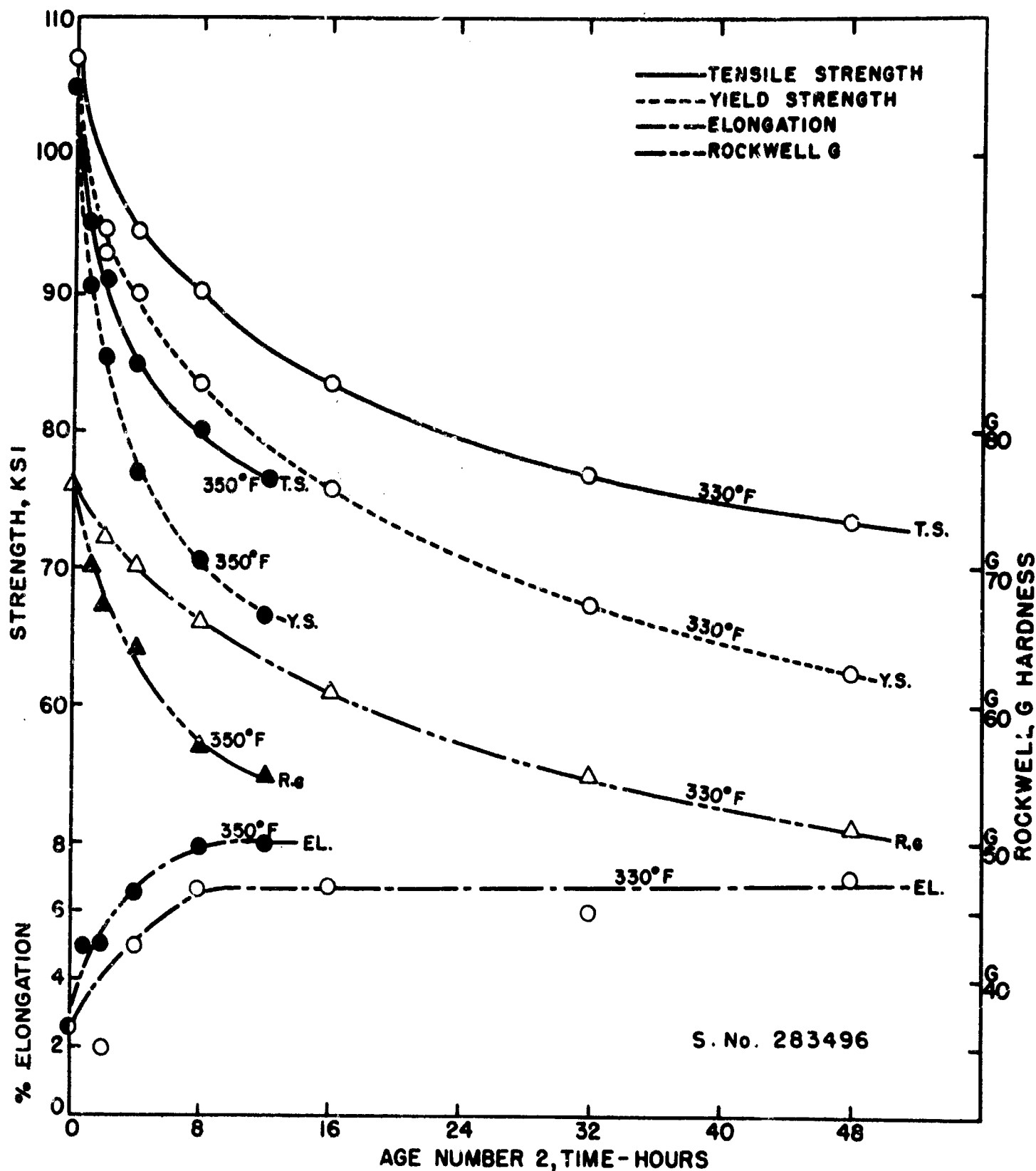


(All specimens SHT 2 hours at 860°F, CWQ, Age number 1 24 hours at 250°F)

FIGURE 7

THE EFFECT OF STEP AGING ON ALLOY 38 - TRANSVERSE





(All specimens SHT 2 hours at 860°F, CWQ, Age number 1, 24 hours at 250°F)

FIGURE 8  
THE EFFECT OF STEP AGING ON ALLOY 52 - LONGITUDINAL

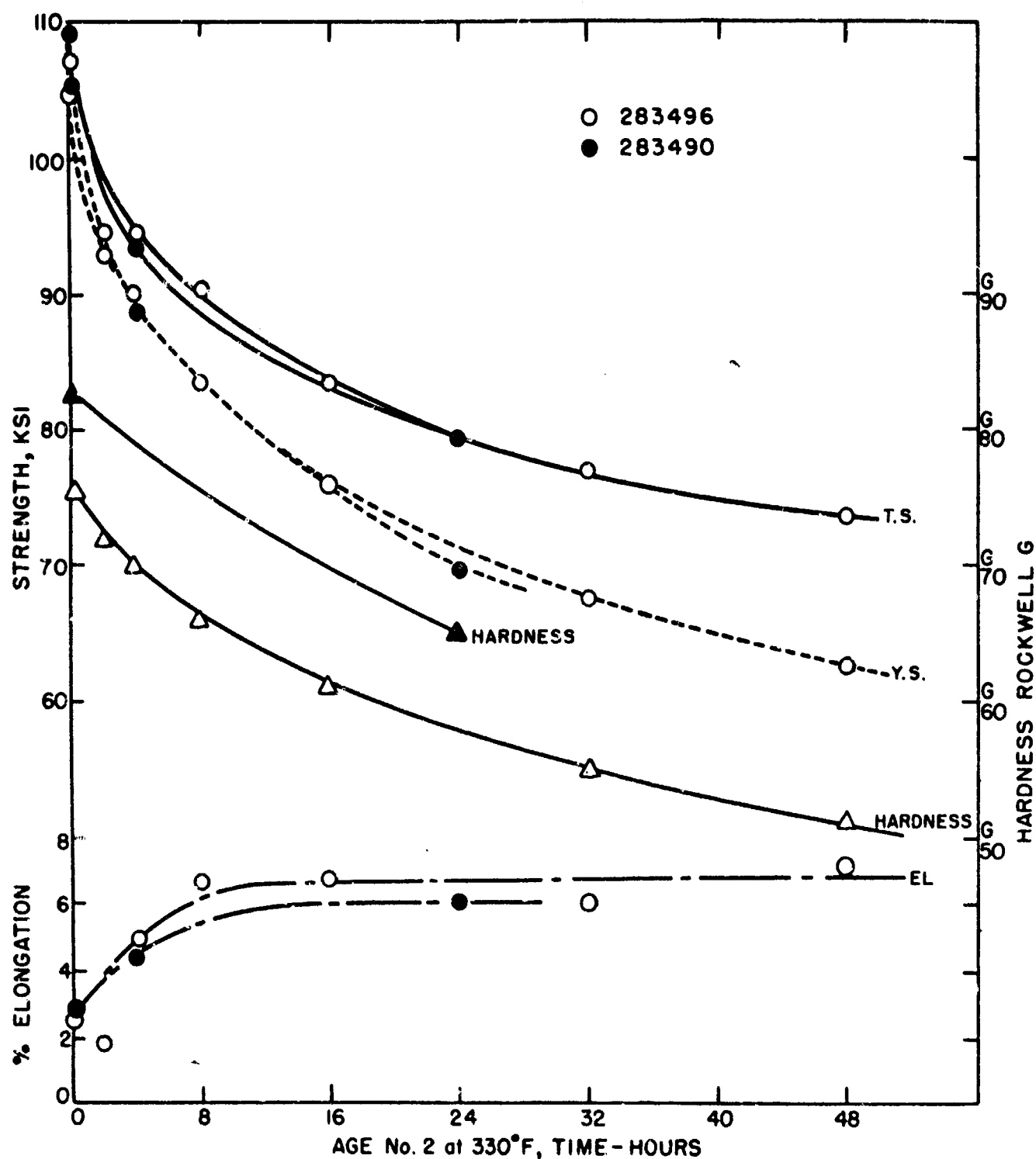
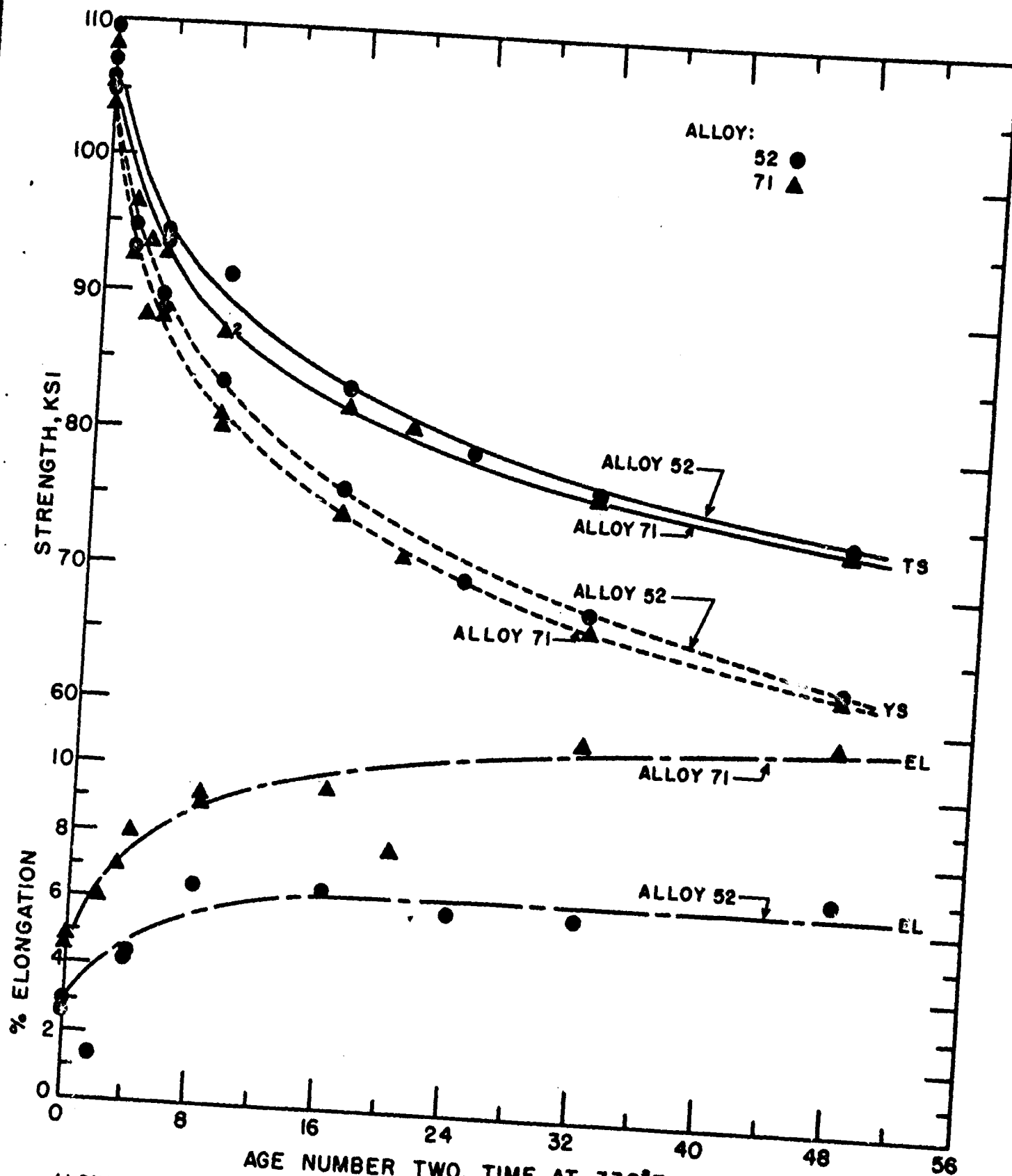
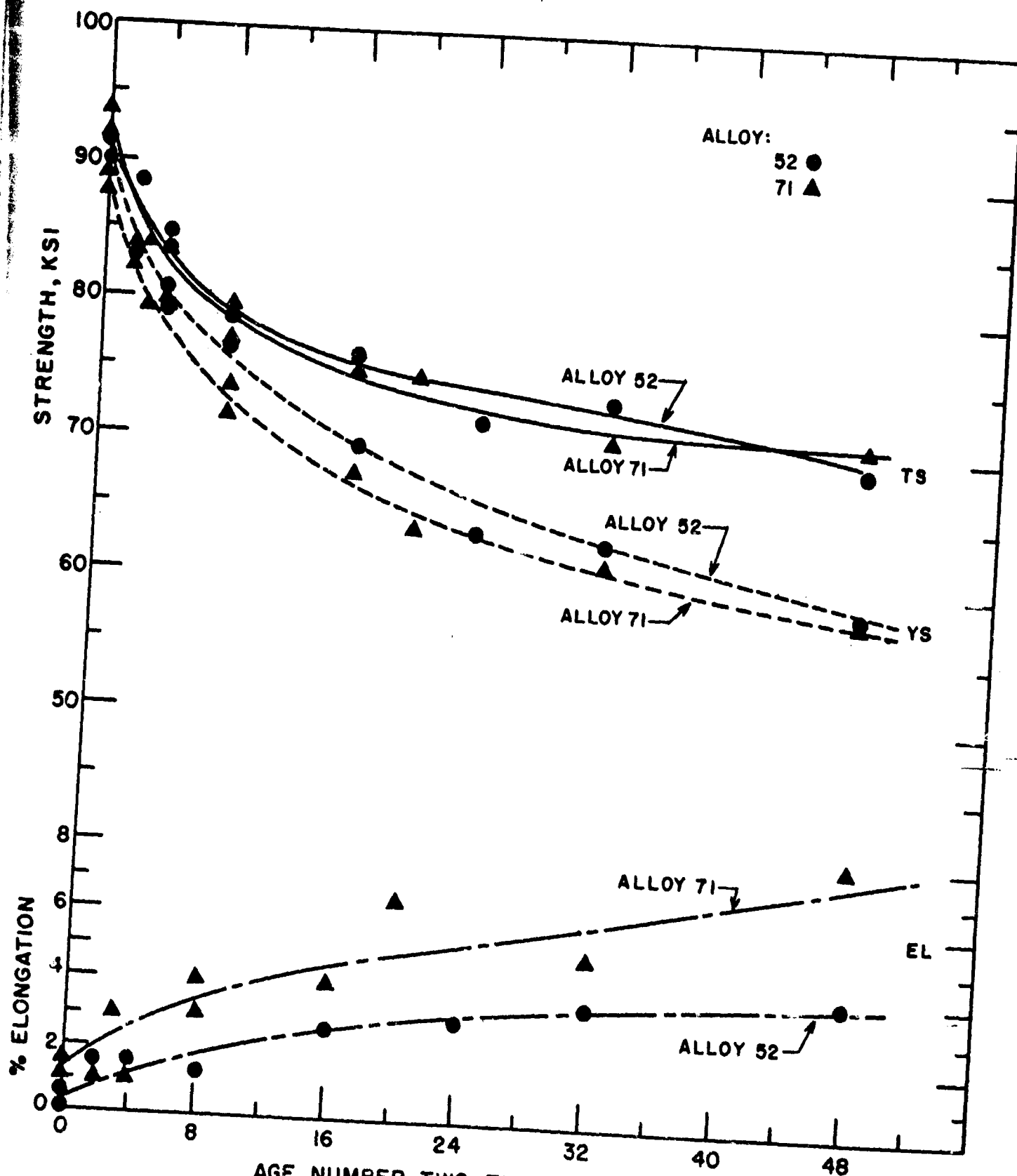


FIGURE 9  
THE REPRODUCIBILITY OF STEP AGING ON ALLOY 52 - LONGITUDINAL



(LONGITUDINAL PROPERTIES: ALL SPECIMENS SHOT 2 HRS AT 860°F, CWQ, AGE NUMBER ONE, 24 HRS AT 250°F.)

FIGURE 10  
THE EFFECT OF STEP AGING ON ALLOYS 52 AND 71 -  
LONGITUDINAL



(TRANSVERSE PROPERTIES: ALL SPECIMENS SHOT 2 HRS AT 860°F, CWQ, AGE NUMBER ONE, 24 HRS AT 250°F)

FIGURE 11  
THE EFFECT OF STEP AGING ON ALLOYS 52 AND 71 -  
TRANSVERSE

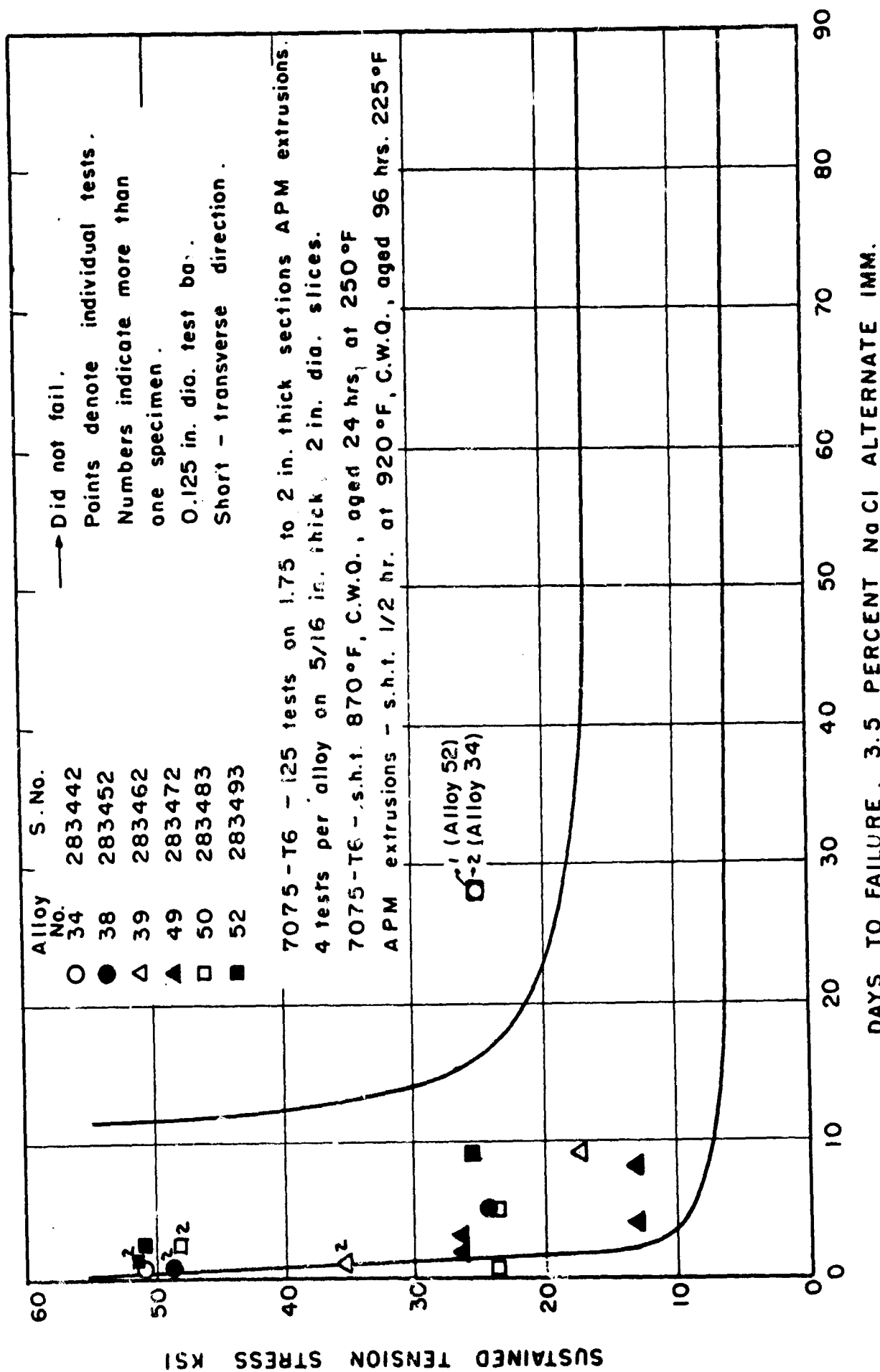


FIGURE 12

COMPARISON OF RELATIVE RESISTANCE TO STRESS-CORROSION CRACKING OF ALUMINUM  
 POWDER METALLURGY EXTRUSIONS AND 7075-T6 EXTRUDED SECTIONS

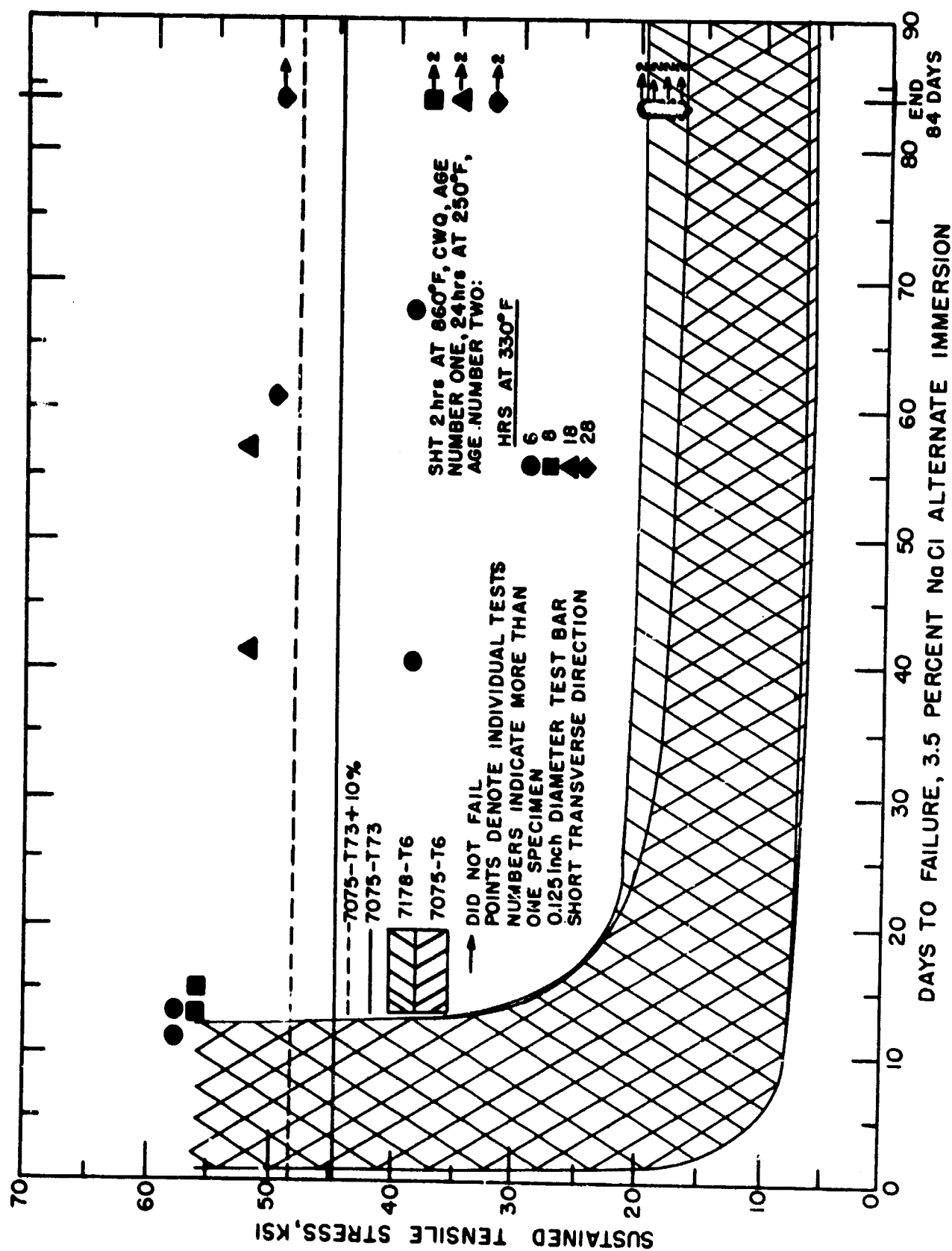


FIGURE 13  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 2

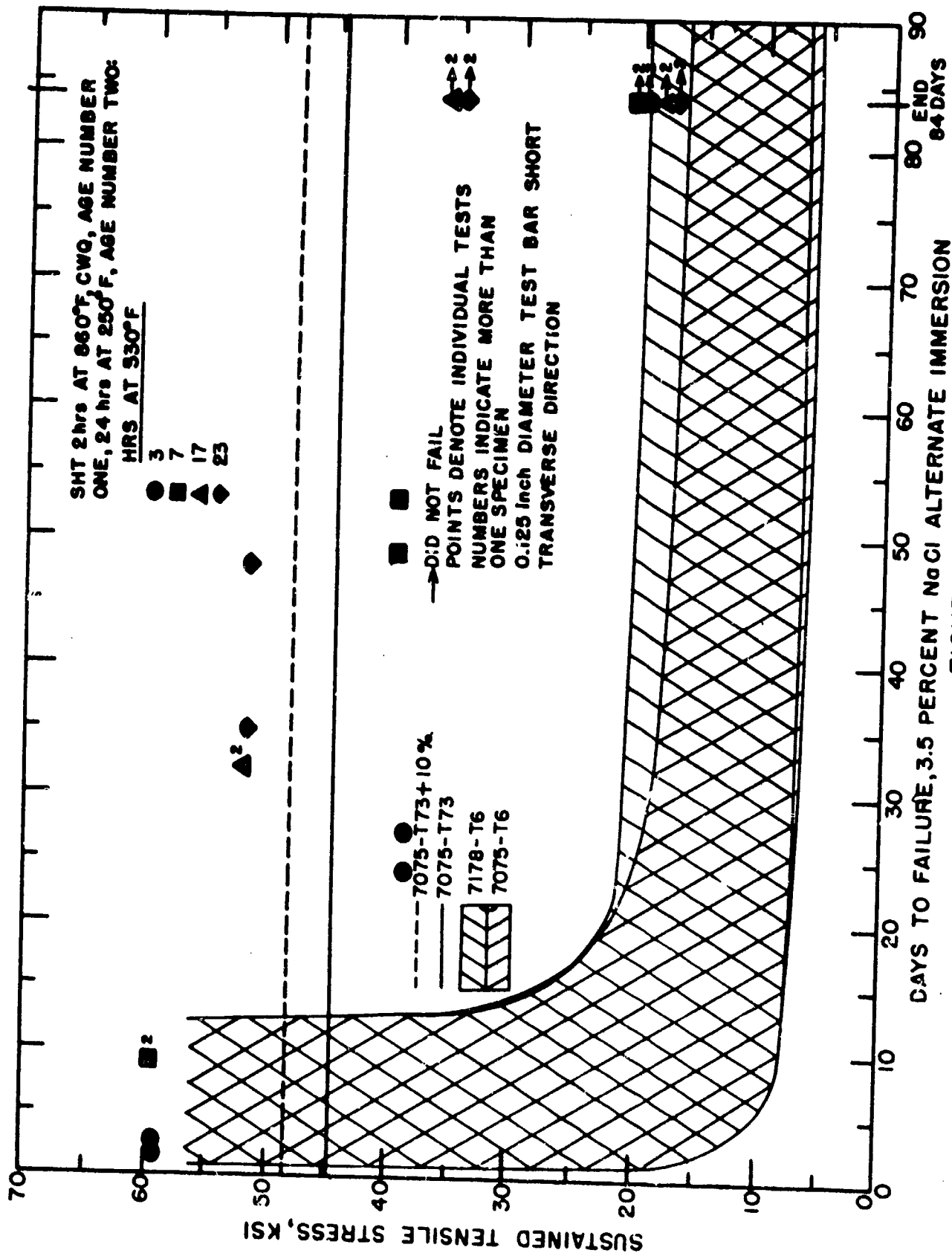


FIGURE 14  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 3

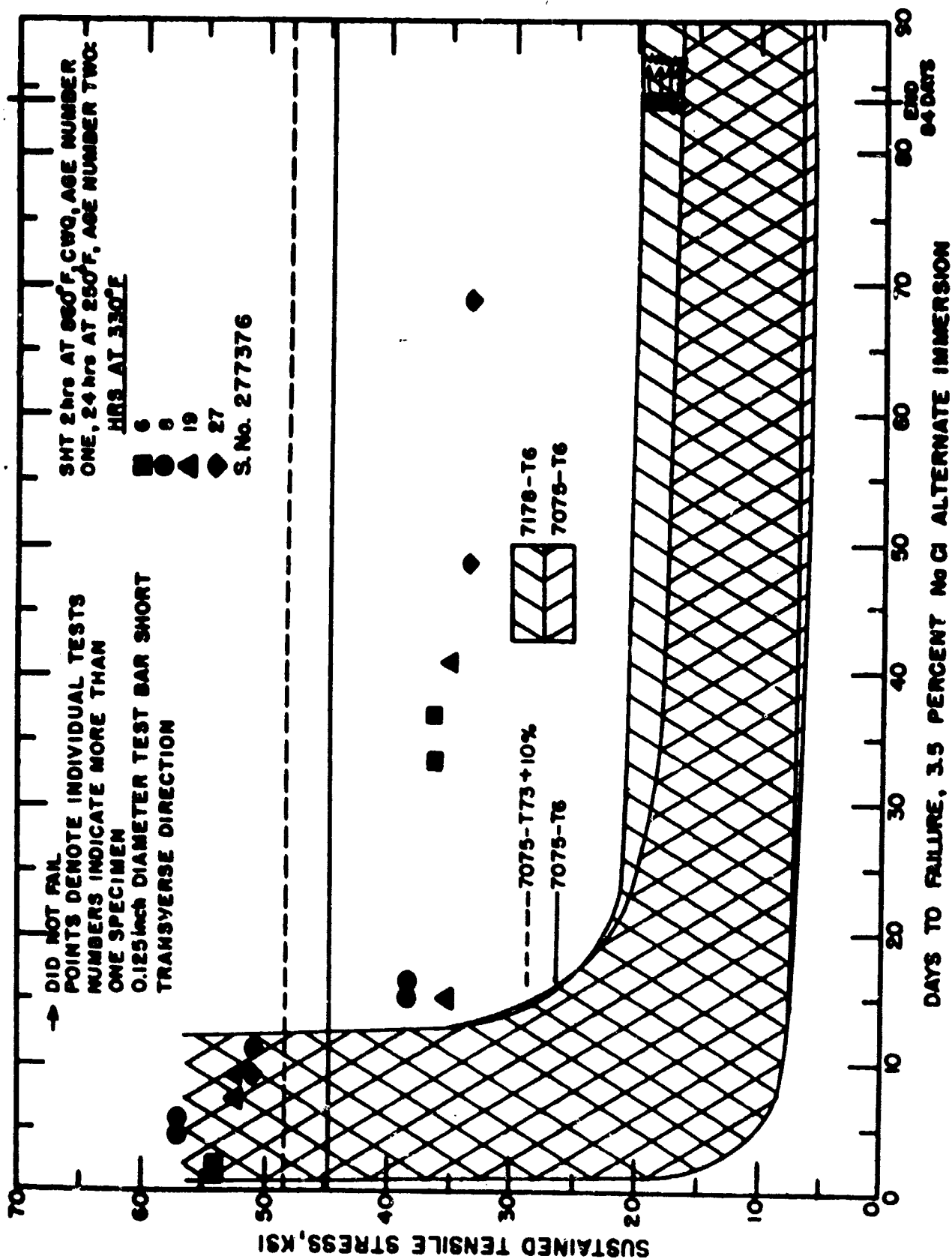
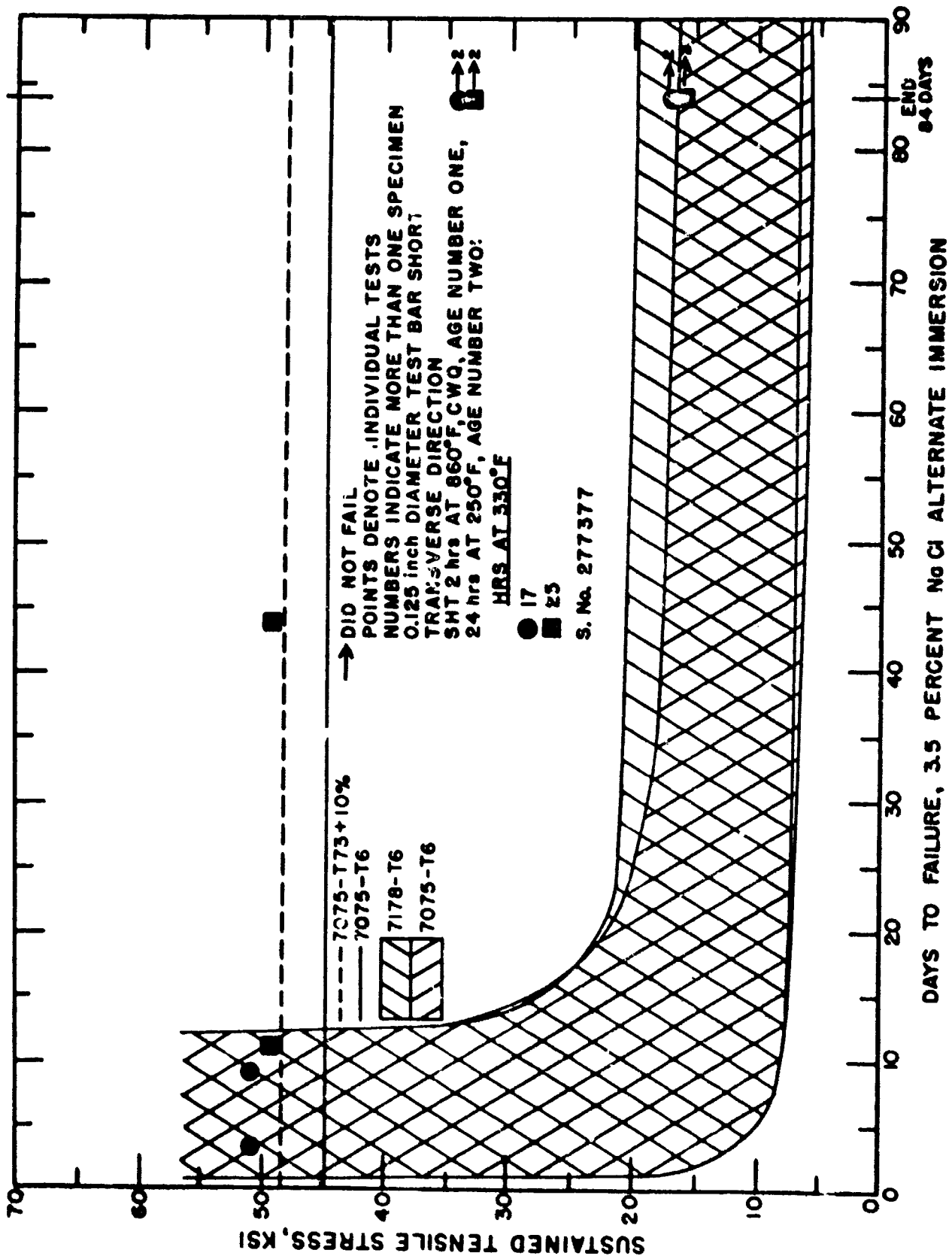


FIGURE 15  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 4





**FIGURE 16**  
**COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS**  
**CORROSION CRACKING FOR ALLOY 5**

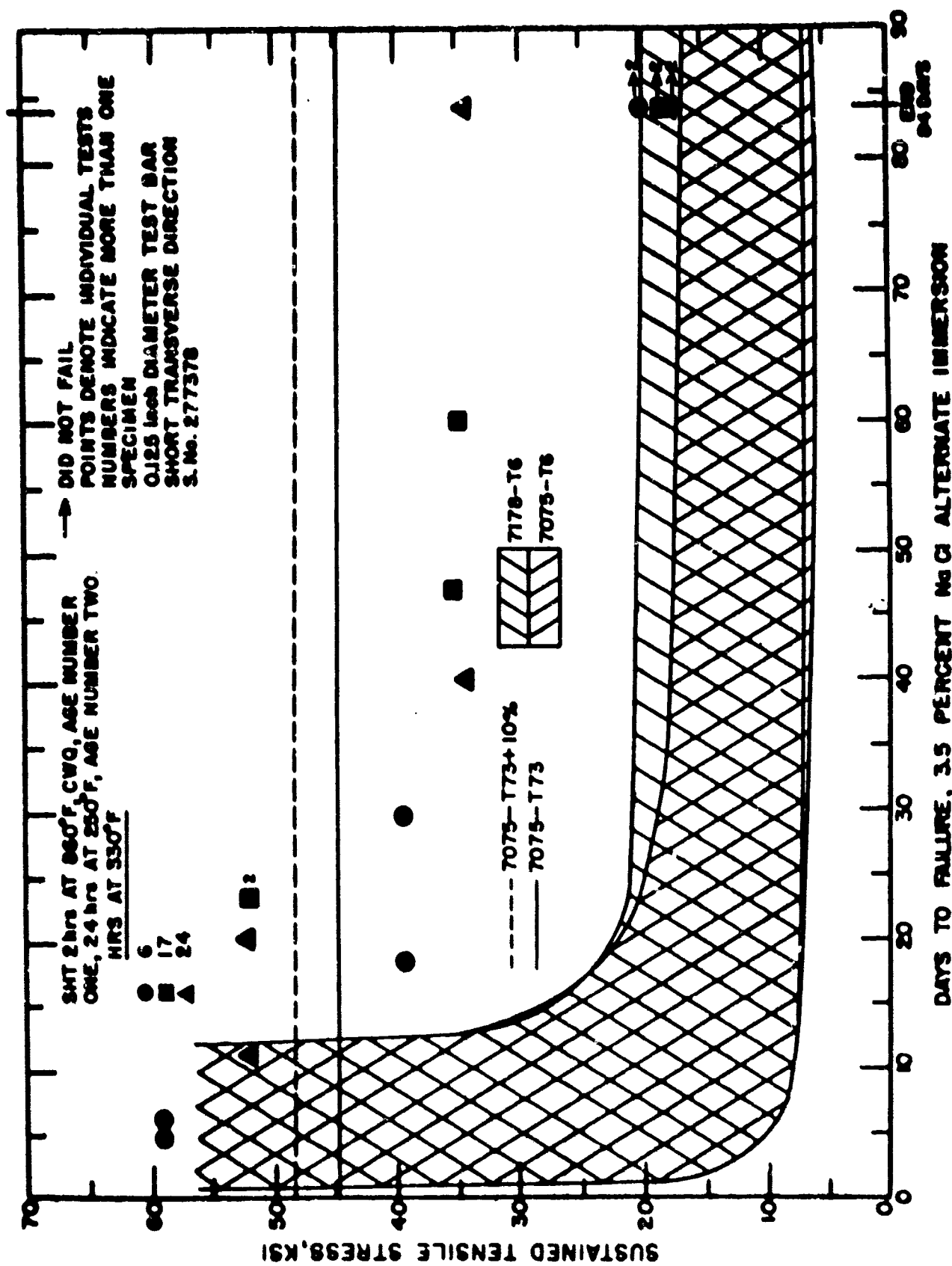


FIGURE 17  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 6

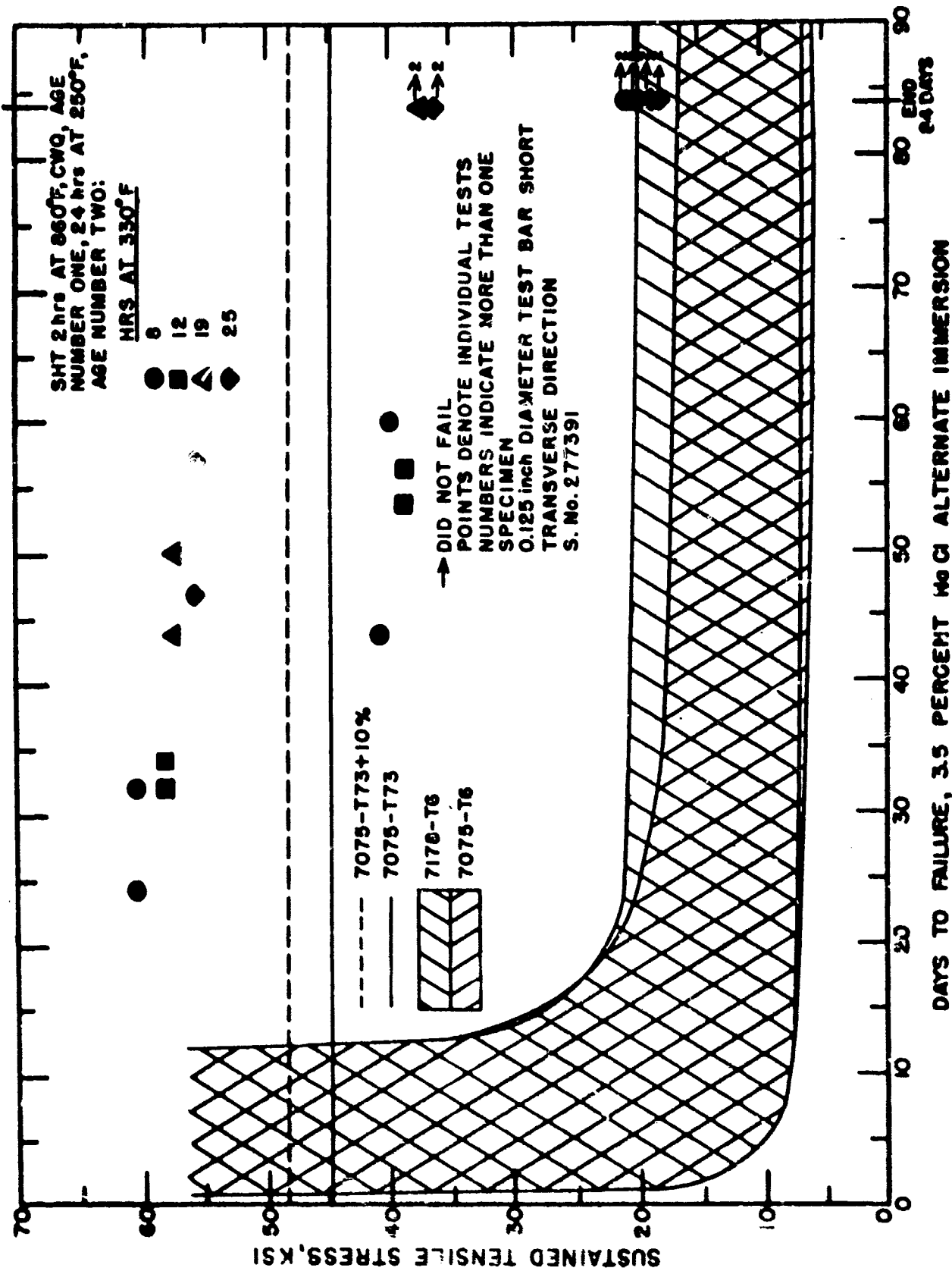


FIGURE 18  
 COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
 CORROSION CRACKING FOR ALLOY 19

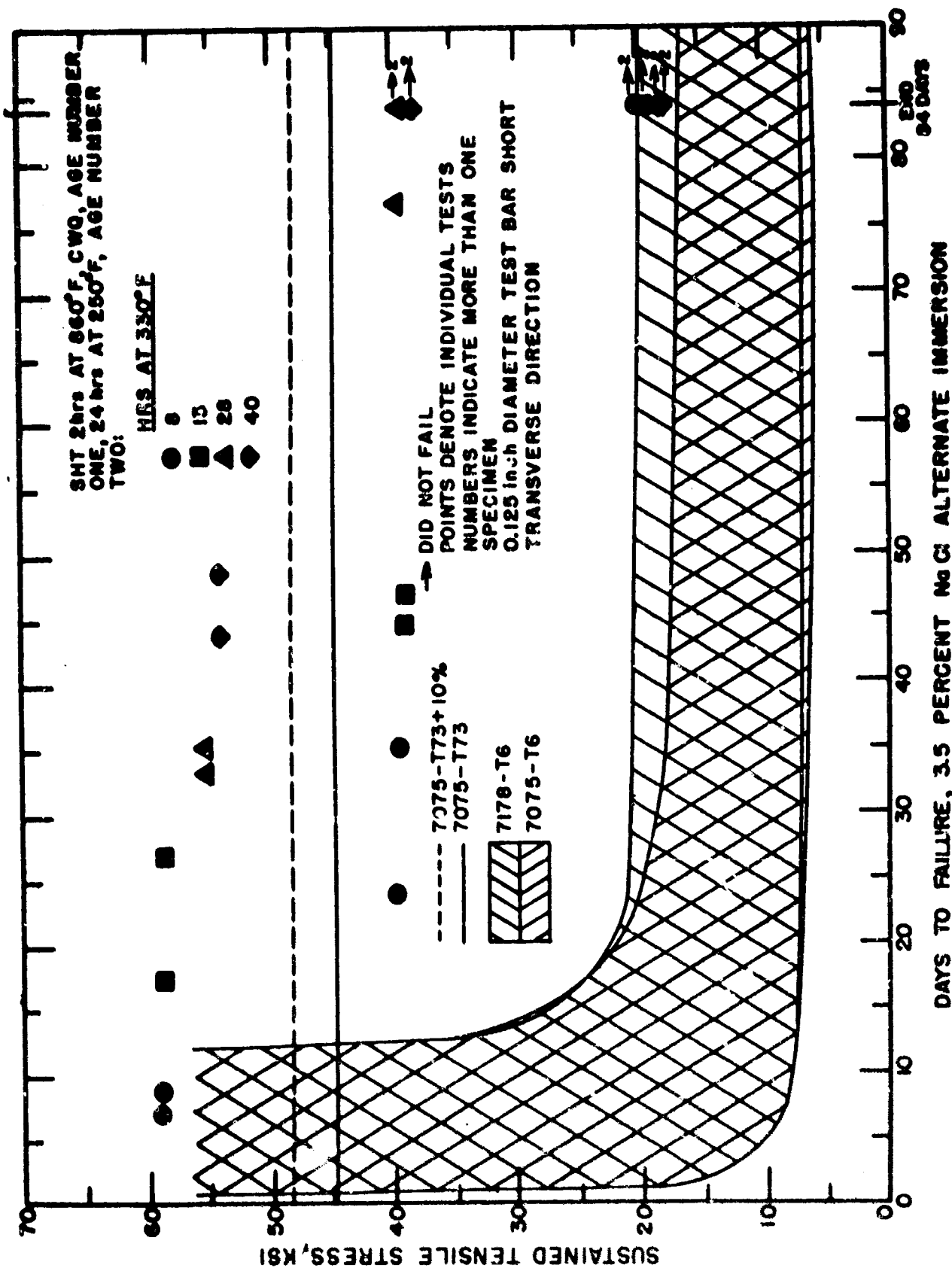


FIGURE 19  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 20

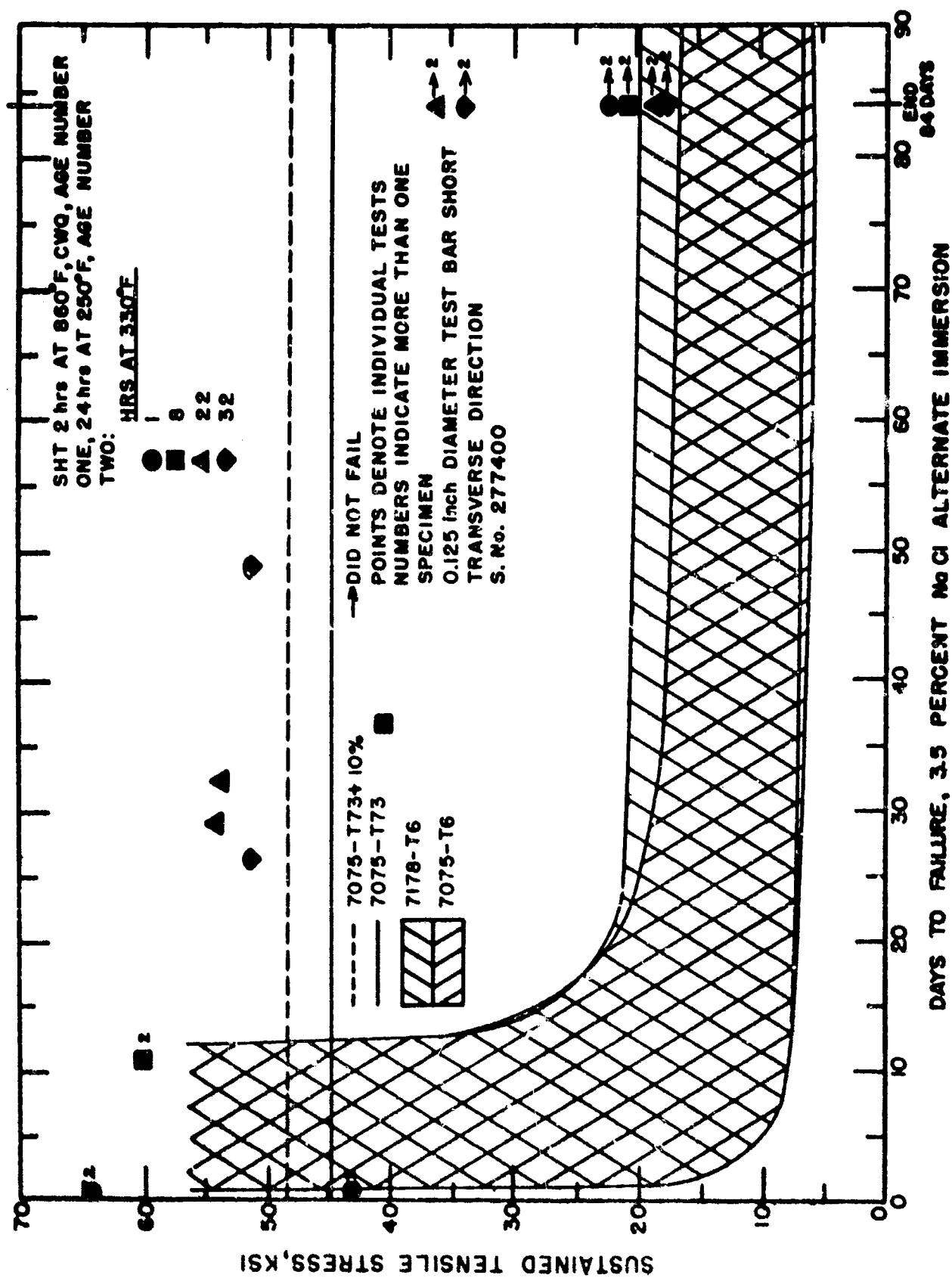


FIGURE 20  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 28

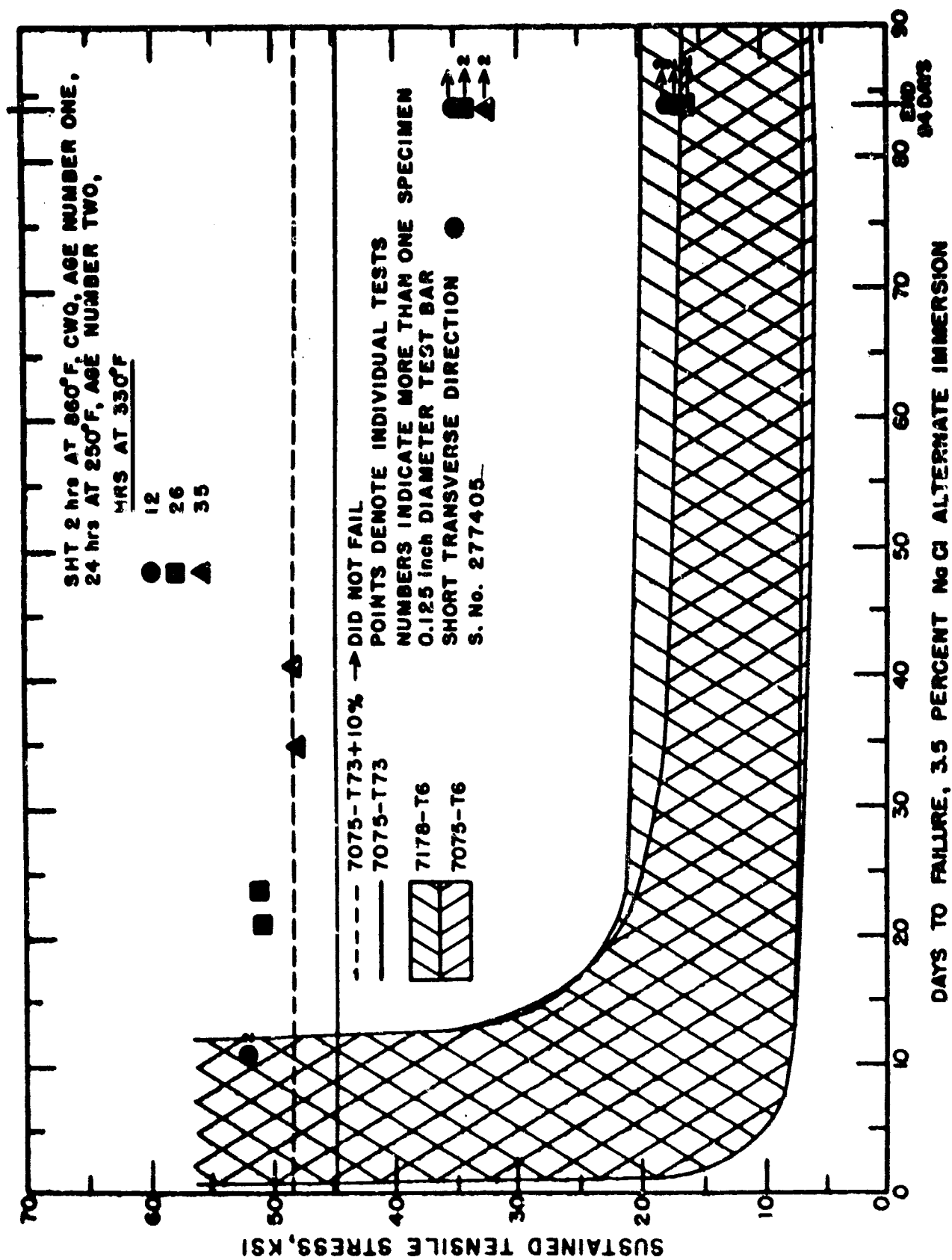


FIGURE 21  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 33

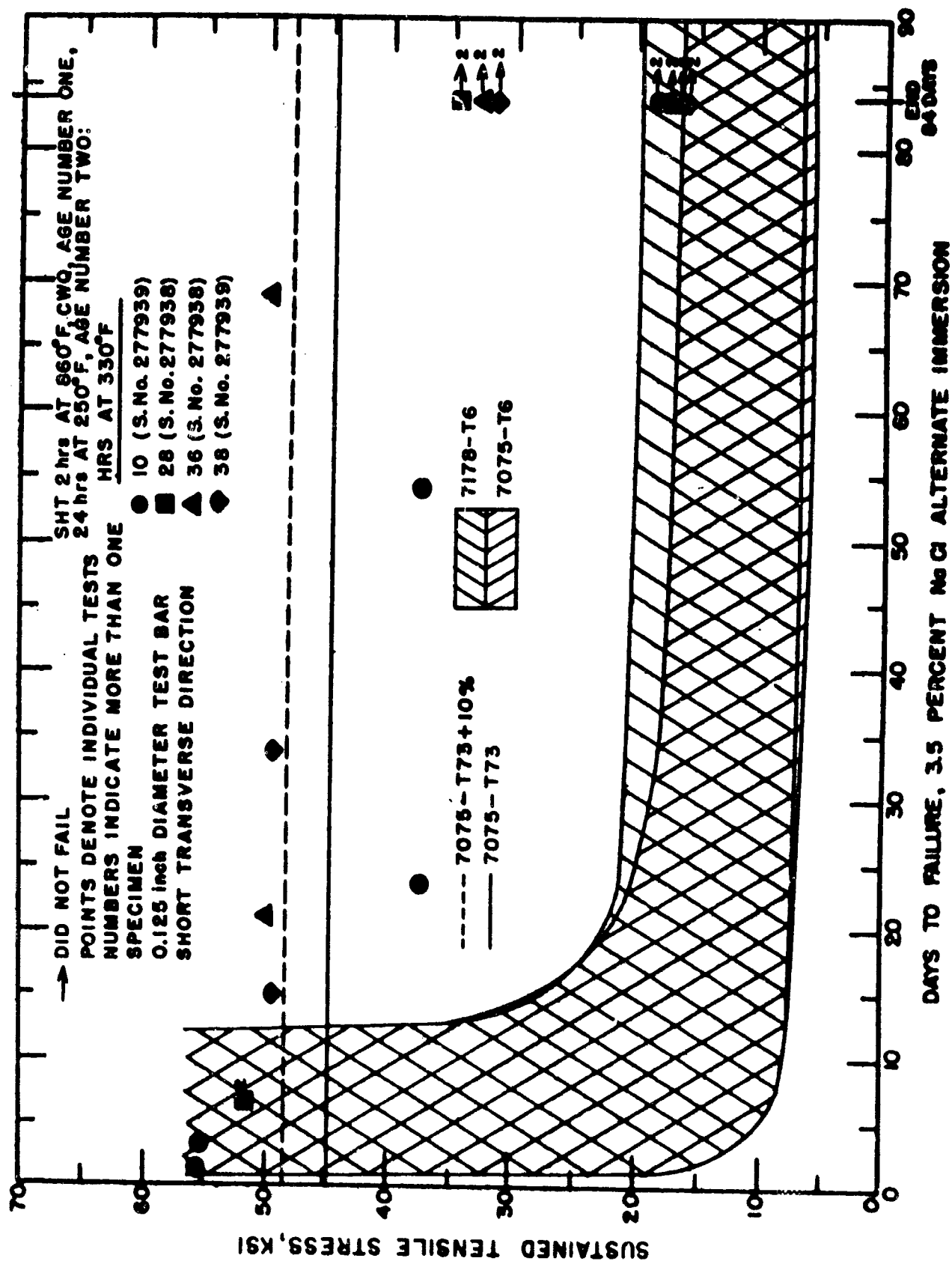


FIGURE 22

COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 36

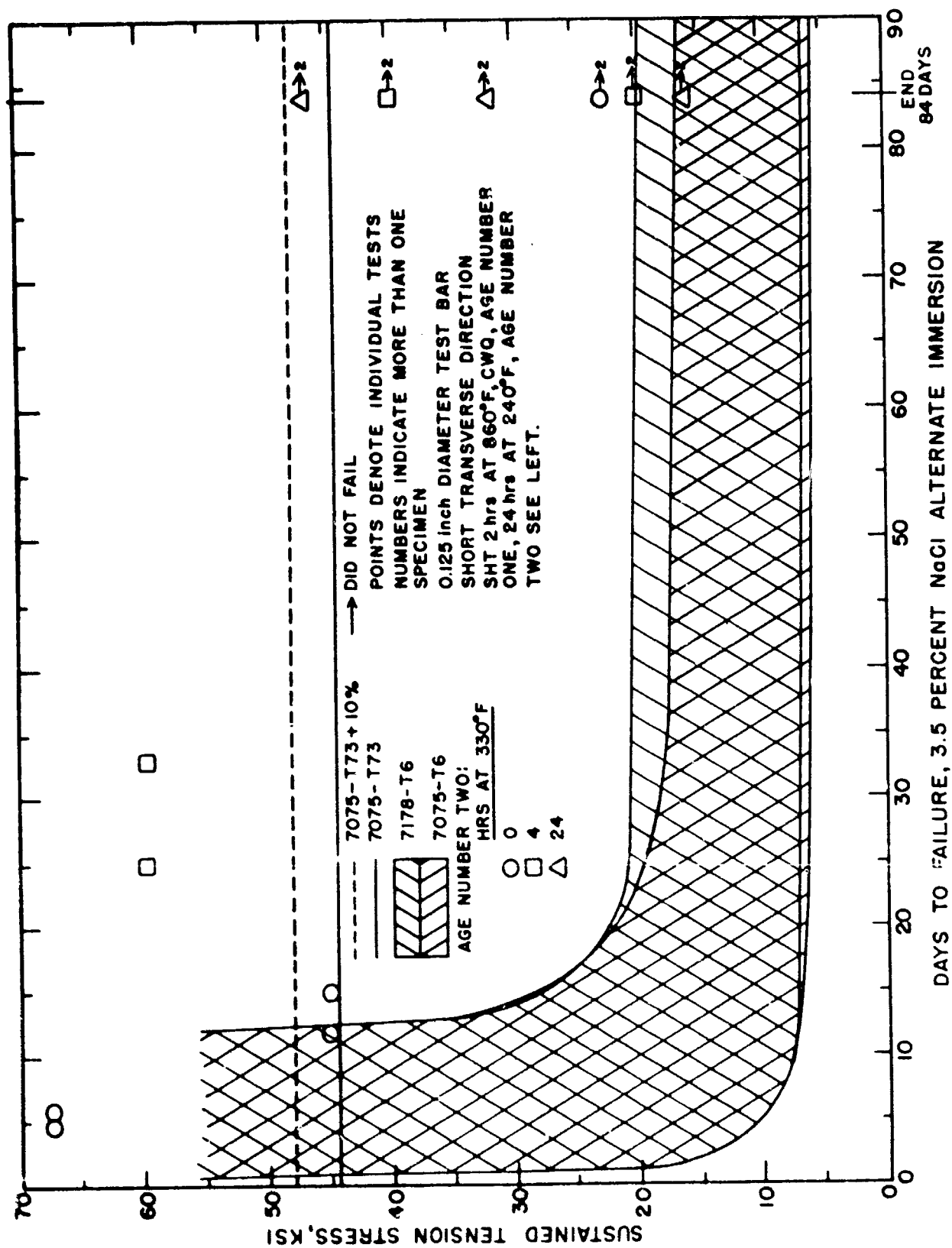


FIGURE 23

COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 52.



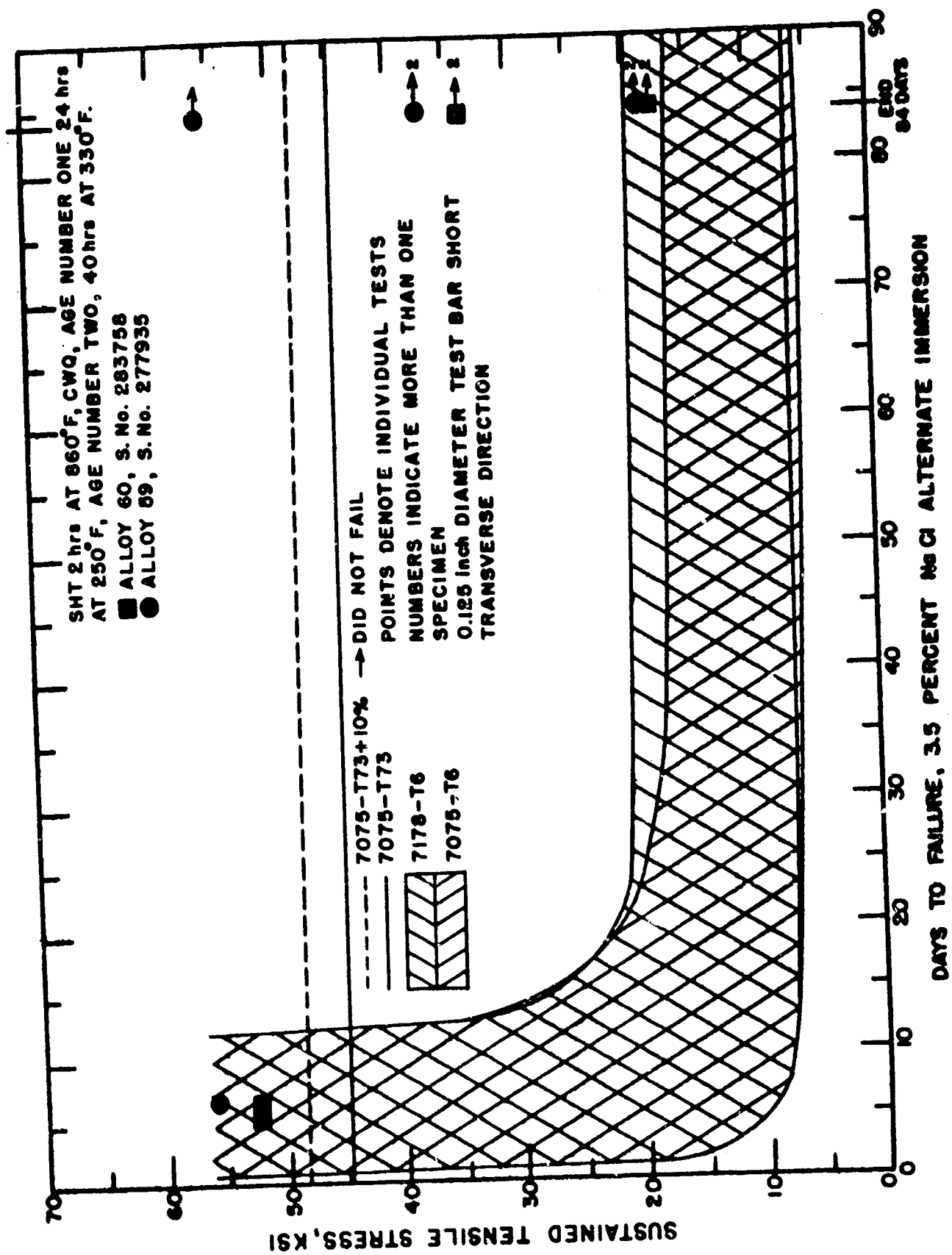


FIGURE 24  
 COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
 CORROSION CRACKING FOR ALLOYS 59 AND 60

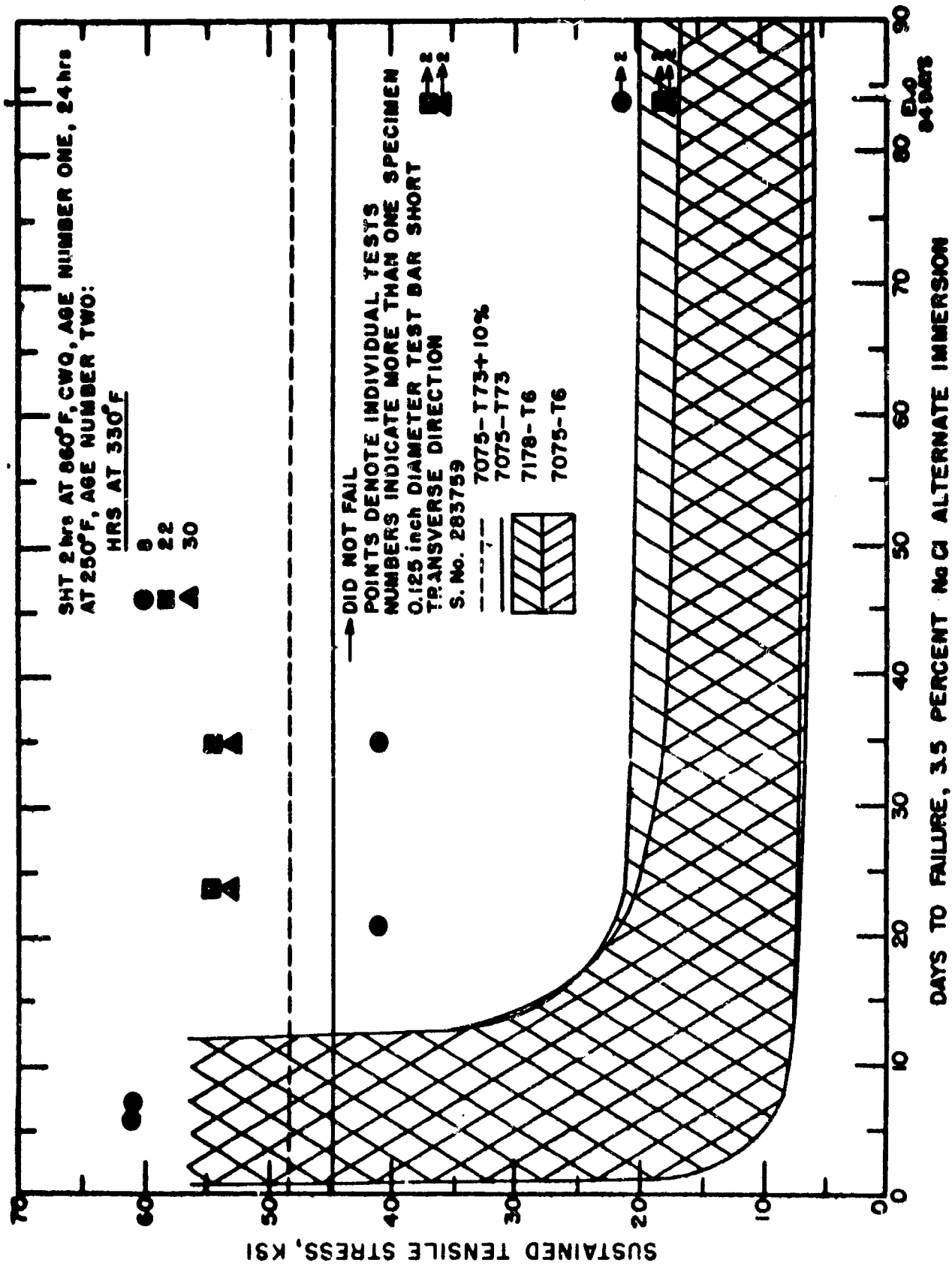


FIGURE 25  
COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
CORROSION CRACKING FOR ALLOY 61

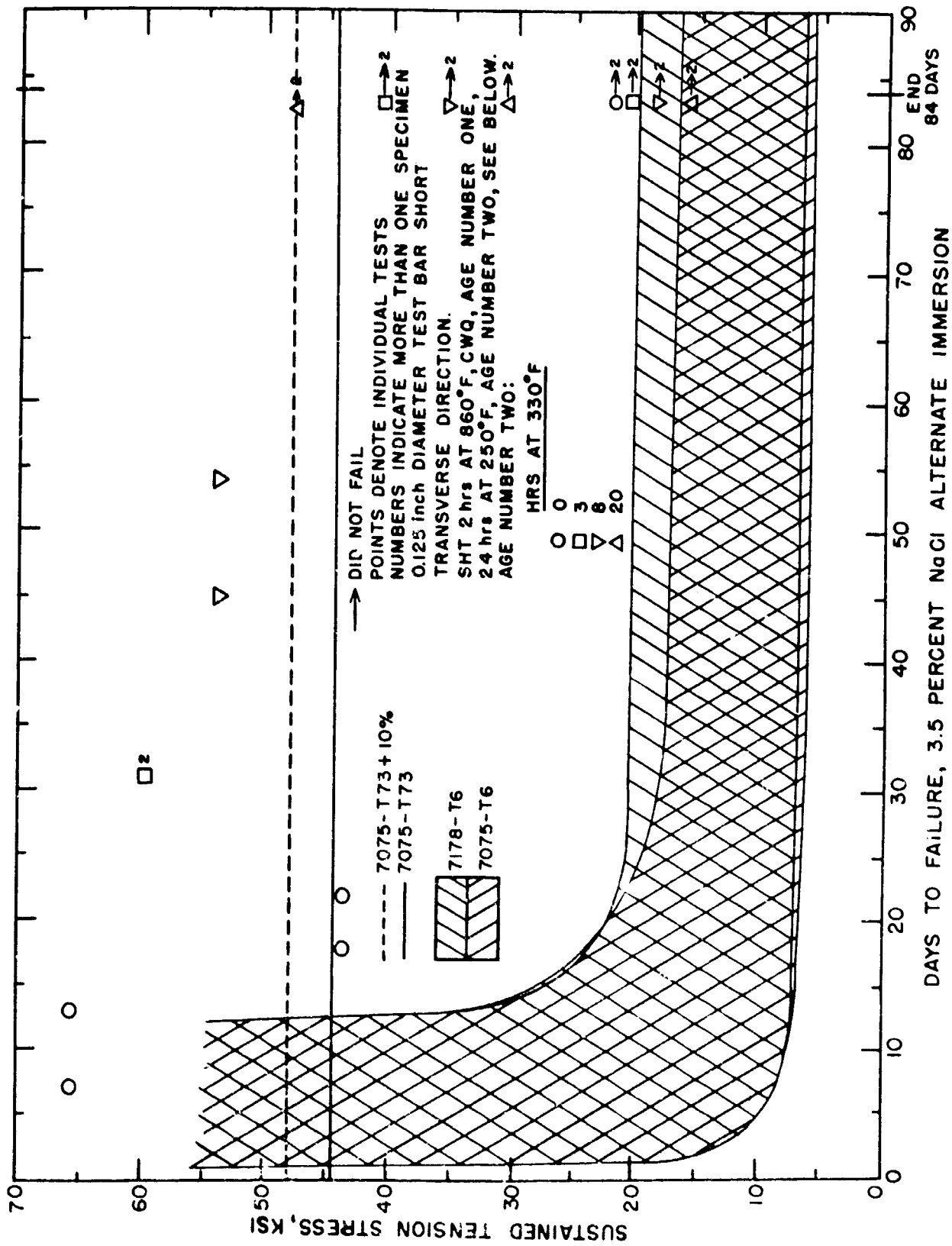


FIGURE 26  
 COMPARISON OF AGE NUMBER TWO ON RELATIVE RESISTANCE TO STRESS  
 CORROSION CRACKING FOR ALLOY 71.

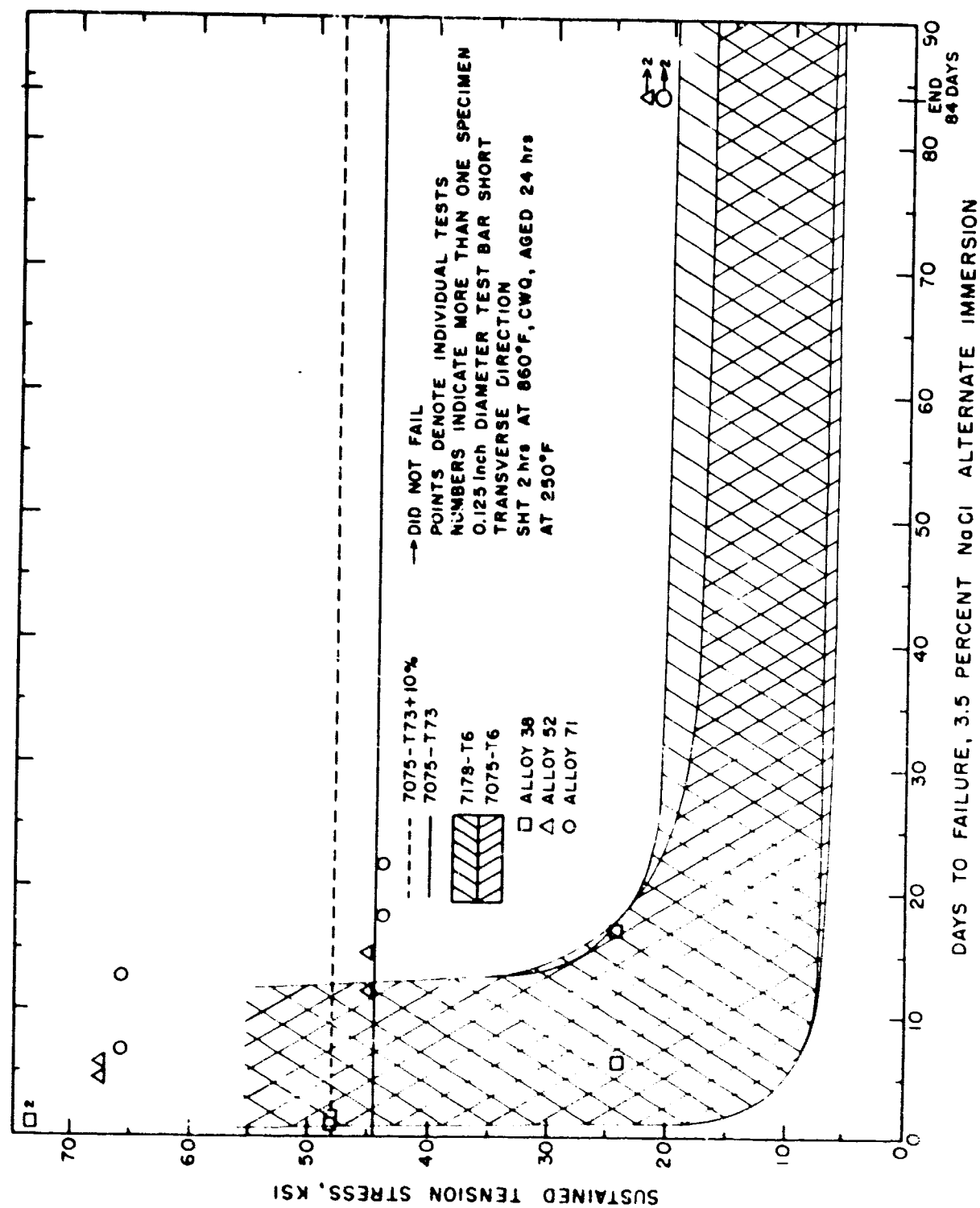


FIGURE 27

RELATIVE RESISTANCE OF APM ALLOYS 38, 52 AND 71 TO STRESS CORROSION CRACKING  
FOR TENSILE STRENGTHS OF APPROXIMATELY 10% GREATER THAN 7178-T6

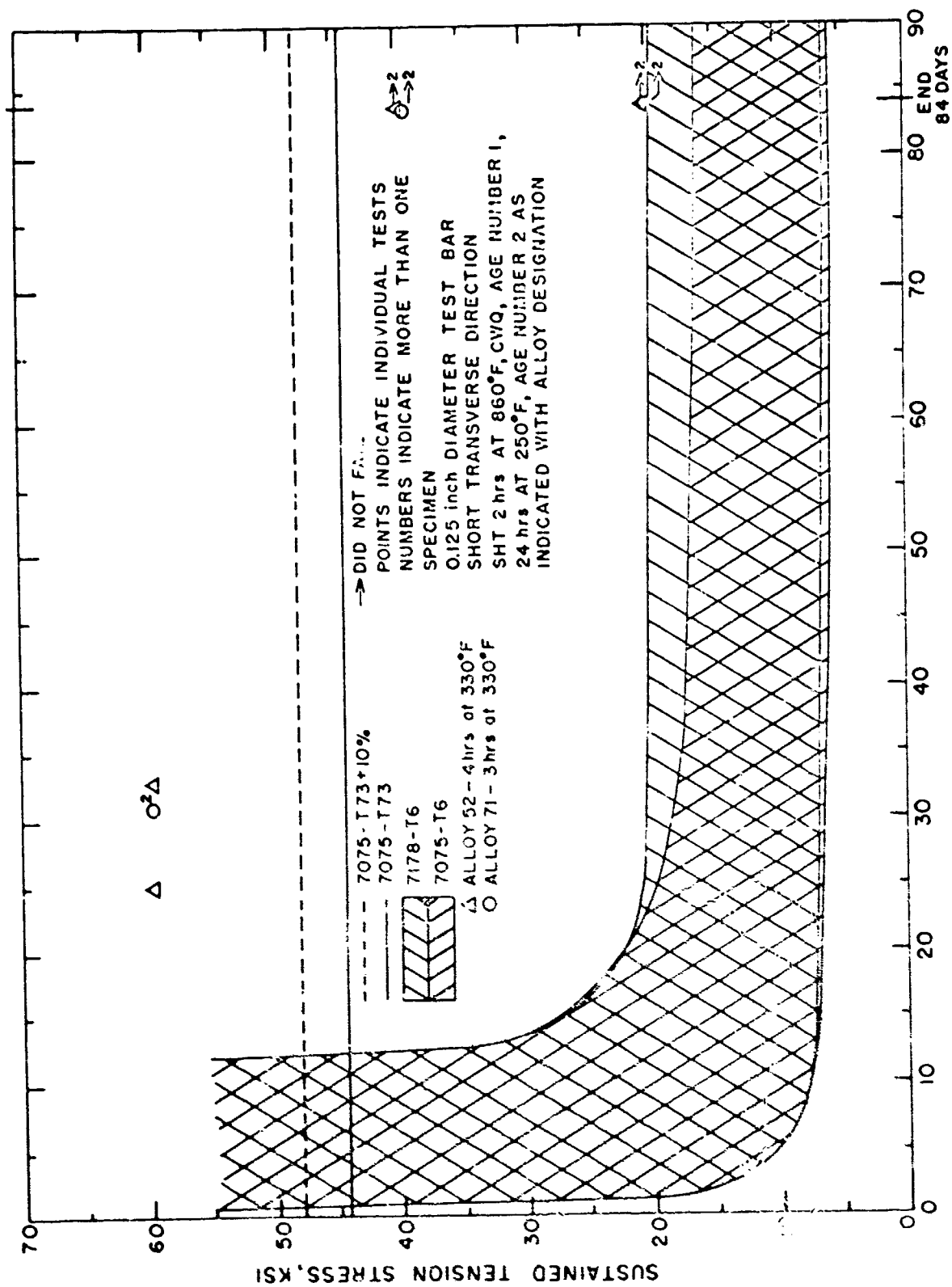


FIGURE 28  
DAYS TO FAILURE, 3.5 PERCENT NaCl ALTERNATE IMMERSION

FIGURE 28

RELATIVE RESISTANCE OF ALUM ALLOYS 52 AND 71 TO STRESS CORROSION CRACKING  
FOR TENSILE STRENGTHS OF APPROXIMATELY 10% GREATER THAN 7075-T6

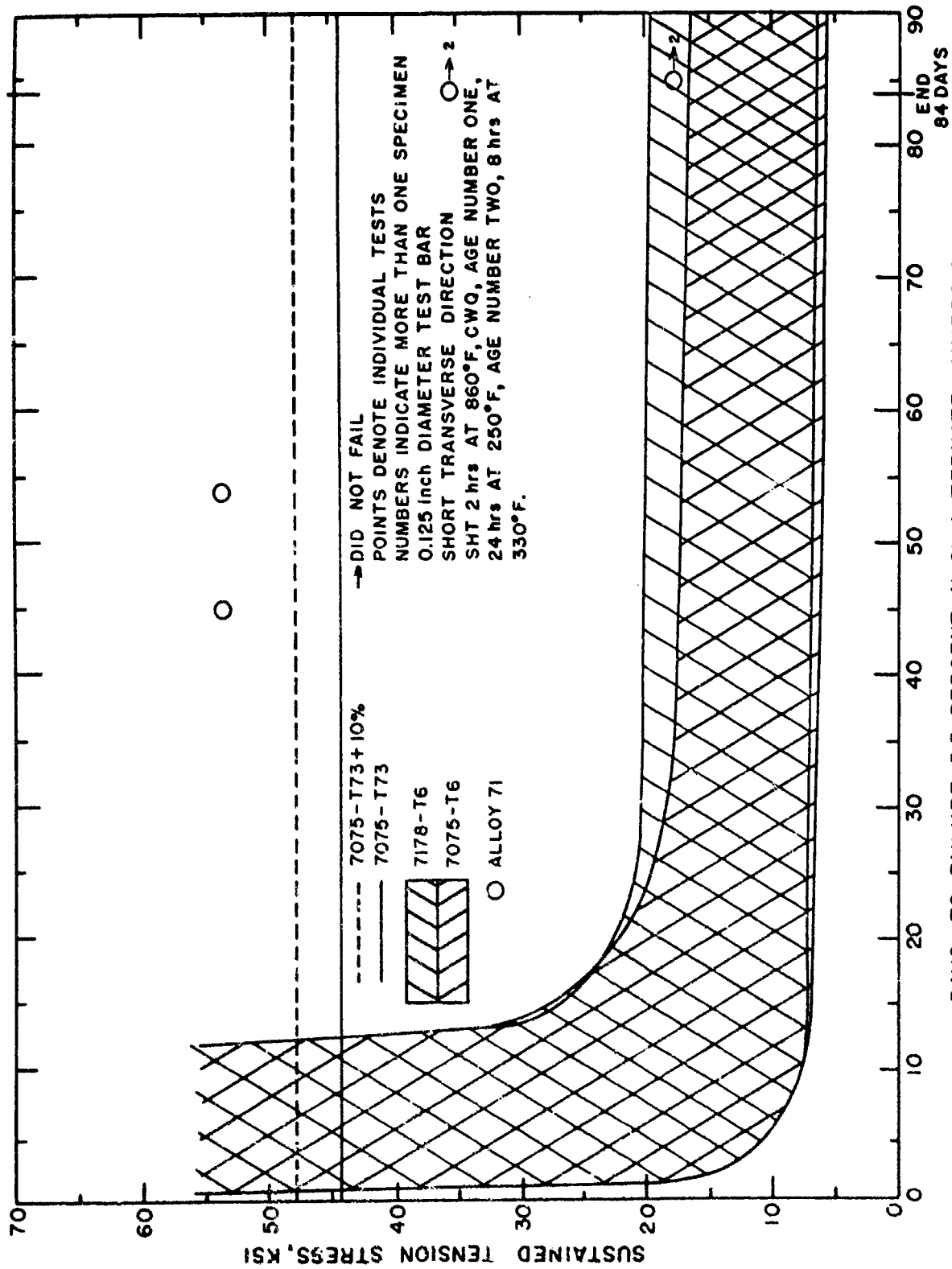
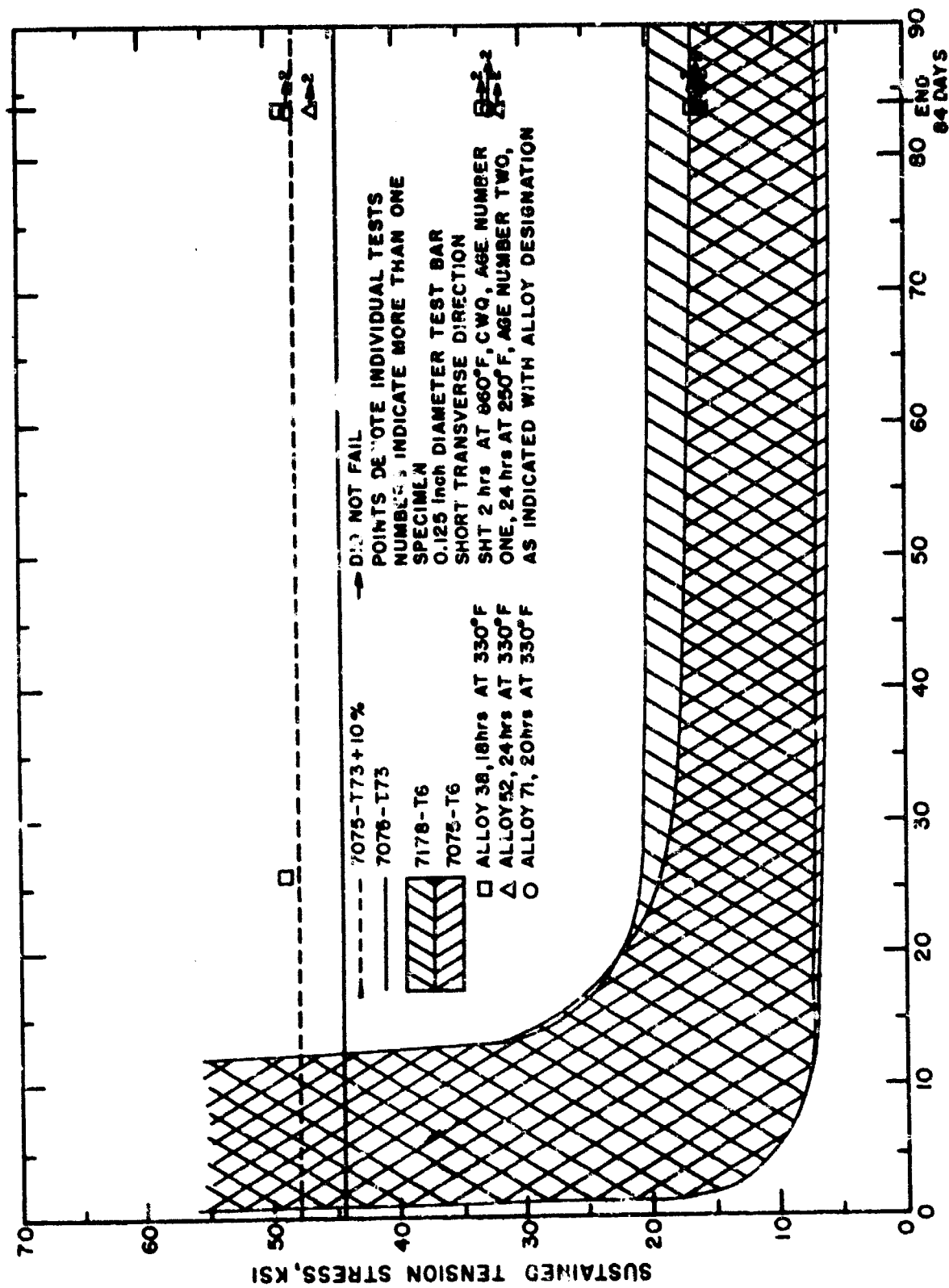


FIGURE 29

RELATIVE RESISTANCE OF APM ALLOY 71 TO STRESS CORROSION CRACKING  
 FOR TENSILE STRENGTHS GREATER THAN 7075-T73 BUT LESS THAN 7075-T6



DAYS TO FAILURE, 3.5 PERCENT NaCl ALTERNATE IMMERSION

FIGURE 30

RELATIVE RESISTANCE OF APM ALLOYS 38, 52 AND 71 TO STRESS CORROSION CRACKING  
FOR TENSILE STRENGTHS OF APPROXIMATELY 10% GREATER THAN 7075-T7351

**FIGURE 31**  
**COMPARISON OF RELATIVE RESISTANCE TO STRESS CORROSION CRACKING**  
**OF APM ALLOY 34, 79, 87 AND 90 EXTRUSIONS**



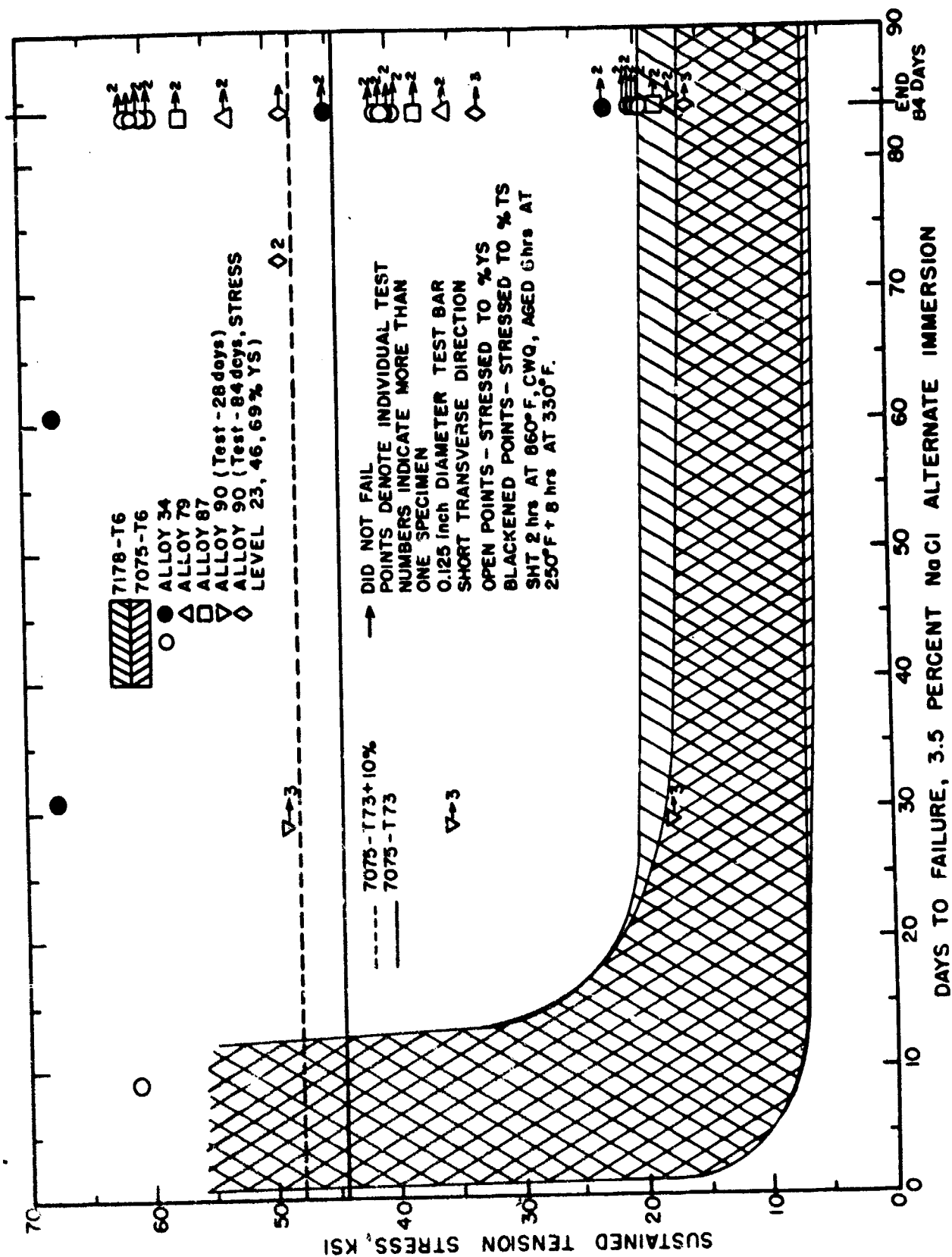


FIGURE 32  
COMPARISON OF RELATIVE RESISTANCE TO STRESS CORROSION CRACKING  
OF APM ALLOY 34, 79, 87 AND 90 EXTRUSIONS

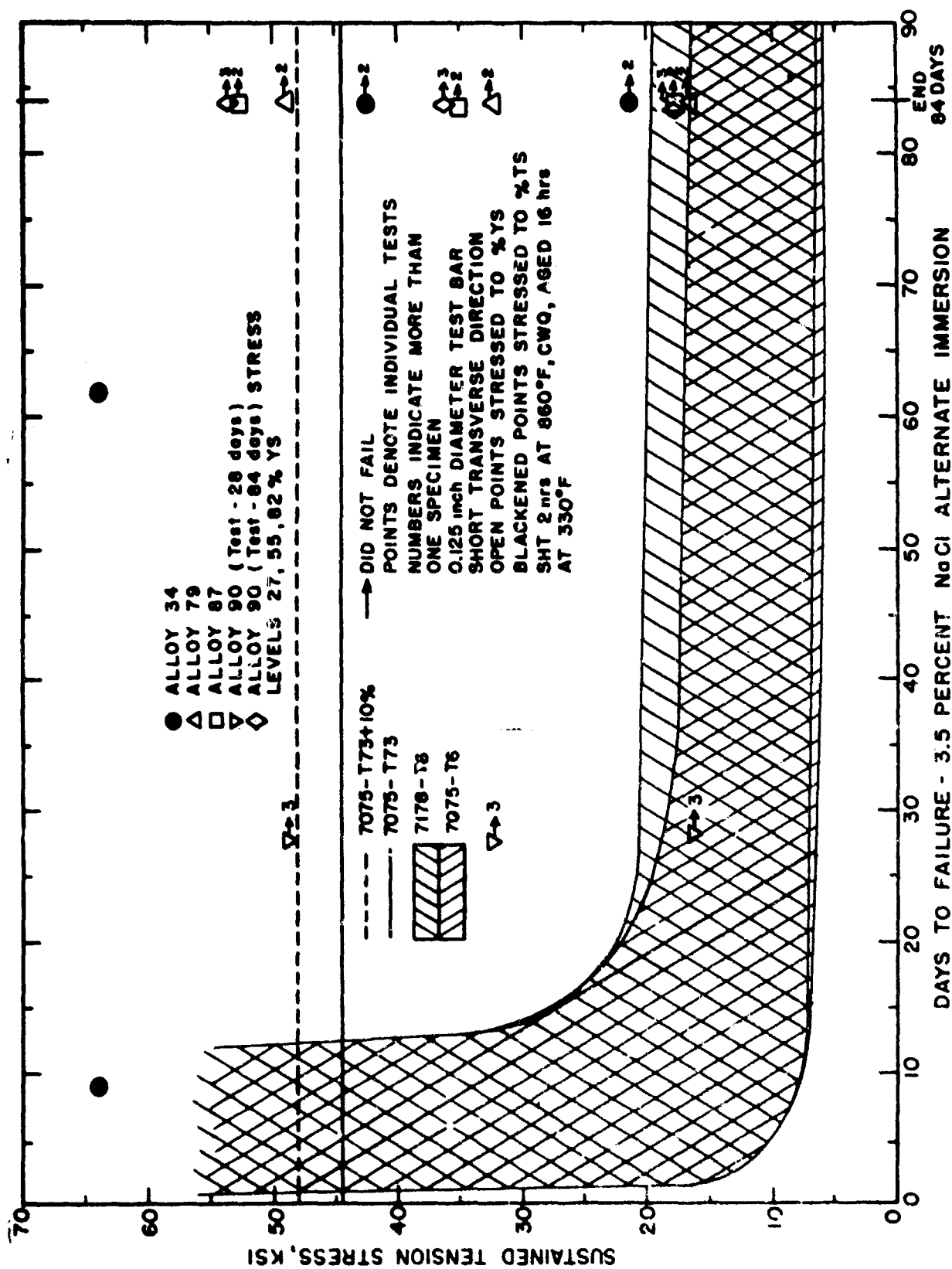


FIGURE 33  
COMPARISON OF RELATIVE RESISTANCE TO STRESS CORROSION CRACKING  
OF APM ALLOY 34, 79, 87 AND 90 EXTRUSIONS

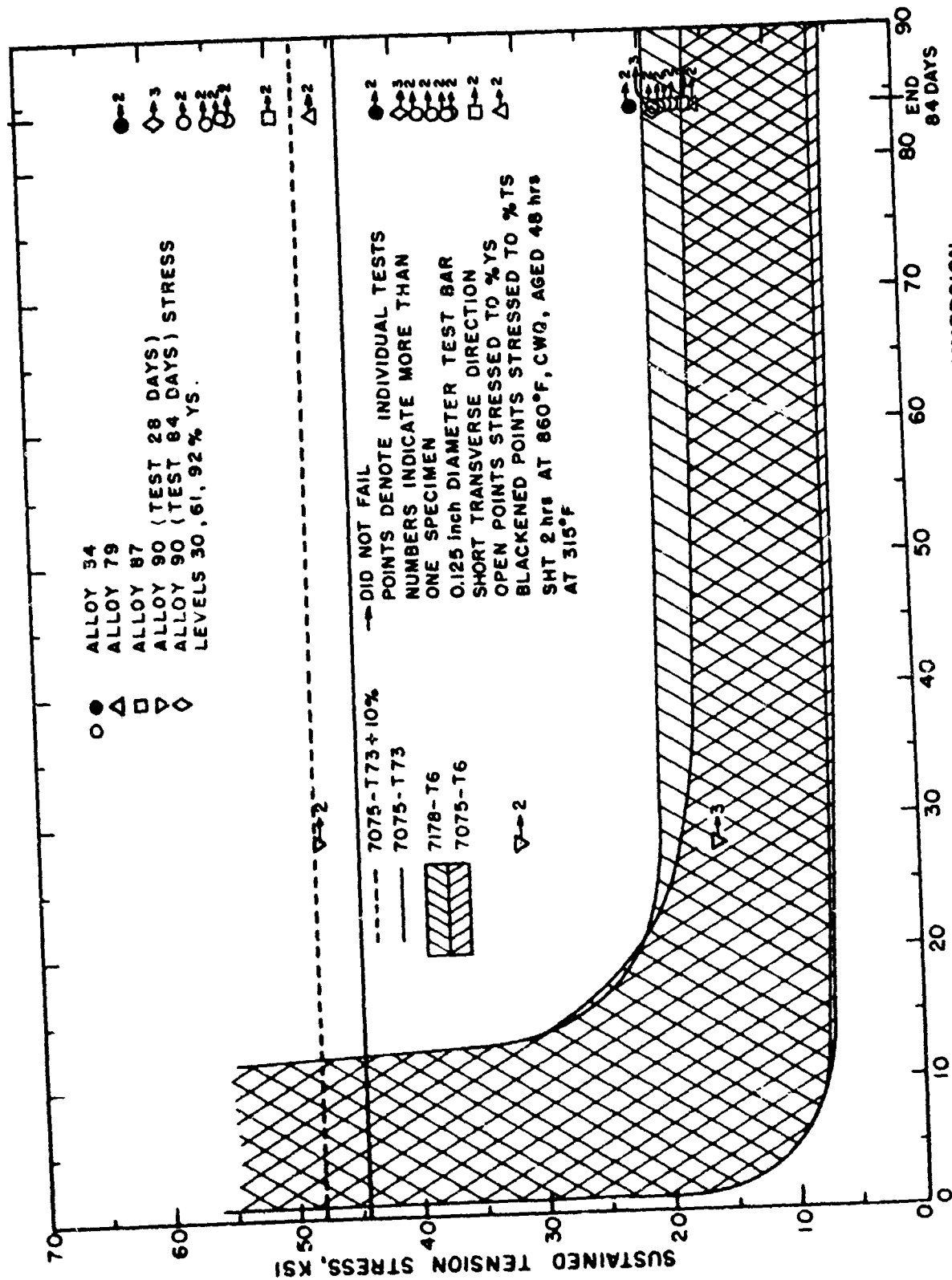
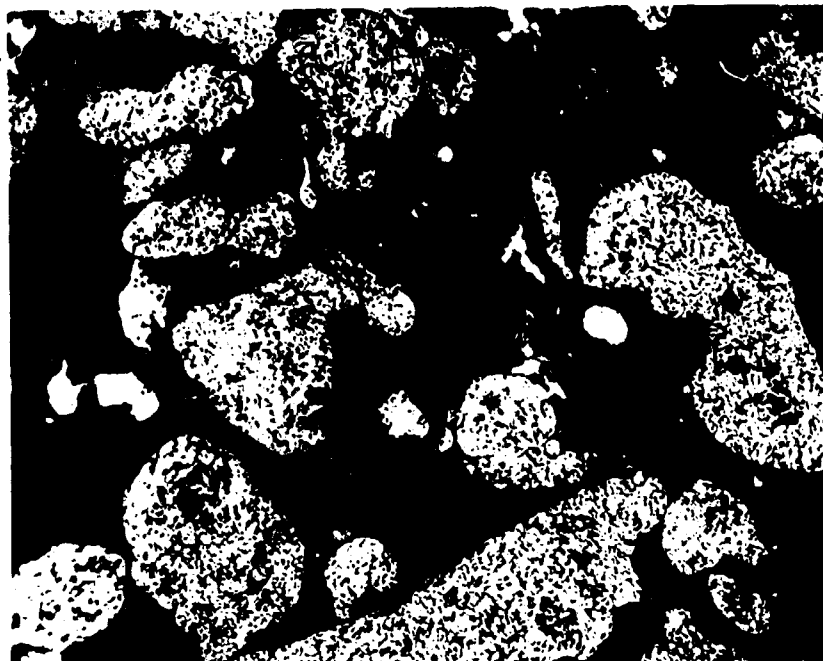


FIGURE 34  
COMPARISON OF RELATIVE RESISTANCE TO STRESS CORROSION CRACKING  
OF APM ALLOY 34, 79, 87 AND 90 EXTRUSIONS



POWDER  
KELLER'S ETCH

500X

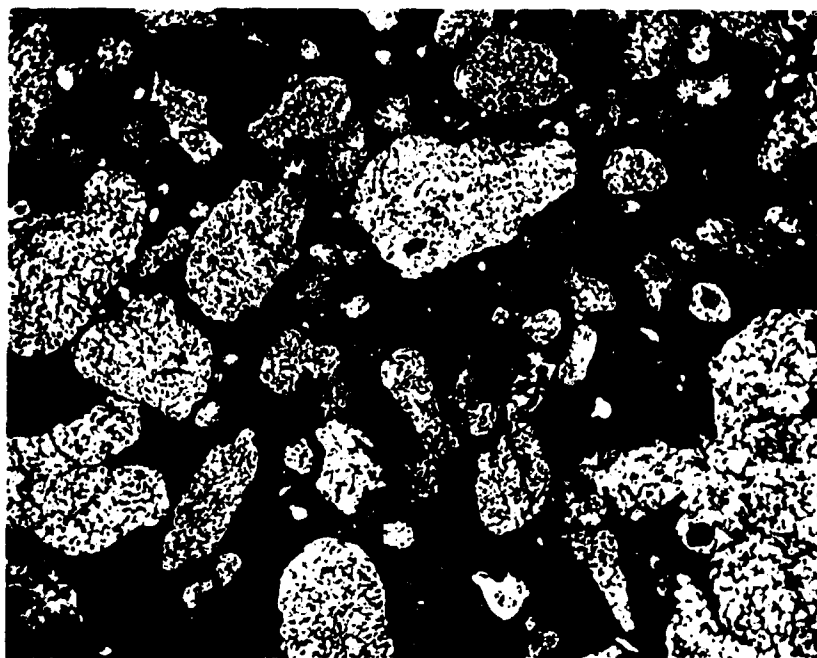


EXTRUSION IN -T6 TEMPER  
KELLER'S ETCH

500X  
LONG

131452J  
1352972

FIGURE 35: ALLOY 38. Al-8.4 Zn - 3.6 Mg - 1.0 Mn - 0.8 Fe - 2.8 Ni



POWDER  
KELLER'S ETCH

500X

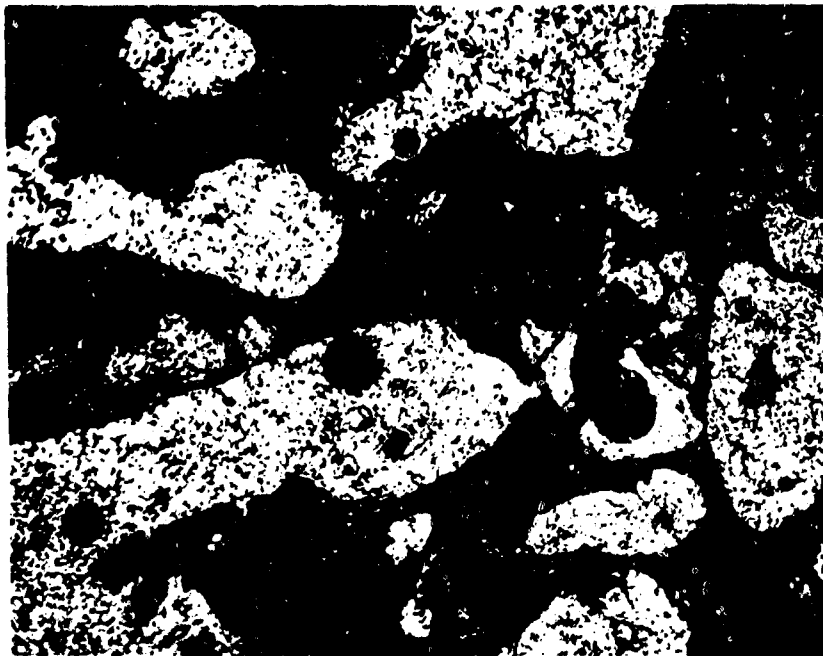


EXTRUSION IN -T6 TEMPER  
KELLER'S ETCH

500X  
LONG

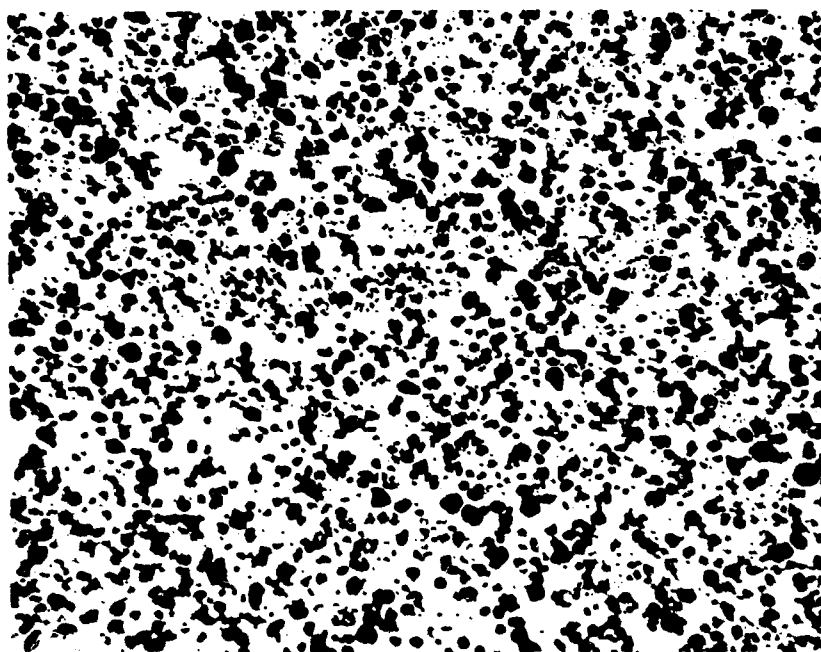
135201J  
135202J

FIGURE 36: ALLOY 15. Al-11.4 Zn-3.4 Mg-1.5 Cu-.5 Cr-.5 Ti-.5 V-.6 Zr



POWDER  
KELLER'S ETCH

500X

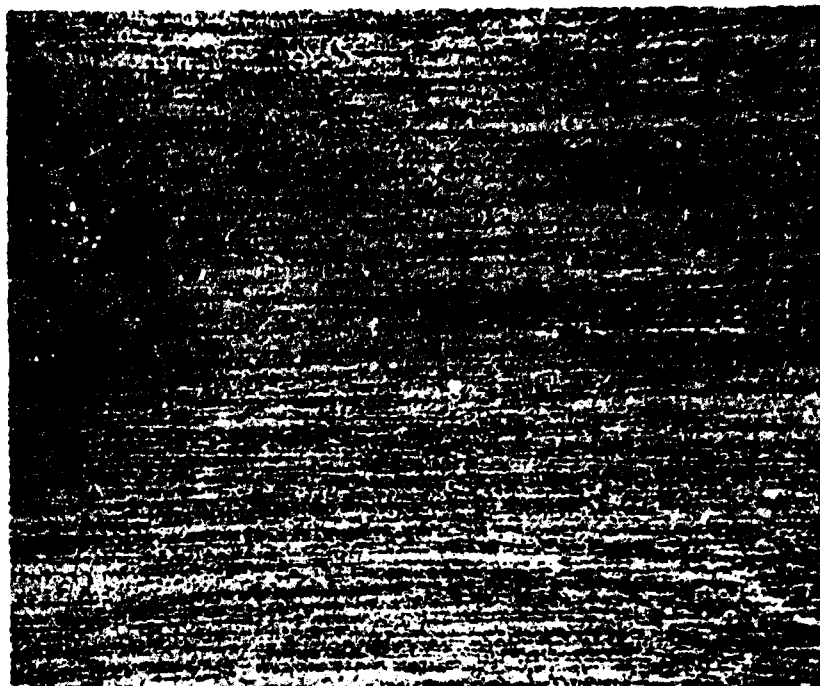


EXTRUSION IN -T6 TEMPER  
KELLER'S ETCH

500X  
LONG

FIGURE 37: ALLOY 22. Al - 7.9 Zn - 8.4 Mg - 14.9 Cu

1352967  
1352968

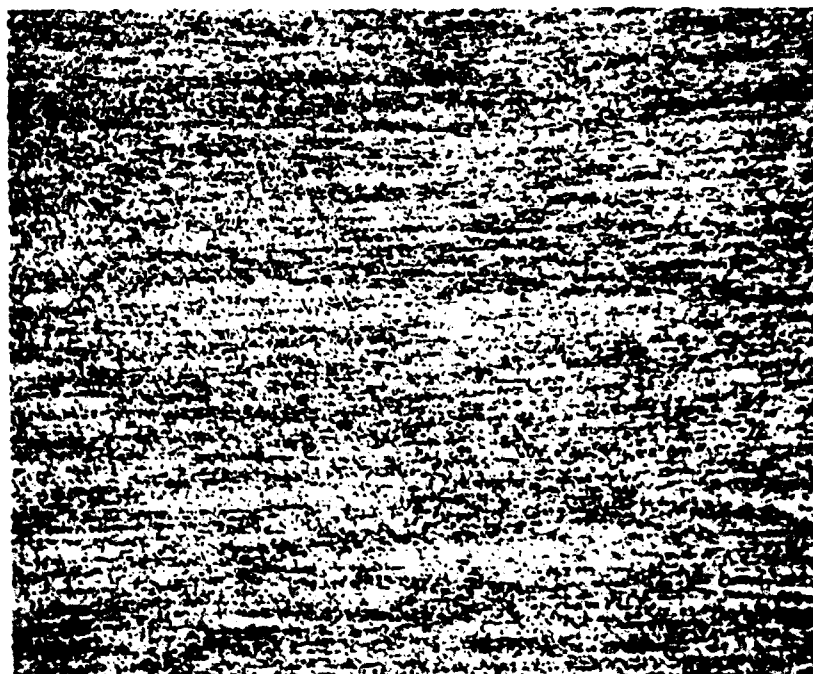


Extrusion in -T6 Temper  
Keller's Etch

500X

Figure 38

Alloy 71. Al - 9.3 Zn - 3.6 Mg - 0.5 Cu - 0.7 Co.



Extrusion in -T6 + 20 hrs at 330°F Temper  
Keller's Etch

500X

Figure 39

Alloy 71. Al - 9.3 Zn - 3.6 Mg - 0.5 Cu - 0.7 Co.



S-283269

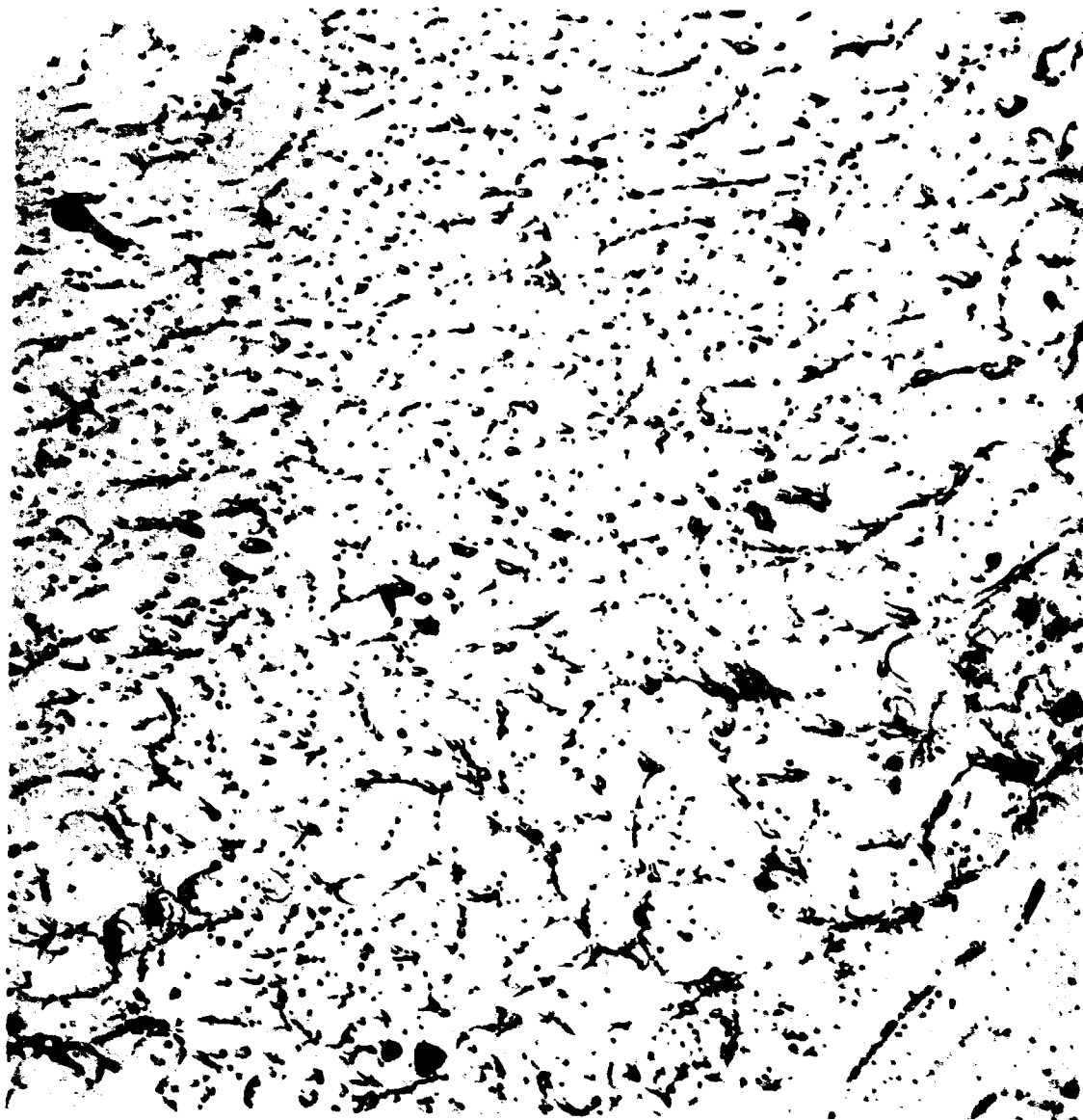
Carbon Replica

20,000X

Figure 40

Electron micrograph showing the structure  
of Alloy 34 powder.





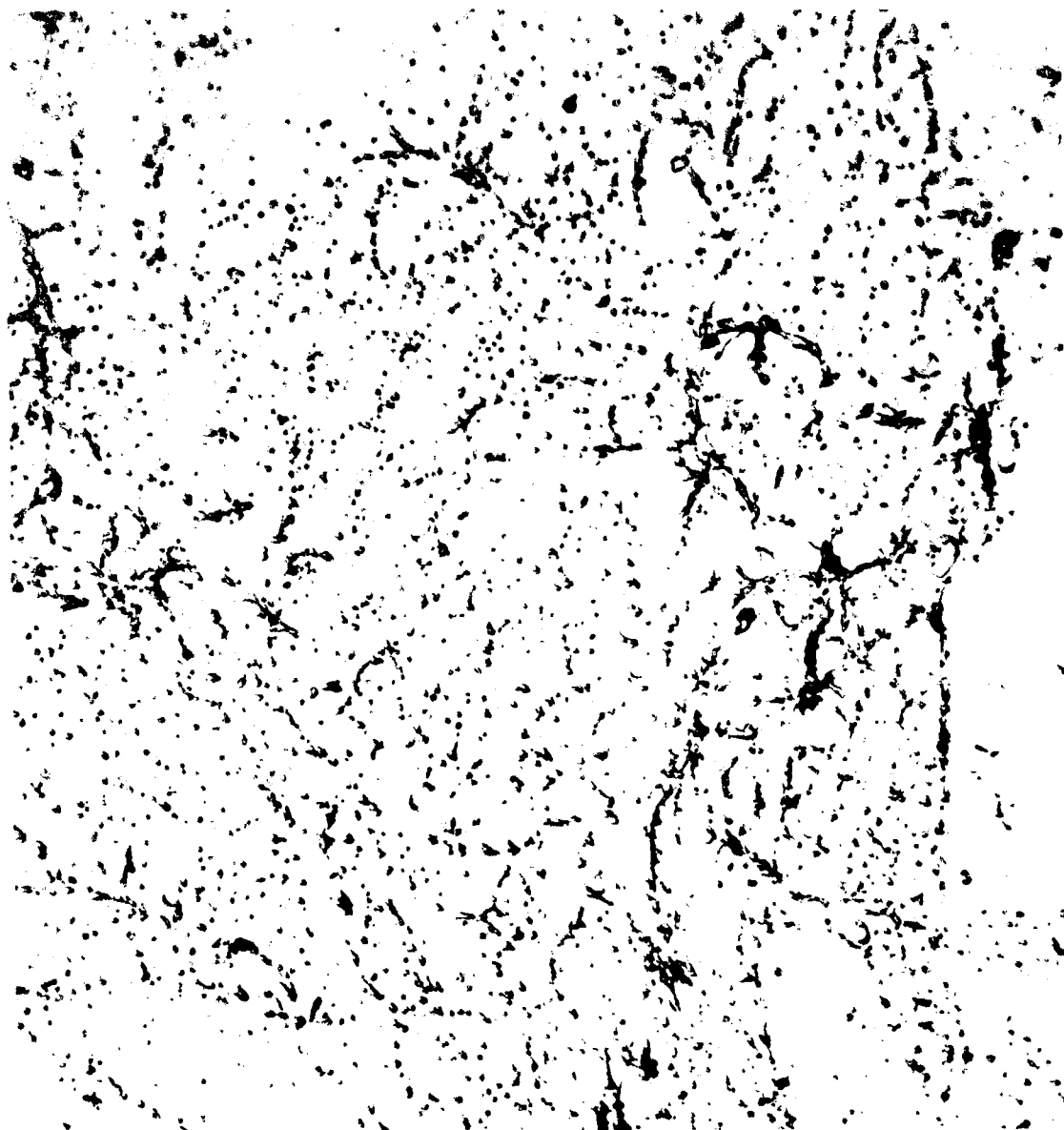
S-307598

Carbon Replica

20,000X

Figure 41

Electron micrograph showing the structure  
of Alloy 87 powder.



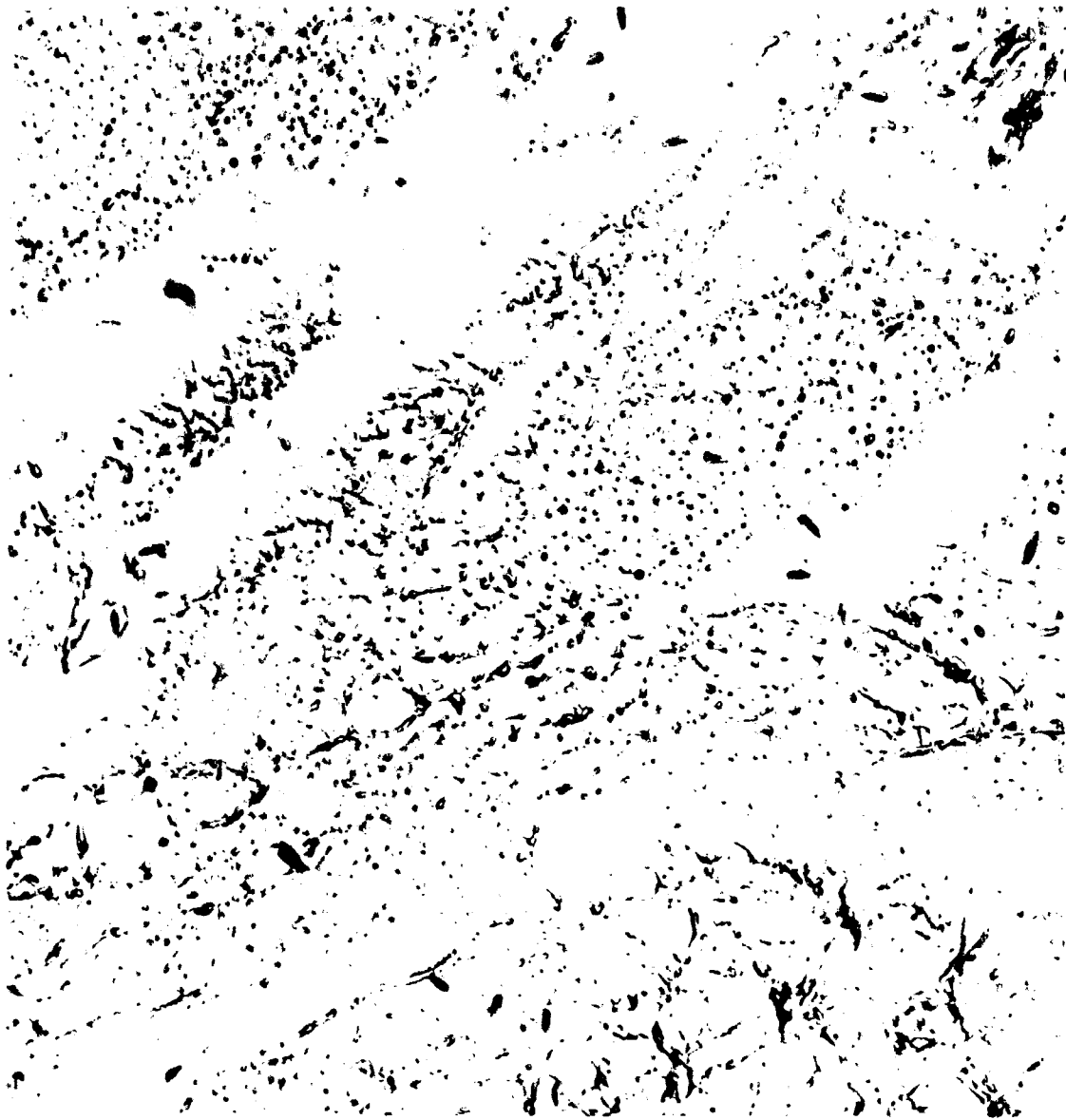
S-283274

Carbon Replica

20,000X

Figure 42

Electron micrograph showing the structure  
of Alloy 52 powder.



S-293303

Carbon Replica

20,000X

Electron micrograph showing the structure of Alloy 71 powder. Particles in this powder were slightly less uniformly dispersed than in the other powders.

Figure 43



S-283441-A

Oxide Replica

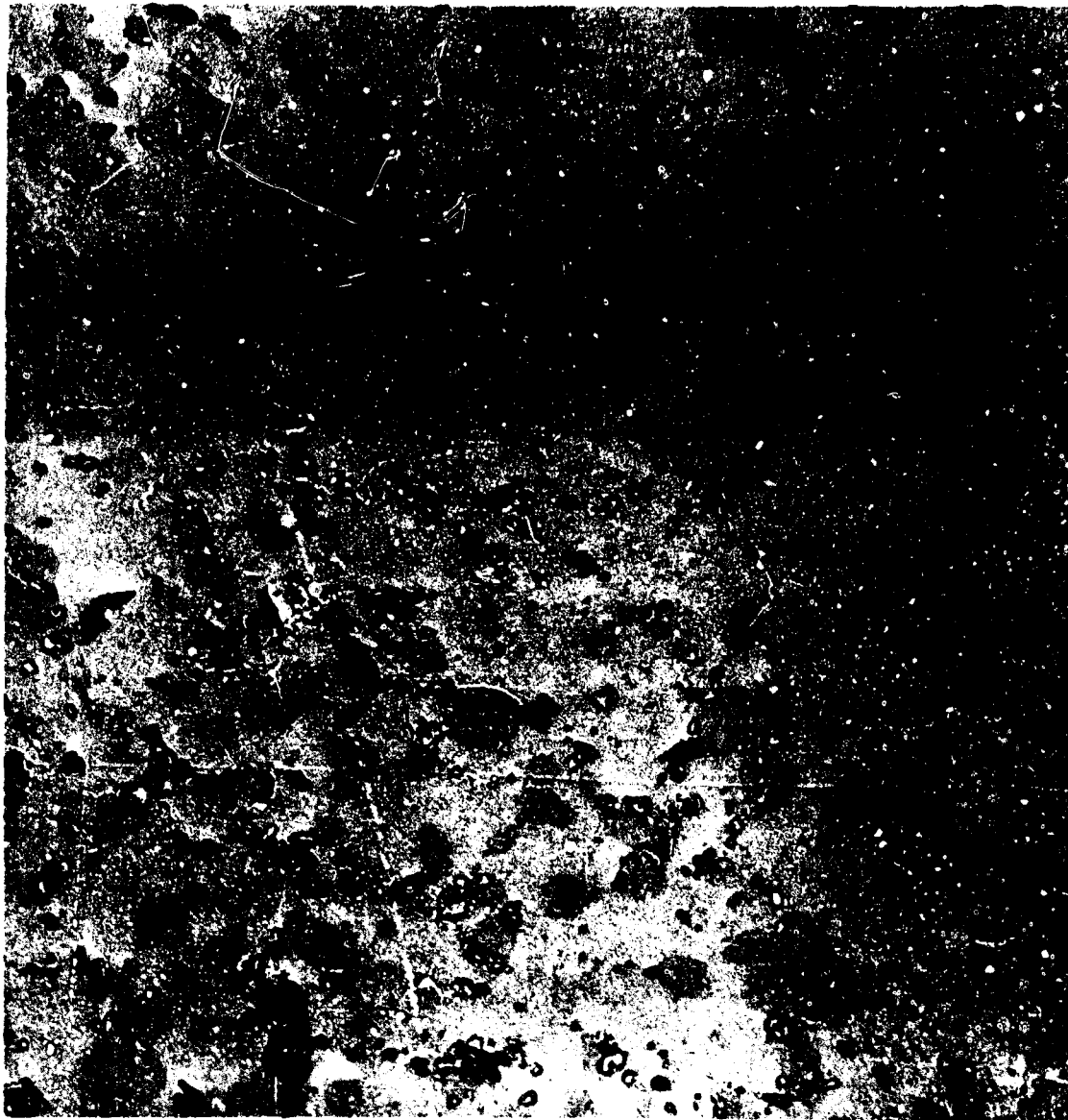
10,000X

Alloy 34-T6

Powder Extrusion

Figure 44

Shows the structure (longitudinal section)  
of the extrusion made from Alloy 34 powder.  
Extrusion has been S.H.T. 2 hrs at 860°F,  
C.W.Q., Aged 24 hrs at 250°F.



S-283441-B

Oxide Replica

10,000X

Alloy 34

Prolonged Age

Powder Extrusion

Shows the structure (longitudinal section) of the extrusion made from Alloy 34 powder. Extrusion has been S.H.T. 2 hrs at 860°F, C.W.Q., Aged 48 hrs at 315°F.

Figure 45



S-293196-A

Oxide Replica

10,000X

Alloy 87-T6

Powder Extrusion

Shows the structure (longitudinal section) of the extrusion formed from Alloy 87 powder. The extrusion had been S.H.T. 2 hrs at 860°F, C.W.Q. Aged 24 hrs at 250°F.

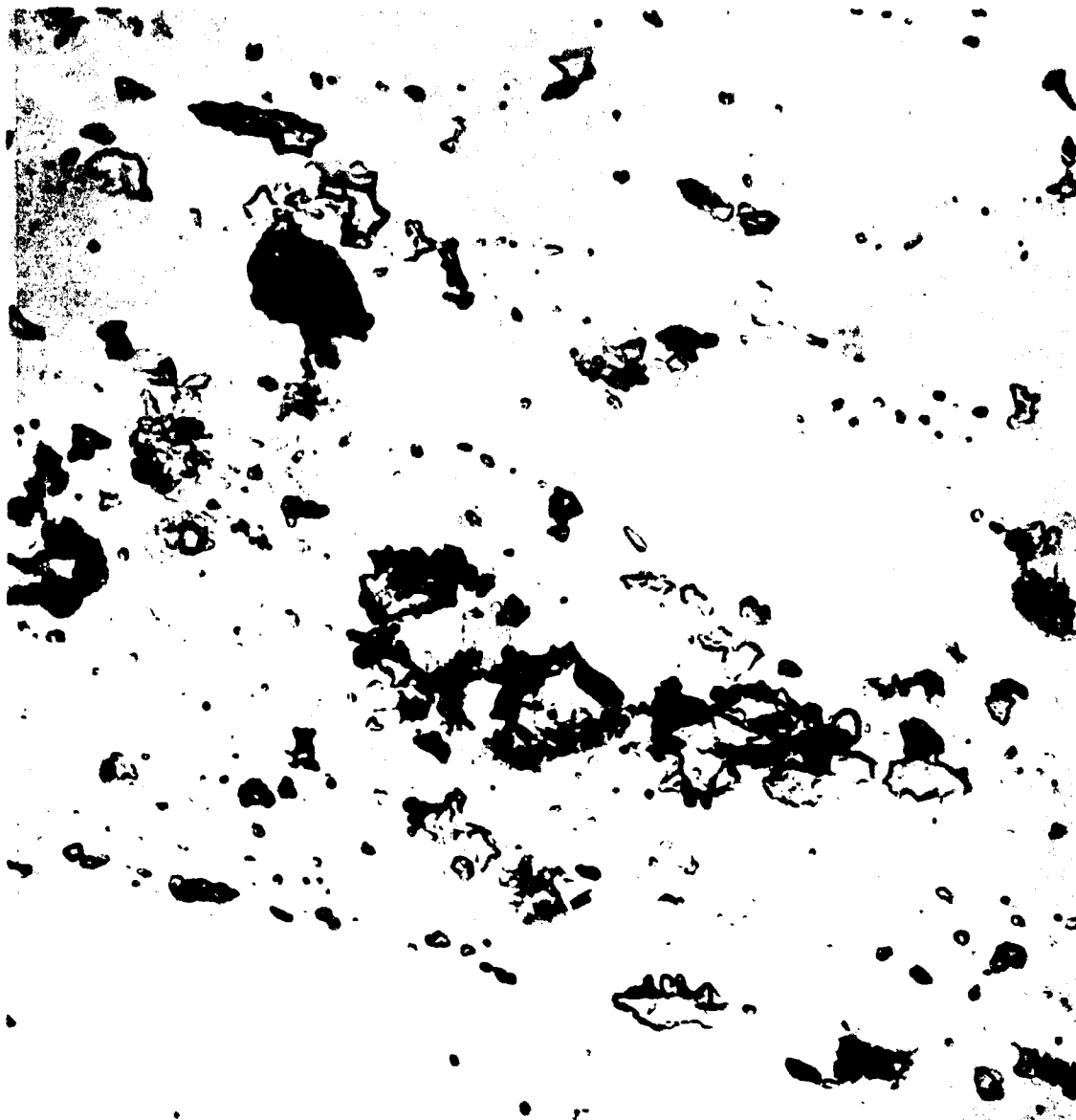
Figure 46



S-293196-B      Oxide Replica      10,000X  
Alloy 87      Prolonged Age      Powder Extrusion

Figure 47

Shows the structure of Alloy 87 powder extrusion that had been S.H.T. 2 hrs at 860°F, C.W.Q. and Aged 48 hrs at 315°F. Probable identification of dispersoid and precipitate phases is noted.



S-283490-A

Oxide Replica

10,000X

Alloy 52-T6

Powder Extrusion

Figure 48

Shows the structure of the extruded section made from the Alloy 52 powder that was S.H.T. 2 hrs at 860°F, C.W.Q. and Aged 24 hrs at 250°F. Note the large amount of (Mg-Zn) precipitate relative to other sections of similar temper.





S-283490-C

Oxide Replica

10,000X

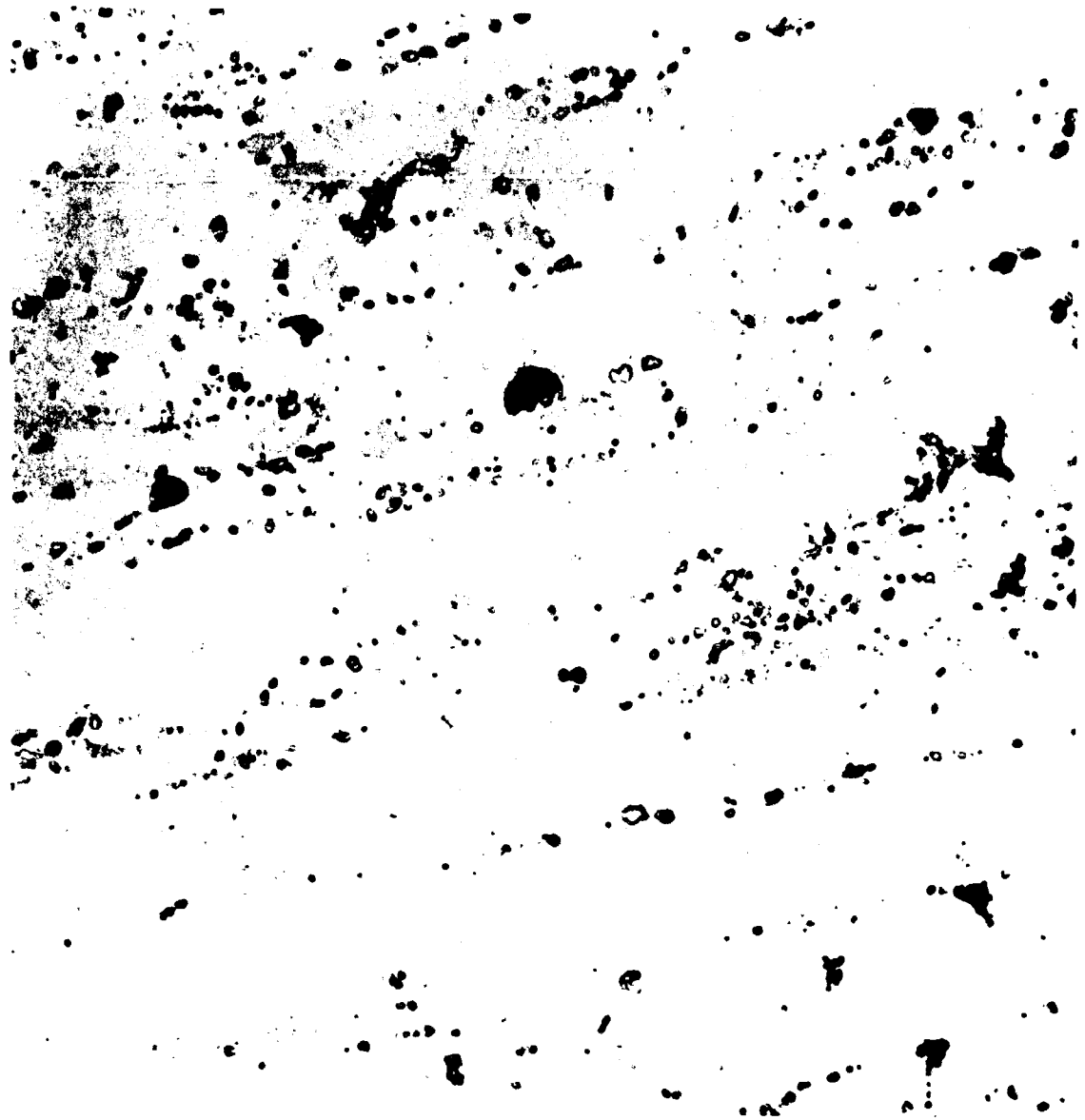
Alloy 52

Step Aged

Powder Extrusion

Figure 49

Shows the structure of Alloy 52 powder extrusion that had been step-aged 24 hrs at 250° + 20 hrs at 330°F after solution heat treatment.



S-293387-A

Oxide Replica

10,000X

Alloy 71-T6

Powder Extrusion

Figure 50

Shows the structure (longitudinal section) of the extrusion made from Alloy 71 powder. The extrusion was S.H.T. 2 hrs at 860°F, C.W.Q. and Aged 24 hrs at 250°F.



S-293387-C

Oxide Replica

10,000X

Alloy 71

Step Aged

Powder Extrusion

Figure 51

Extrusion from Alloy 71 powder that had been step-aged 24 hrs at 250°F + 20 hrs at 330°F after S.H.T. Note increased amount of grain boundary precipitation.

## APPENDIX A

### Statistical Analyses of Alloy Compositions

Attempts at mathematically selecting compositions resulting in a certain strength have been conducted at times during the course of this contract. The initial evaluation was made on data compiled before the contract began. Two alloy groups were investigated, the Al-Zn-Mg-Cu-Mn-Cr and the Al-Zn-Mg-Cu-Fe-Ni-Mn-Cr series. Equations were obtained from the data and the predicted optimum tensile strength for each group was calculated. The predicted optimum tensile values, the resulting actual tensile values and the equations are listed in Appendix A, Table I. The predicted and actual values are not in close agreement, especially for the latter series. More data are required.

The statistical analyses of the initial 39 alloys resulted in the selection of the compositions of Alloys 43 through 59. These data were obtained using several applicable statistical tools. According to the evaluation, V, Zr, Co, Mo and W appear to have no significant effect on tensile strength. When these elements appeared, they were generally all present and the apparent lack of effect could have been an inability to extract information from the combined groups. The ranges worth investigating were Zn, 8 to 11%; Mg, 3 to 6%; Cu, 1 to 2.5%; Mn, 1 to 2%; Fe, 0 to 1.5%; Ni, 2 to 5%; Cr, 0 to 0.1%; and Ti, 0 to 0.1%. The analyses also indicated the "best" composition, which was atomized as Alloy 59. The results are listed below.

Alloy	Zn	Mg	Cu	Mn	Fe	Ni	Cr	Ti	Predicted	Actual	
									T.S. ksi	T.S., ksi	
										H.T.#1	H.T.#2
--	10.0	4.0	1.5	1.5	1.0	4.0	0.0	0.0	123.0	--	--
59	10.2	3.9	1.6	1.6	1.0	4.1	-	-	--	99.3	83.9

The values are much lower than the predicted values; the reasons for this large difference are not known.

An evaluation of the Al-Zn-Mg-Cu-Mn and the Al-Zn-Mg-Cu-Mn-Fe-Ni alloys was made to determine if a higher strength alloy could be statistically selected. Alloys 1 to 3, 33, 35, 36, 39, 48 and 49 were used for the former group; these same alloys being given Heat Treatments #1 and #2 and evaluated by the optimum point equation. The resulting composition selected to give the highest predicted properties was compared to alloys with similar compositions and heat treatments along with actual properties and is given in Appendix A, Table II. The actual and predicted properties for compositions given Heat Treatment #1 were good. The composition required to give the optimum properties has already been studied. The multiple regression model was used to analyze all alloys having a yield strength of 90,000 psi or higher. The equation obtained is

$$\begin{aligned}
 \frac{Y.S., ksi}{10} = & 5.932 + .5029 Zn - .0215 Zn^2 + .7205 Mg \\
 & - .0930 Mg^2 - .0962 Cu + .0046 Cu^2 + .5335 Mn \\
 & - .1426 Mn^2 - .1826 Fe + .1622 Fe^2 + .1031 Ni \\
 & - .0437 Ni^2 - .4624 Cr - .0639 Ti - .1903 \\
 & (V+Zr+Mo+W) + .0485 Co - .0015 (Cr+Ti)^2 \\
 & + .0082 [Mg \times (Cu+Fe+Mn)] - .0154 (Mg \times Mn) \\
 & - .0010 (Mg \times Zn) - .00008 (Al \times Mg \times Cu) \\
 & - .0011 (Al \times Cu \times Fe)
 \end{aligned}$$

Higher strengths were suggested if the Fe were raised, the Cu were eliminated or maintained at a low level and using Zn ~ 11.7,

Mg = 3.87, Mn = 1.87, and Ni = 1.18. All other alloying additions with the exception of perhaps Co are detrimental. A sample of the predicted yield strengths for several alloys follows:

Composition, %						Predicted Yield, ksi
<u>Zn</u>	<u>Mg</u>	<u>Cu</u>	<u>Mn</u>	<u>Fe</u>	<u>Ni</u>	
11.7	3.87	0.0	1.87	4.0	1.18	127.2
11.7	3.87	0.6	1.87	4.0	1.18	127.0
11.7	3.87	0.0	1.87	3.0	1.18	117.4
11.7	3.87	0.6	1.87	3.0	1.18	117.3

Between the initiation of this survey and the reporting of the results, the object of this contract was changed. This resulted in no alloys being made to obtain the high strengths predicted. Comparisons with existing data are not feasible since the Fe content is high and the ratio of 3 to 4 Fe to 1 Ni has not been investigated.

In conjunction with the revised objective, an optimum point program was conducted on Alloys 52 and 62 to 72. Each heat treatment, Nos. 1 through 3, was evaluated separately to give the highest elongation and the yield strength. The results are given in Appendix A, Table III. Also included are data for Alloy 71 which has a similar composition. The agreement between the predicted and actual yield strengths is quite good; the agreement between the elongations leaves much to be desired. Regression analysis was obtained also, the predicted elongation and yield strength for two alloys calculated from the equation. These results are given in Appendix A, Table IV. An alloy with a higher elongation is possible by lowering the Zn, Mg and Co and setting the Cu at

1.3%. Alloys 19 and 20 have a somewhat similar composition and were included for comparison. The elongation model predictions appear quite reasonable but the yield strength predictions may be high. In comparing the results of the optimum point program and the regression analysis, it is interesting to note that though the indicated alloys differ, the general conclusions of lower Zn and lower Co result from both approaches.

Selection of alloys by statistical methods as a means of obtaining desired properties is possible. The chances of obtaining a reliable value depend on the phases present in the samples and the logical selection of composition.

# APPENDIX A

## Table I

### INITIAL STATISTICAL SELECTION OF COMPOSITION TO GIVE HIGH TENSILE PROPERTIES

Alloy	S. No.	Composition, %						Tensile Strength, ksi		Equation	
		Zn	Mg	Cu	Mn	Cr	Fe	Ni	Actual Avg.		Predicted
--		9.2	3.4	1.2	1.1	0.0	--	--	--	117.5	1
33	277405	9.0	3.5	1.2	1.1	--	--	--	108.0	--	
--		6.5	2.7	0.6	0.0	0.1	0.0	2.0	--	116.2	2
32	277404	6.6	2.7	0.6	--	0.1	--	2.3	97.0	--	

Equation (1)

117.5

$$TS = 1 + (.075 Zn - .685)^2 + (.218 Mg - .750)^2 + (.370 Cu - .434)^2 + (.321 Mn - .354)^2 + (.416 Cr - .078)^2$$

Equation (2)

116.2

$$TS = 1 + (.083 Zn - .537)^2 + (.052 Mn - .141)^2 + (.169 Cu - .103)^2 + (.027 Ni - .053)^2 + (.029 Mn)^2 + (1.42 Cr - .144)^2$$



# APPENDIX A

## Table II

### SUMMARY OF HIGH STRENGTH STATISTICAL ANALYSES PROGRAMS (a)

Alloy	Composition, %							H.T. No.	Avg. Tensile Properties, ksi			
	Zn	Mg	Cu	Mn	Fe	Ni	Cr		Actual		Predicted	
									T.S.	Y.S.	T.S.	Y.S.
36(b)	10.7	3.8	2.0	1.7	--	--	--	1	--	106.0	--	105.3
--	10.88	4.9	2.0	1.71	--	--	--	1	--	--	108.3	--
39	10.9	4.9	2.0	1.7	--	--	--	1	110.0	--	--	--
--	8.4	3.6	0.006	0.99	0.80	2.81	0.0	2	--	--	109.6	--
38	8.4	3.6	0.0	1.00	0.80	2.80	0.0	2	119.6	--	--	--

-149-

(a) Optimum point analysis.

(b) Predicted alloy had same composition as Alloy 36.

# APPENDIX A

Table III

## SUMMARY OF RESULTS ON OPTIMUM POINT PROGRAMS OF HIGH ELONGATION ALLOYS

Heat Treatment	Composition, %			Yield Strength, ksi		Elongation, %	
	Zn	Mg	Cu	Predicted	Actual Avg. (b)	Predicted	Actual Avg. (b)
1	9.43	NE (3.6-4.7) (a)	0.72	103.2	105.0	2.57	3.1
2	9.29	NE (3.6-4.7)	NE (0.6-1.5)	112.1	112.0	2.13	3.7
3	9.37	4.10	0.55	109.8	111.4	1.80	2.4
(b)	9.2	3.6	0.6	--	--	--	--

(a) NE = Negligible effect in range studied.

(b) Alloy 71 has a similar composition and actual properties to the predicted optimums.

H.T.#1

$$YS = \frac{104.03}{1 + (.0017)(Zn-11.10)^2 + (.0438)(Cu-1.008)^2}$$

$$EL = \frac{2.568}{1 + (.0209)(Zn-9.43)^2 + (.1186)(Cu-.725)^2}$$

H.T.#2

$$YS = \frac{112.09}{1 + (.0021)(Cu-1.31)^2}$$

$$EL = \frac{2.133}{1 + (.2827)(Zn-9.29)^2 + (.9181)(Co-.88)^2}$$

H.T.#3

$$YS = \frac{109.79}{1 + (.0074)(Mg-4.1)^2 + (.0389)(Cu-.60)^2}$$

$$EL = \frac{1.803}{1 + (.1564)(Zn-9.37)^2 + (.0369)(Cu-.551)^2 + (.4999)(Co-.737)^2}$$

# APPENDIX A

## Table IV

### SUMMARY OF RESULTS OF REGRESSION ANALYSES ON HEAT TREATMENT #1 HIGH ELONGATION ALLOYS

Alloy	Composition, %			Yield Strength, ksi		Elongation, %	
	Zn	Mg	Co	Predicted	Actual Avg.	Predicted	Actual Avg.
--	8.0	3.0	1.3	0.5	107.7	--	6.6
--	9.0	3.0	1.3	0.5	107.1	--	5.8
19	7.5	3.0	1.7	0.0	--	95.3	7.2
20	7.8	3.6	2.3	0.0	--	94.8	4.0

$$YS = 284.91 - 20.53 Zn - 39.30 Mg - 17.37 Cu + .17 Co \\ + 1.17 Zn^2 + 4.61 Mg^2 + 6.52 Cu^2 - .28 Co^2$$

$$EL = 60.70 - 4.89 Zn - 15.81 Mg + 1.13 Cu - 2.06 Co \\ + .24 Zn^2 + 1.92 Mg^2 + .42 Cu^2 + .30 Co^2$$

APPENDIX B

Hardness Value

The heat treated APM alloys have hardnesses much higher than usually associated with aluminum, Table I. During the step aging evaluation, a possible correlation between the strength and hardness was sought. A number of alloys were given Heat Treatment #1 with Age #2 at 330°F or 350°F, the hardness values being taken at predetermined intervals, Tables II and III. The results of these and also earlier tests were evaluated graphically. The poor reproducibility of several lots of Alloy 52 is seen in Figure 1. The excellent strength reproducibility is contrasted to the poor hardness reproducibility in the main text in Figure 9, also for Alloy 52. Work was discontinued.

Appendix B

Table I

HARDNESS VALUES OF SELECTED APM ALLOYS

<u>Alloy Number</u>	<u>Sample Number</u>	<u>H.T. No. (a)</u>	<u>Test Piece (b)</u>	<u>Rockwell Hardness</u>	<u>Brinell Hardness (c)</u>
34	283442	2	A	G-79	(189)
	283449	2	B	----	230 (e)
	283449	3	B	----	229 (e)
38	283454	2	A	G-80	(192)
	283459	2	B	----	208 (e)
	283459	3	B	----	210 (e)
39	283463	1	A	G-71	(167)
	283467	2	A	G-80	(192)
49	283471	2	A	G-84	(210)
	283472	2	A	G-79	(189)
50	284130	F	C	B-56	(90)
	284130	2	C	G-86	(220)
	283484	2	A	G-84	(210)
52	307454	1	E	----	190 (e)
	307454	1	F	----	189 (e)
	307455	1	E	----	188 (e)
	307455	1	F	----	186 (e)
	307454	2	E	----	195 (e)
	307454	2	F	----	204 (e)
	307455	2	E	----	182 (e)
	307455	2	F	----	190 (e)
	283497	2	A	G-73	(172)
	284131	2	D	G-87	226 (d)
	284131	2	D	----	213 (e)
	284131	3	D	G-84	204 (e)
	284131	3	D	----	219 (f)
	307321	1	E	----	186 (e)
	307321	1	F	----	181 (e)
62	307321	2	E	----	190 (e)
	307321	2	F	----	188 (e)
64	307323	1	E	----	176 (e)
	307323	1	F	----	186 (e)
	307323	2	E	----	176 (e)
	307323	2	F	----	192 (e)
71	307330	1	E	----	191 (e)
	307330	1	F	----	192 (e)
	307330	2	E	----	185 (e)
	307330	2	F	----	190 (e)

(a) Heat Treatment

F - As Fabricated.

#1 - SHT 2 hrs. at 860°F, CWQ, Aged 24 hrs. at 250°F.

#2 - SHT 1/2 hr. at 920°F, CWQ, Aged 96 hrs. at 225°F.

#3 - SHT 2 hrs. at 860°F, CWQ, Aged 96 hrs. at 225°F.

(b) Test Pieces

A - Broken tensile specimens from 2 in. dia. extruded rod.

B - Rolled plate from hot press forged compacts.

C - Slices from hot compacted billets.

D - Section of rolled plate from 1" x 4-1/4" extrusion 1/2" thick.

E - Section of rolled sheet from 1" x 4-1/4" extrusion 1/10" thick.

F - Section of rolled plate from 1" x 4-1/4" extrusion 1/4" thick.

(c) Values in parenthesis are converted from Rockwell to Brinell Hardness values using a 500 kg load and a 10 mm ball.

Source: ASM Metals Handbook, 1948 Edition, p. 101

(d) 1500 kg load and 10 mm ball.

(e) 500 kg load and 10 mm ball.

(f) 3000 kg load and 10 mm ball.

APPENDIX B

Table II

ROCKWELL C HARDNESS VALUES OF SELECTED ALLOYS AFTER PROLONGED AGING AT 330°F

Alloy/S. No.

Additional Aging Time (a) at 330°F, Hours	5	7	8	14	33	38	39	40	49	50	51	52	54	57	66
2	277177	277379	277380	277386	277405	283452	283464	277428	283470	283482	278249	283492	278252	278255	307244
4	C	C	C	C	C	C	C	C	C	C	C	C	C	C	C
6	73	76	78	77	73	75	77	75	76	77	77	73	80	80	77
8	71	74	75	75	71	72	74	74	74	75	75	70	78	79	71
10	68	72	74	73	69	70	74	73	72	73	74	69	76	77	69
12	66	70	71	71	67	69	73	72	70	70	71	66	75	76	67
14	65	69	70	69	65	67	72	70	68	69	71	65	74	75	65
16	62	68	70	68	63	64	71	70	66	68	70	62	73	73	63
18	62	67	70	67	63	64	71	70	66	69	68	64	73	73	63
20	61	66	69	66	62	64	69	69	65	65	67	62	72	72	62
22	60	65	68	65	60	63	67	68	63	64	68	61	71	70	59
24	57	62	67	64	59	61	67	67	62	63	66	58	69	70	59
26	56	61	66	63	57	61	65	66	61	63	64	57	68	69	58
28	54	60	64	62	56	60	64	65	60	62	65	55	66	68	57
30	53	59	63	61	55	59	61	65	60	61	64	54	65	69	56
32	53	58	62	60	54	58	60	64	59	60	63	53	64	68	55
34	53	57	61	59	53	57	59	63	58	59	62	52	63	67	54
36	53	56	60	58	52	56	58	62	57	58	61	51	62	66	53
38	53	55	59	57	51	55	57	61	56	57	60	50	61	65	52
40	53	54	58	56	50	54	56	60	55	56	59	49	60	64	51
42	53	53	57	55	49	53	55	59	54	55	58	48	59	63	50
44	53	52	56	54	48	52	54	58	53	54	57	47	58	62	49
46	53	51	55	53	47	51	53	57	52	53	56	46	57	61	48
48	53	50	54	52	46	50	52	56	51	52	55	45	56	60	47
50	53	49	53	51	45	49	51	55	50	51	54	44	55	59	46

NOTE: (a) S.P.T. 2 hrs. at 860°F, C.W.Q., Age #1 - 24 hrs. at 250°F

C - Center  
M - Midway

APPENDIX B

Table III

ROCKWELL C HARDNESS VALUES OF SELECTED ALLOYS AFTER PROLONGED AGING AT 350°F

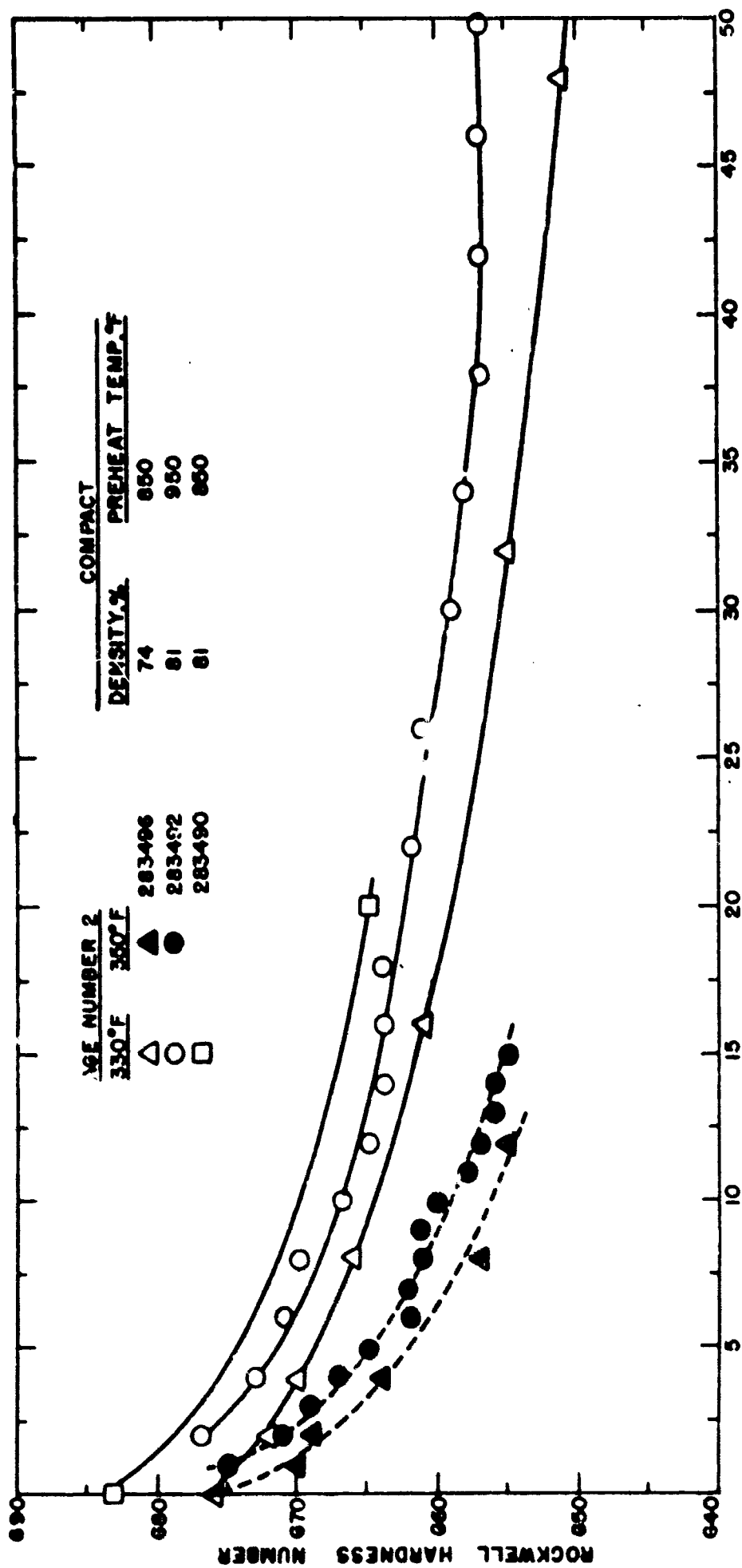
Alloy/S. No.

Additional Aging Time  
at 350°F, Hours (a)

	5	7	8	11	33	38	39	40	49	50	51	52	54	57	66
1	277377	277379	277380	277386	277405	283452	283464	277428	283470	283482	278219	283492	278252	278255	307314
2	70	71	75	74	70	72	76	75	74	75	76	71	78	79	71
3	69	70	72	73	68	71	75	71	72	73	73	69	76	77	71
4	66	67	70	69	66	67	72	69	69	70	70	66	74	75	68
5	64	66	69	71	64	66	71	69	66	69	70	64	73	74	65
6	61	63	67	68	62	63	65	69	66	68	69	62	71	72	64
7	59	60	66	67	61	62	65	68	65	66	67	62	72	73	63
8	59	60	66	67	59	61	67	68	64	66	67	61	70	72	62
9	58	60	64	64	58	63	64	67	63	65	65	60	69	70	59
10	58	59	64	63	57	62	66	67	62	64	64	57	68	69	58
11	56	58	63	62	56	61	65	67	61	63	64	57	68	70	57
12	55	57	63	62	56	60	65	67	61	63	64	56	68	70	55
13	54	57	63	62	54	59	64	66	60	62	64	56	68	69	55
14	54	57	63	62	54	59	64	66	60	62	64	56	67	68	54
15															

NOTE: (a) S.H.T. 2 hrs. at 860°F, C.M.Q., Age #1 - 24 hrs. at 250°F

C - Center  
M - Midway



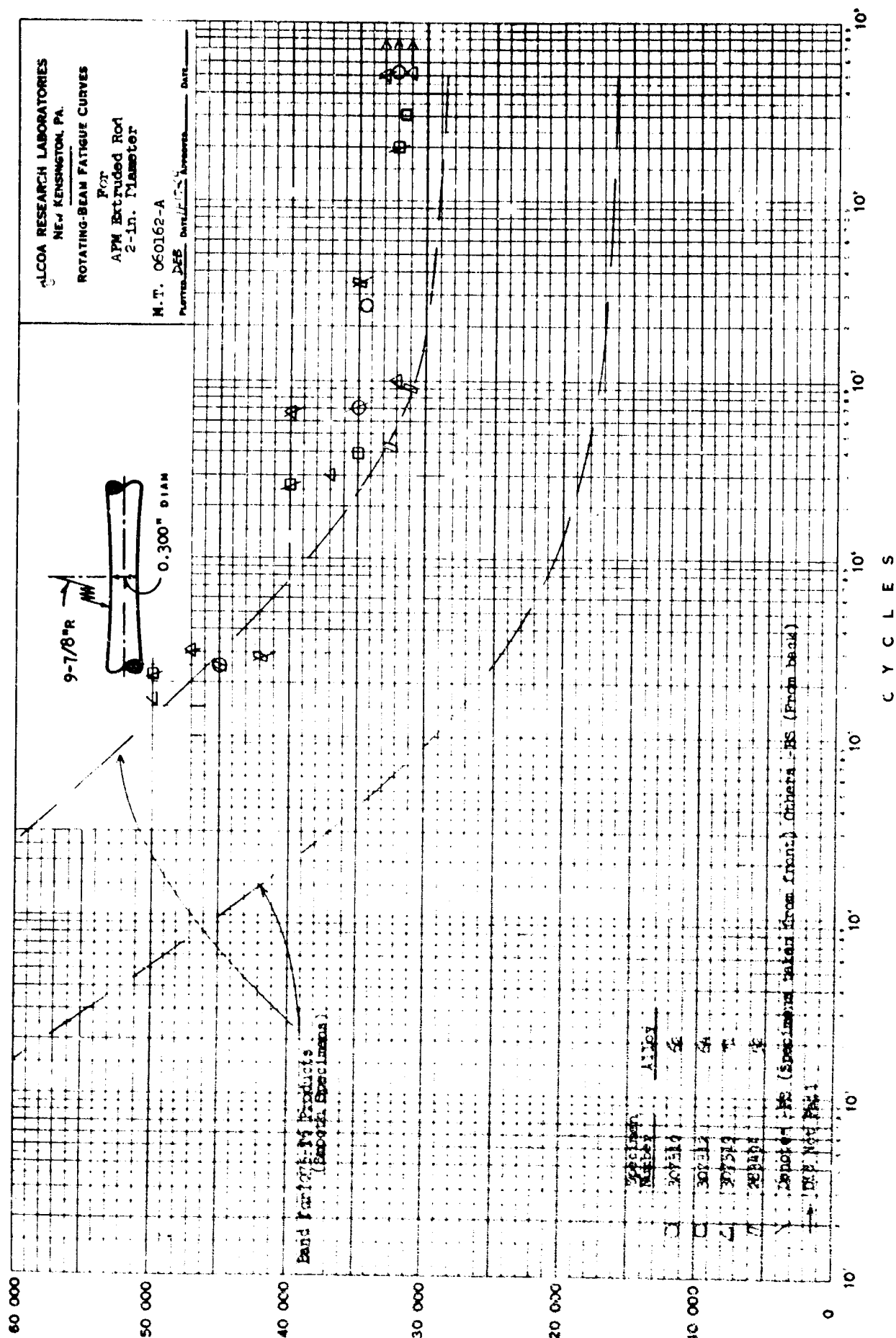
THE EFFECT OF PROLONGED AGING ON HARDNESS OF ALLOY 52



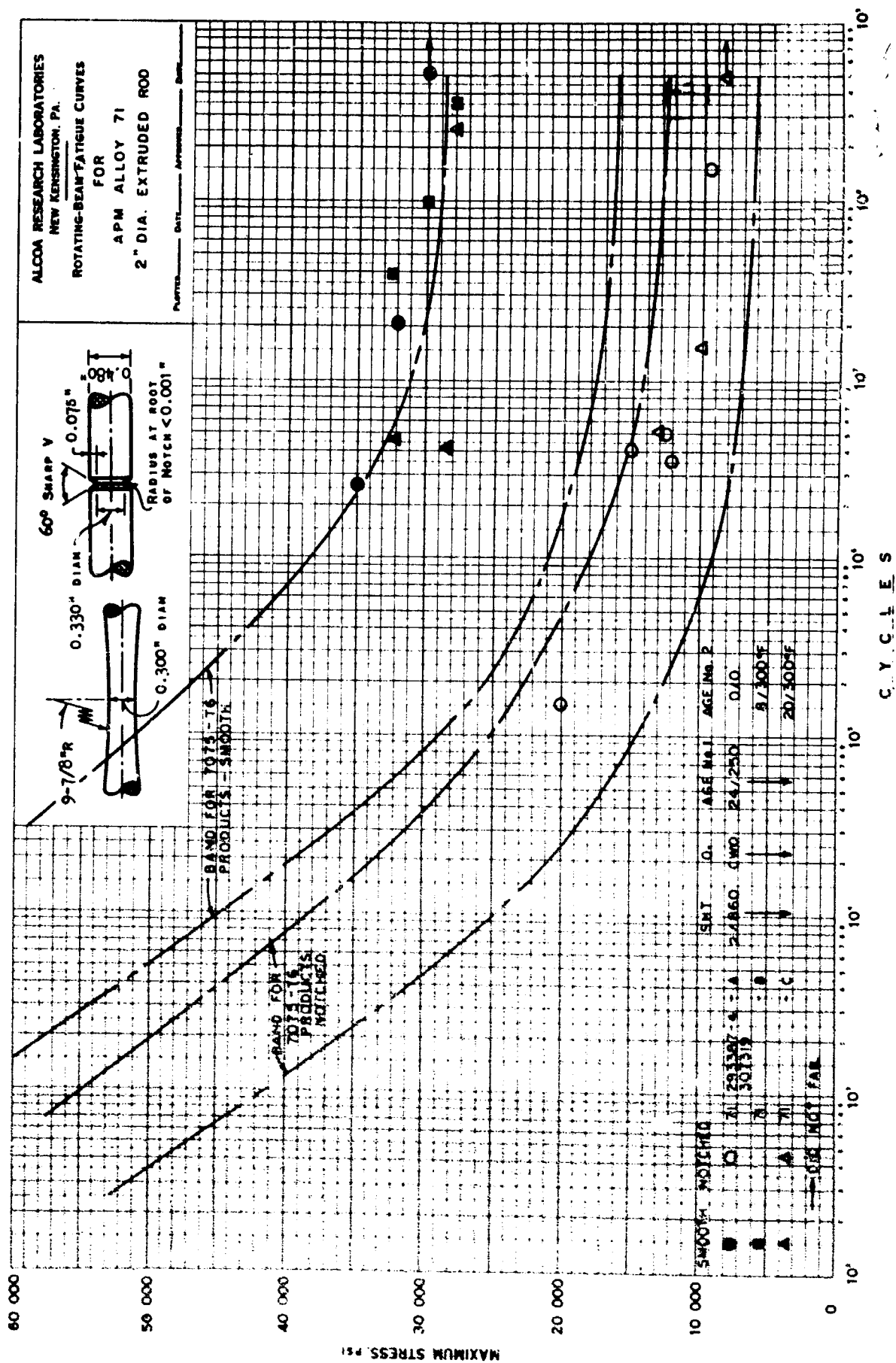
#### APPENDIX C

Fatigue limits were determined on smooth (unnotched) specimens machined from extruded stock of Alloys 52, 62, 64 and 71. The specimens were given Heat Treatment #1. The results are compared with the 7075-T6 product scatter band, Figure 1. The fatigue limits of these alloys generally fall above the 7075-T6 band with a few falling on or just below the upper limit of the band. The fatigue limit for Alloy 62 is approximately 32 ksi, for Alloy 64 approximately 31 ksi, and for Alloy 71 approximately  $32 \pm 1$  ksi. No fatigue limit was determined for Alloy 52 because of testing difficulties.

Additional tests were made outside the contract but are included in this report. Smooth (unnotched) and notched specimens of Alloy 71 extrusions were given Heat Treatment #1 and also step aged, Figure 2. The notched specimens are at least as good as notched 7075-T6. The smooth specimens are better than 7075 smooth specimens. Step aging does not appear to improve or have an adverse effect on the fatigue endurance limit. Alloys 85 and 86 also show similar fatigue endurance limits in the smooth and notched configurations, Figure 3.



APPENDIX C  
FIGURE 1



**APPENDIX C**  
**FIGURE 2**

